

N O T I C E

THIS DOCUMENT HAS BEEN REPRODUCED FROM
MICROFICHE. ALTHOUGH IT IS RECOGNIZED THAT
CERTAIN PORTIONS ARE ILLEGIBLE, IT IS BEING RELEASED
IN THE INTEREST OF MAKING AVAILABLE AS MUCH
INFORMATION AS POSSIBLE



National Aeronautics and Space Administration

JSC 16303

**Lyndon B. Johnson Space Center
Houston, Texas 77058**

November 1979

(NASA-TM-80946) EVALUATION OF USE OF MPAD
TRAJECTORY TAPE AND NUMBER OF ORBIT POINTS
FOR ORBITER MISSION THERMAL PREDICTIONS

FOR ORBITER MISSION VEHICLE
(NASA) 126 p HC A06/MF A01

CSCL 22A

N80-16081

Unclas

G3/13

12280

EVALUATION OF USE OF MPAD TRAJECTORY TAPE AND NUMBER OF ORBIT POINTS FOR ORBITER MISSION THERMAL PREDICTIONS

STRUCTURES AND MECHANICS DIVISION
THERMAL TECHNOLOGY BRANCH
ES3-79-6



Evaluation Of Use Of MPAD Trajectory Tape
And Number Of Orbit Points For
Orbiter Mission Thermal Predictions

Prepared by: Robert A. Vogt *R.A.V.*

Approved by: E.T. Chimenti
E. T. Chimenti, Head
Thermal Analysis Section

J.A. Smith
J. A. Smith, Chief
Thermal Technology Branch

Table of Contents

	Page
Table of Contents	1
1.0 Summary	1
2.0 Trajectory Versus Current Method	3
2.1 Introduction	3
2.2 Results	6
3.0 Effects of Number Of Orbit Points	7
3.1 Introduction	7
3.2 Results	7
4.0 Conclusions	9
References	11
Figures	12

1.0 SUMMARY

The TRASYS¹ computer program has the capability to compute orbital heating rates utilizing the JSC Mission Planning and Analysis Division (MPAD) common format trajectory data tape². This tape defines the on-orbit flight profile for a given flight. In the past, this option has not been utilized for Shuttle thermal analysis because initially the analysis performed was for design and mission planning purposes. In these situations, the current method of specifying nominal orbital and attitude parameter data and letting TRASYS determine the position and orientation is more suitable and convenient.

The subject of this report is to document an evaluation of the application of this simplified approach to predicting temperatures for preflight and post flight mission analysis. All of the analysis presented utilized the latest STS-1 MPAD trajectory tape³, and the simplified "136 node" midsection/payload bay thermal math model⁴. For the first 6.7 hours of the STS-1 flight profile, transient temperatures are presented for selected nodal locations with the current standard method, and the trajectory tape method. Whether the differences are considered significant or not depends upon the view point. For a specified time, temperature differences of well over 100°F on externally exposed surfaces, and up to 15°F on the Thermal Protection System (TPS) bondline locations, can be noted by comparing the results. On the other hand, if the cyclic extremes per orbit at the bondline are the real interest, the maximum difference is less than 10°F. The cyclic extremes on the surface nodes approach a maximum difference of 100°F.

Other transient temperature predictions are also presented. These results were obtained to investigate an initial concern that perhaps the predicted temperature differences between the two methods would not only be caused by the inaccuracies of the current method's assumed nominal attitude profile but also be affected by a lack of a sufficient number of orbit points in the current method. The current practice for an attitude hold profile is to

utilize 12 orbit points plus the entry and exit points for the earth's shadow. Data is presented where 24 and 6 orbit points were used in place of 12 orbit points. The comparison between the 6, 12 and 24 orbit point parameter shows a surprising insensitivity to the number of orbit points. For the example used, the temperature differences between the current method and the trajectory tape method are primarily related to the differences in the attitude time line.

2.0 TRAJECTORY VERSUS CURRENT METHOD

This section defines the baseline conditions and methods utilized to permit an objective comparison of the trajectory tape method, and the current method used to generate the on-orbit external environmental heat loads for JSC Shuttle Orbiter/payload thermal analysis.

An essential part of the comparison are the SINDA⁵ temperature-time plots and environmental heat rates. This data is also presented for both approaches.

2.1 INTRODUCTION

The trajectory tape utilized by TRASYS to compute the environmental heat loads was the latest available MPAD tape for the STS-1 Shuttle flight. It supposedly has a predominant nominal +Z LV attitude (top to earth) hold. A JSC program which prints the pertinent data of interest to a TRASYS user revealed, however, that a strict +Z LV attitude is seldom adhered to. How these attitude deviations from the nominal, feed back into the temperatures is basically the subject of this report. Table 1 summarizes the trajectory tape timeline analyzed from the printout of the data on the trajectory tape and reference 6. It should be noted that only 139 minutes of actual +Z LV attitude was included.

Full advantage of several new features which have recently been implemented in TRASYS to make it more convenient and efficient to utilize the trajectory tape input were used. Only the first 6.7 hours of elapsed on-orbit time was evaluated because of the desire to make this study available in a timely manner, so its significance may be evaluated with respect to on-going Orbiter/payload thermal analysis. To prepare the trajectory tape input for maximum program efficiency, the procedure is still very time consuming and error prone.

TABLE 1 - STS-1 ATTITUDE TIME LINE

TIME H:M:S	EVENT	LOOK ANGLES TO: [*]				ATTITUDE MAINTENANCE		REMARKS	
		SUN		EARTH		MODE	MIN		
		CLOCK	CONE	CLOCK	CONE				
0:45:30	OMS-2 CutOff	328.2	93.4	90.0	0.0	Inertial	17		
1:02:00		328.2	93.4	0.0	66.0				
1:08:00	+ZLV Hold	276.6	31.9	N/A	0.0	LV	79	Open Payload Bay Doors	
2:27:00	Maneuver	320.7	55.4	N/A	0.0			Maneuver to COAS Calibration	
2:33:00	COAS Calibration	97.6	120.4	140.6	145.1	Inertial	6		
2:39:00	Maneuver	97.5	120.4	95.5	151.9			Maneuver to IMU Alignment Attitude	
2:45:00	IMU Alignment Attitude	97.5	120.3	53.9	142.3	Inertial	15		
3:19:00	Maneuver	145.5	95.2	2.5	87.5				
3:25:00	+ZLV Hold	275.4	148.4	N/A	0.0	LV	60		
4:25:00	Maneuvers	214.2	67.2	N/A	0.0				
6:27:00	Maneuver	108.1	8.1	326.9	141.3			Maneuver to IMU Alignment Attitude	
6:33:00	IMU Alignment Attitude	213.2	123.8	32.3	101.3	Inertial	29		
6:53:00	COAS Verification	213.2	123.8	119.3	135.1				
6:57:00	COAS IMU Alignment	213.2	123.8	140.7	131.4				
7:02:00	Maneuver	213.2	123.8	162.9	122.0			Maneuver to COAS IMU Alignment #2	
7:05:00	COAS IMU Alignment #2	118.9	133.7	143.6	105.4	Inertial	22		
7:13:00	COAS Verification #2	118.9	133.7	165.4	147.4				
7:27:00	COAS Verification #3	118.9	133.7	253.6	145.9				

* LOOK ANGLE DEFINITIONS AS DEFINED IN REFERENCE 1

The TRASYS computational charges to compute all the required direct incident fluxes for the 78 node geometric math model (GMM) was approximately 1 hour and 5 minutes for the trajectory tape method and 18 minutes for the run depicting the current method. The time interval between points does not exceed 10 minutes on the trajectory tape generated by MPAD. If events of interest occur between these 10 minute intervals, these points are also included on the tape. If justified, it is believed MPAD would honor a user's request to output at smaller nominal time intervals. Some attitude changes on the tape can be noted by double time point entries, in which case two records are given for the same time, but have a different orientation specified. These orientation changes are assumed instantaneous step changes within TRASYS. For the 6.6916 hours elapsed time, starting at 0.7584 hours and ending at 7.45 hours, the trajectory tape has a total of 80 records. These 80 records were reduced to 51 points utilizing a criteria to make it as comparable as possible to a 12 point parameter TRASYS ORBGEN option run. The criteria limited the maximum time interval to 7.5 minutes unless all heat loads were constant. To the 51 points, 20 entry and exit points of the earth's shadow are added because shadow points are not given on the trajectory tape. This gave a total of 71 discrete heat rate computation points. Conversely, the current method with a 12 point parameter TRASYS ORBGEN run has only 16 heat rate computation points computed. On a point bases the computational charges of both methods are comparable, indicating that the trajectory point method achieved the computational efficiencies of the ORBGEN option. The greater computational charges can thus be accounted to the greater number of points and the resulting fidelity of the trajectory method.

In setting up the 12 point parameter ORBGEN run, the objective was to make it as comparable as possible with the trajectory tape orbit parameters, and also with the currently accepted method of executing TRASYS. All of the required orbit parameters were specified based upon the trajectory tape data. Since the nominal orientation for the STS-1 trajectory tape is defined as +Z LV, this orientation was used in the ORBGEN run. The compatibility of both TRASYS runs was verified where similarity should exist. For

example, the initial fluxes are identical and the shadow entry and exit points times are within a fraction of a minute for the first orbit.

The differences in the TRASYS runs have been noted. All inputs were specified to minimize to the maximum extent possible any differences not related to an objective comparison. Both methods used the identical radiation exchange matrices, and all the SINDA runs used identical models including the radiation conductors. The only thing that was changed was the tape containing the heating rates computed by TRASYS and converted by the JSC TOTALQ2 TRASYS/SINDA interface program for the SINDA FLXRD subroutine which inputs the heat loads to the SINDA thermal analyzer which predicts the temperatures.

2.2 RESULTS

The transient temperatures ($^{\circ}$ F) predictions for the trajectory tape method are shown in Figures 1-17. The nodal locations are defined by reference 4. For the external exposed nodes, Figures 18-28 show the total environmental heat loads (Btu/hr) as supportive data. Immediately following these figures the same data set is shown in Figures 29-45 for the temperatures, and Figures 46-56 for the heat loads of the currently utilize method with an ORBGEN orbit point parameter of 12. Some discontinuties may be noted in the temperature results for the current method (for example Figure 35). This was verified to be because of the computation and/or output interval in SINDA was too large to pick up all the peaks accurately. Although all the SINDA run data presented, utilized a 0.1 hours compute and output interval, a test using a compute and output interval of 0.05 hours showed improved symmetry in the cyclic data. The differences occur only when the input heat rates change very rapidly with respect to time. The reader should be aware of the problem when evaluating the results although its true significance is not considered a serious problem since this occurs only over a short time pulse. Because symmetry is not generally expected in the STS-1 trajectory data, the detection of this type of error is not obvious, although it no doubt occurs.

3.0 EFFECT OF NUMBER OF ORBIT POINTS

The effect the number of orbit points used in the ORBGEN option of TRASYS has on temperature predictions was evaluated utilizing the nominal attitude hold of the STS-1 mission (+Z LV) and the simplified midsection/pay-load bay thermal model defined in reference 4. This analysis is described and the SINDA temperature time plots and environmental heat rates for a true anomaly increment of 15°, 30°, and 60° are presented for comparisons.

3.1 INTRODUCTION

The ORBGEN option has a variable called NPT, which is the number of equal true anomaly increments desired. Currently in the application of TRASYS for Shuttle thermal analysis the standard practice is to set NPT equal to 12 for fixed nominal attitude hold flight profiles. An additional four points are automatically added if TRASYS determines the vehicle enters and exits the earth's shadow. These four points are just each side of the terminators. With NPT equal to 12, the true anomaly increment is 30° exclusive of the shadow points. To bracket the standard 30° true anominaly increment, the temperatures were also computed for a 60° (NPT=6) and 15° (NPT=24) time anomaly increment.

3.2 RESULTS

The NPT=12 results have already been presented in Figures 29 through 56. A one-to-one comparison with NPT=6 can be made by examining Figures 57 through 84. Similarly, these may be compared with the NPT=24 results shown in Figures 85-112. The temperature results show very little difference between the three cases. Except for a momentary temperature spike, surface temperature all agree within 3 to 4°F, and for all practical purposes the bondline temperatures are identical in each case. Even if one is concerned by the temperature spikes there is only a maximum difference in surface temperature of approximately 20°F for a very short period of time. These spikes

occur because the rapid transition in the heating rates of some surfaces as they get a glimpse of the sun before the fixed earth orientation causes it to block itself, or be blocked by another intervening surface, from the sun. The heating rates differ in these instances for the three cases because the variation in the true anomaly interval used. These differences can be seen by comparing the "Impressed Heat" curves for each data set.

4.0 CONCLUSIONS

The comparison of transient on-orbit temperature predictions based on the TRASYS computed heat load with the latest STS-1 trajectory tape and assumed nominal +Z LV ORBGEN run shows large differences do occur. This study has determined that these differences occur primarily because the STS-1 mission attitude profile can not be simulated by a nominal fixed attitude hold for the elapsed time period investigated. The largest differences occurred on the externally exposed surfaces. On the TPS bondline nodes, the differences are much smaller. A longer period of time should be analyzed, however, to determine if the bondline differences will exceed the 15°F maximum variance noted thus far. A longer time is required because the structural nodes have not reached cyclic equilibrium for the fixed nominal attitude data case. Whether the necessity for utilizing a trajectory tape is established by this study, lies in the response to the questions; for what purpose the predictions will be used, and what vehicle locations are of interest? For example, if only the bondline temperatures are of interest, the 15°F maximum difference would probably not justify the added difficulty and expense of the more vigorous treatment of the trajectory tape method. Thinking in terms of the more widely utilized larger 400 to 520 node math models and the 52 hours mission duration, this has to be an important consideration. In considering the accuracy-run time relation of this study, one thing to keep in mind is that we are comparing the extremes of the spectrum. The real solution may be in the fact that the strength of both methods can be compromised to a degree to improve their respective weaknesses. For the trajectory this would mean sacrificing some accuracy to reduce run time by skipping more points on the trajectory tape. In addition, more attitude holds could be assumed whenever the attitude changes are judged to be insignificant. The nominal fixed attitude method could have improved accuracy by subdividing orbits and specifying more applicable attitudes for each segment.

The ORBGEN temperature predictions are shown in this study to be rather insensitive to the number of orbit points. Not documented herein, the results from a -Z LV (bottom to earth, +Y direction of velocity vector) further substantiates the belief that for any fixed attitude holds; 12 points and possibly 6 points are adequate for heat flux calculations with the subject model. The -Z LV attitude hold case is judged to be the severest test to evaluate the sensitivity to the number of orbit points. Sun orientation in general will be less sensitive than earth orientations. Selection of the severest STS-1 trajectory and ORBGEN sensitivity tests must be qualified with respect to the model being utilized. Whether other models will give significantly different results is uncertain since only one model was evaluated. In general, it is expected a more detail external midsection model would show greater differences relative to the simplified midsection model used, because the coarse nodal breakdown of the simplified model will smear out discontinuities in heating rates caused by shadowing, and also the sharp gradients caused by localized radiation entrapment. It is further anticipated that a detail model of the external surfaces of the Orbiter's nose section will not show differences as great as the simplified midsection model, because it is not affected by these factors to any great extent.

REFFERENCE

1. Jensen, Carl L. and Goble, Richard G.; "Thermal Radiation Analysis System - User's Manual," MCR 73-105 (Revision 2), June 1979.
2. Conway, H. L.; "MPAD Common Format Data Tape," (Revision 5), FM17 (79-120), August 27, 1979.
3. JSC/MPAD Non-format 7 Track Tape; Bin Number X18573.
4. Proctor, B. W., Ting, P. C. and Russel, D. J.; "Simplified '136 Node' On-orbit Orbiter Midsection/Payload Bay Thermal Math Model Description," ES3-76-7 (Revision C), April 1979.
5. Smith, James P.; Systems Improved Numerical Differencing Analyzer SINDA User's Manual, April 1971.
6. "STS-1 Operational Flight Profile, Appendix A On-orbit Trajectory and Attitude Data Cycle 2," JSC-14483, January 1979.

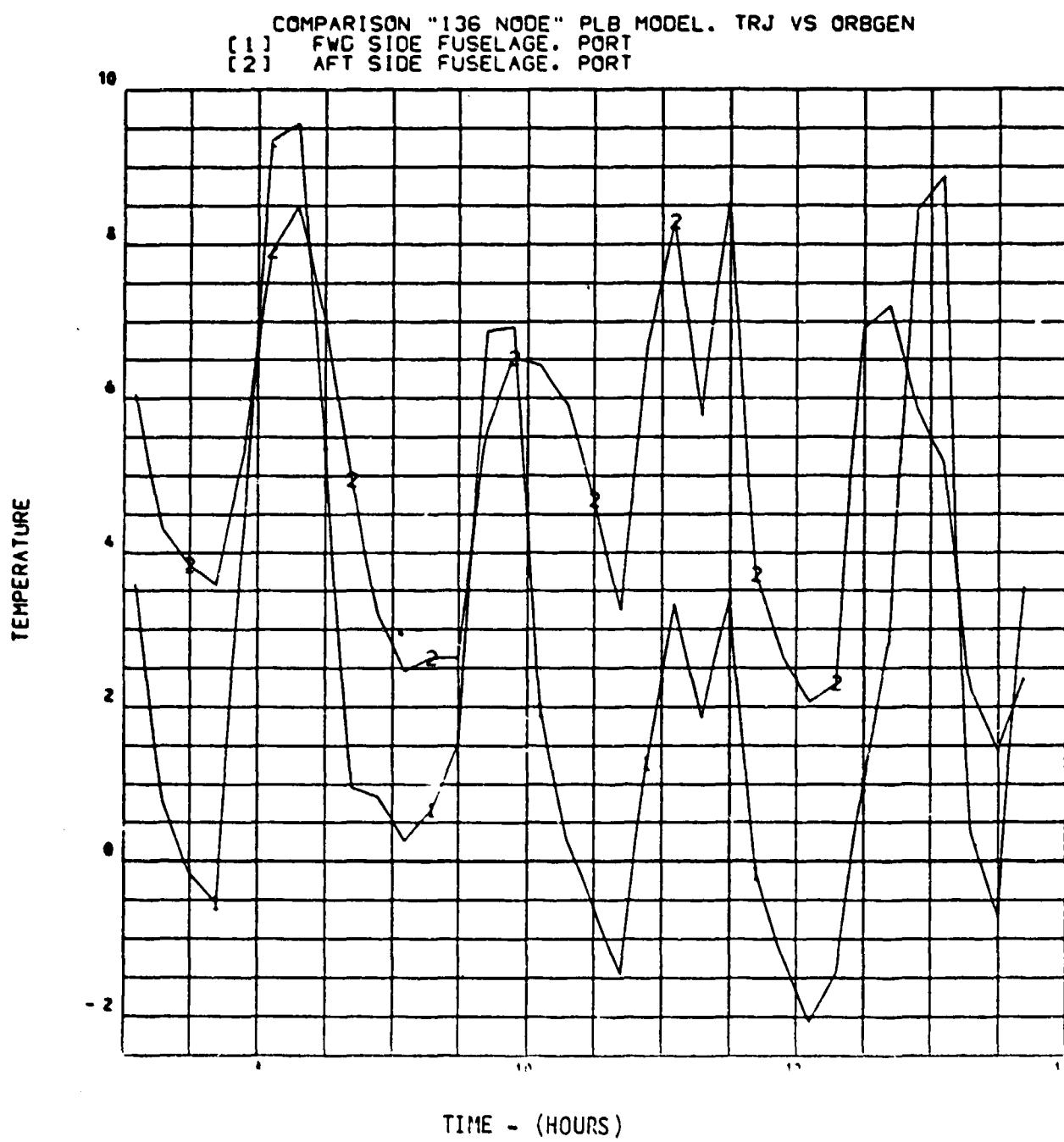
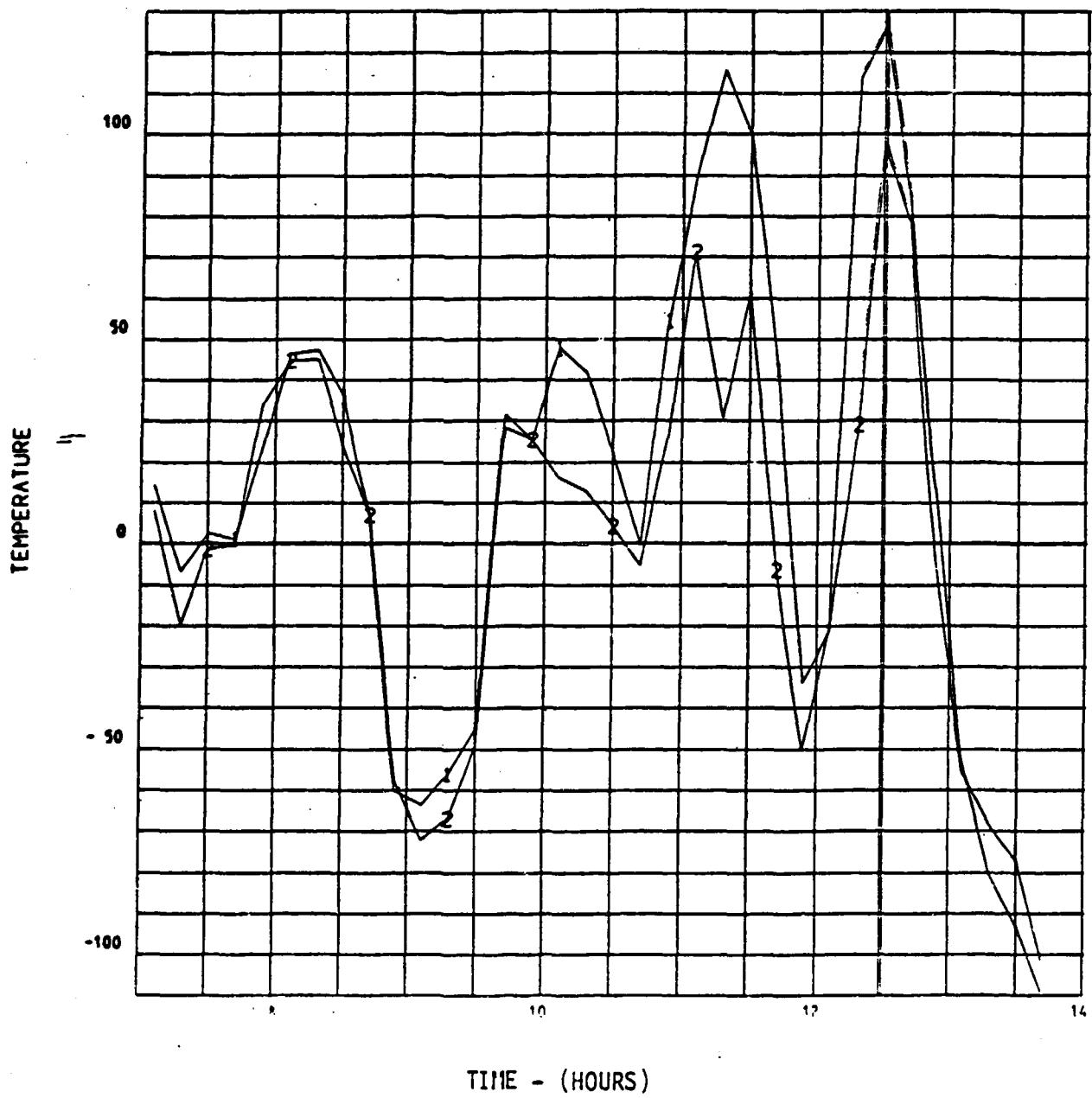


Figure 1

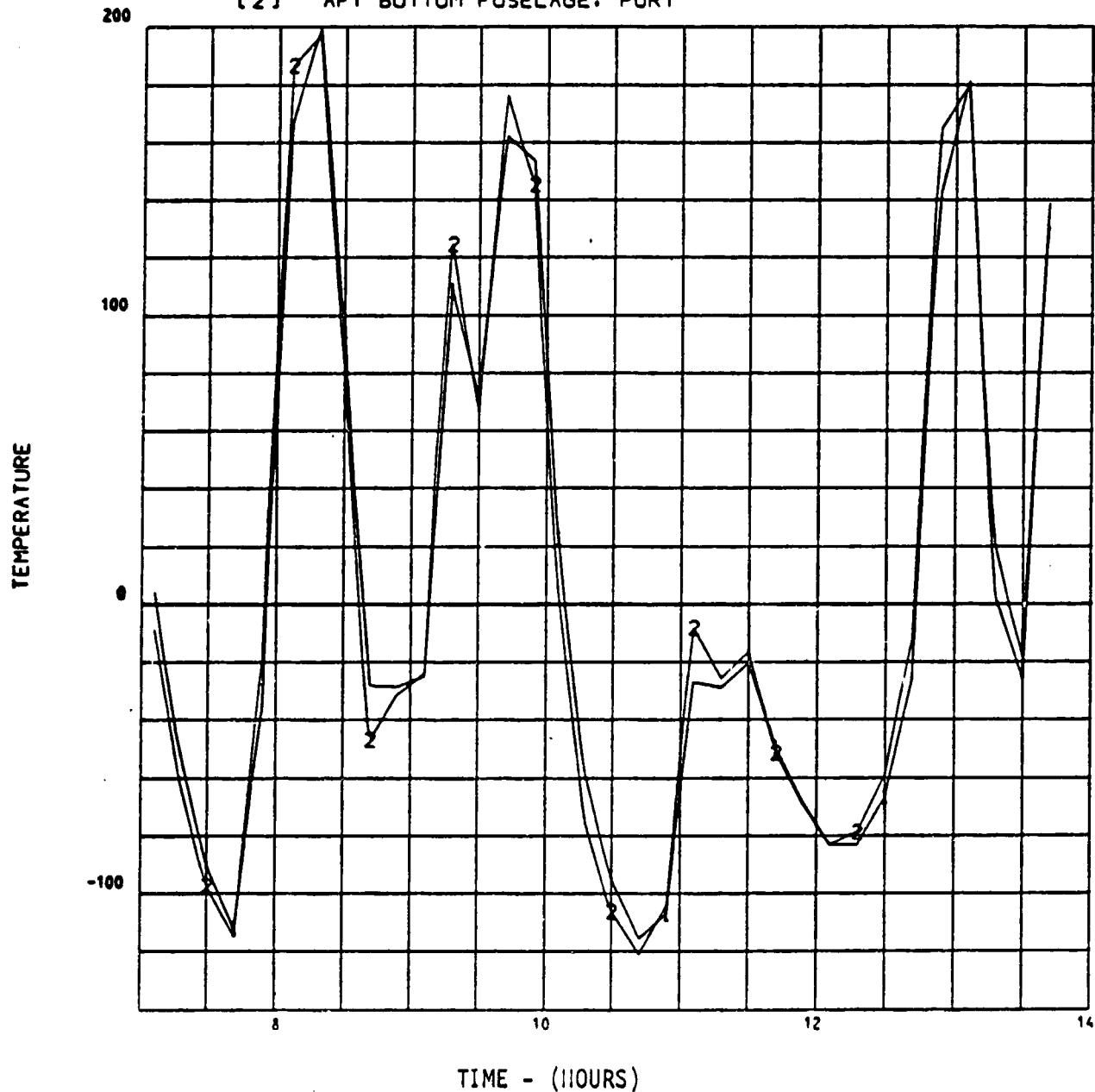
COMPARISON "136 NODE" PLB MODEL. TRJ VS OREGON
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT



TIME - (HOURS)

Figure 2

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD BOTTOM FUSELAGE, PORT
[2] AFT BOTTOM FUSELAGE, PORT



TIME - (HOURS)

Figure 3

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD PLB DOORS. PORT
[2] FWD PLB DOORS. PORT
[3] AFT PLB DOORS. PORT
[4] AFT PLB DOORS. PORT

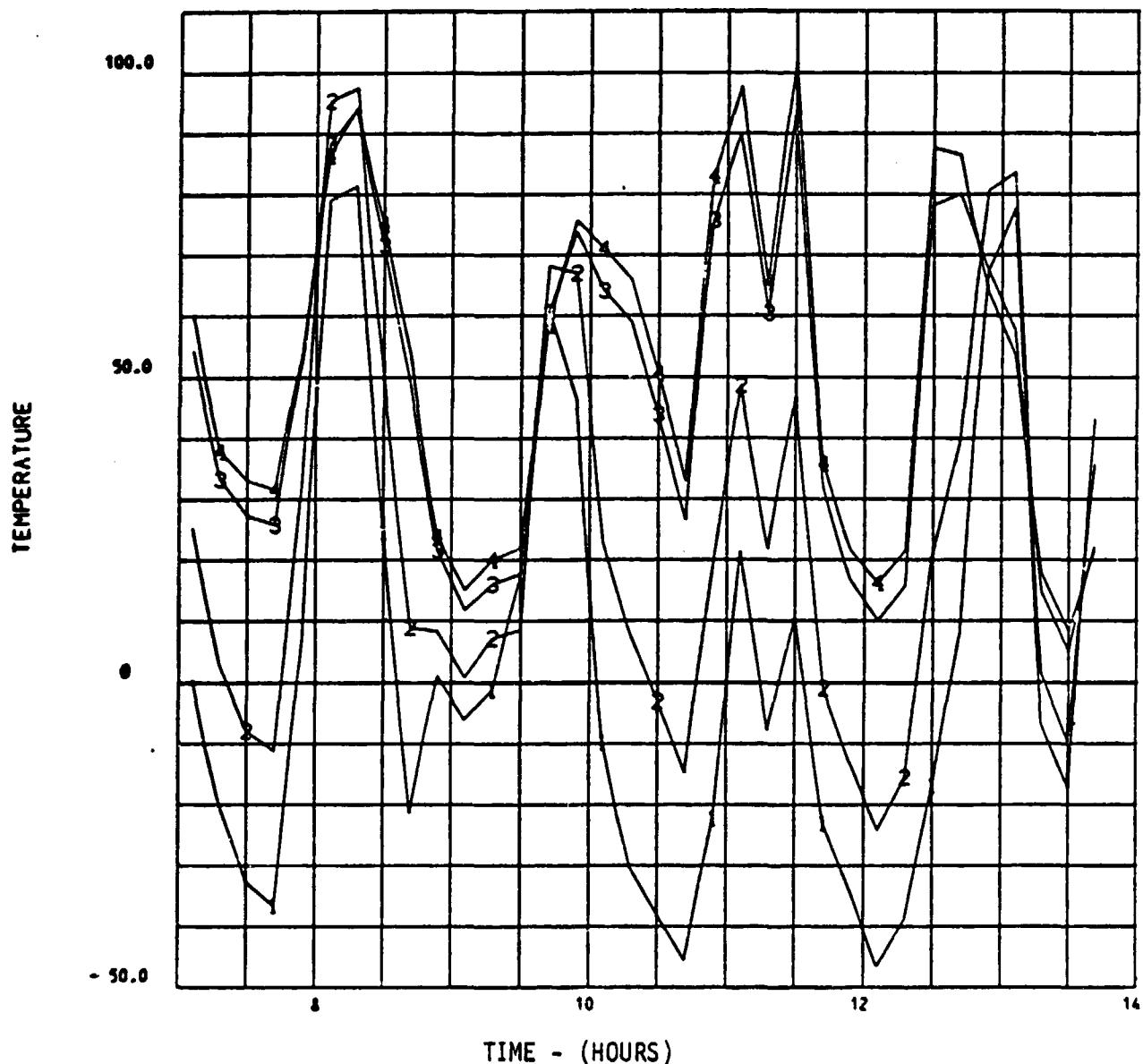


Figure 4

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD RADIATOR. PORT
[2] FWD RADIATOR. PORT
[3] AFT RADIATOR. PORT
[4] AFT RADIATOR. PORT

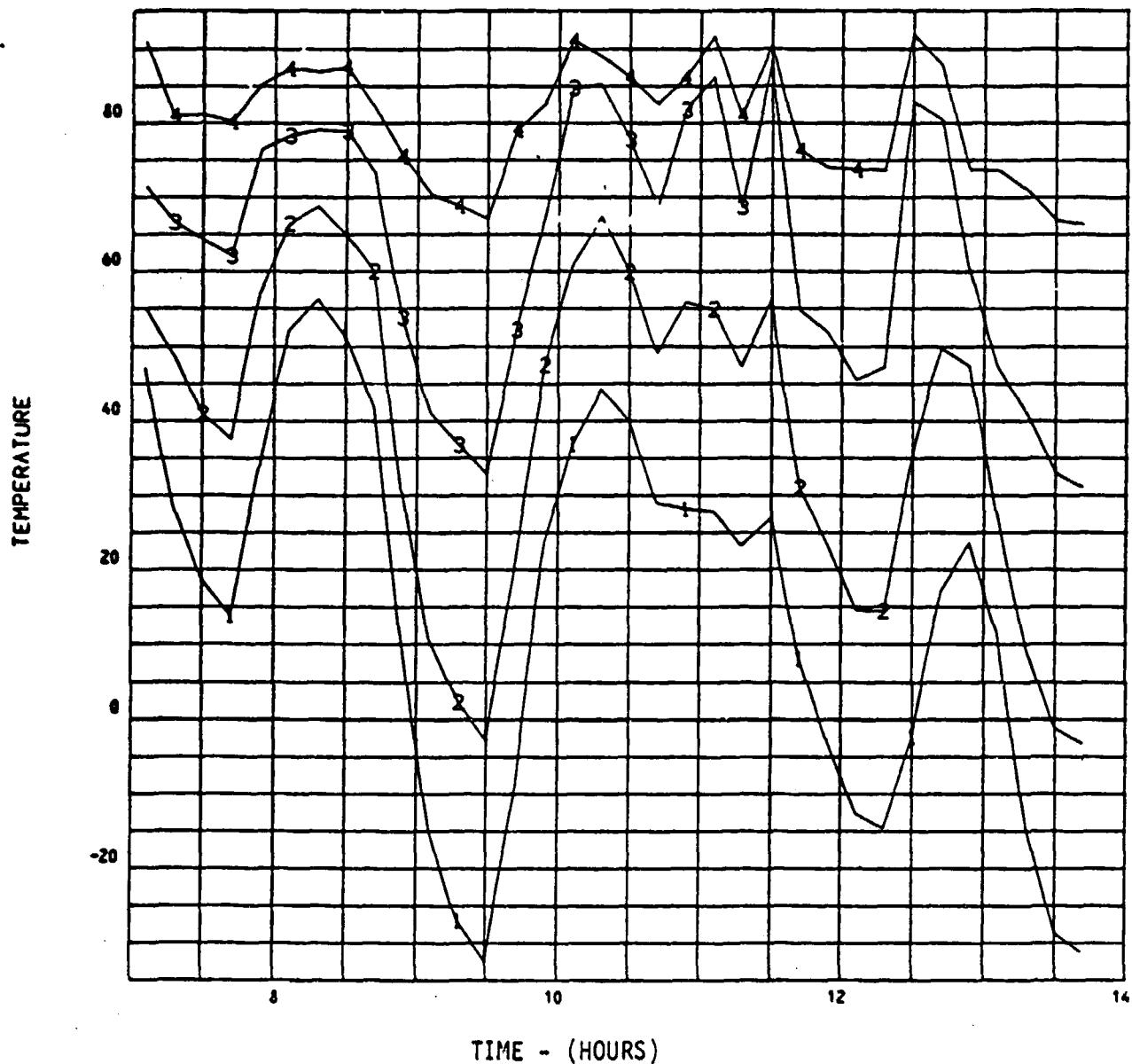


Figure 5

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD BULKHD BOTTOM
[2] FWD BULKHD TOP

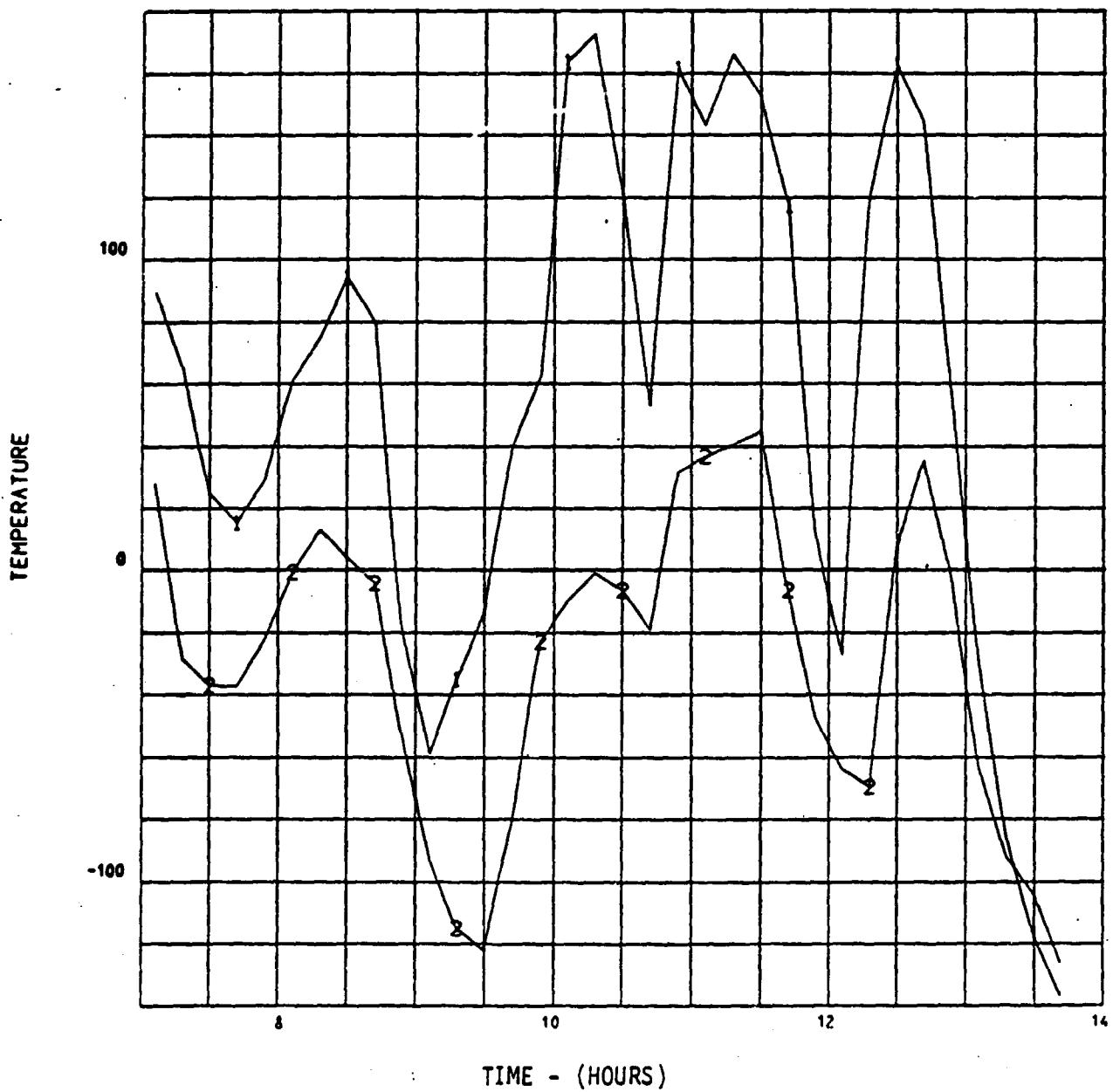


Figure 6

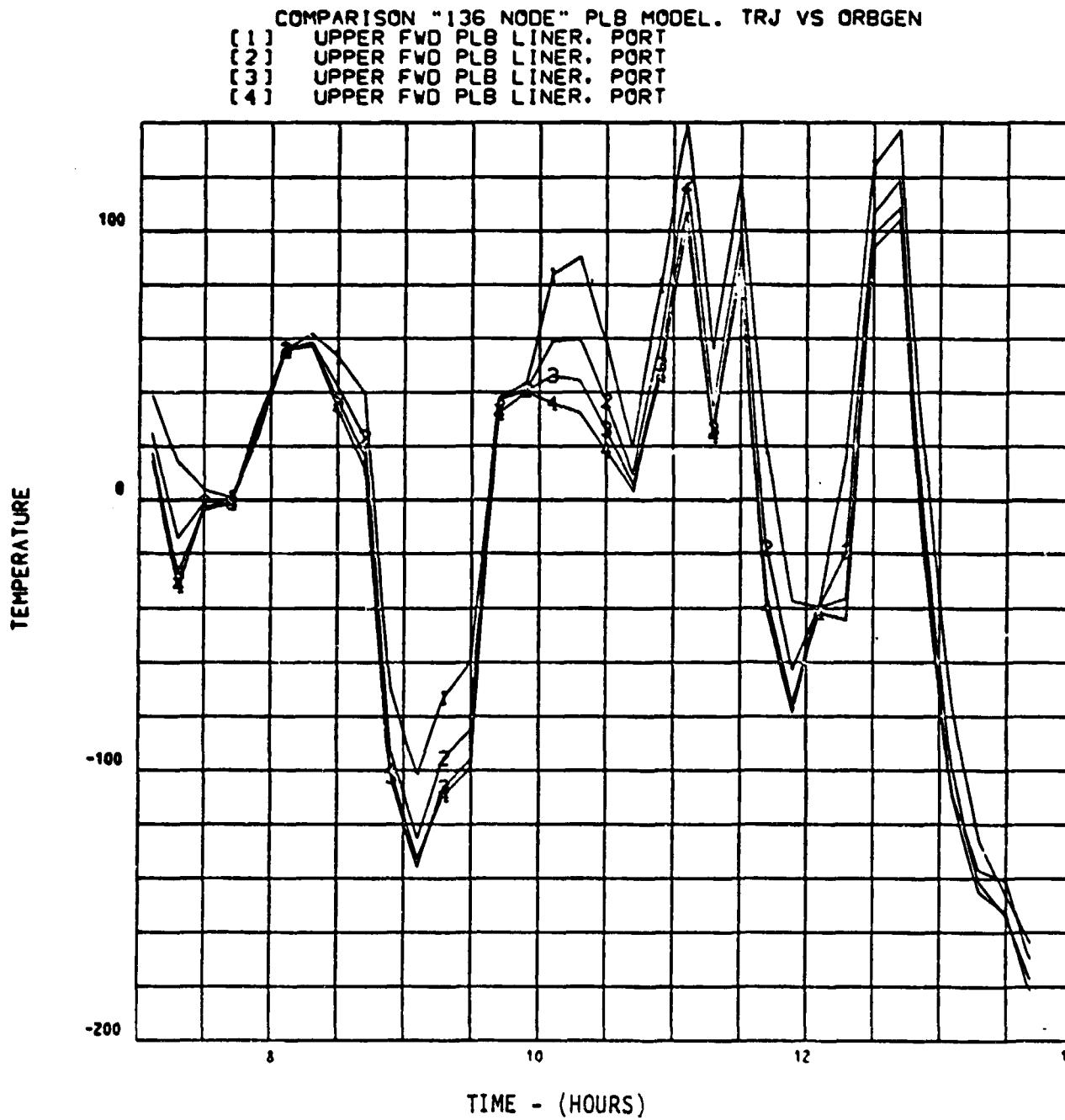


Figure 7

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] UPPER AFT PLB LINER. PORT
[2] UPPER AFT PLB LINER. PORT
[3] UPPER AFT PLB LINER. PORT
[4] UPPER AFT PLB LINER. PORT

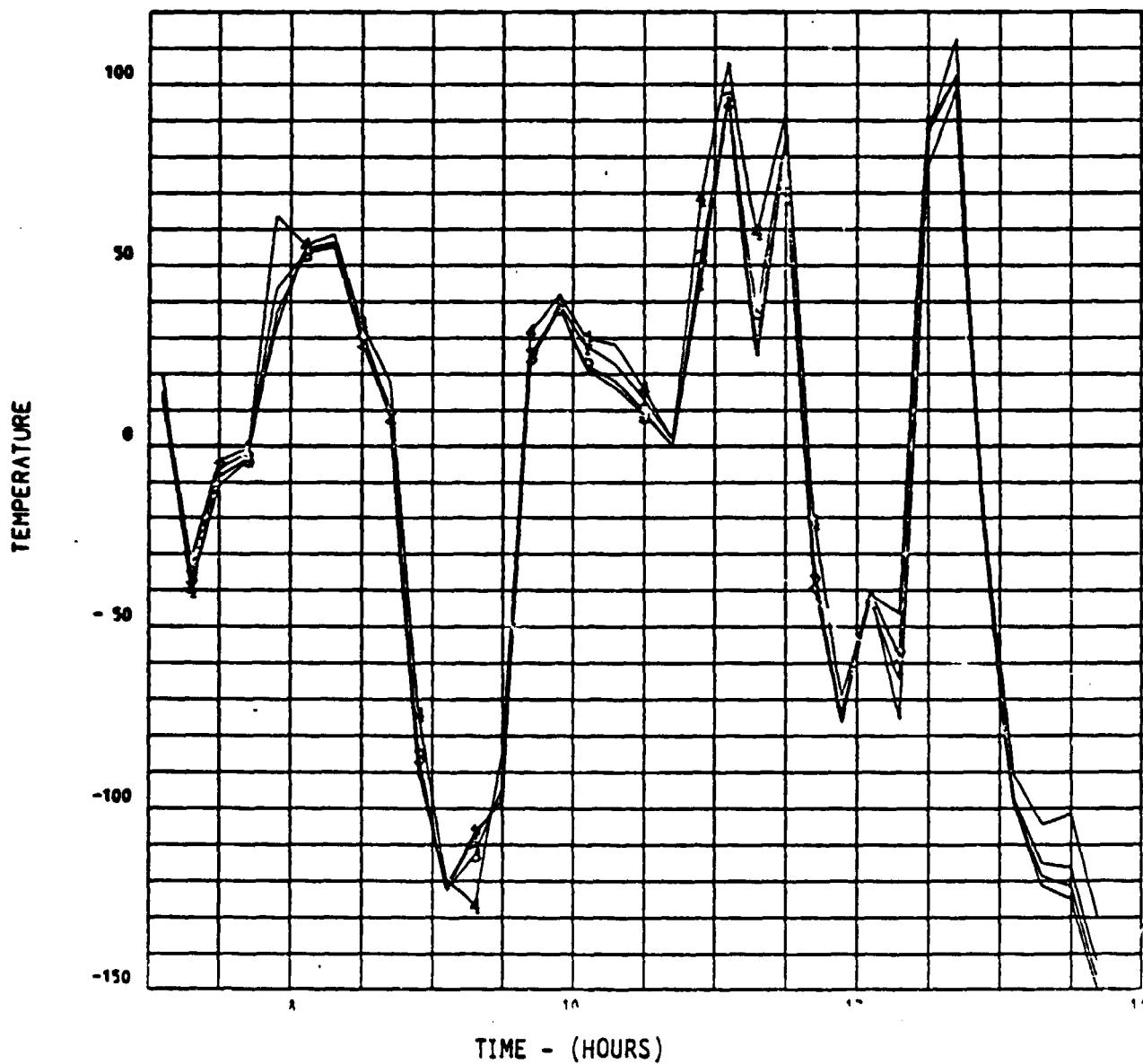
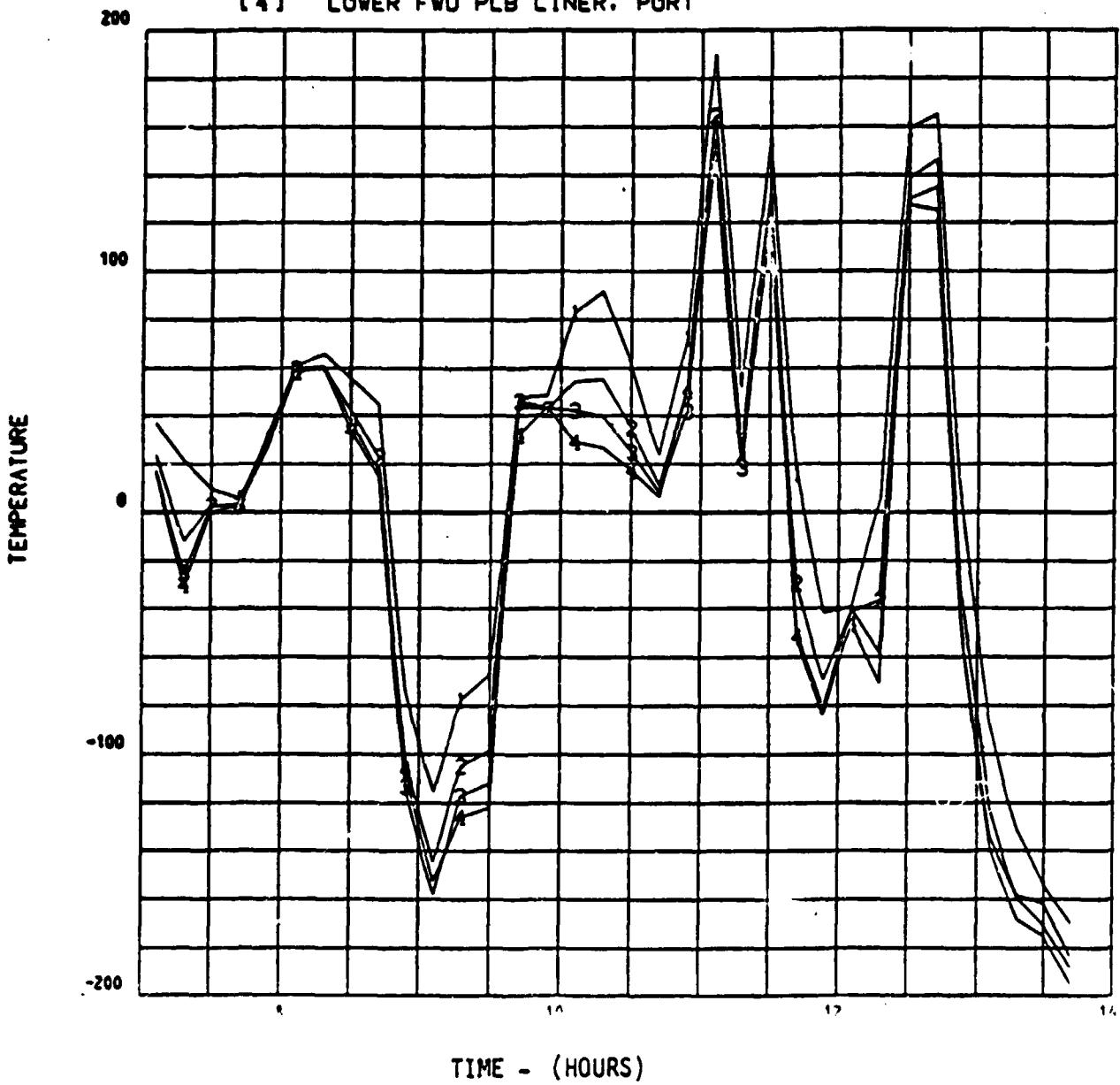


Figure 8

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN

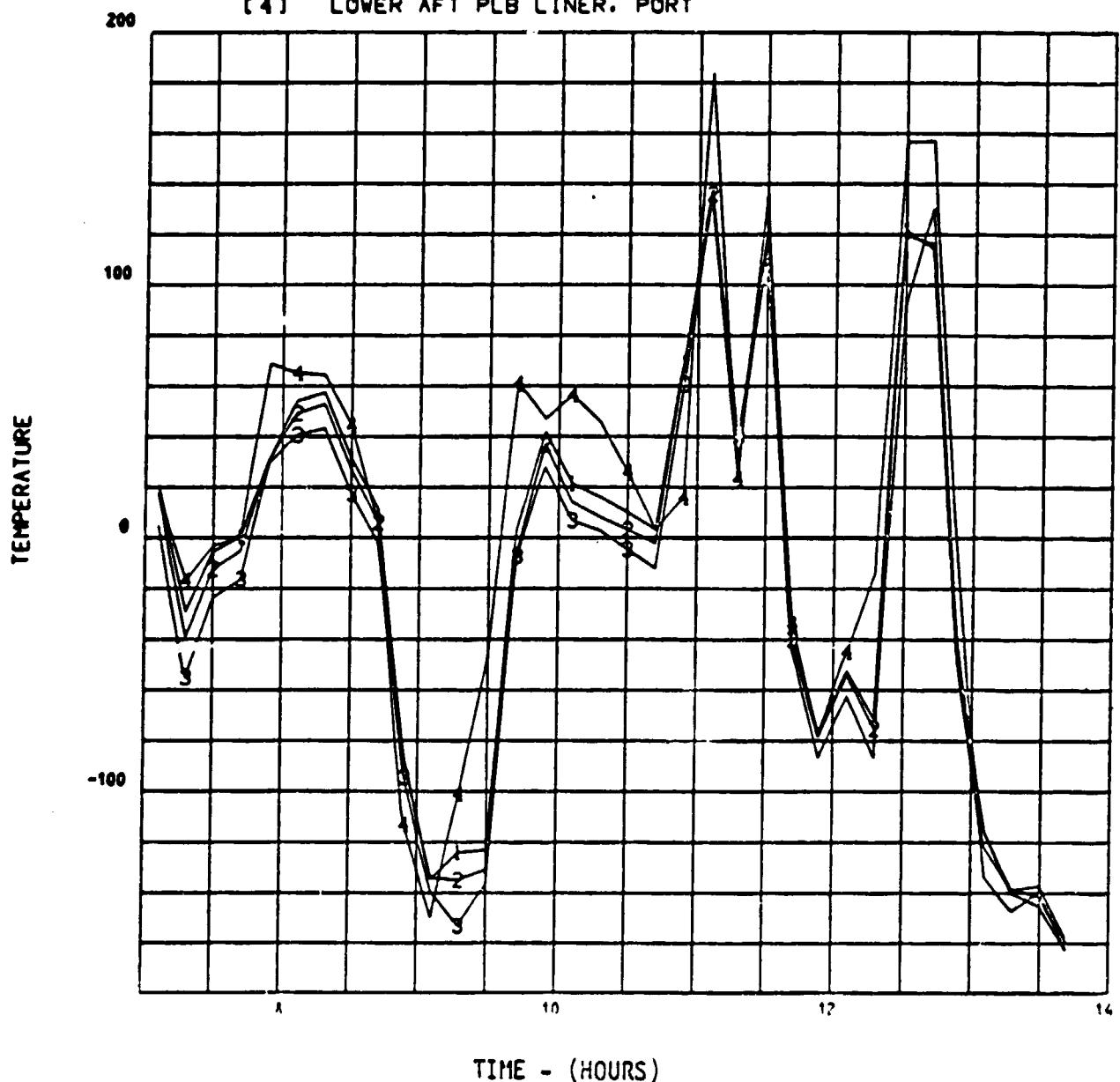
- [1] LOWER FWD PLB LINER. PORT
- [2] LOWER FWD PLB LINER. PORT
- [3] LOWER FWD PLB LINER. PORT
- [4] LOWER FWD PLB LINER. PORT



TIME - (HOURS)

Figure 9

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] LOWER AFT PLB LINER. PORT
[2] LOWER AFT PLB LINER. PORT
[3] LOWER AFT PLB LINER. PORT
[4] LOWER AFT PLB LINER. PORT



TIME - (HOURS)

Figure 10

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
(1) FWD LONGERON. PORT
(2) AFT LONGERON. PORT

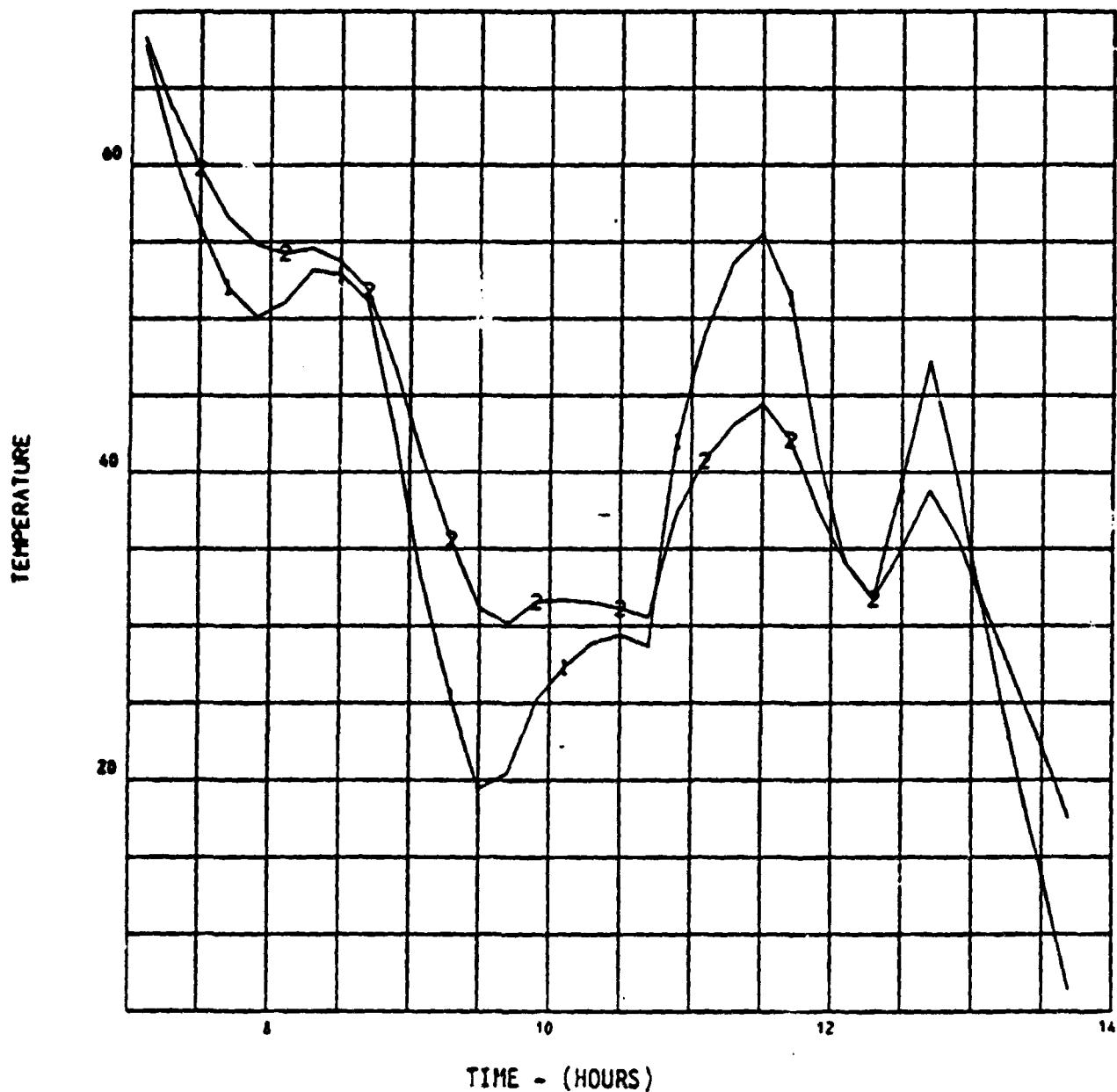


Figure 11

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] AFT BULKHD BOTTOM
[2] AFT BULKHD TOP

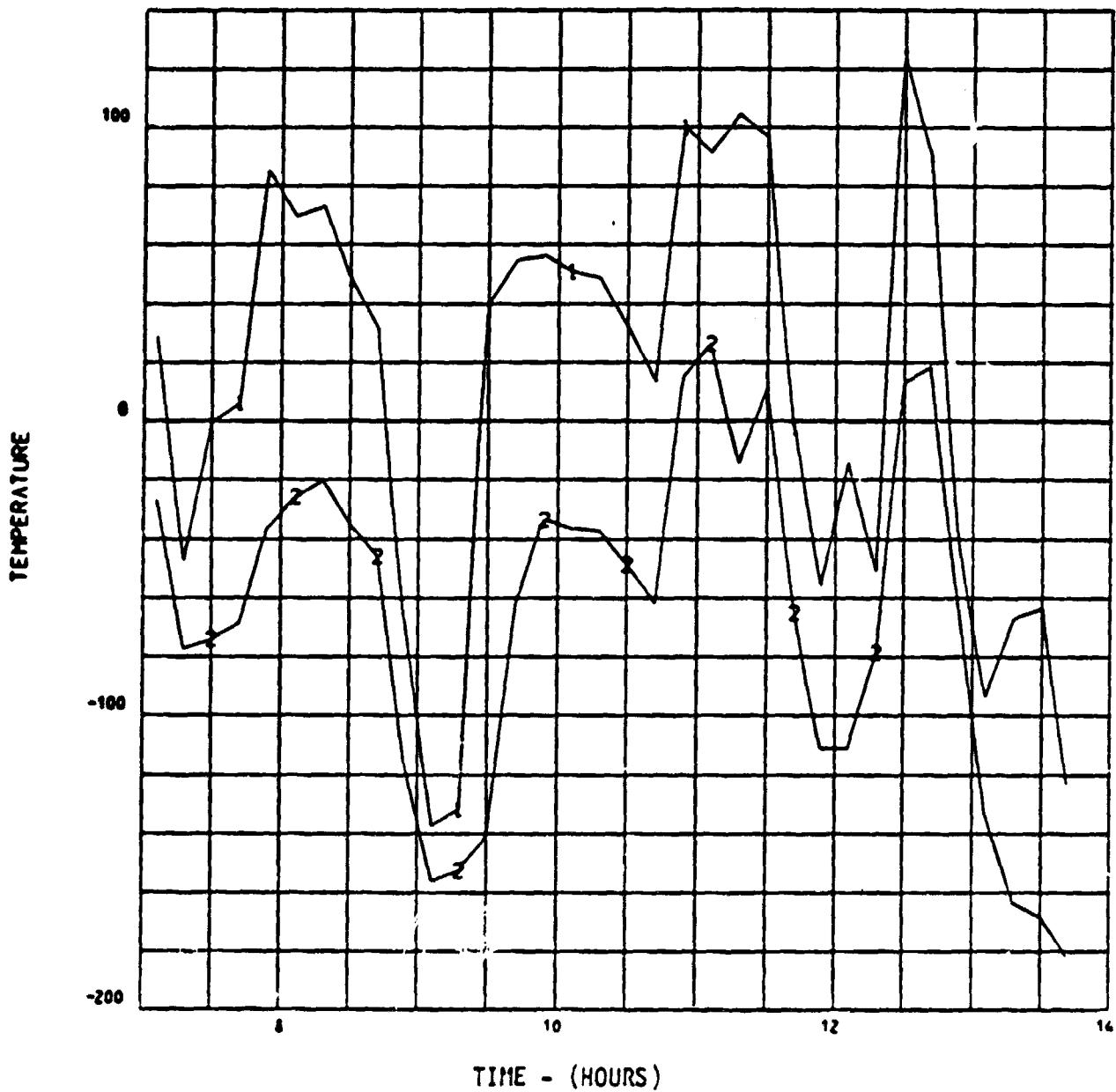


Figure 12

COMPARISON "136 NODE" PLB MODEL. TRJ VS CRBGEN
[1] FWD BULKHO BOTTOM BELOW PLB LINER

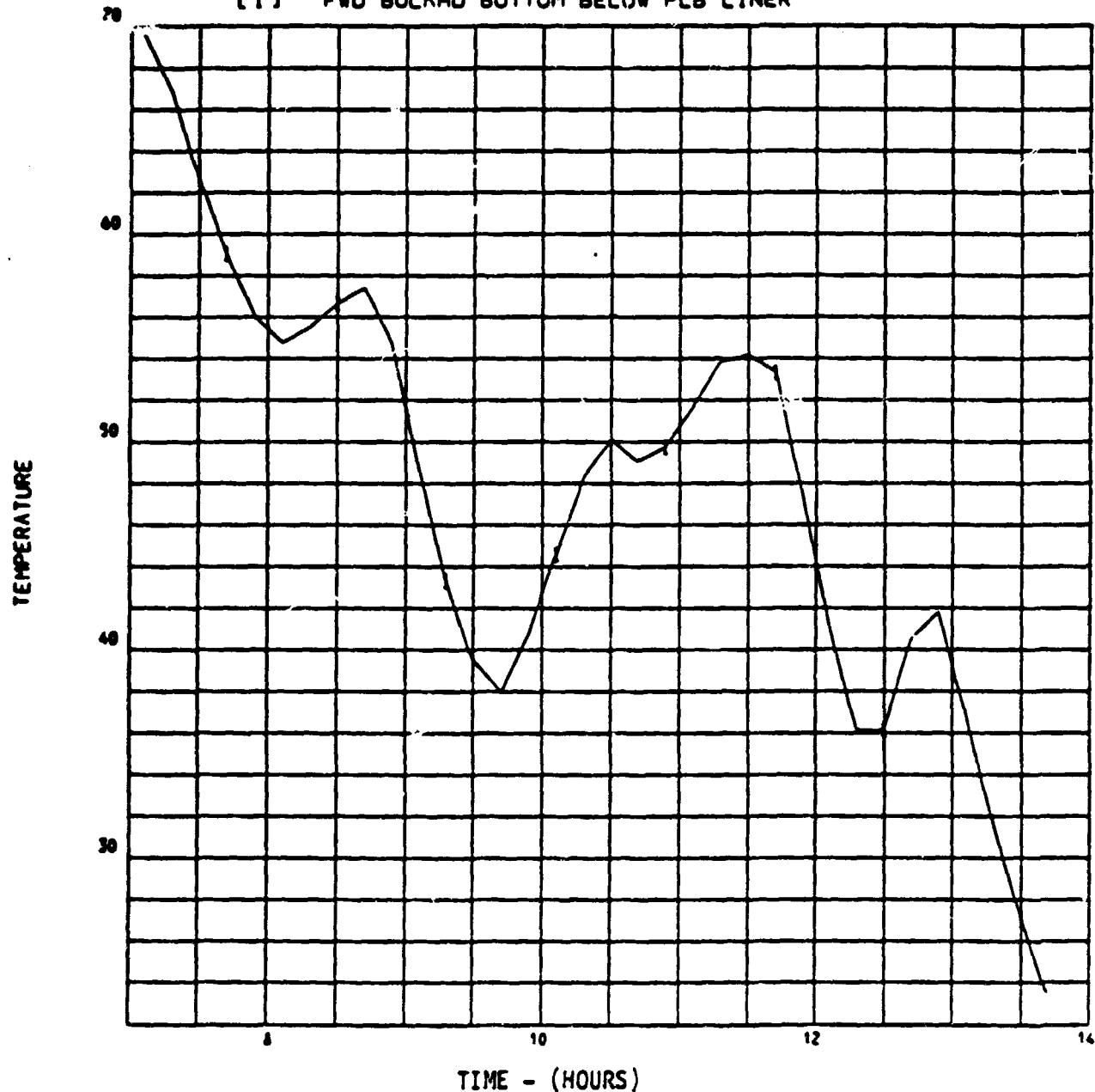


Figure 13

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD SIDE FUSELAGE STRUCTURE. PORT
[2] AFT SIDE FUSELAGE STRUCTURE. PORT

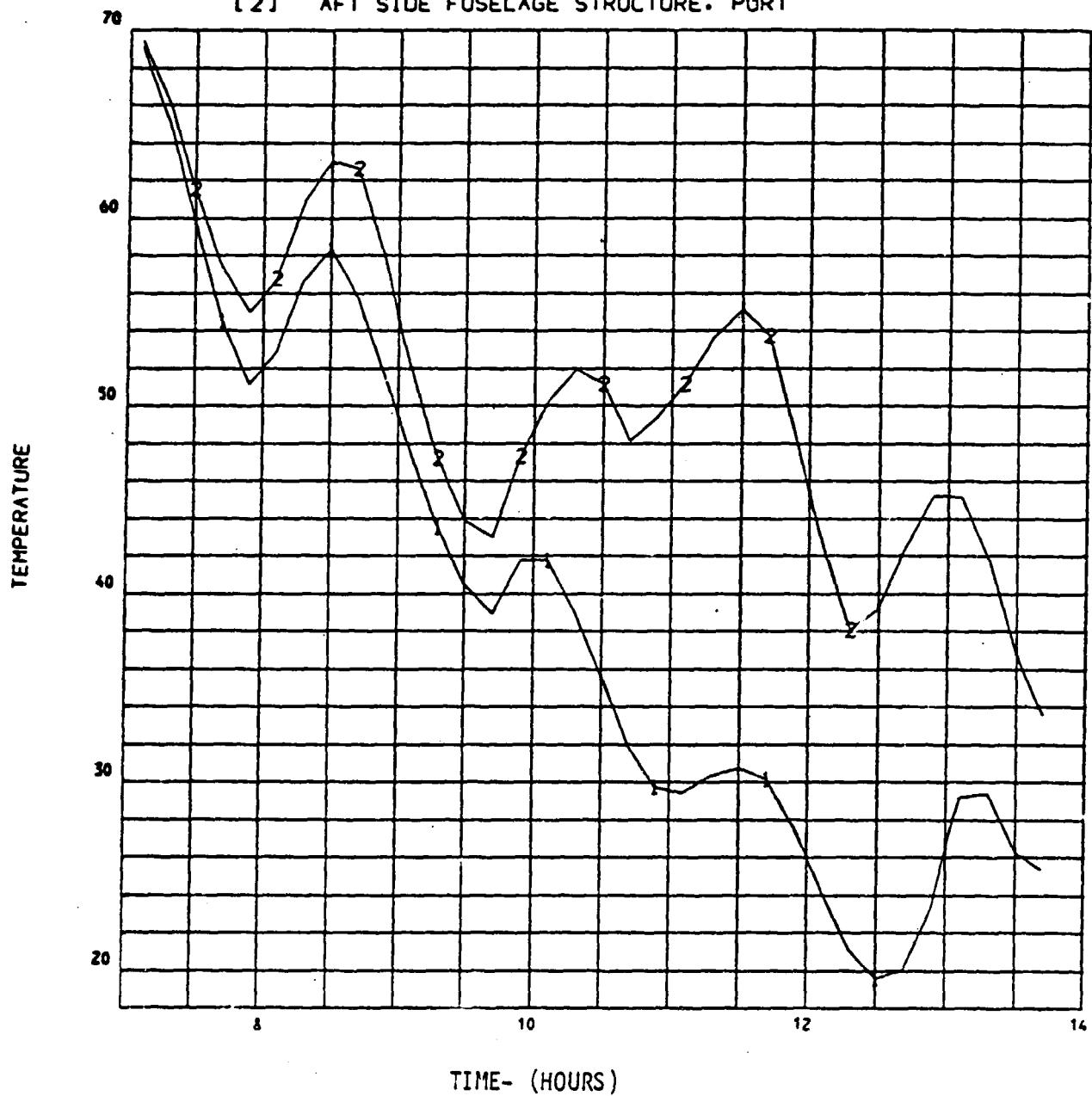


Figure 14

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD BOTTOM FUSELAGE STRUCTURE. PORT
[2] AFT BOTTOM FUSELAGE STRUCTURE. PORT

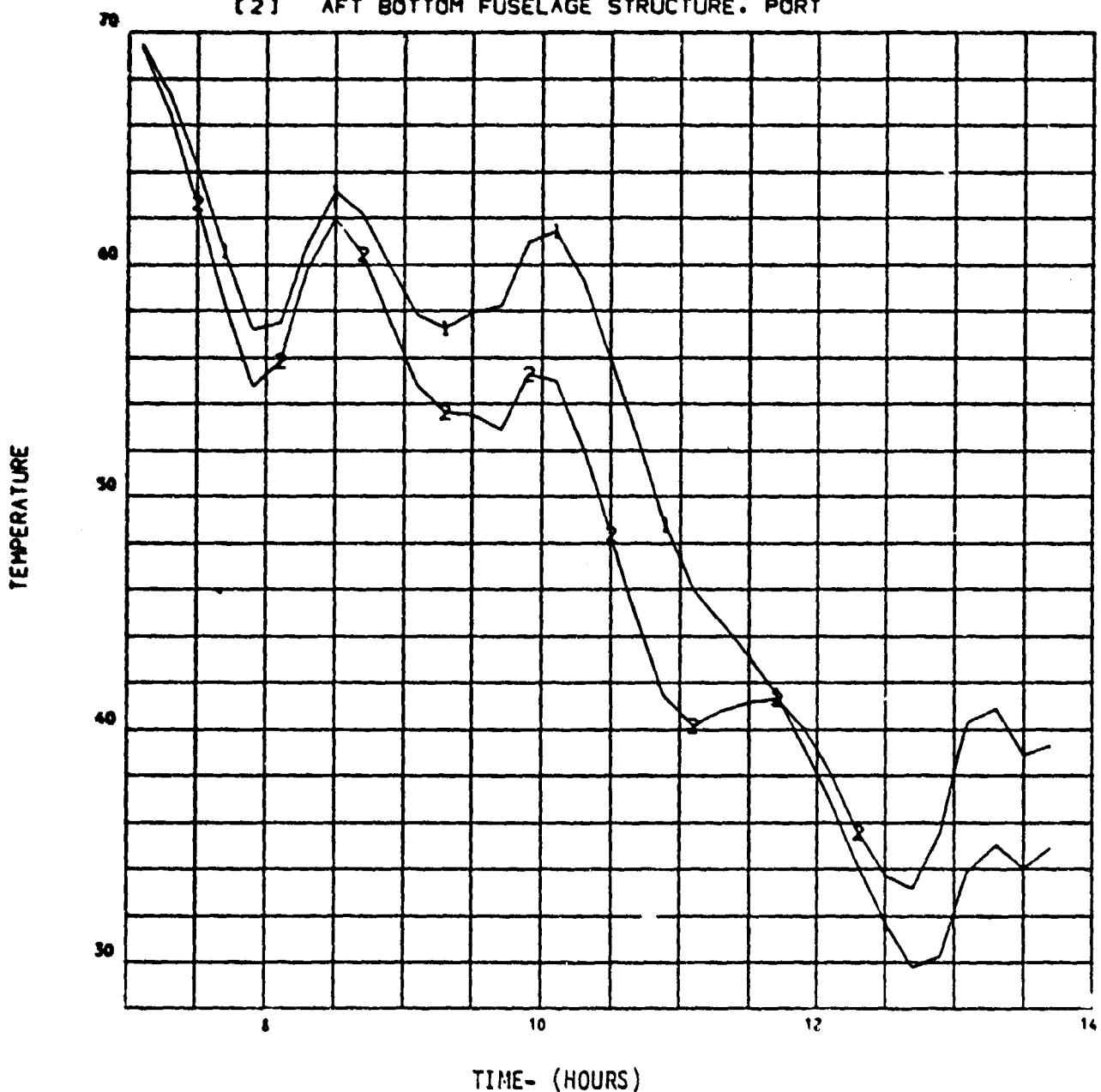


Figure 15

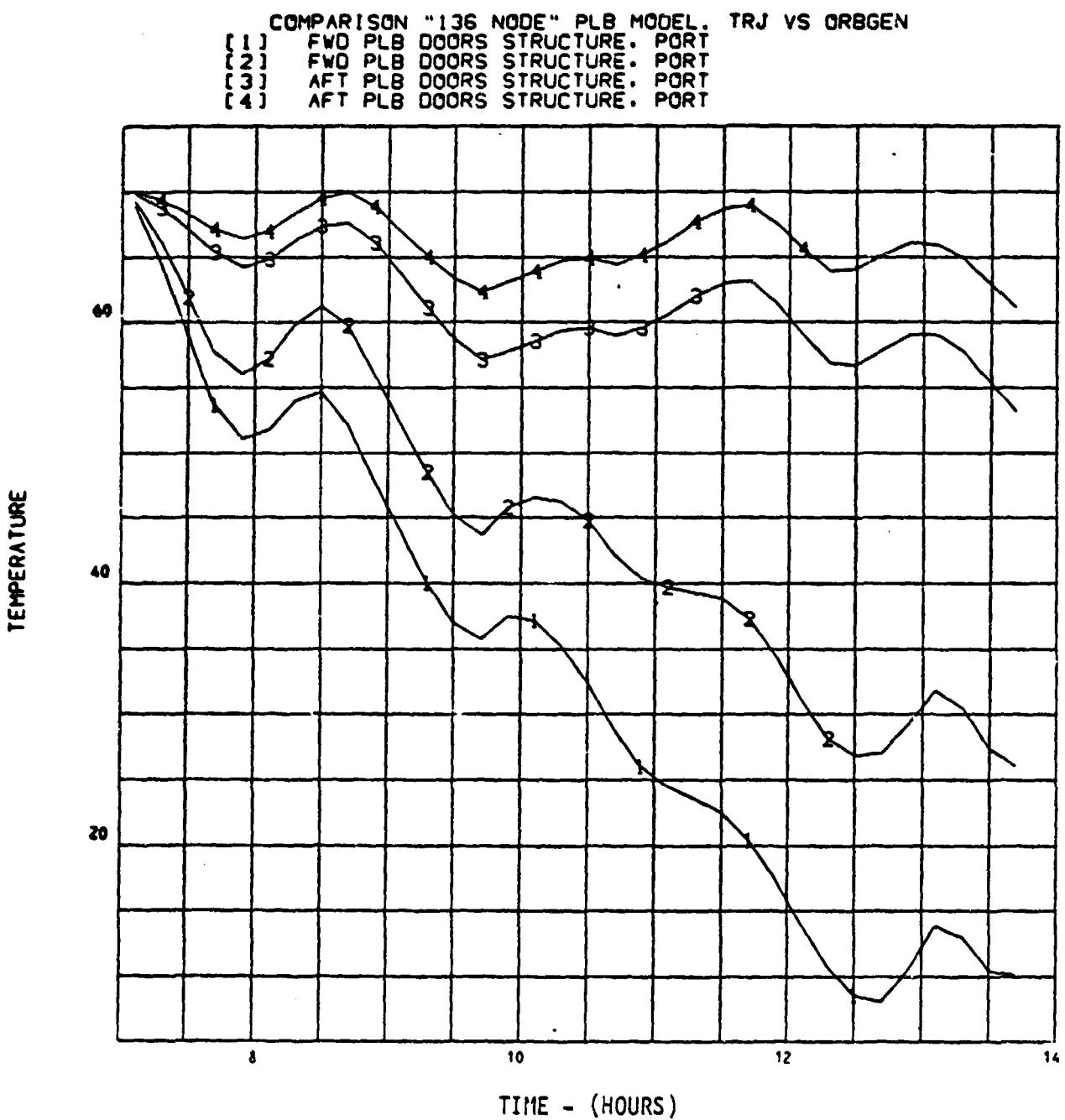


Figure 16

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] AFT BULKHO BOTTOM BELOW PLB LINER

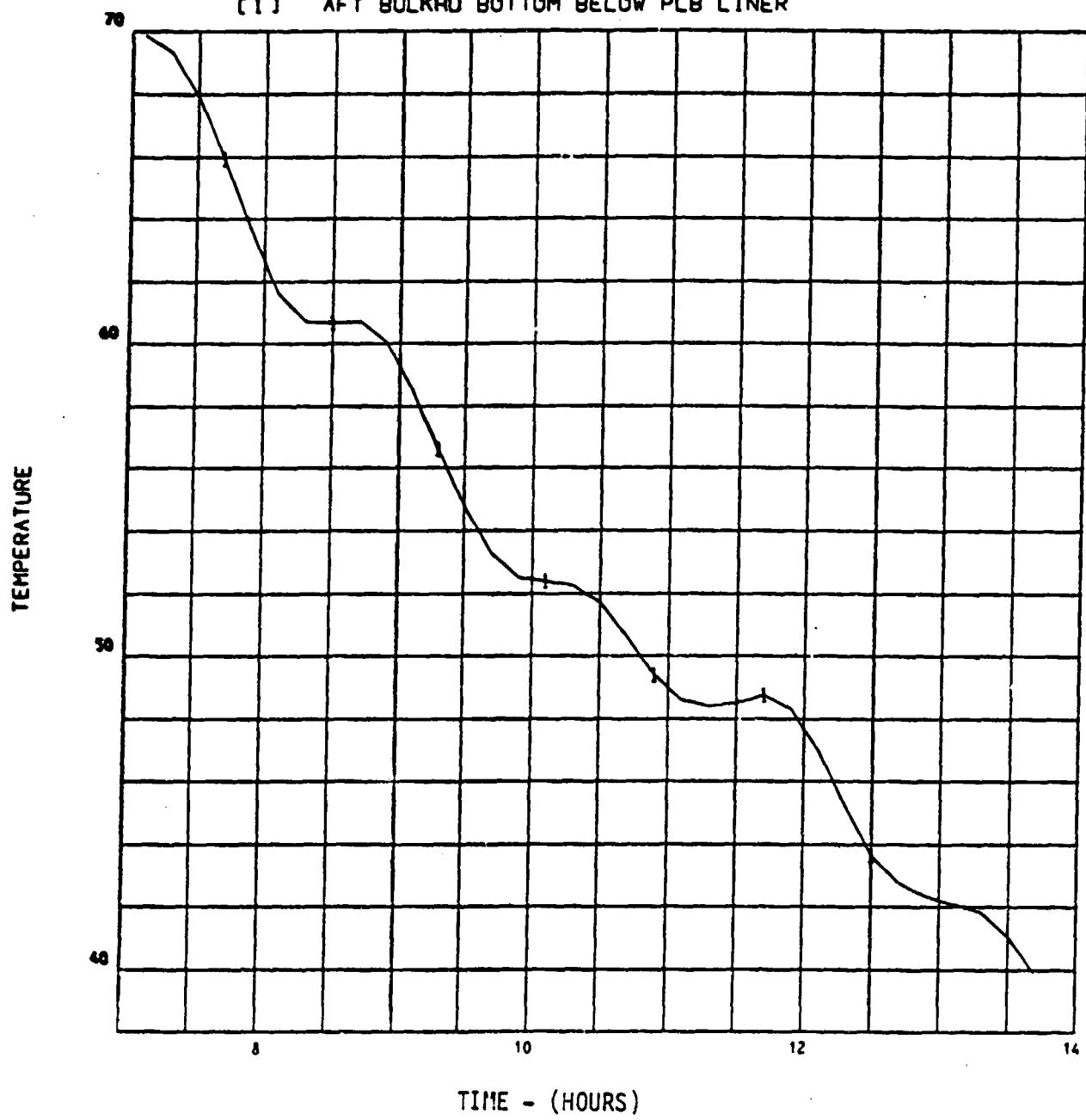
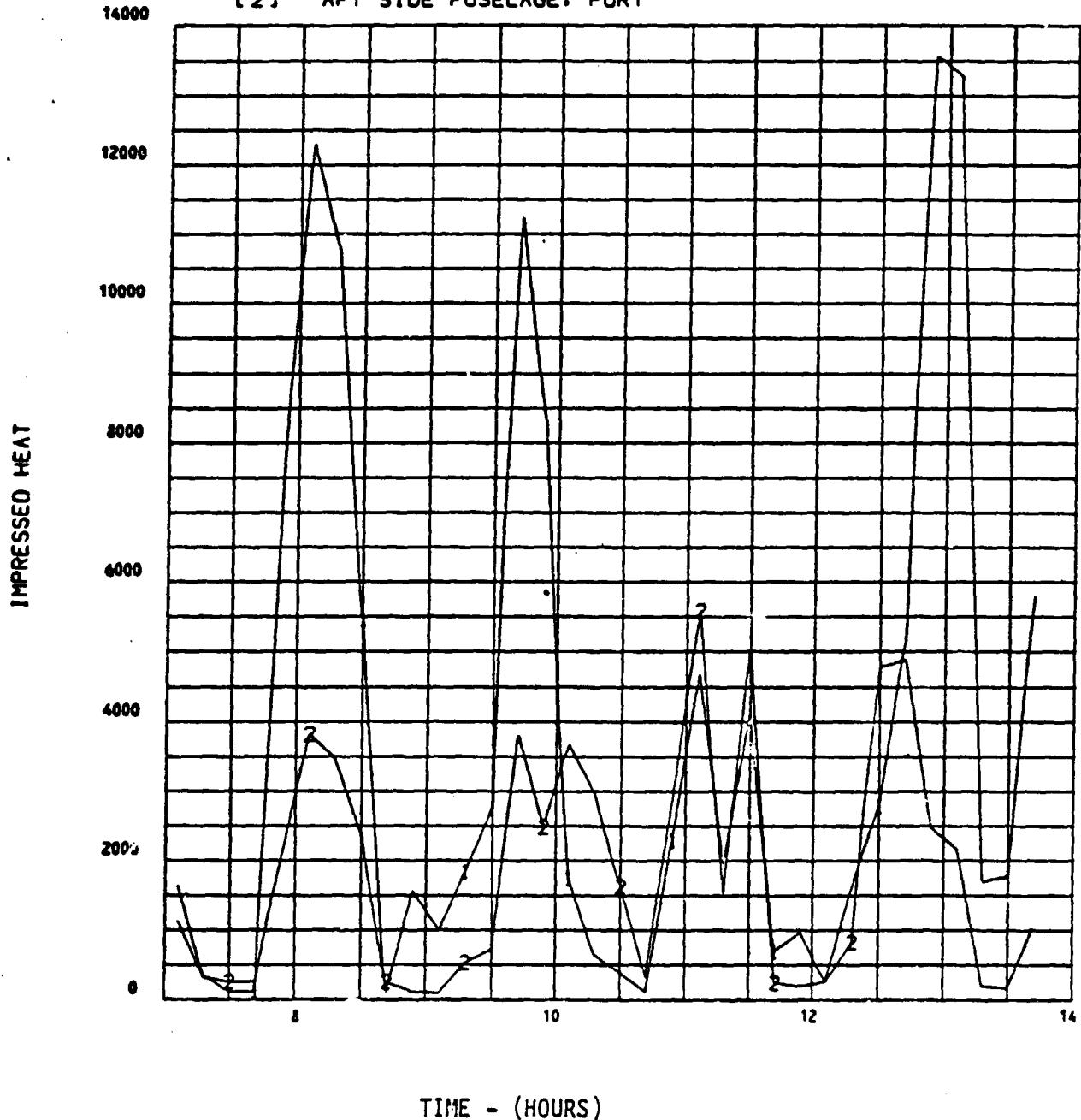


Figure 17

COMPARISON "136 NODE" PLB MODEL. TRJ VS OR8GEN
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT



TIME - (HOURS)

Figure 18

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

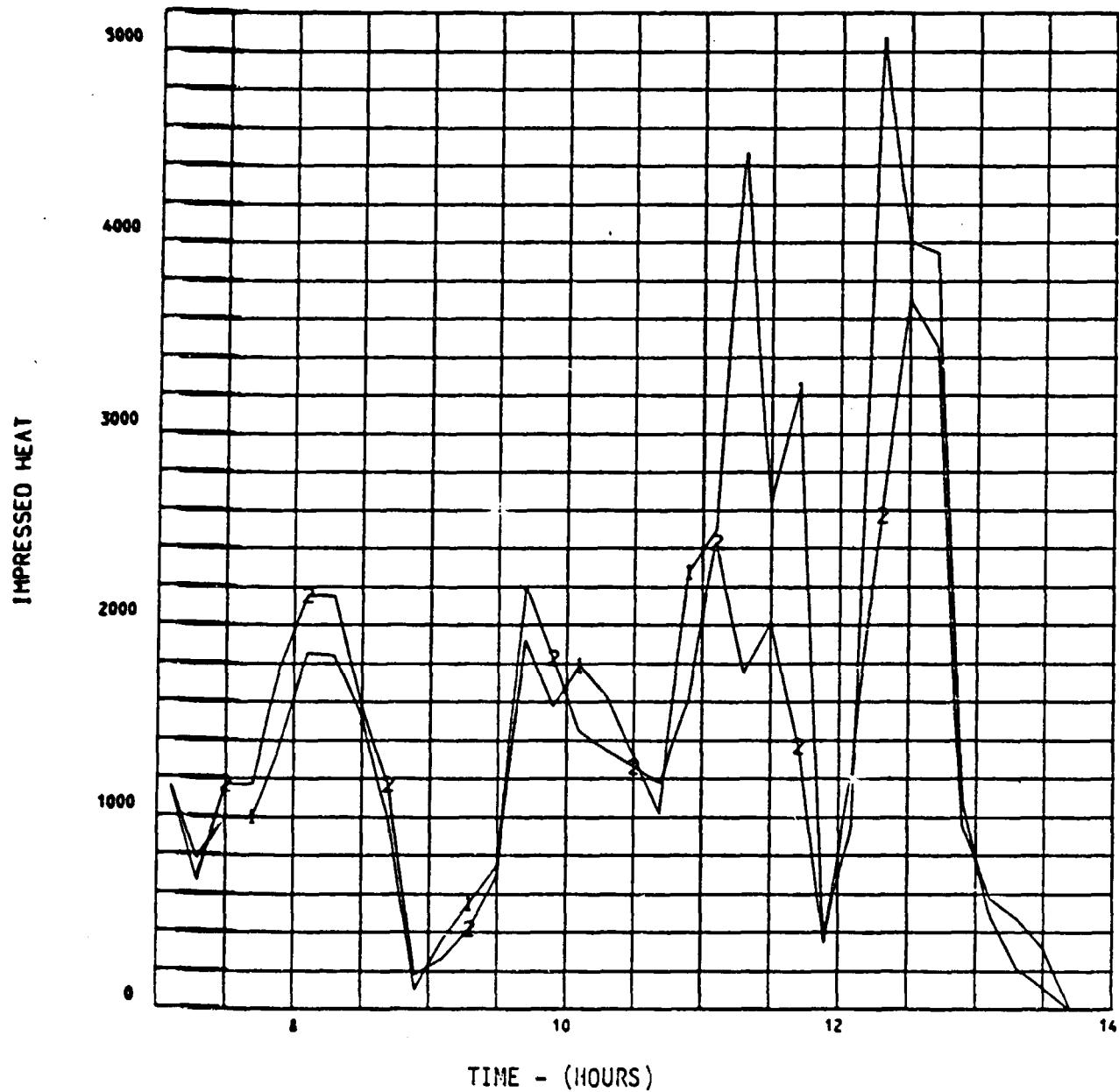


Figure 19

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD BOTTOM FUSELAGE. PORT
[2] AFT BOTTOM FUSELAGE. PORT

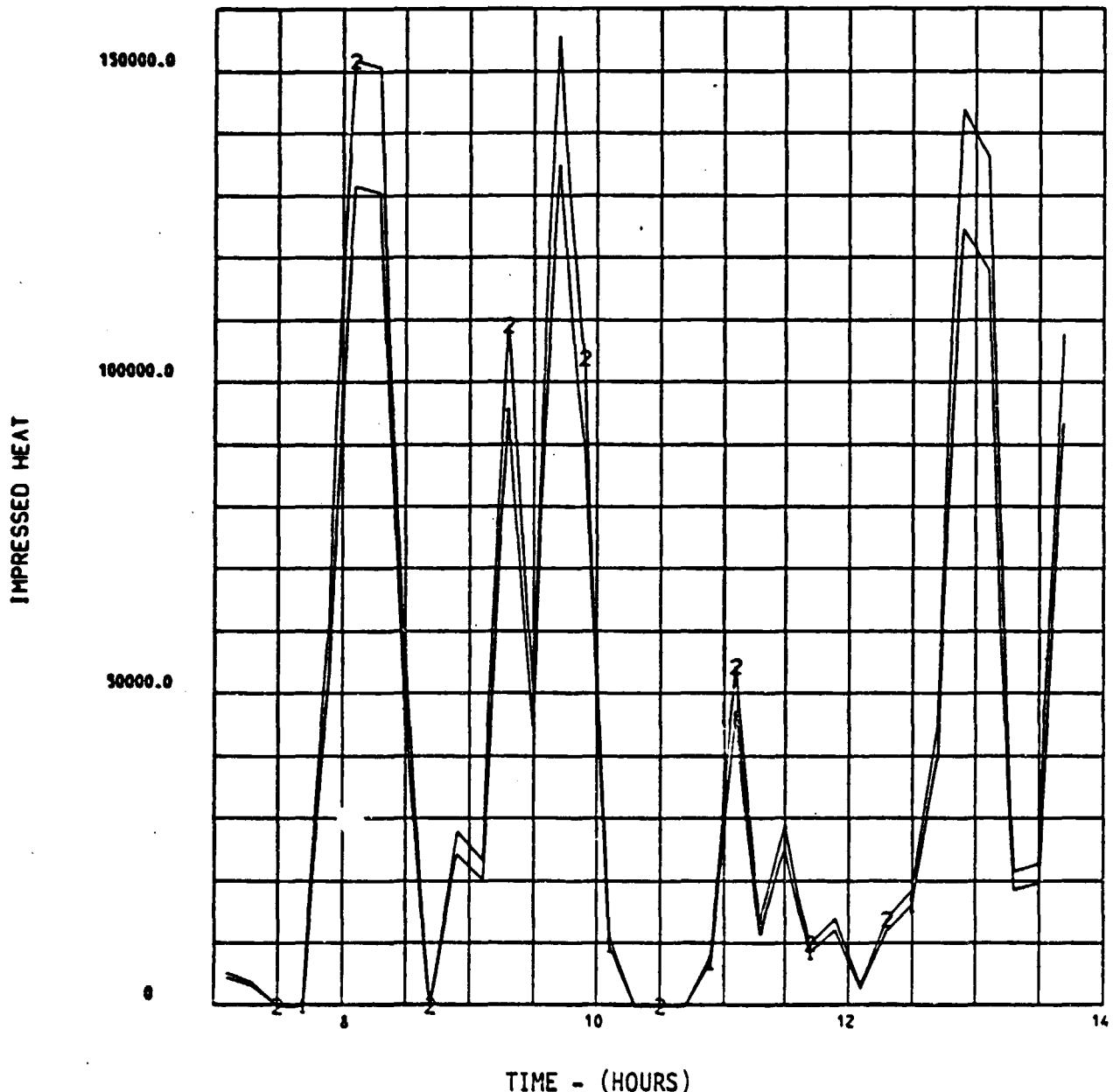


Figure 20

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN

- [1] FWD PLB DOORS. PORT
- [2] FWD PLB DOORS. PORT
- [3] AFT PLB DOORS. PORT
- [4] AFT PLB DOORS. PORT

IMPOSED HEAT

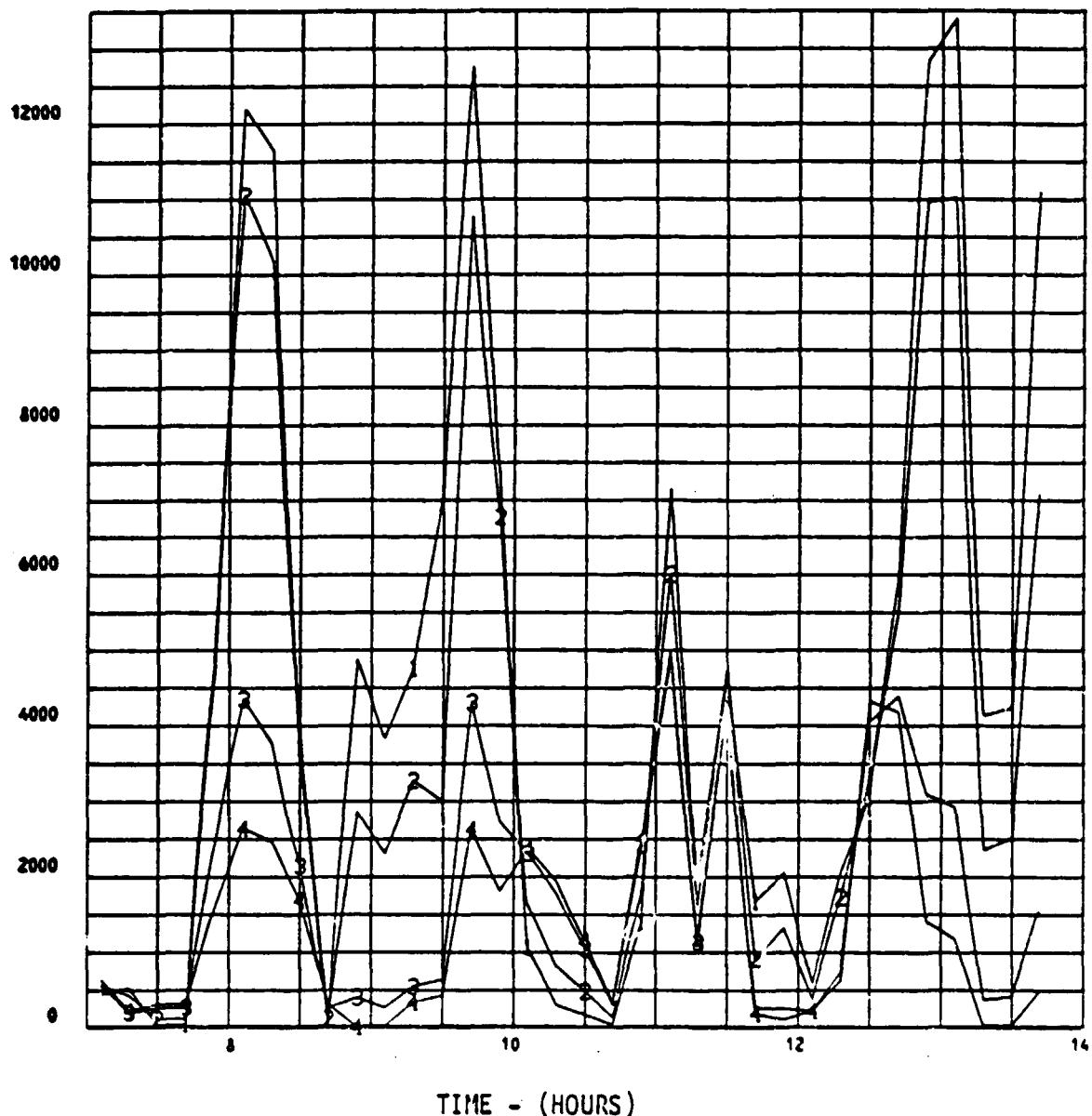


Figure 21

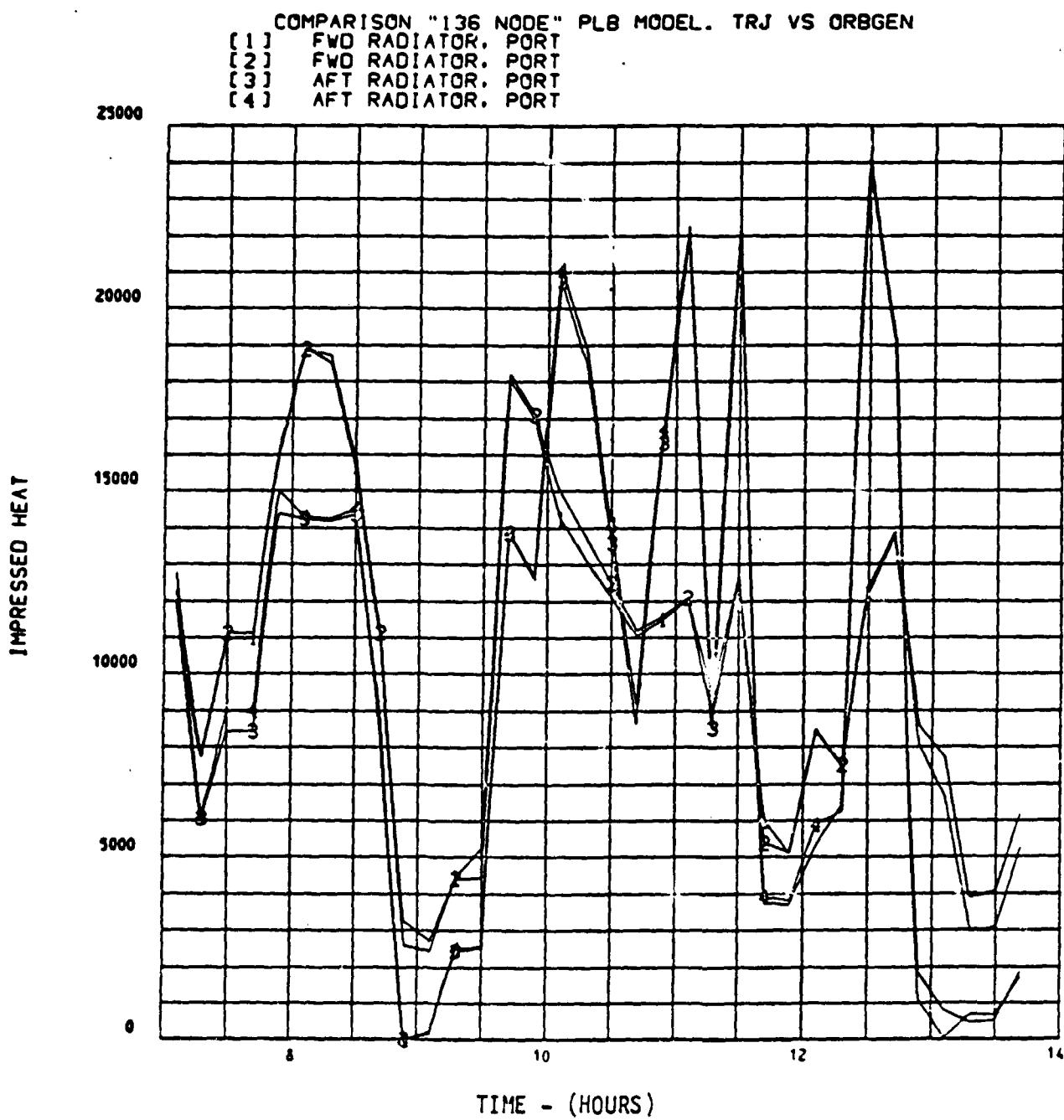
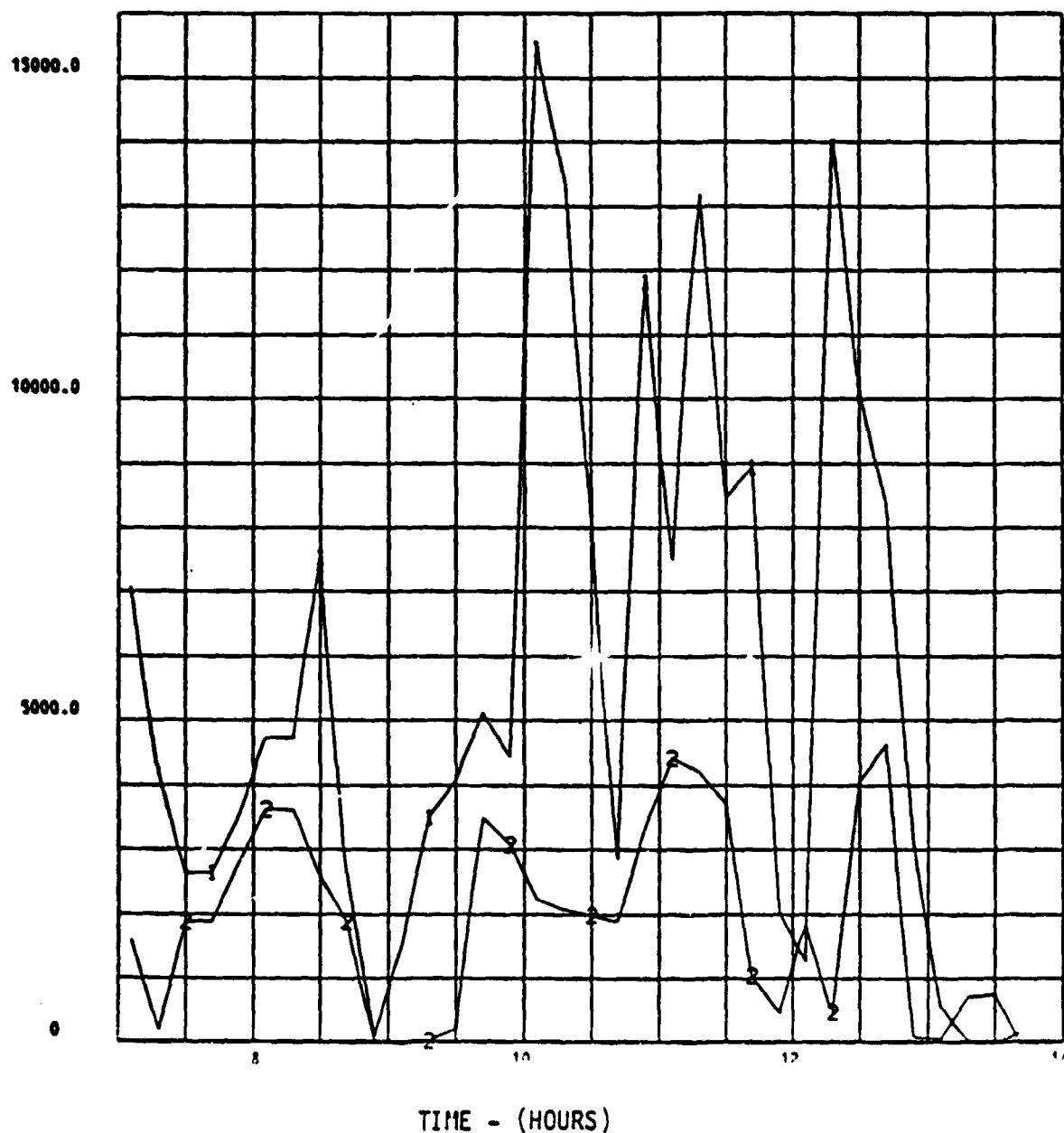


Figure 22

IMPOSED HEAT

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] FWD BULKHO TOP
[2] FWD BULKHO BOTTOM



TIME - (HOURS)

Figure 23

IMPOSED HEAT

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] UPPER FWD PLB LINER. PORT
[2] UPPER FWD PLB LINER. PORT
[3] UPPER FWD PLB LINER. PORT
[4] UPPER FWD PLB LINER. PORT

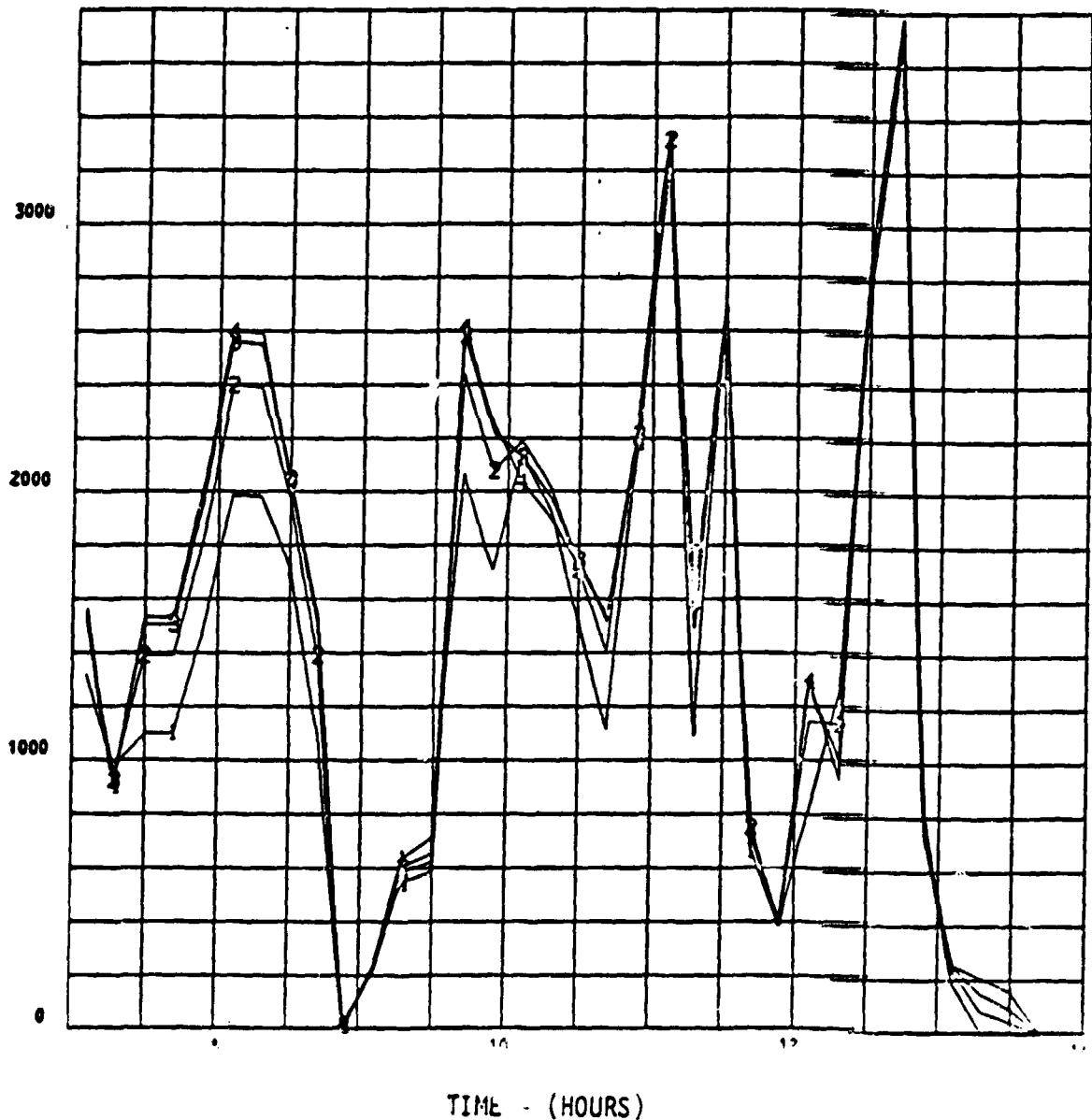


Figure 24

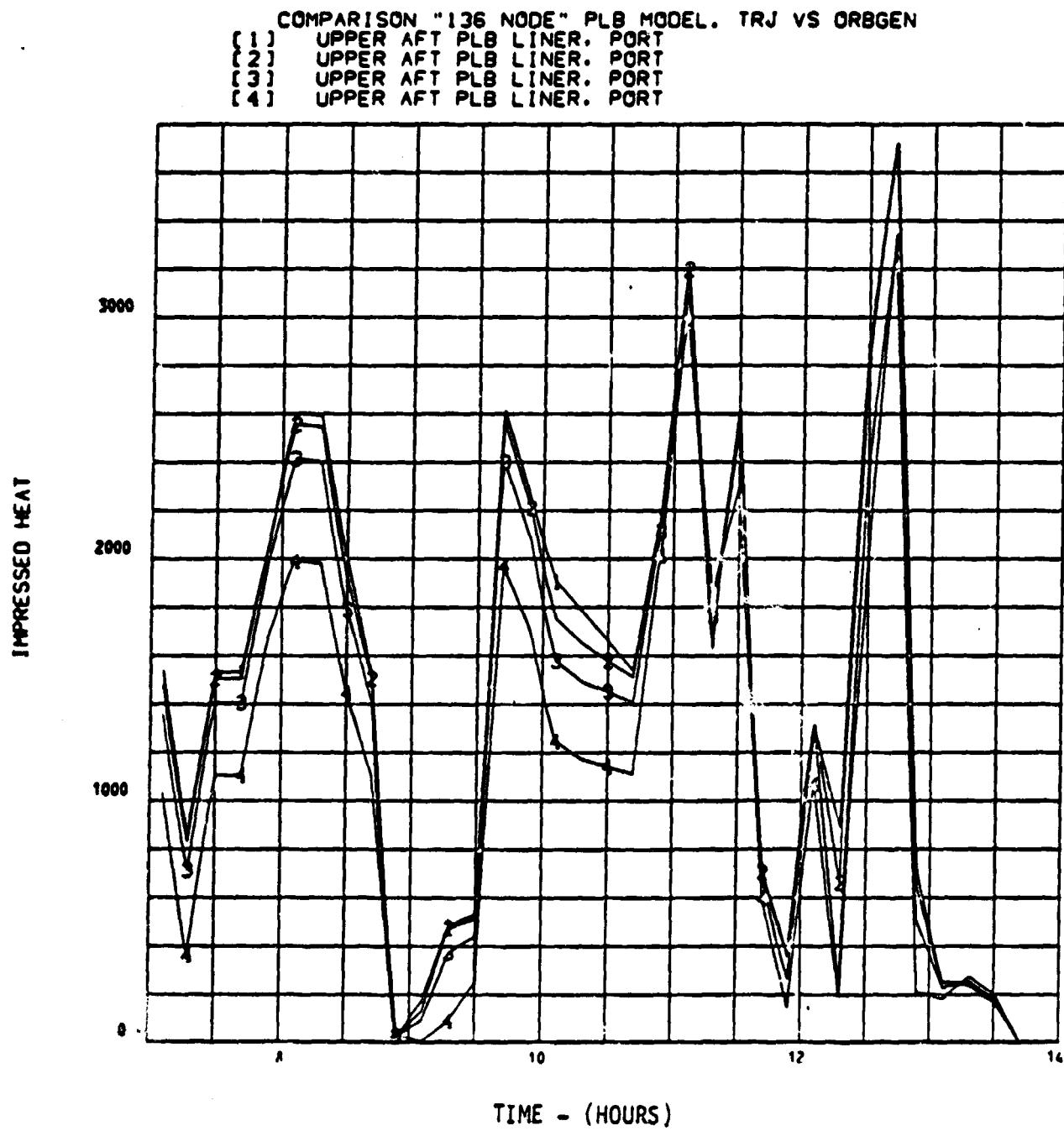


Figure 25

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN

- [1] LOWER FWD PLB LINER. PORT
- [2] LOWER FWD PLB LINER. PORT
- [3] LOWER FWD PLB LINER. PORT
- [4] LOWER FWD PLB LINER. PORT

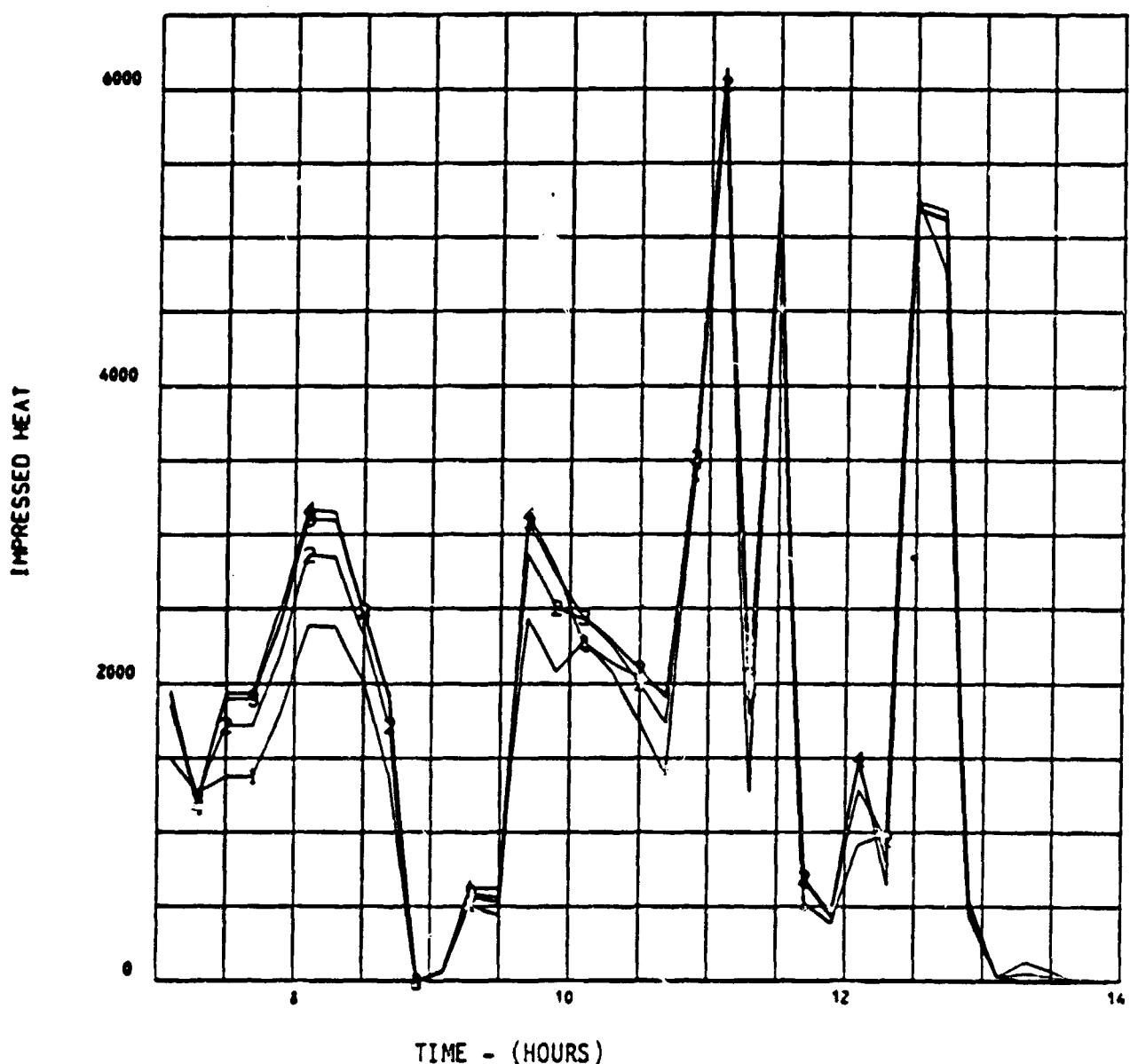


Figure 26

COMPARISON "136 NODE" PLB MODEL. TRJ VS ORBGEN
[1] LOWER AFT PLB LINER. PORT
[2] LOWER AFT PLB LINER. PORT
[3] LOWER AFT PLB LINER. PORT
[4] LOWER AFT PLB LINER. PORT

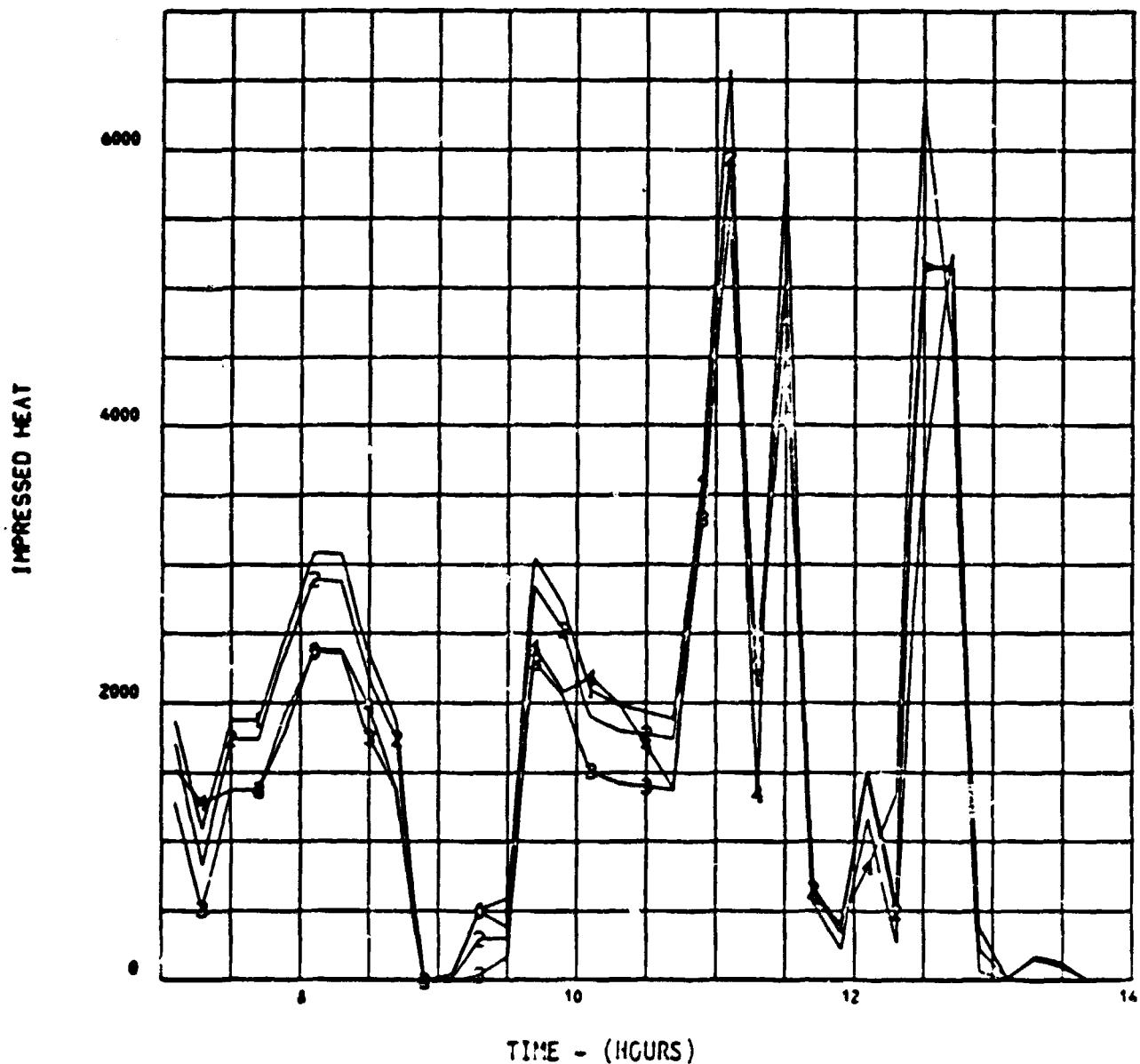


Figure 27

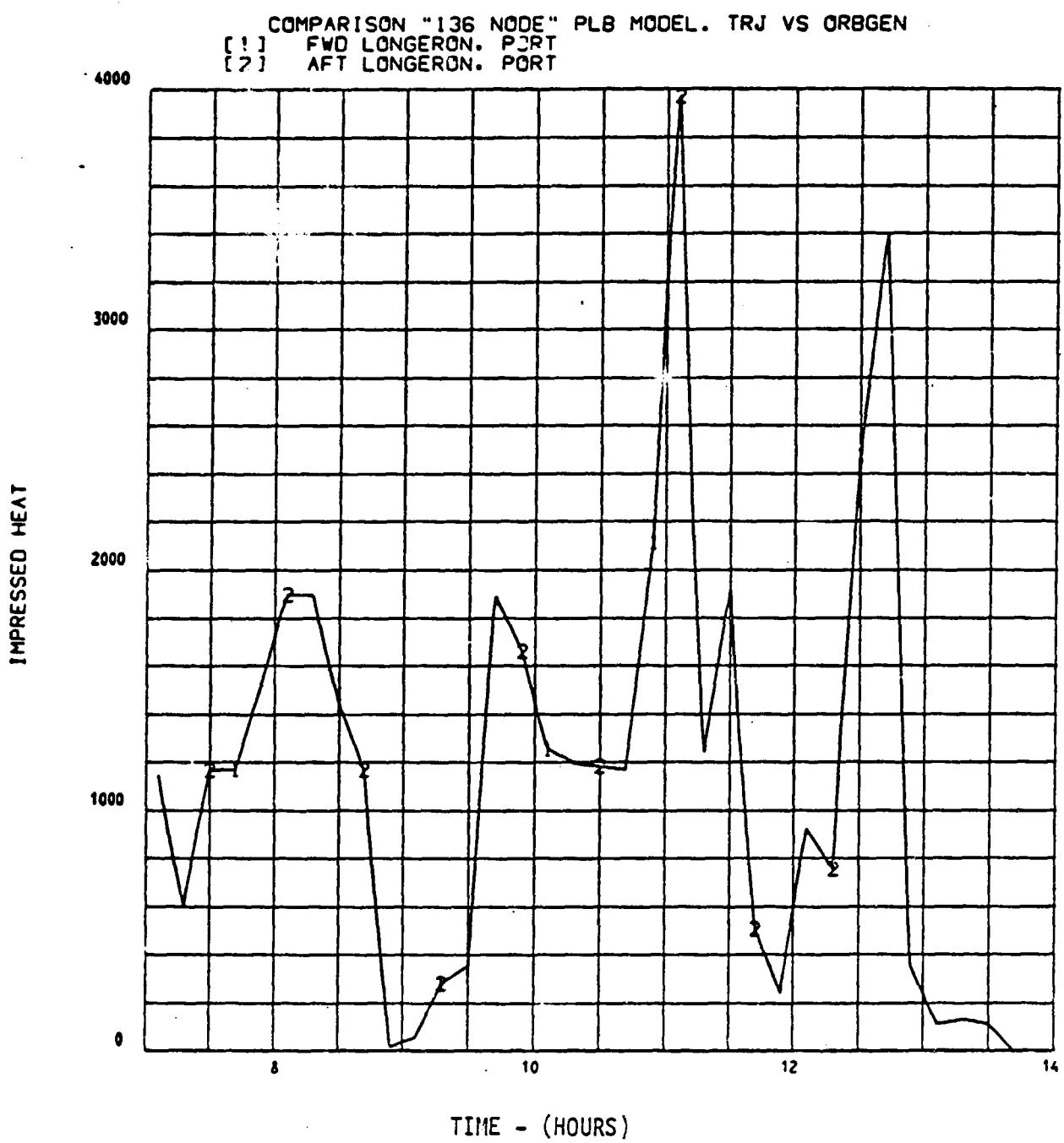
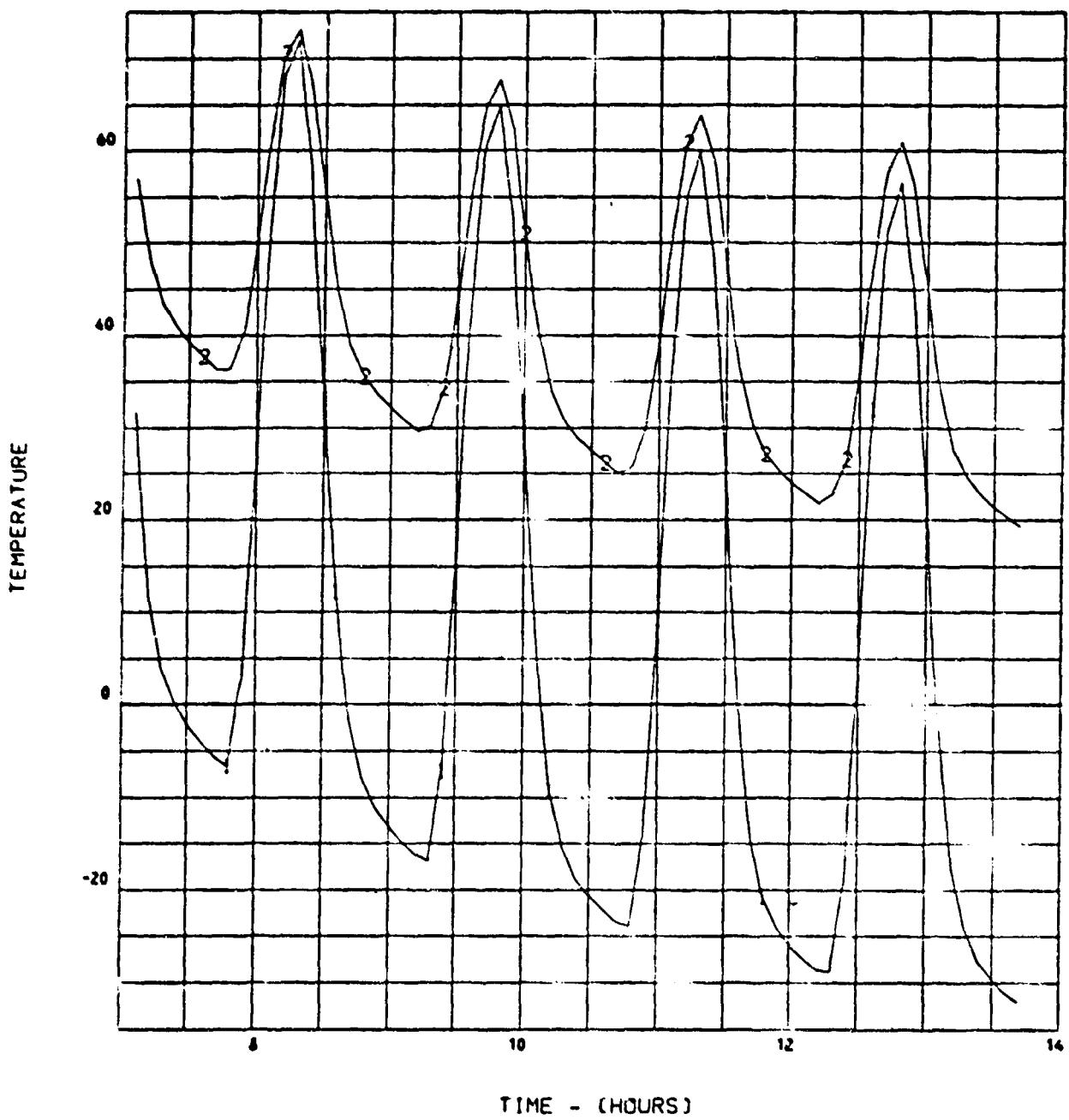


Figure 28

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT



TIME - (HOURS)

Figure 29

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

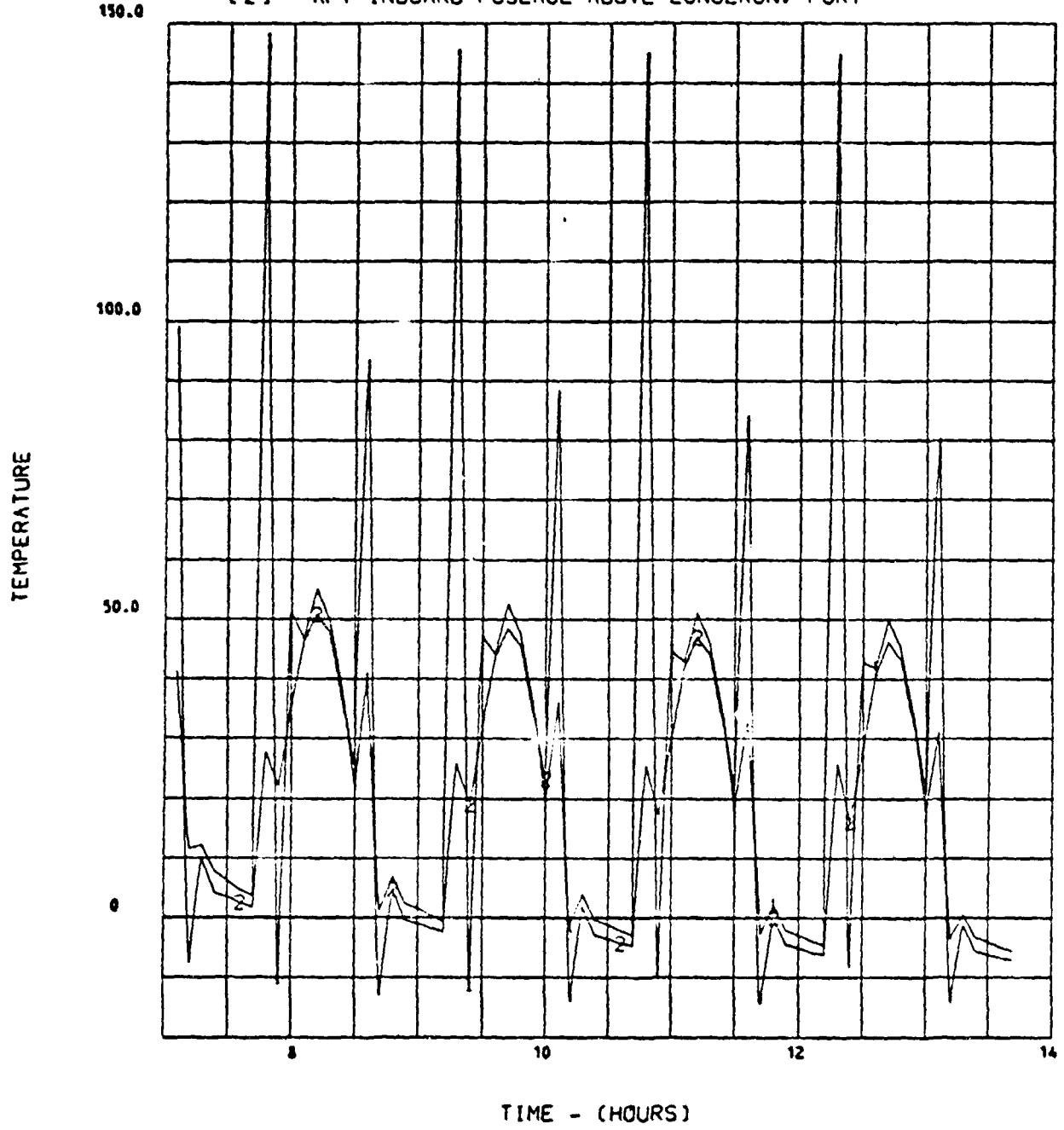


Figure 30

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD BOTTOM FUSELAGE. PORT
[2] AFT BOTTOM FUSELAGE. PCRT

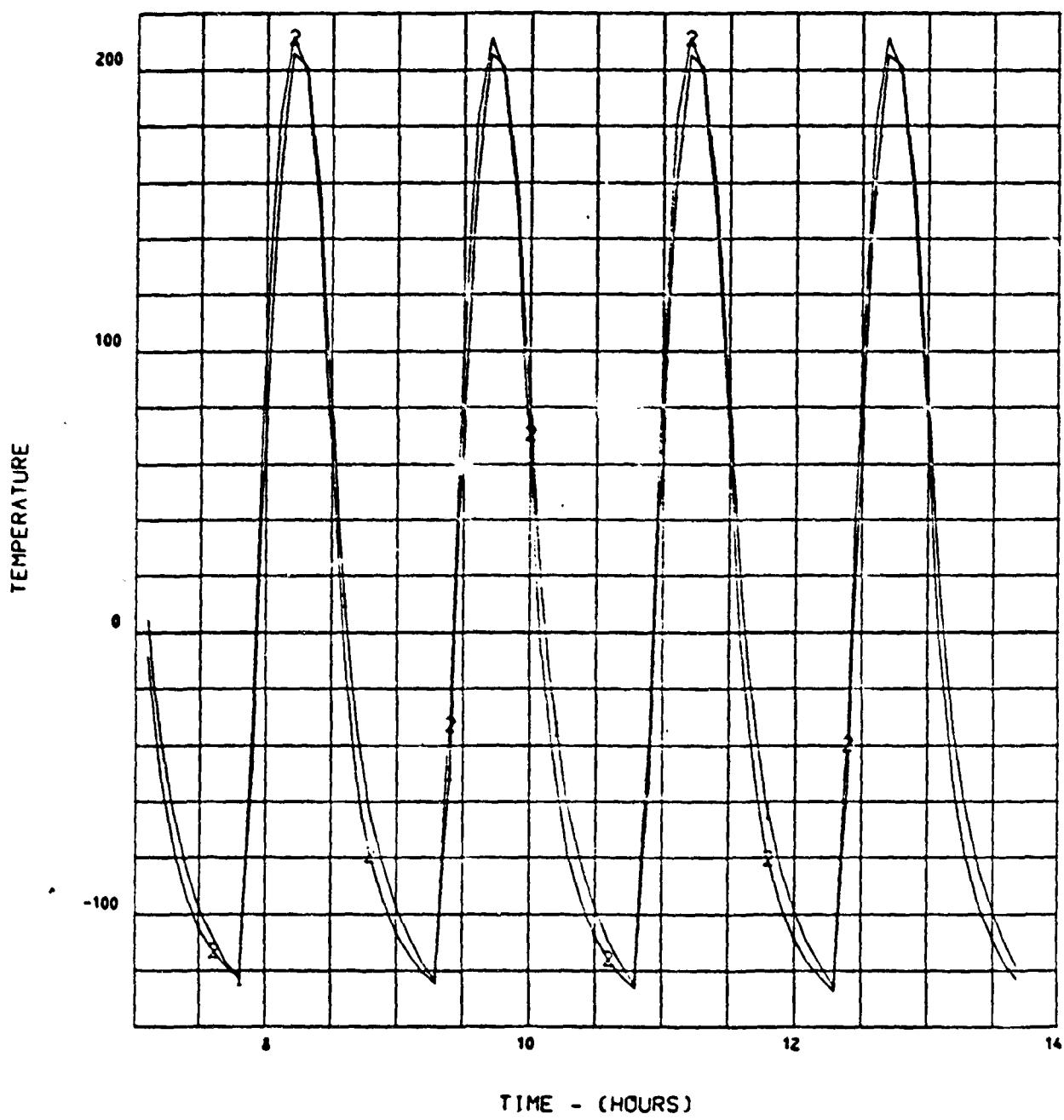


Figure 31

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD PLB DOORS. PORT
[2] FWD PLB DOORS. PORT
[3] AFT PLB DOORS. PORT
[4] AFT PLB DOORS. PORT

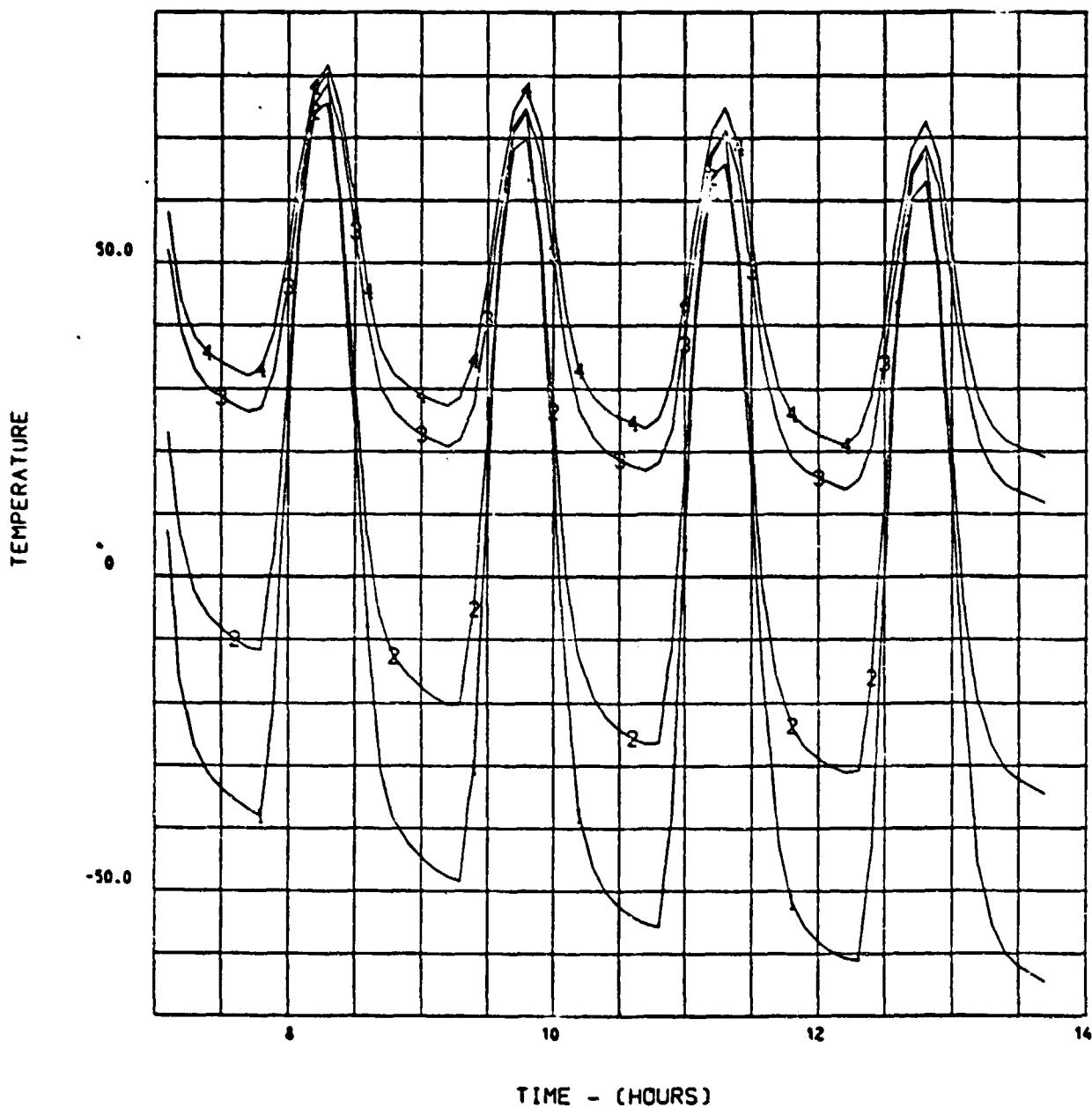
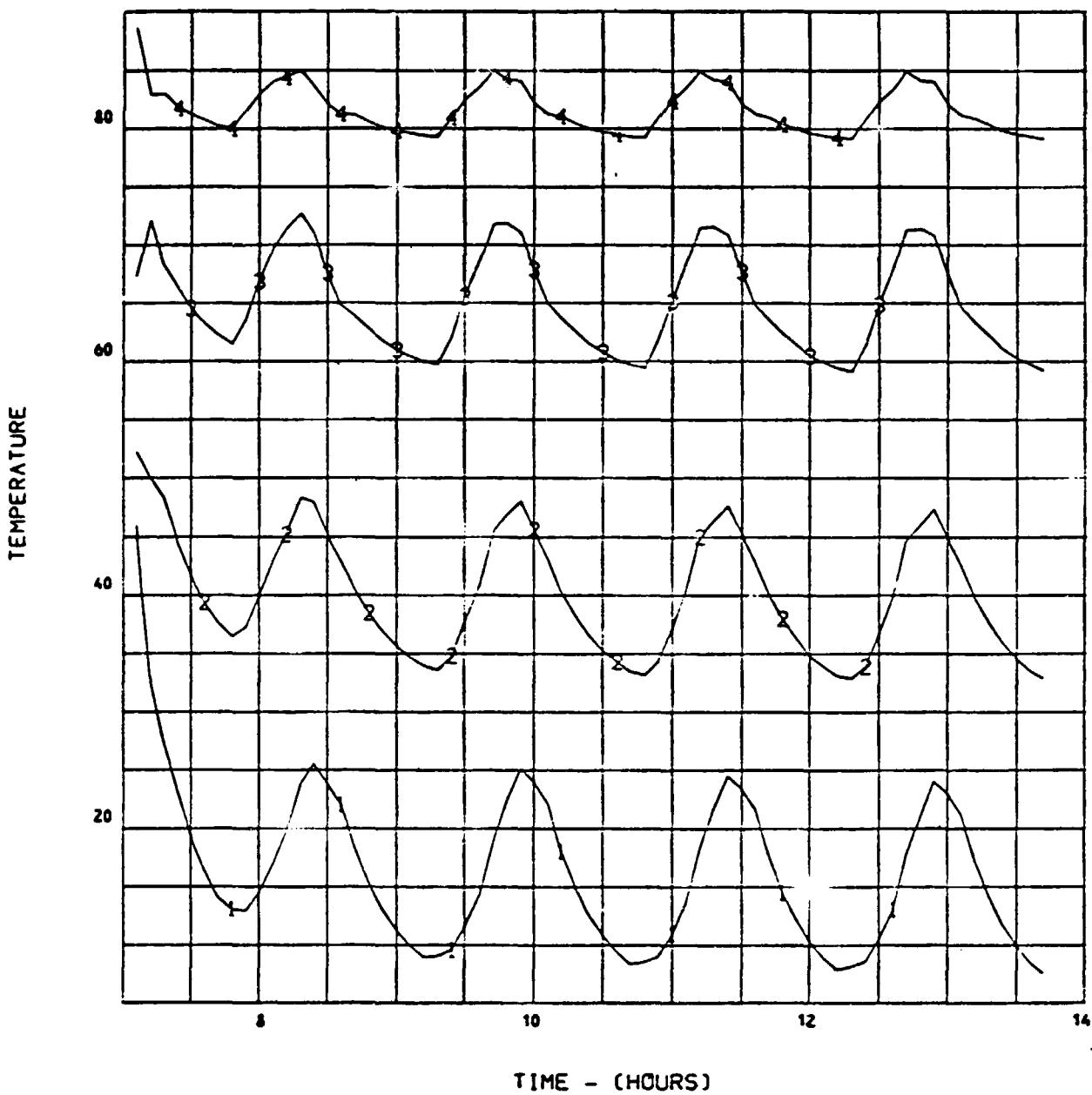


Figure 32

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD RADIATOR. PORT
[2] FWD RADIATOR. PORT
[3] AFT RADIATOR. PORT
[4] AFT RADIATOR. PORT



TIME - (HOURS)

Figure 33

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD BULKHO BOTTOM
[2] FWD BULKHO TOP

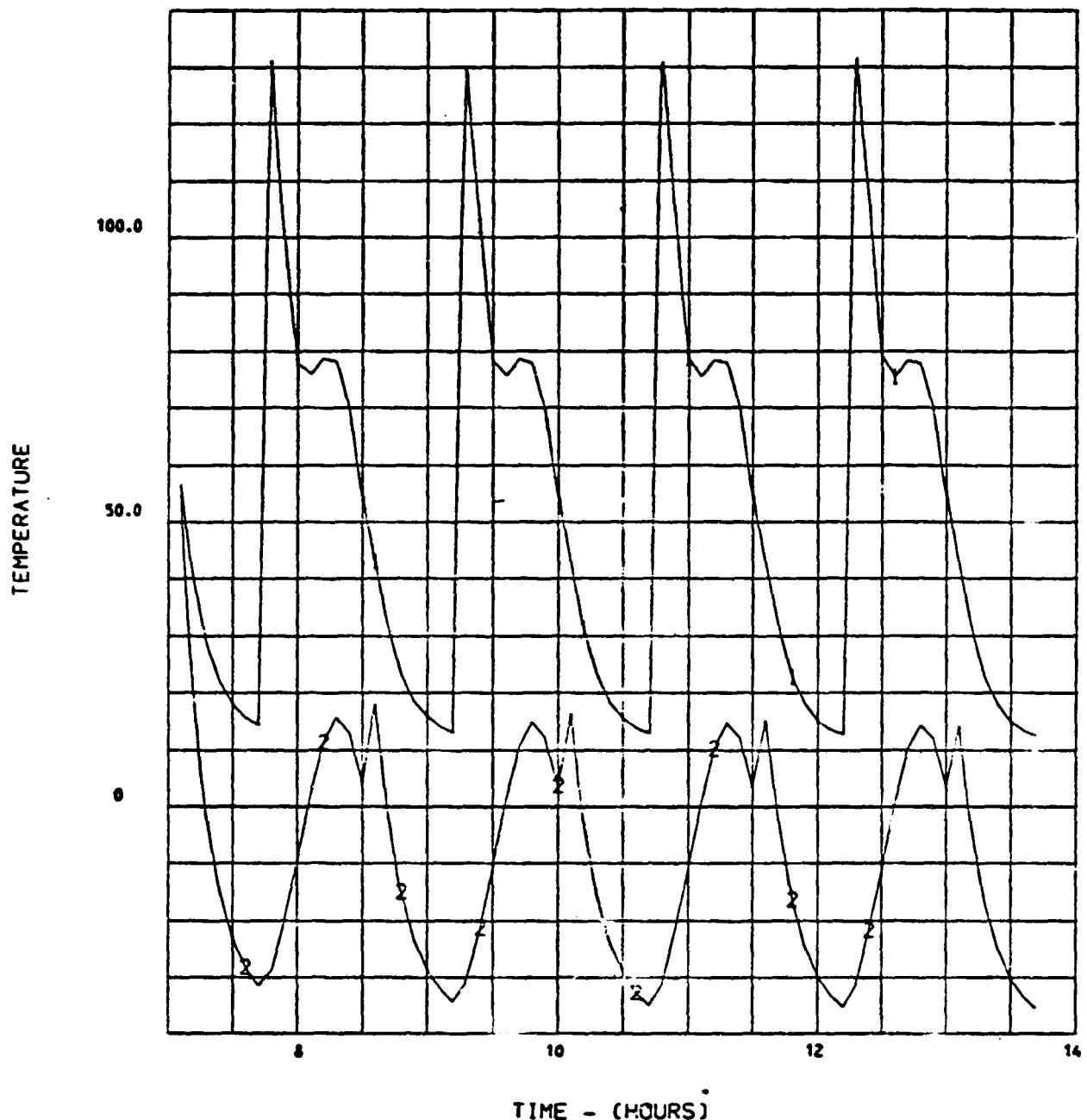


Figure 34

COMPARISON "136 NODE" PLB MODEL. OR8GEN NPT=12 VS TRJ
[1] UPPER FWD PLB LINER. PORT
[2] UPPER FWD PLB LINER. PORT
[3] UPPER FWD PLB LINER. PORT
[4] UPPER FWD PLB LINER. PORT

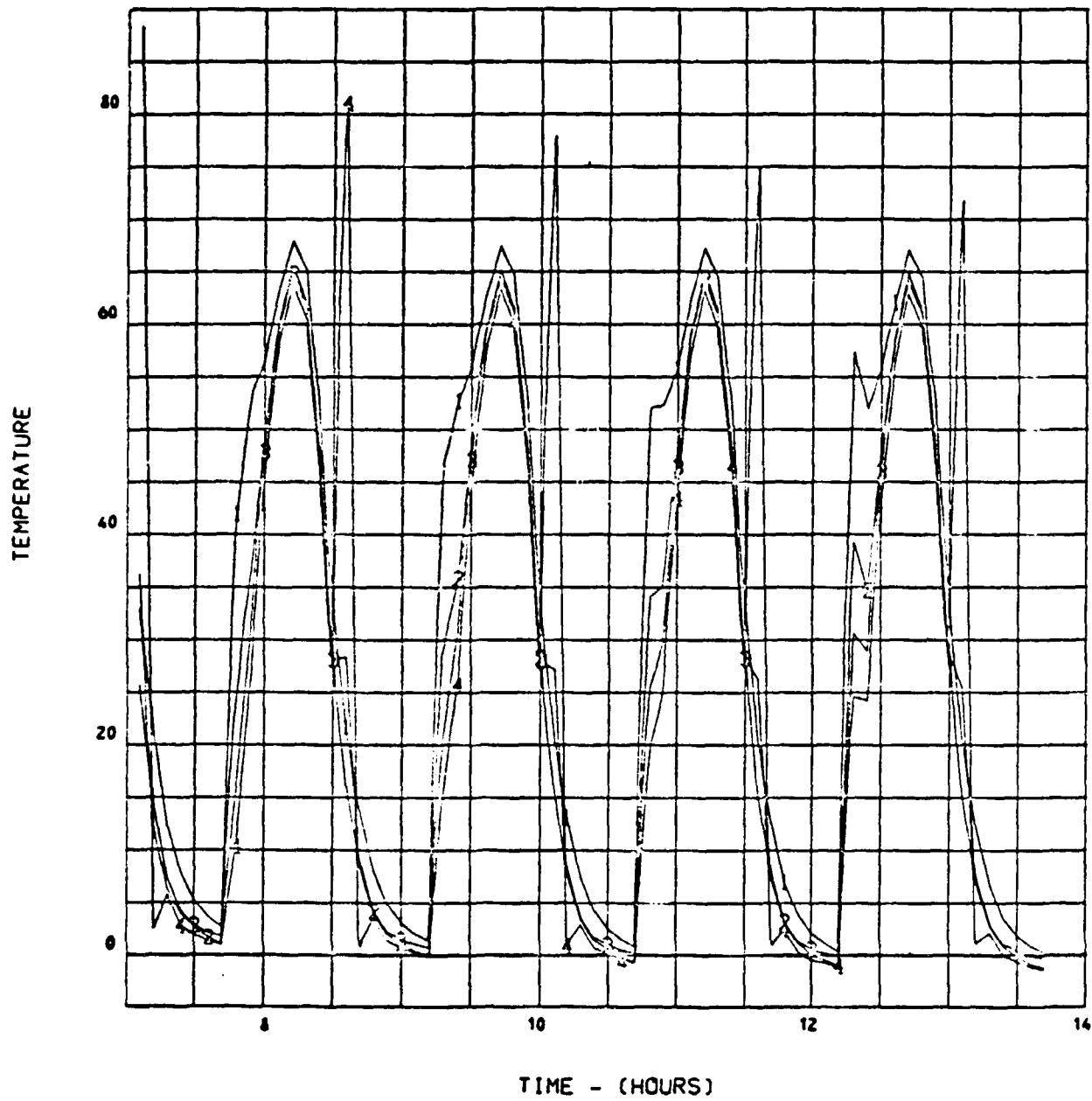
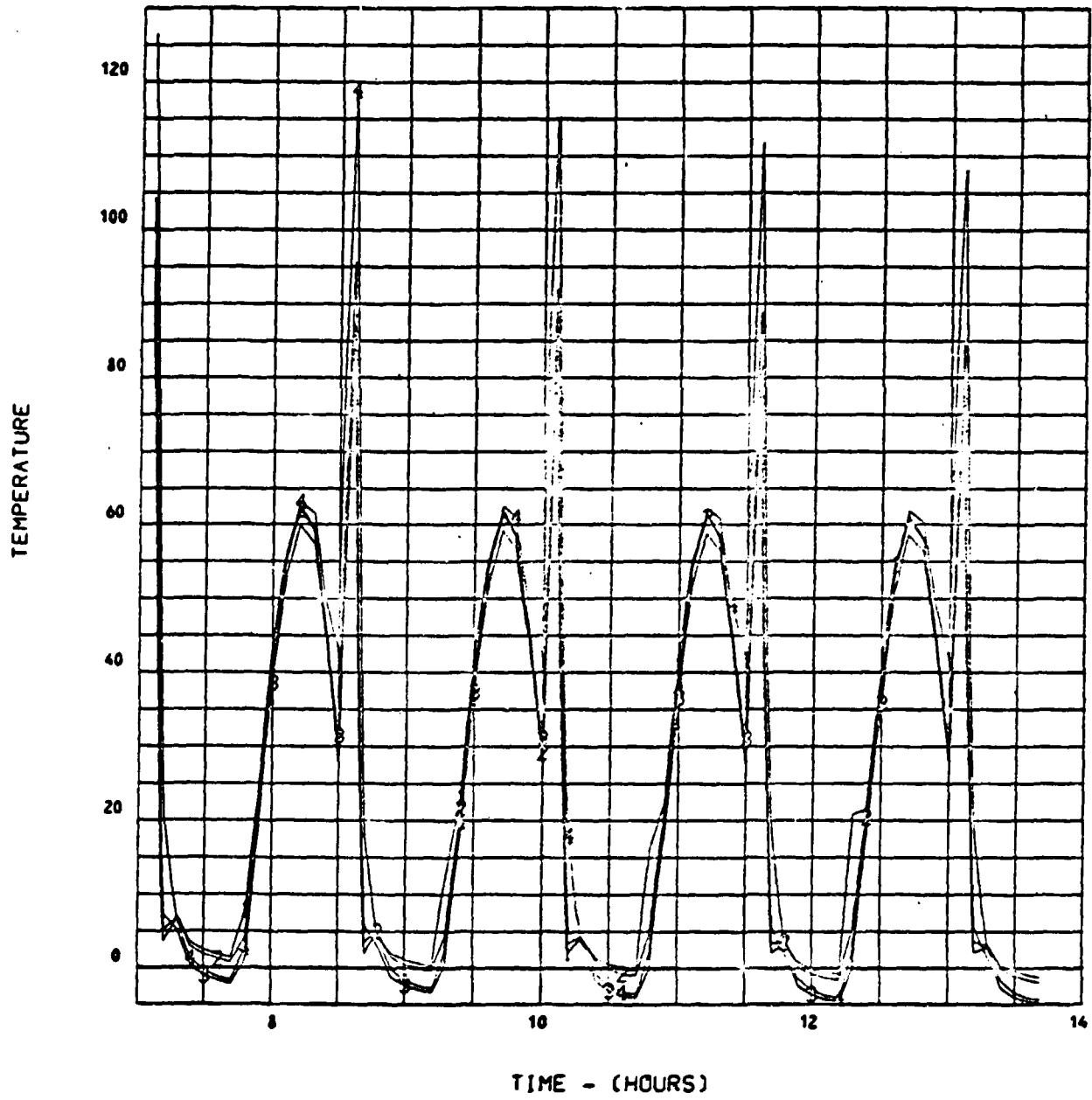


Figure 35

COMPARISON "136 NCDE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] UPPER AFT PLB LINER. PORT
[2] UPPER AFT PLB LINER. PORT
[3] UPPER AFT PLB LINER. PCRT
[4] UPPER AFT PLB LINER. PORT



TIME - (HOURS)

Figure 36

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] LOWER FWD PLB LINER. PORT
[2] LOWER FWD PLB LINER. PORT
[3] LOWER FWD PLB LINER. PORT
[4] LOWER FWD PLB LINER. PORT

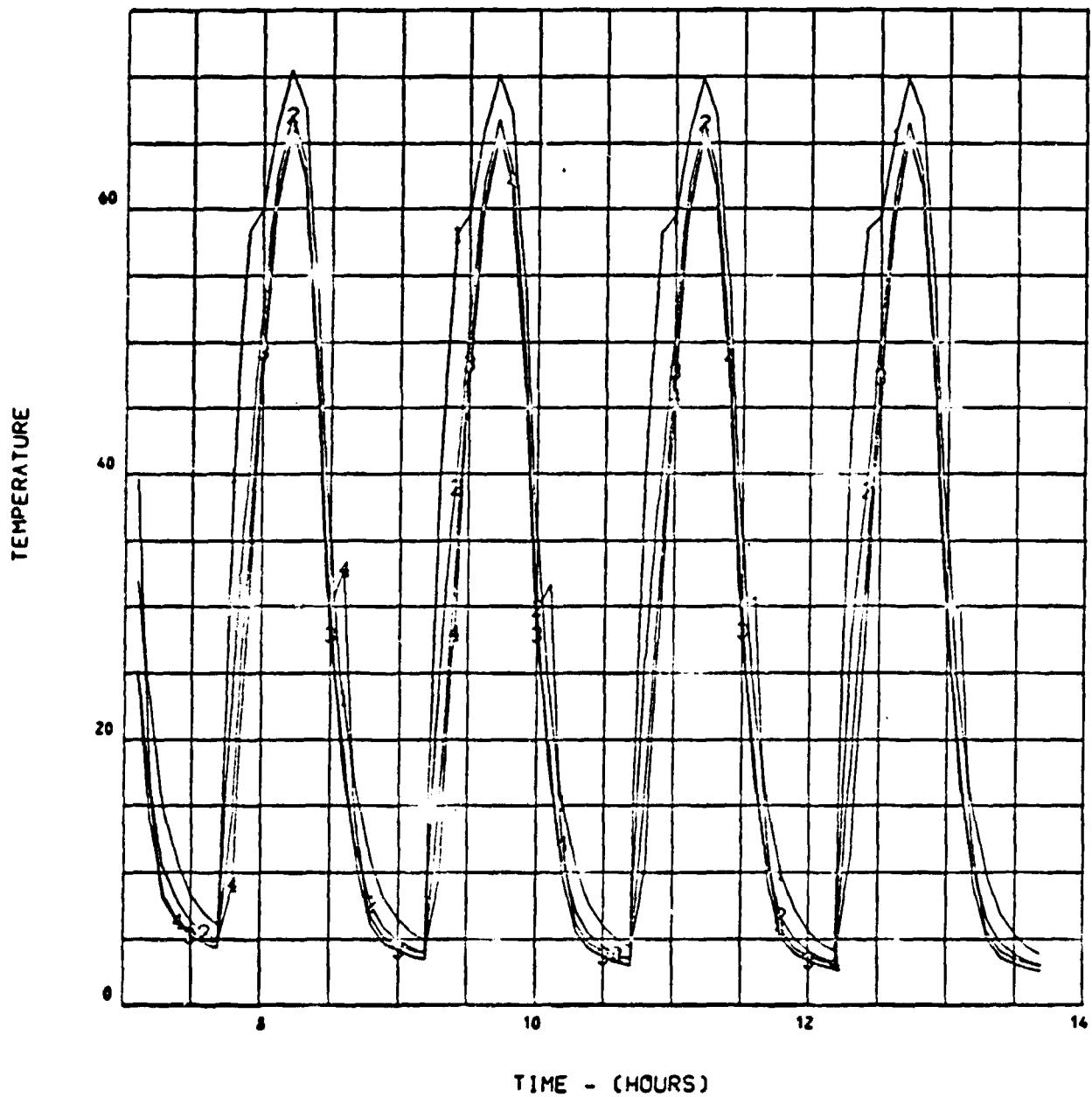
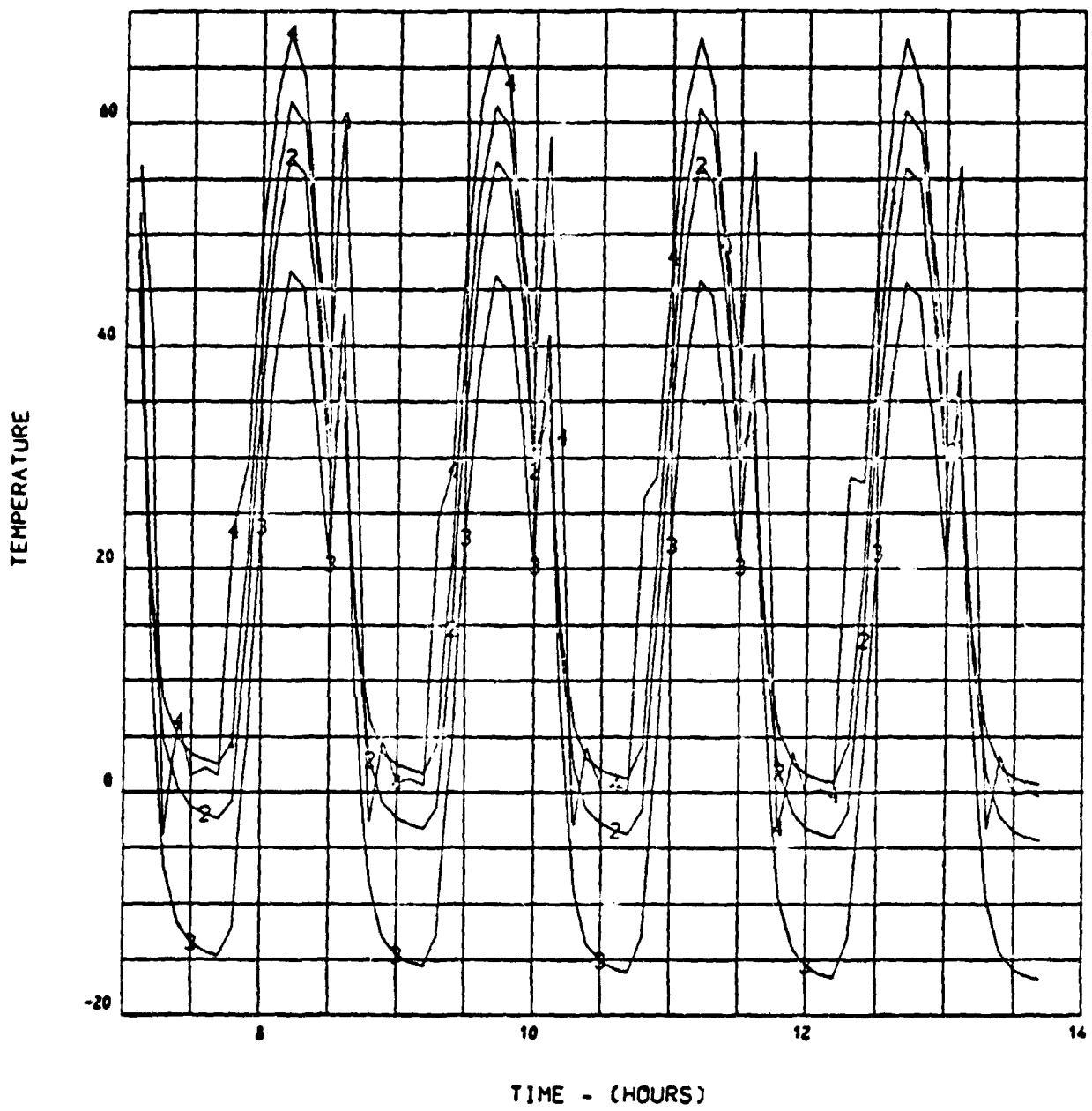


Figure 37

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] LOWER AFT PLB LINER. PORT
[2] LOWER AFT PLB LINER. PORT
[3] LOWER AFT PLB LINER. PORT
[4] LOWER AFT PLB LINER. PORT



TIME - (HOURS)

Figure 38

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD LONGERON. PORT
[2] AFT LONGERON. PORT

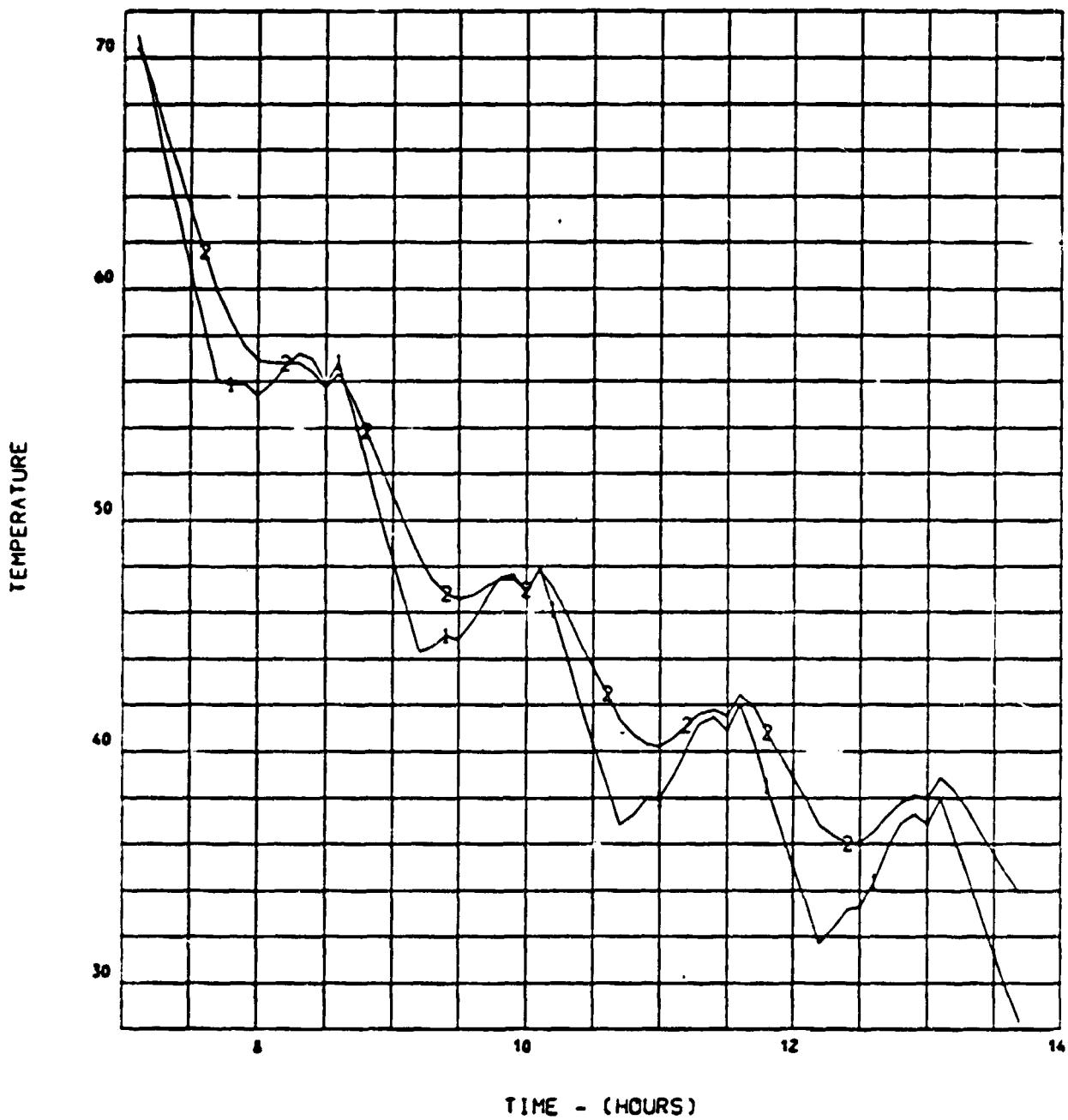


Figure 39

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] AFT BULKHD BOTTOM
[2] AFT BULKHD TOP

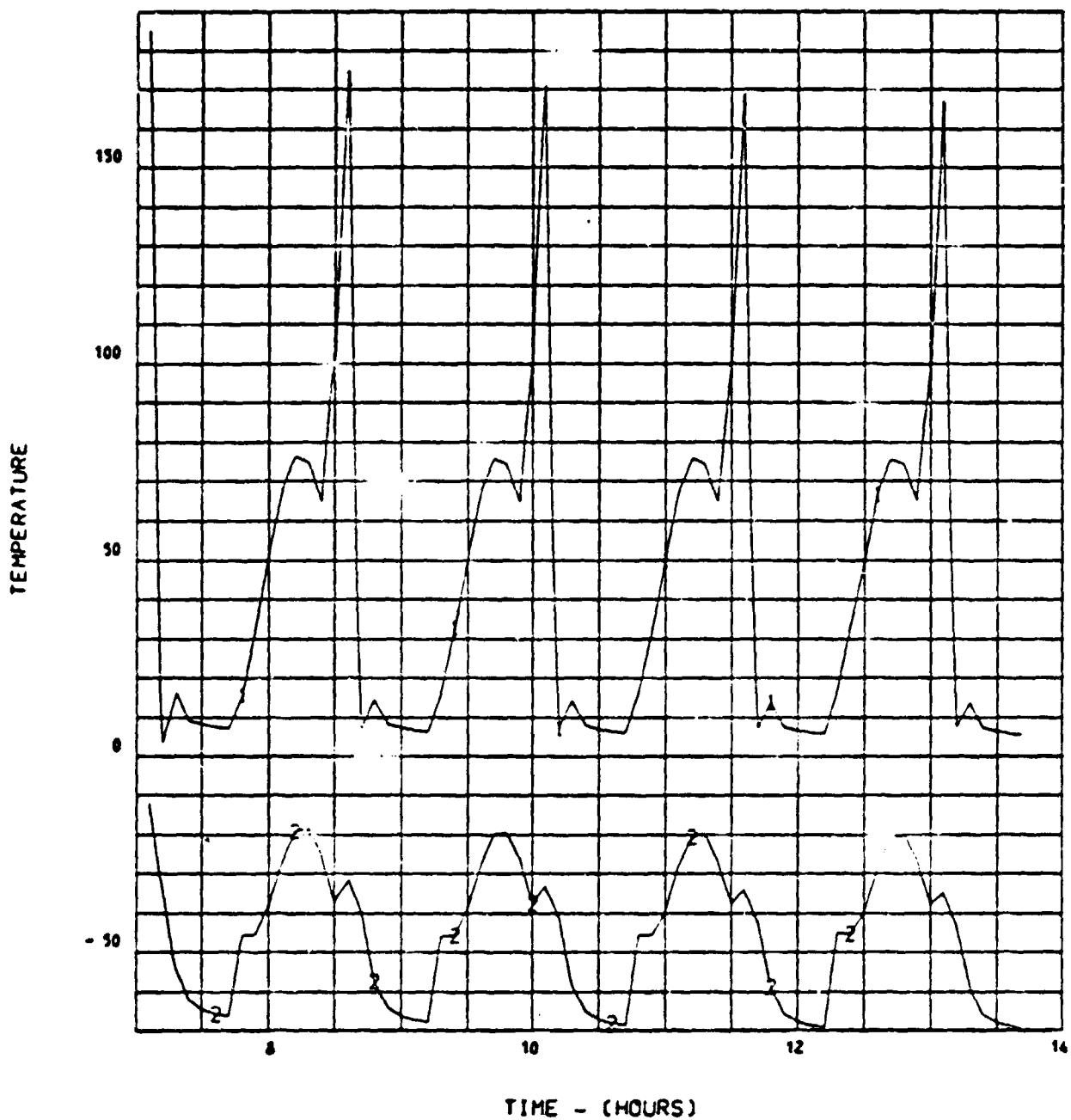


Figure 40

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD BULKHO BOTTOM BELOW PLB LINER

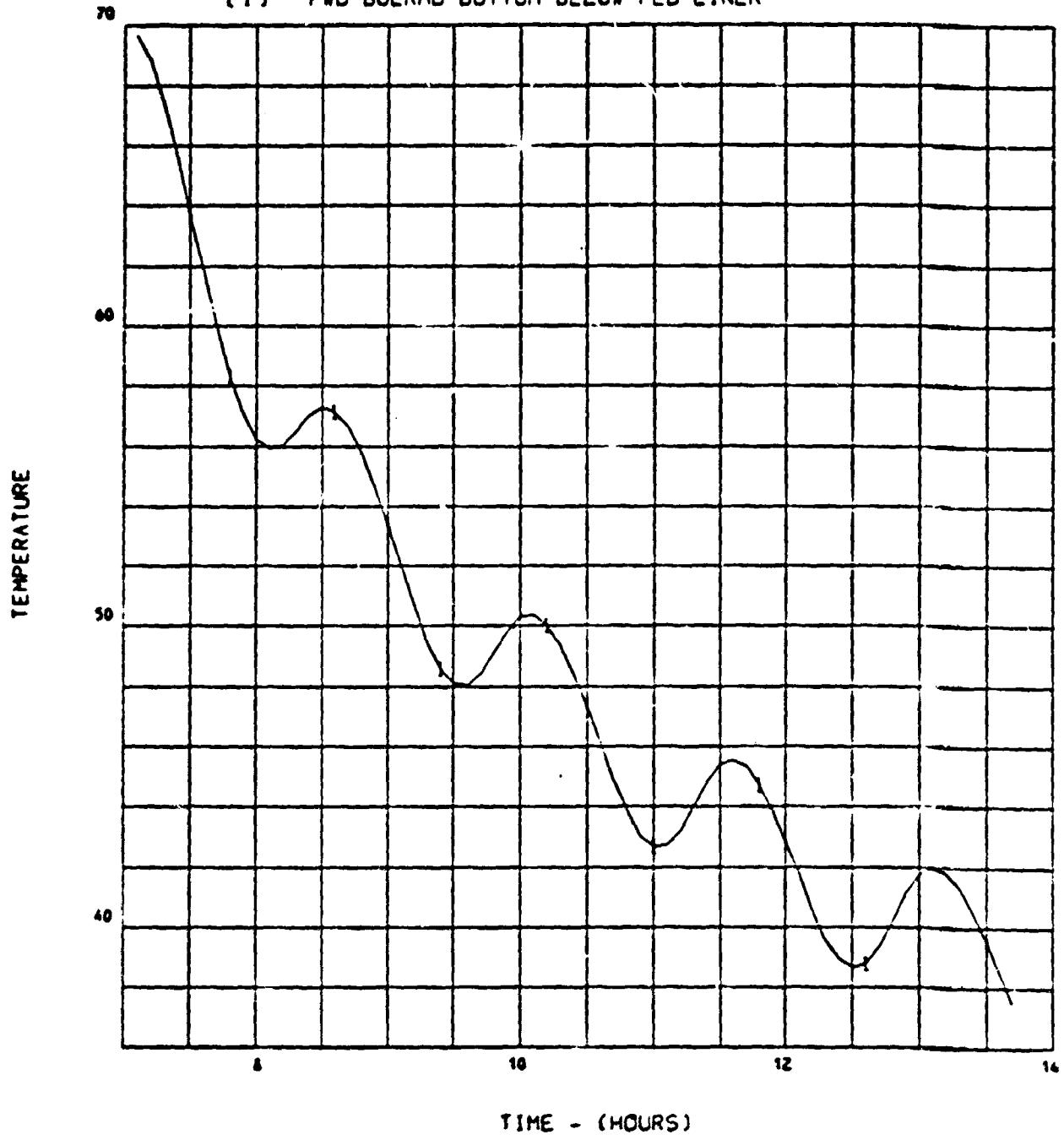


Figure 41

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD SIDE FUSELAGE STRUCTURE. PORT
[2] AFT SIDE FUSELAGE STRUCTURE. PORT

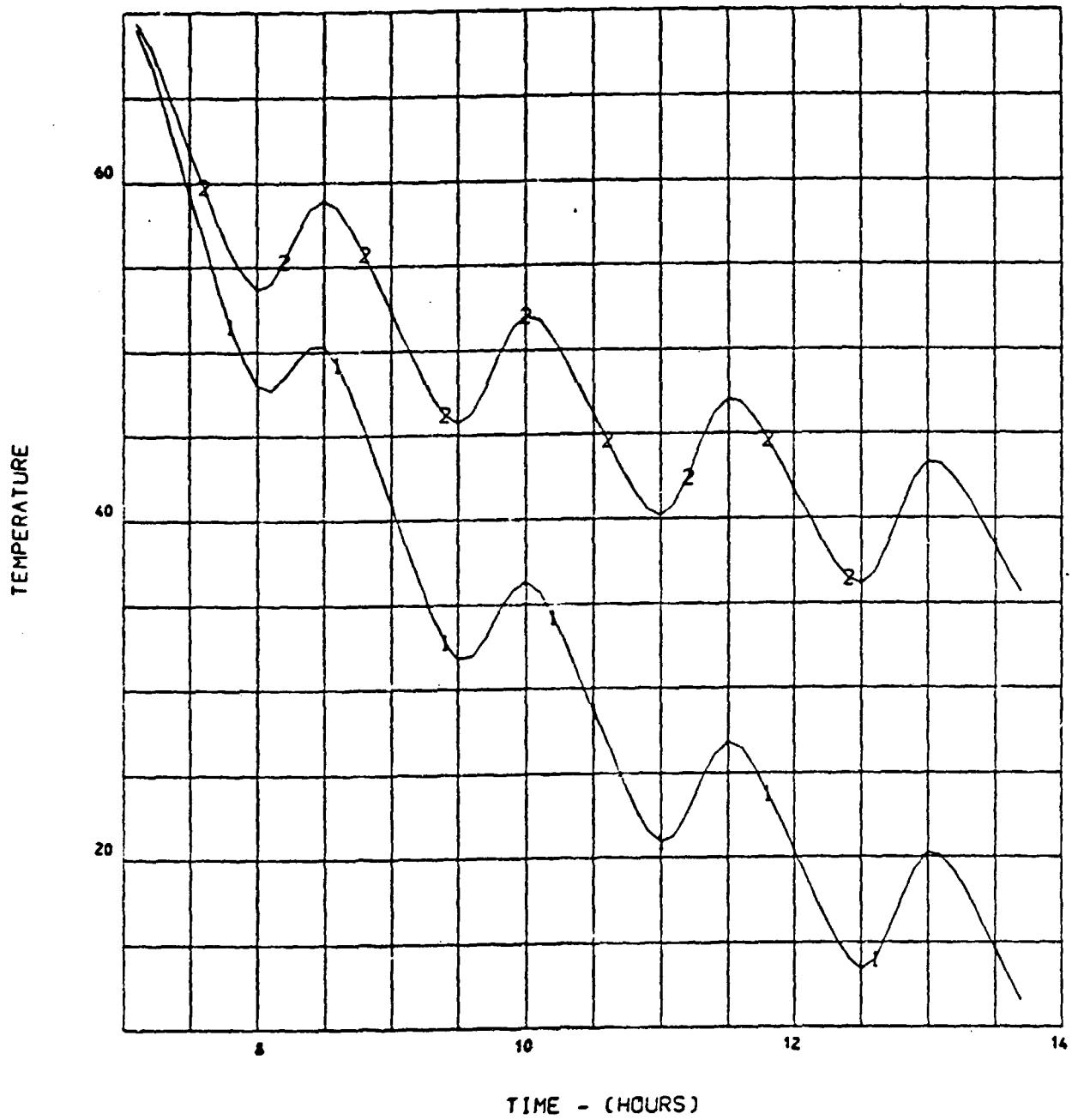


Figure 42

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD BOTTOM FUSELAGE STRUCTURE. PORT
[2] AFT BOTTOM FUSELAGE STRUCTURE. PORT

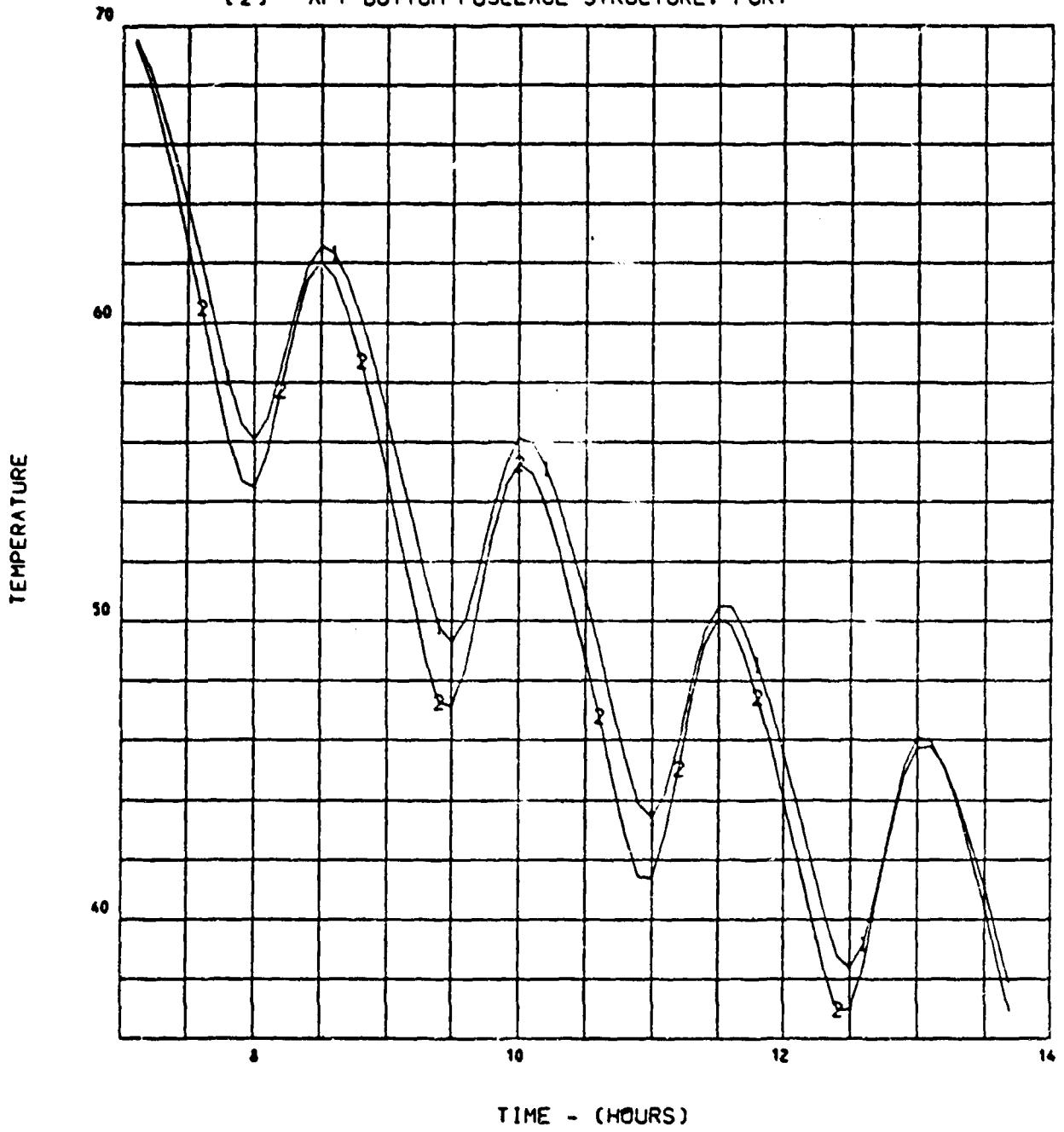


Figure 43

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD PLB DOORS STRUCTURE, PORT
[2] FWD PLB DOORS STRUCTURE, PORT
[3] AFT PLB DOORS STRUCTURE, PORT
[4] AFT PLB DOORS STRUCTURE, PCRT

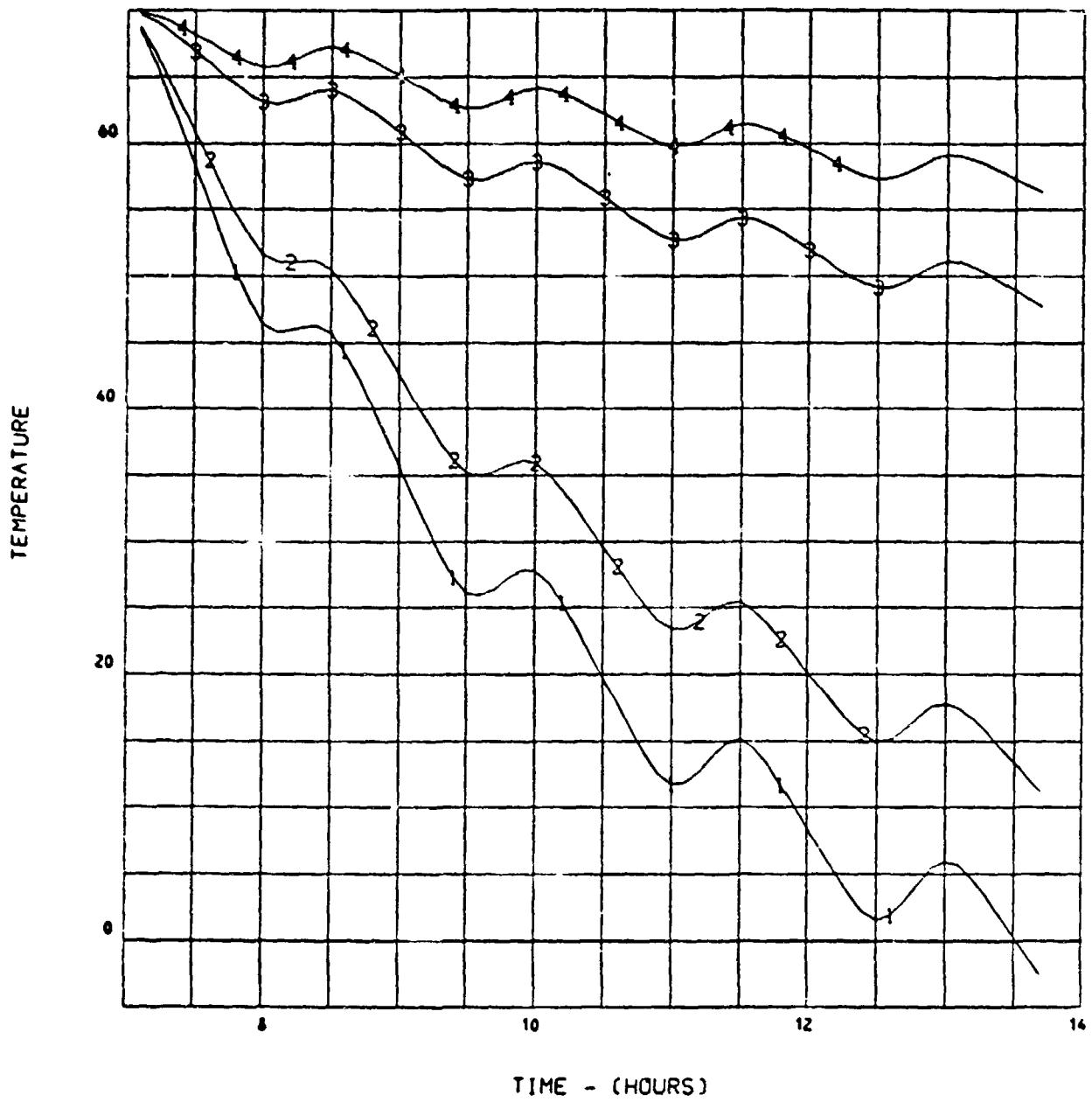


Figure 44

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] AFT BULKHD BOTTOM BELOW PLB LINER

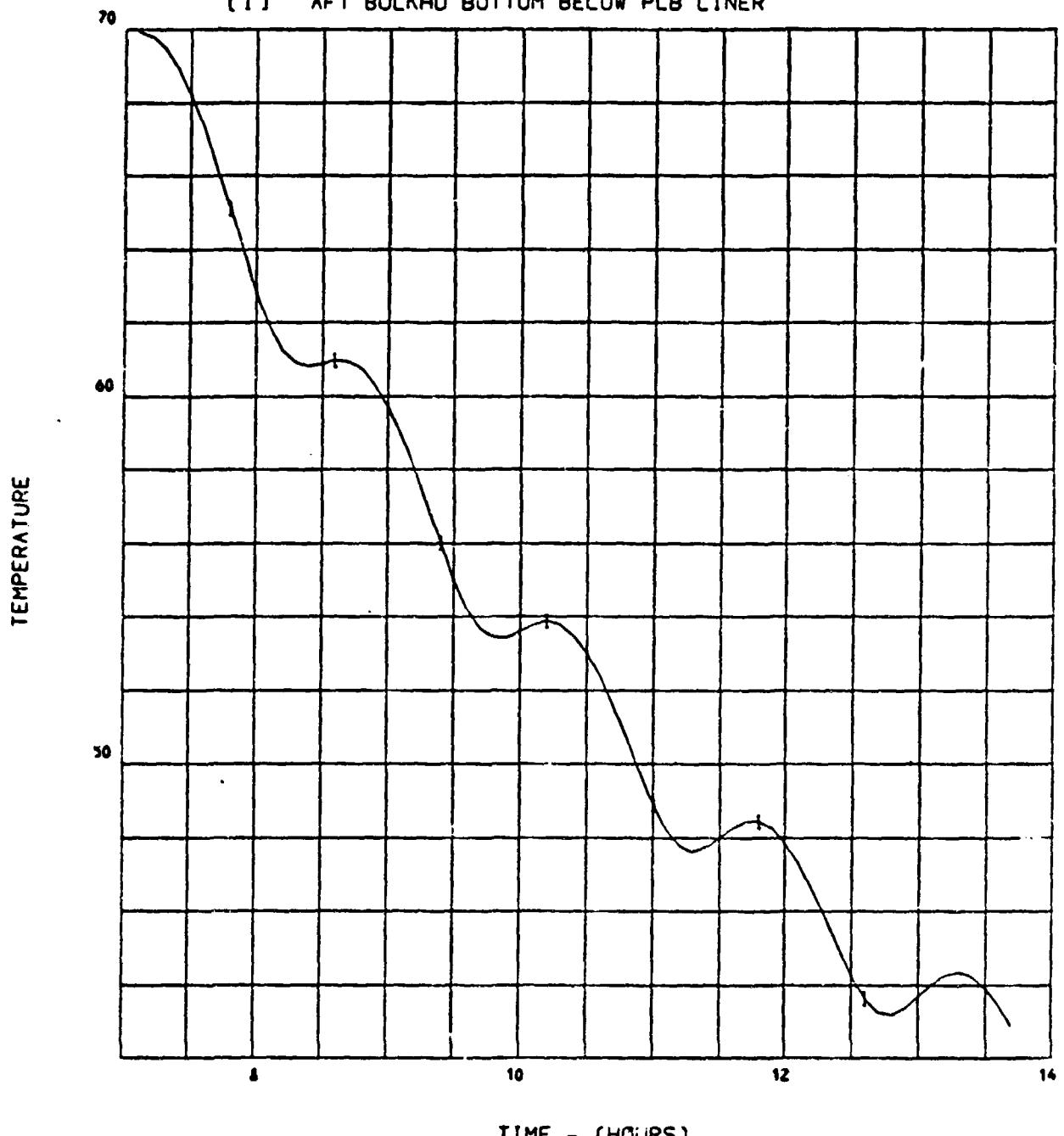


Figure 45

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT

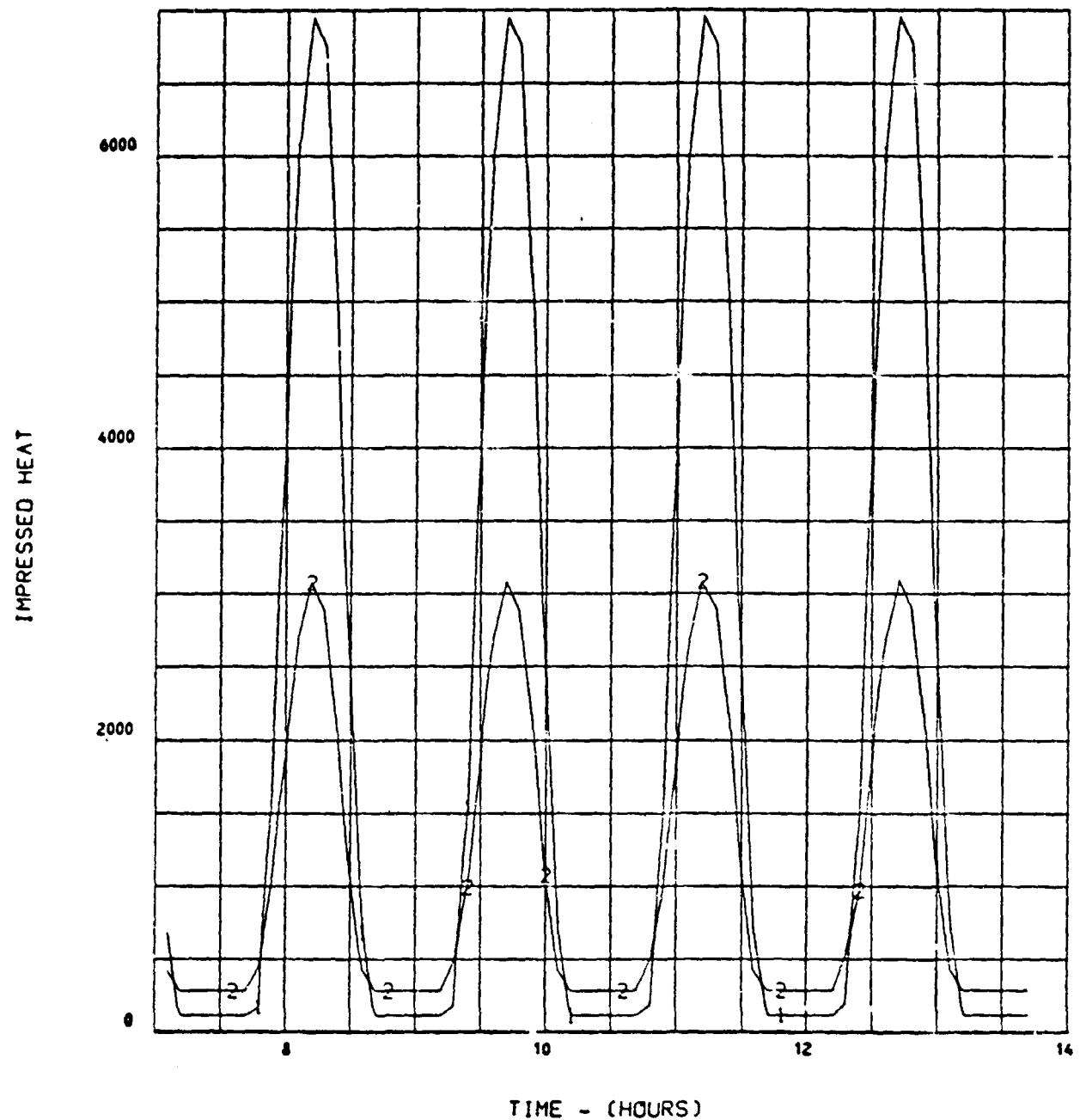


Figure 46

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

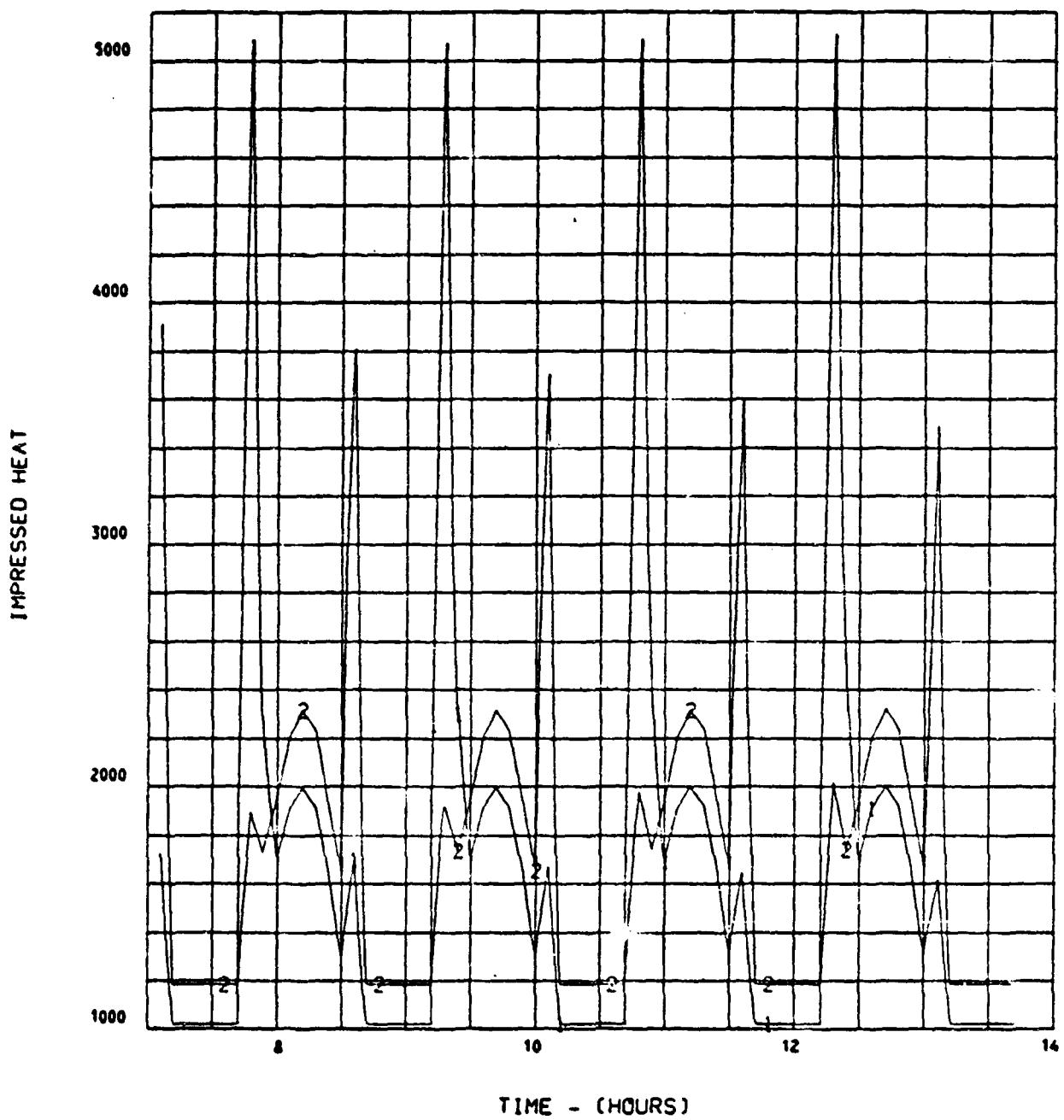


Figure 47

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD BOTTOM FUSELAGE, PORT
[2] AFT BOTTOM FUSELAGE, PORT

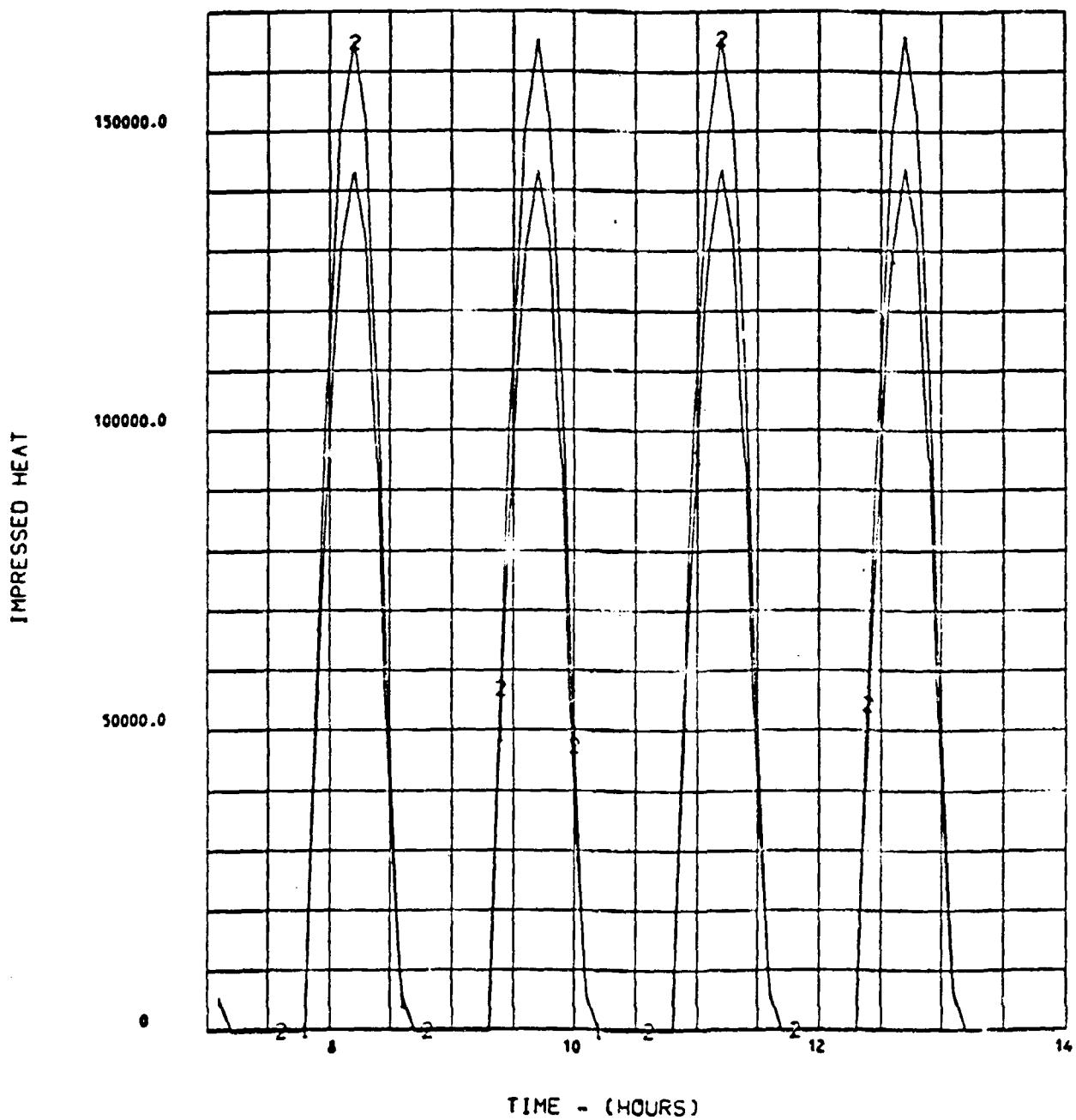


Figure 48

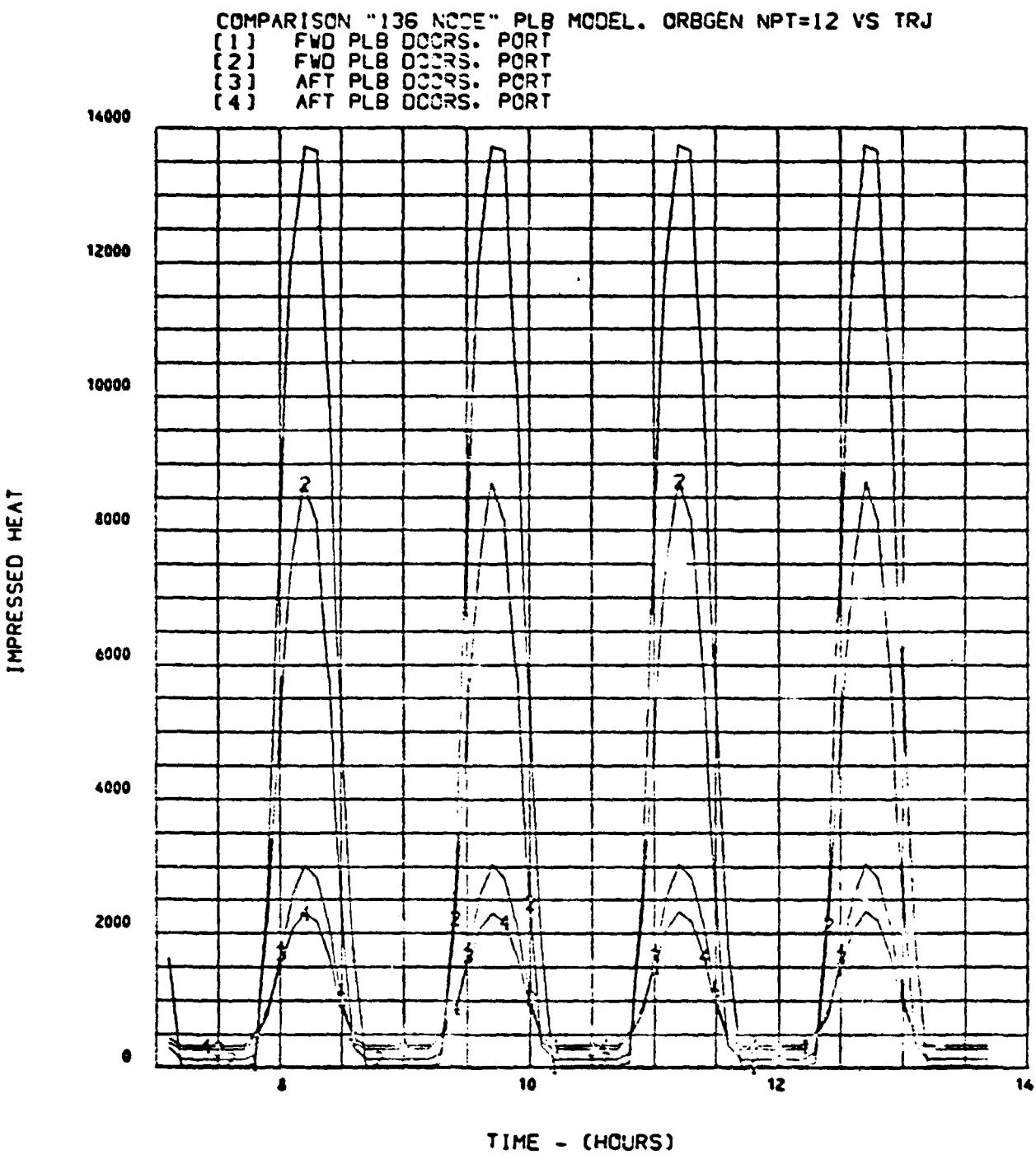


Figure 49

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD RADIATOR. PORT
[2] FWD RADIATOR. PORT
[3] AFT RADIATOR. PORT
[4] AFT RADIATOR. PORT

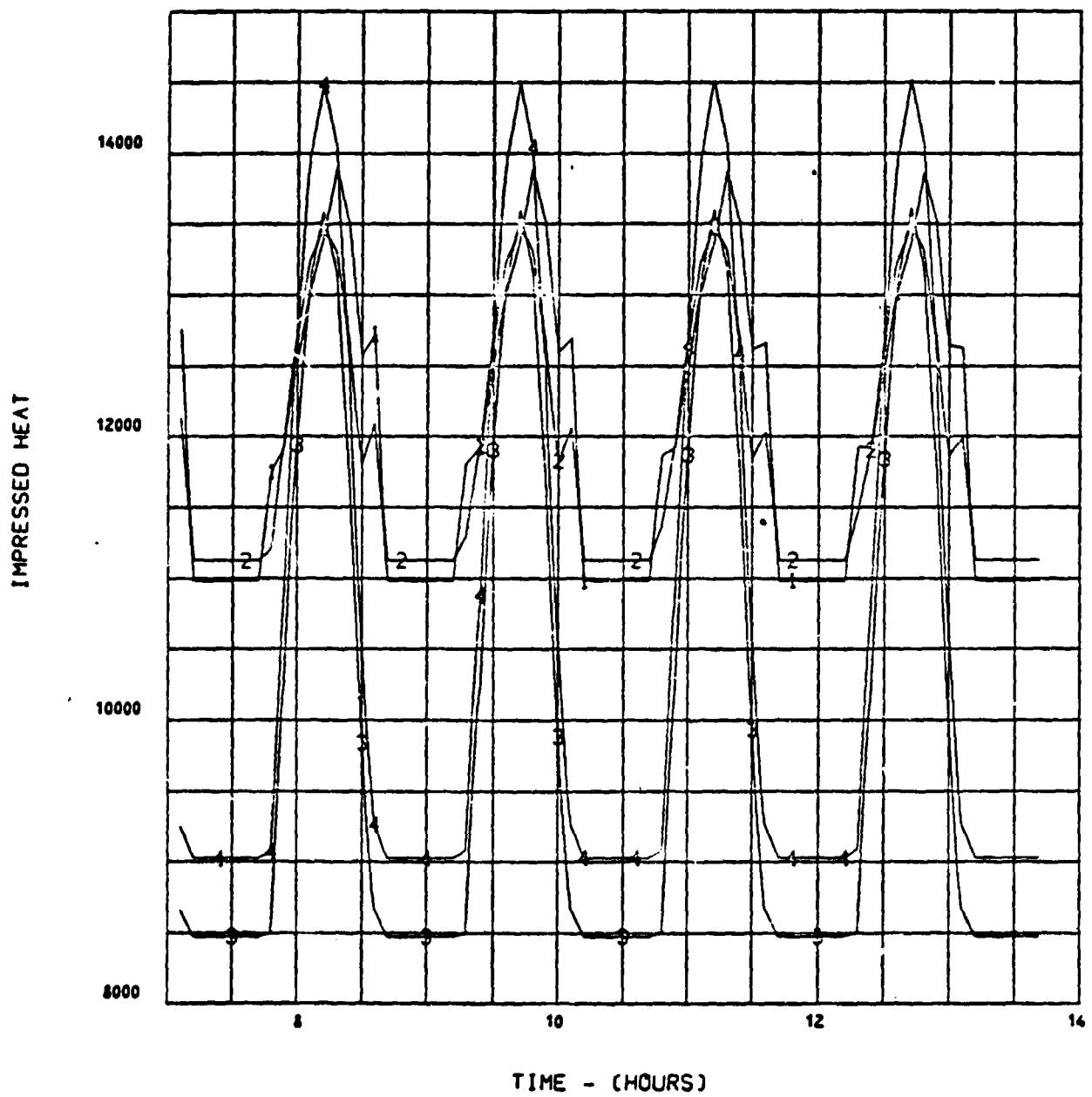


Figure 50

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPY=12 VS TRJ
[1] FWD BULKHO TOP
[2] FWD BULKHO BOTTOM

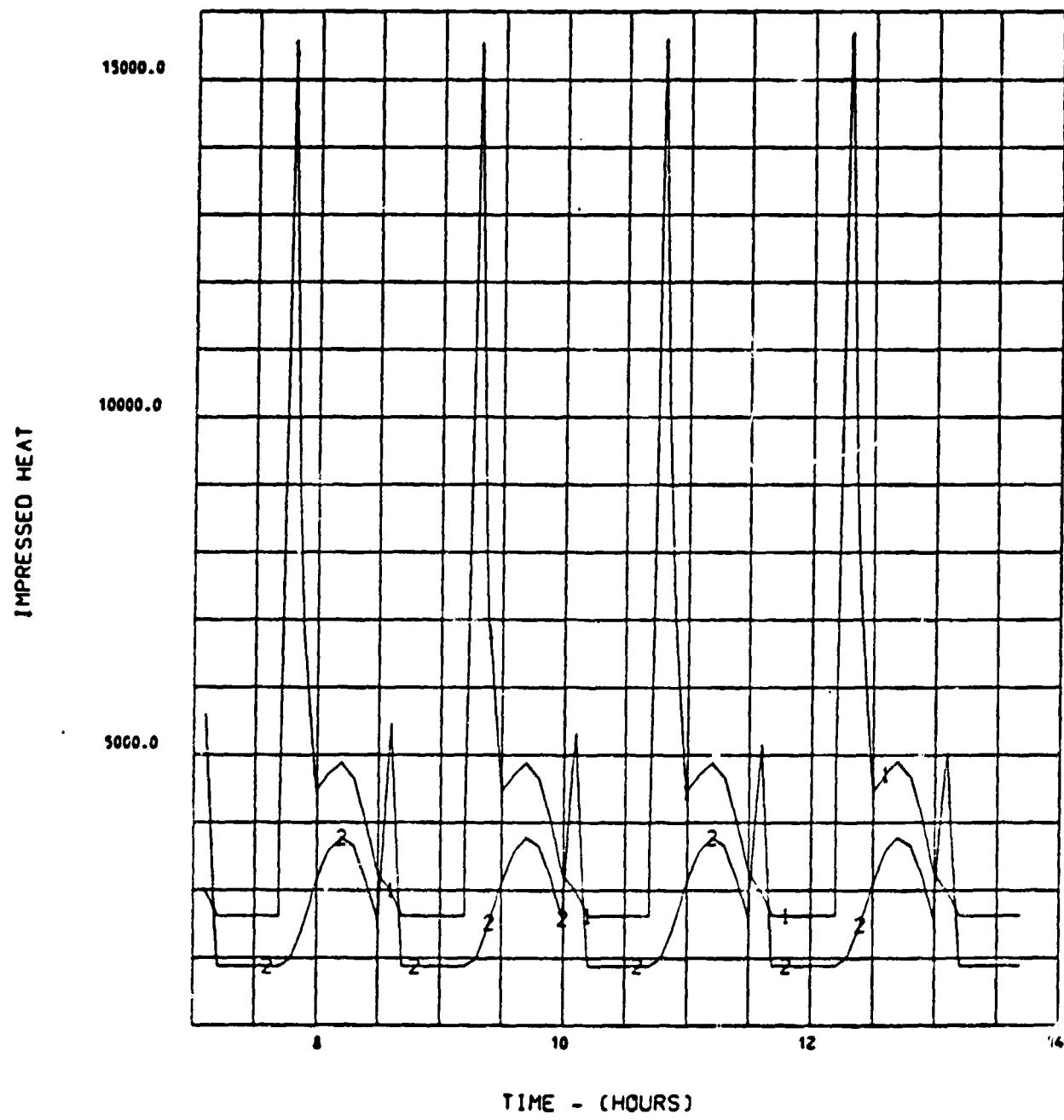
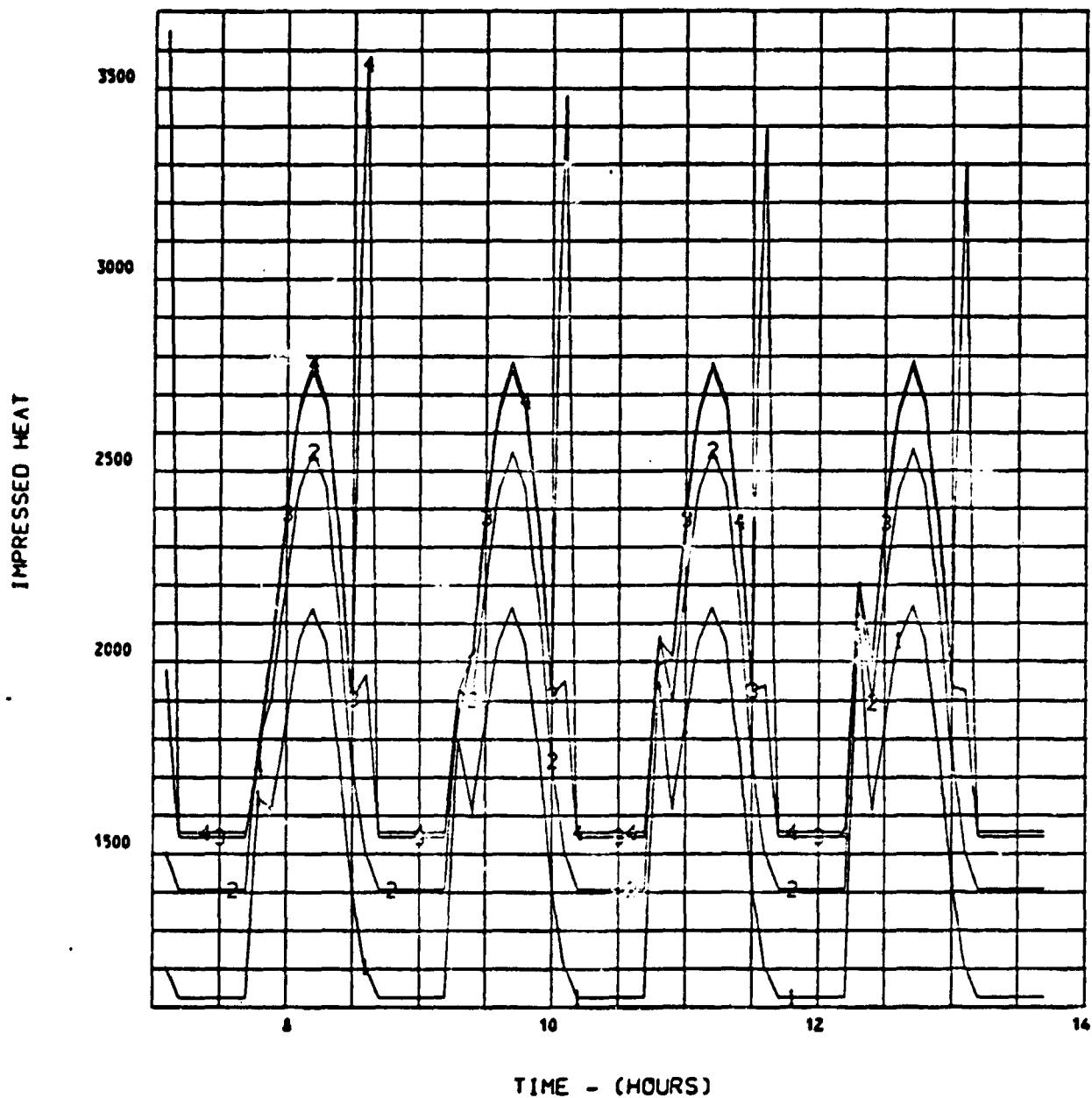


Figure 51

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] UPPER FWD PLB LINER. PORT
[2] UPPER FWD PLB LINER. PORT
[3] UPPER FWD PLB LINER. PORT
[4] UPPER FWD PLB LINER. PORT



TIME - (HOURS)

Figure 52

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] UPPER AFT PLB LINER. PORT
[2] UPPER AFT PLB LINER. PORT
[3] UPPER AFT PLB LINER. PORT
[4] UPPER AFT PLB LINER. PORT

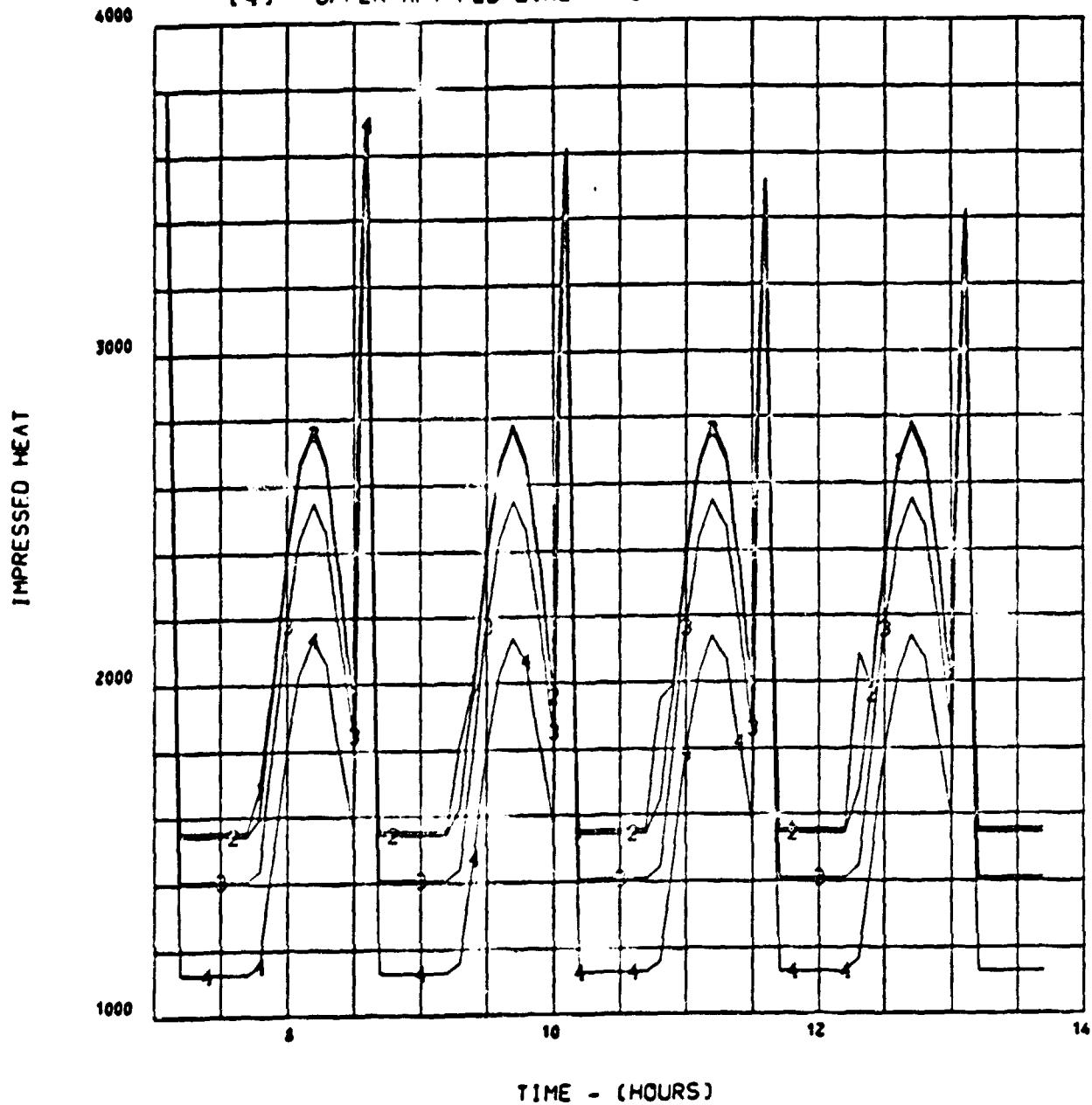


Figure 53

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] LOWER FWD PLB LINER. PORT
[2] LOWER FWD PLB LINER. PORT
[3] LOWER FWD PLB LINER. PORT
[4] LOWER FWD PLB LINER. PORT

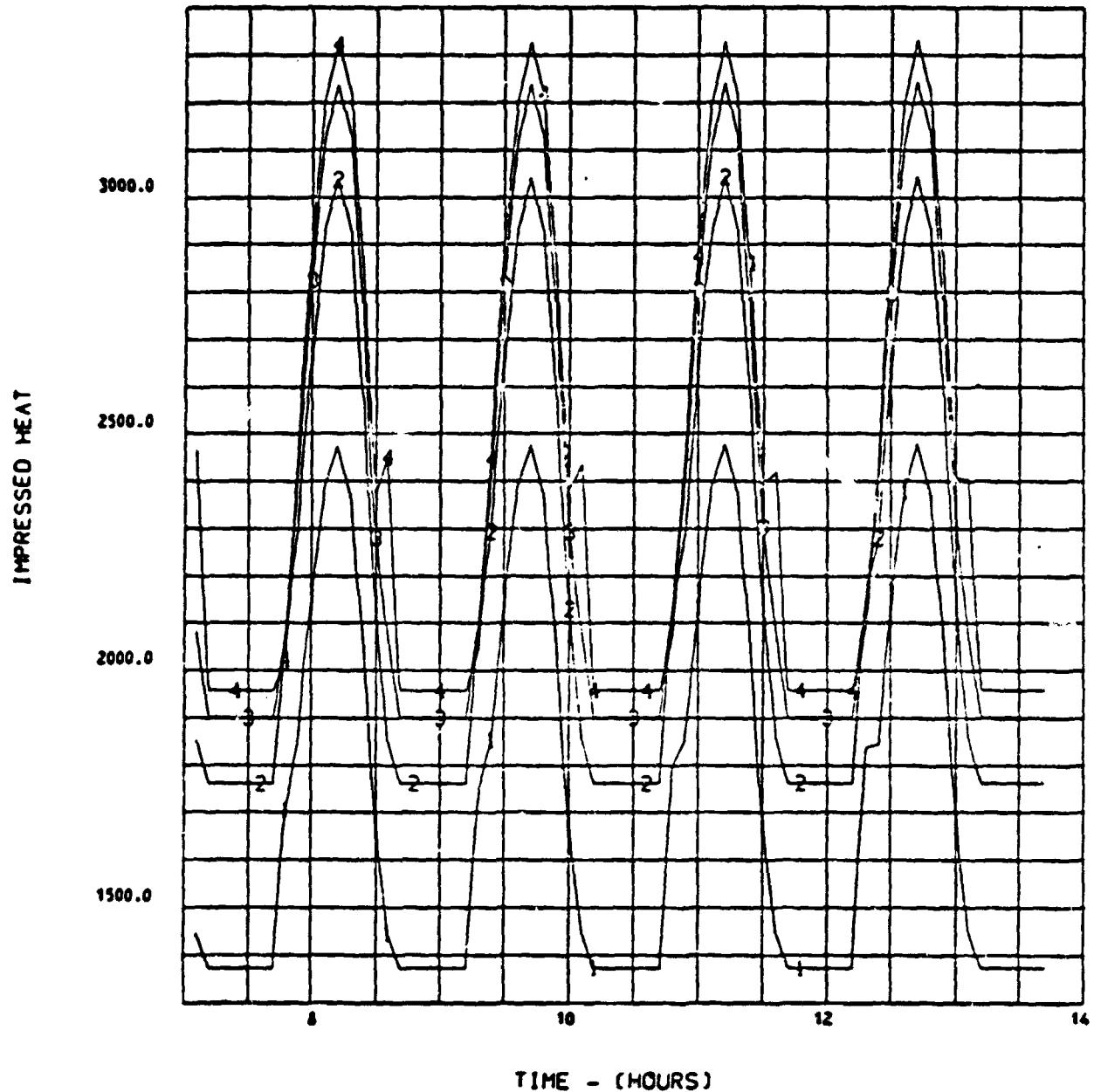


Figure 54

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] LOWER AFT PLB LINER. PORT
[2] LOWER AFT PLB LINER. PORT
[3] LOWER AFT PLB LINER. PORT
[4] LOWER AFT PLB LINER. PORT

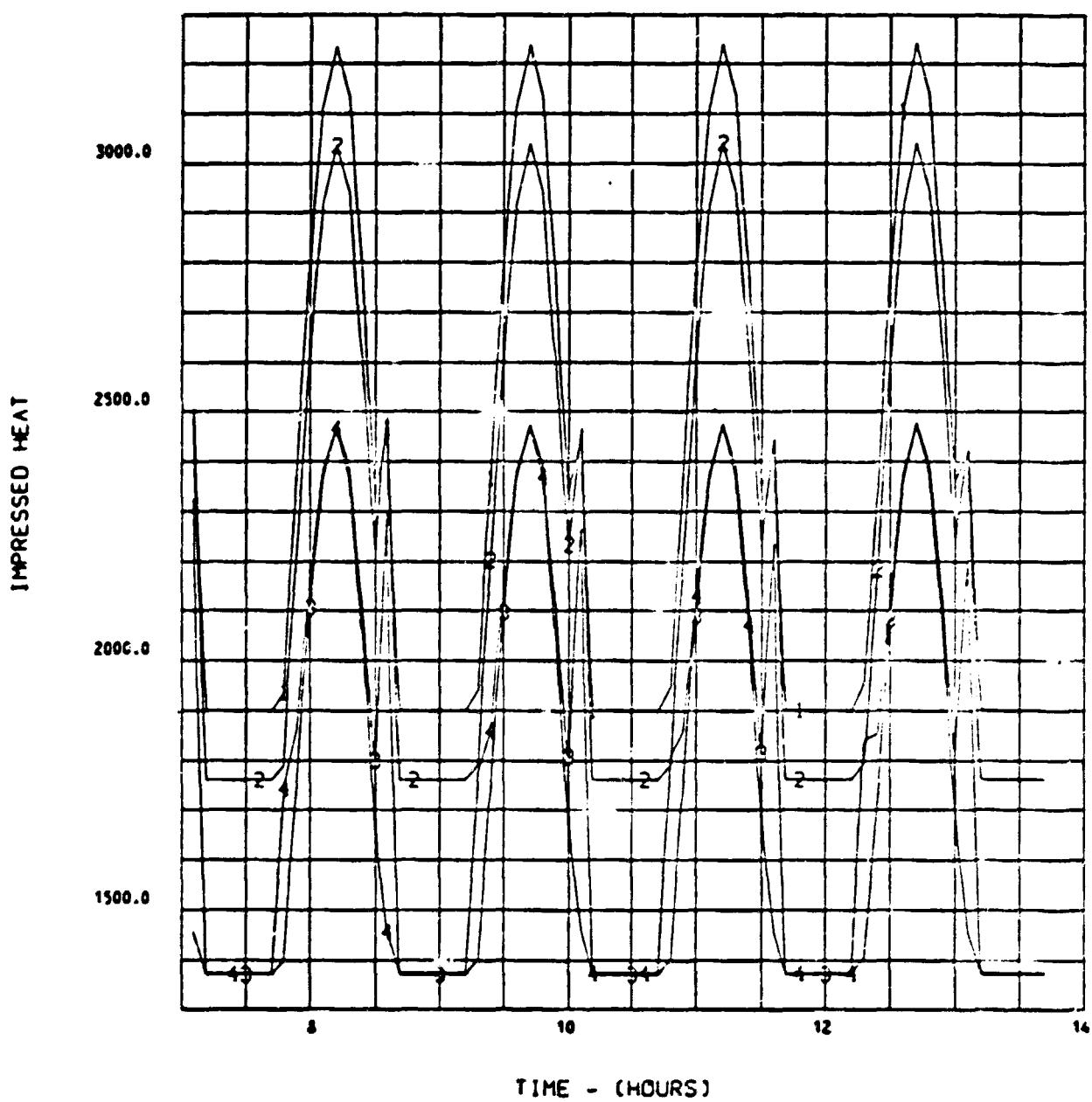


Figure 55

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=12 VS TRJ
[1] FWD LONGERON. PORT
[2] AFT LONGERON. PORT

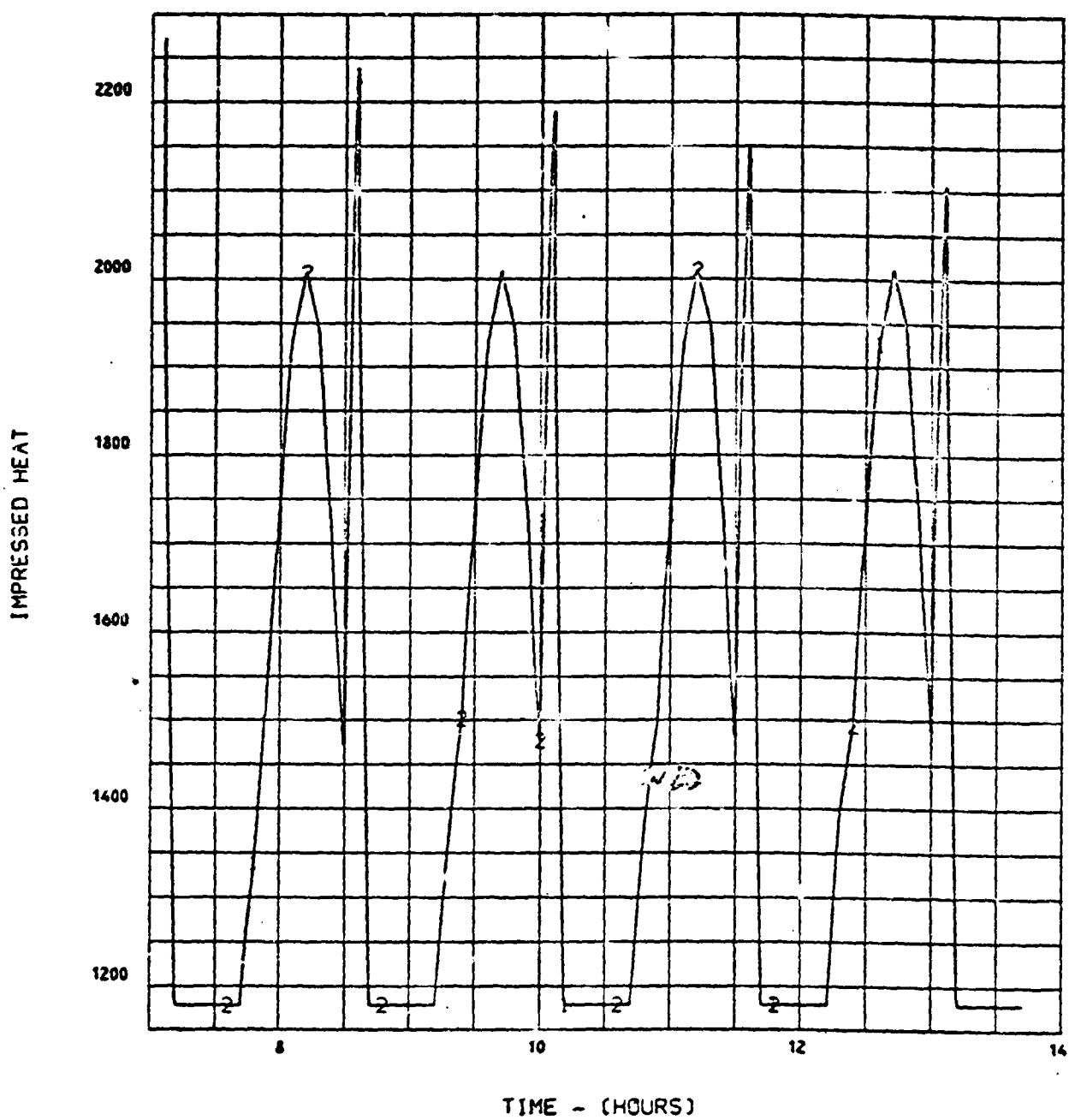


Figure 56

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT

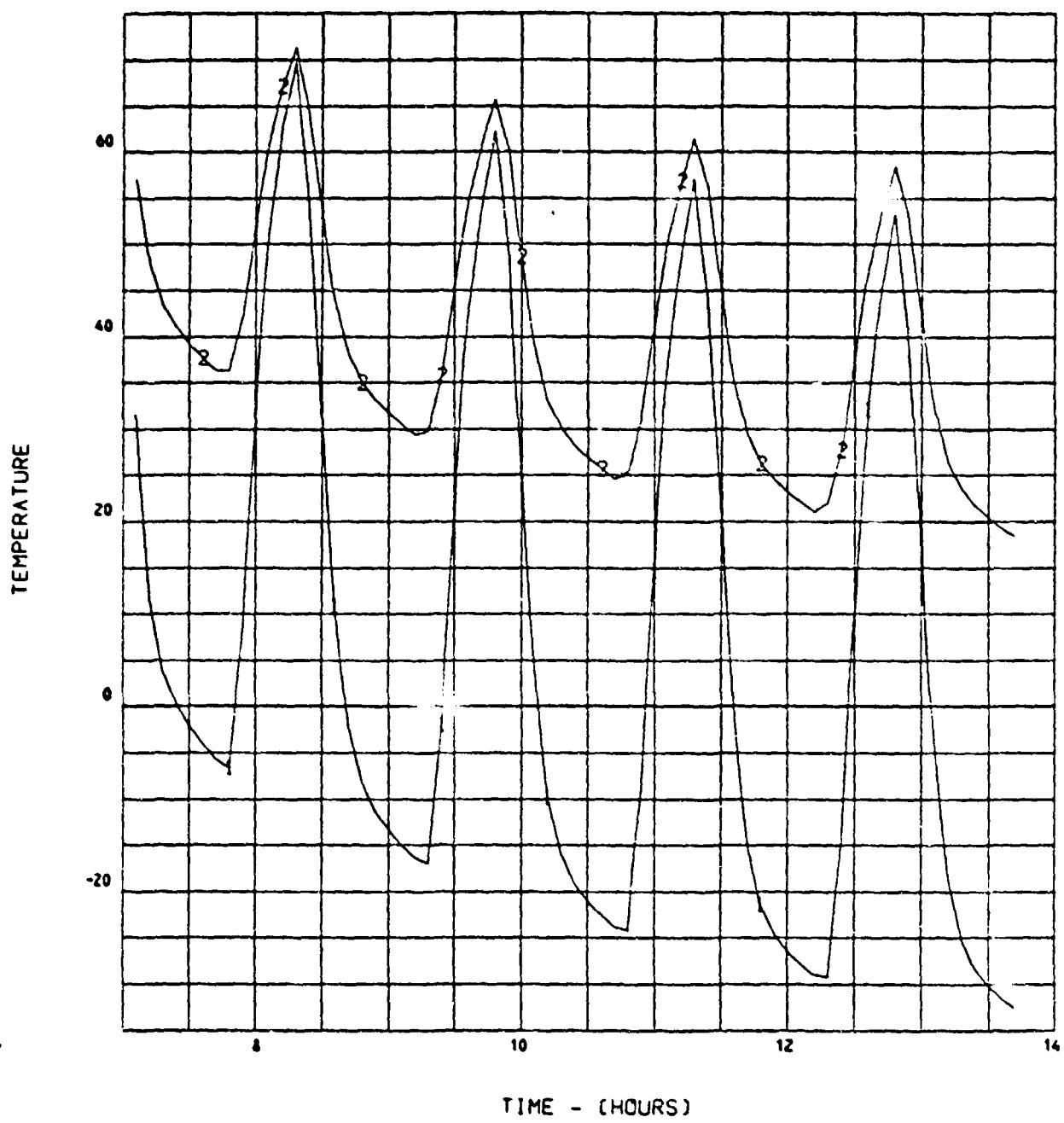


Figure 57

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

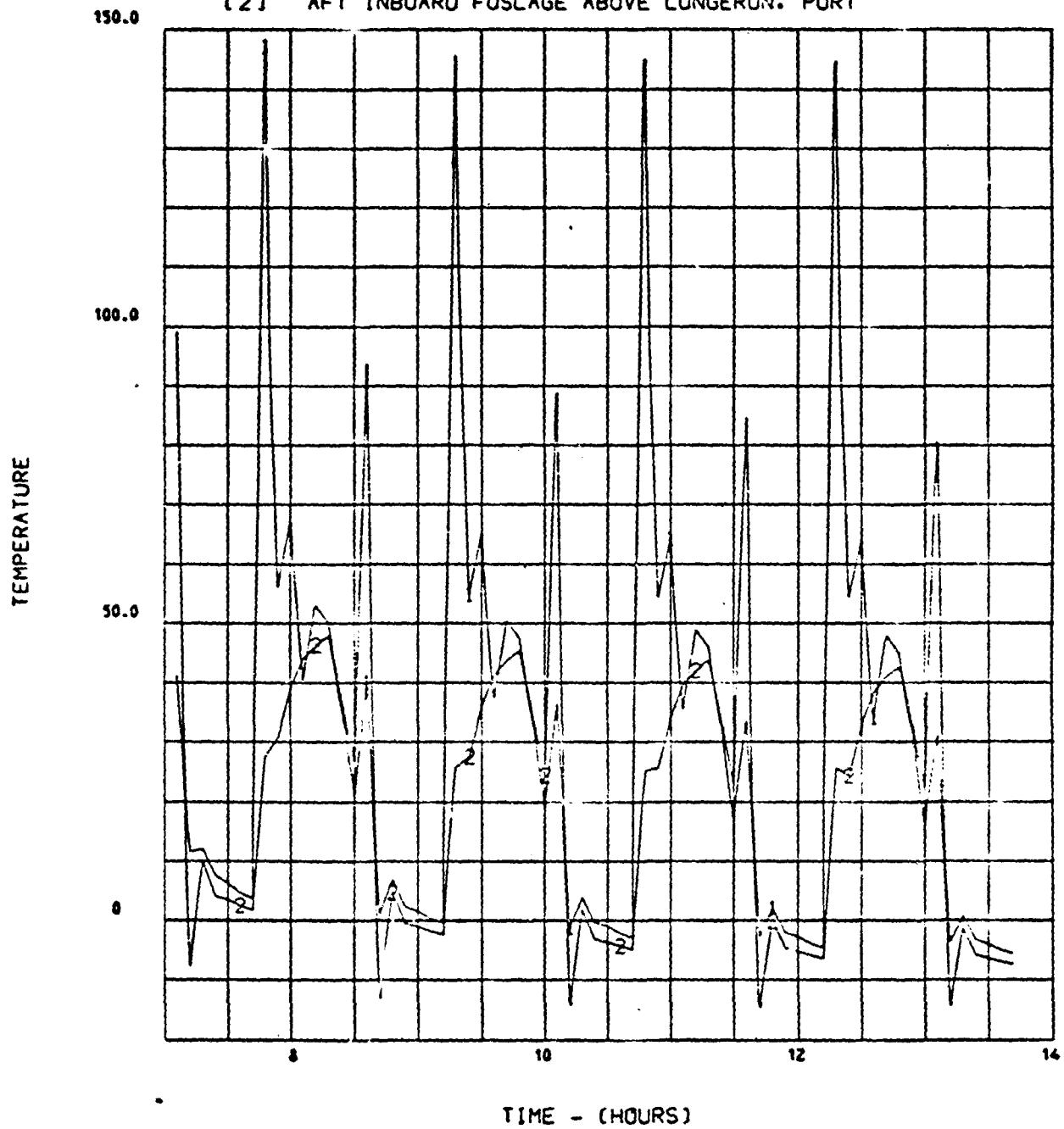


Figure 58

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD BOTTOM FUSELAGE. PCRT
[2] AFT BOTTOM FUSELAGE. PORT

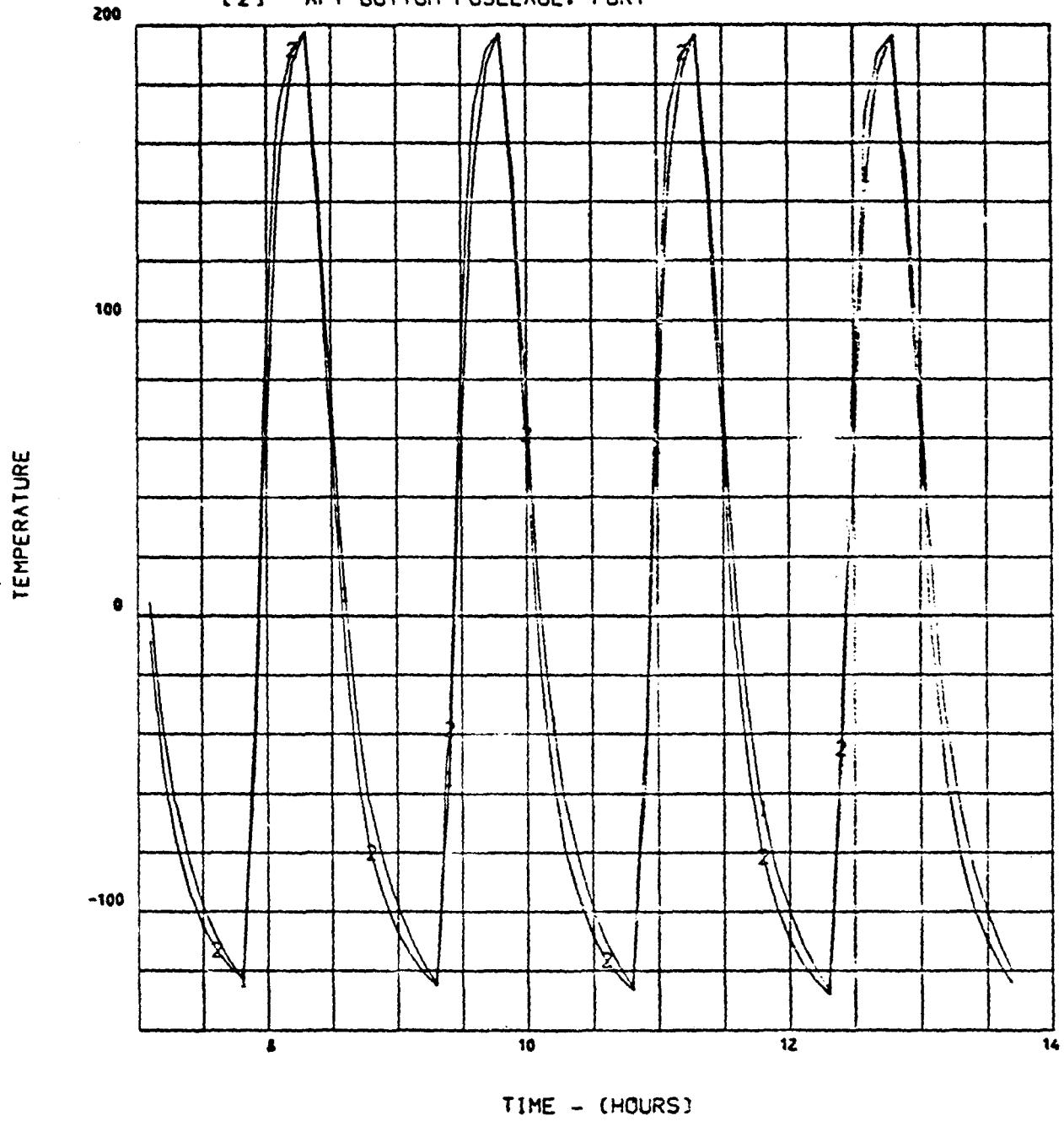


Figure 59

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.

- [1] FWD PLB DOORS. PORT
- [2] FWD PLB DOORS. PORT
- [3] AFT PLB DOORS. PORT
- [4] AFT PLB DOORS. PORT

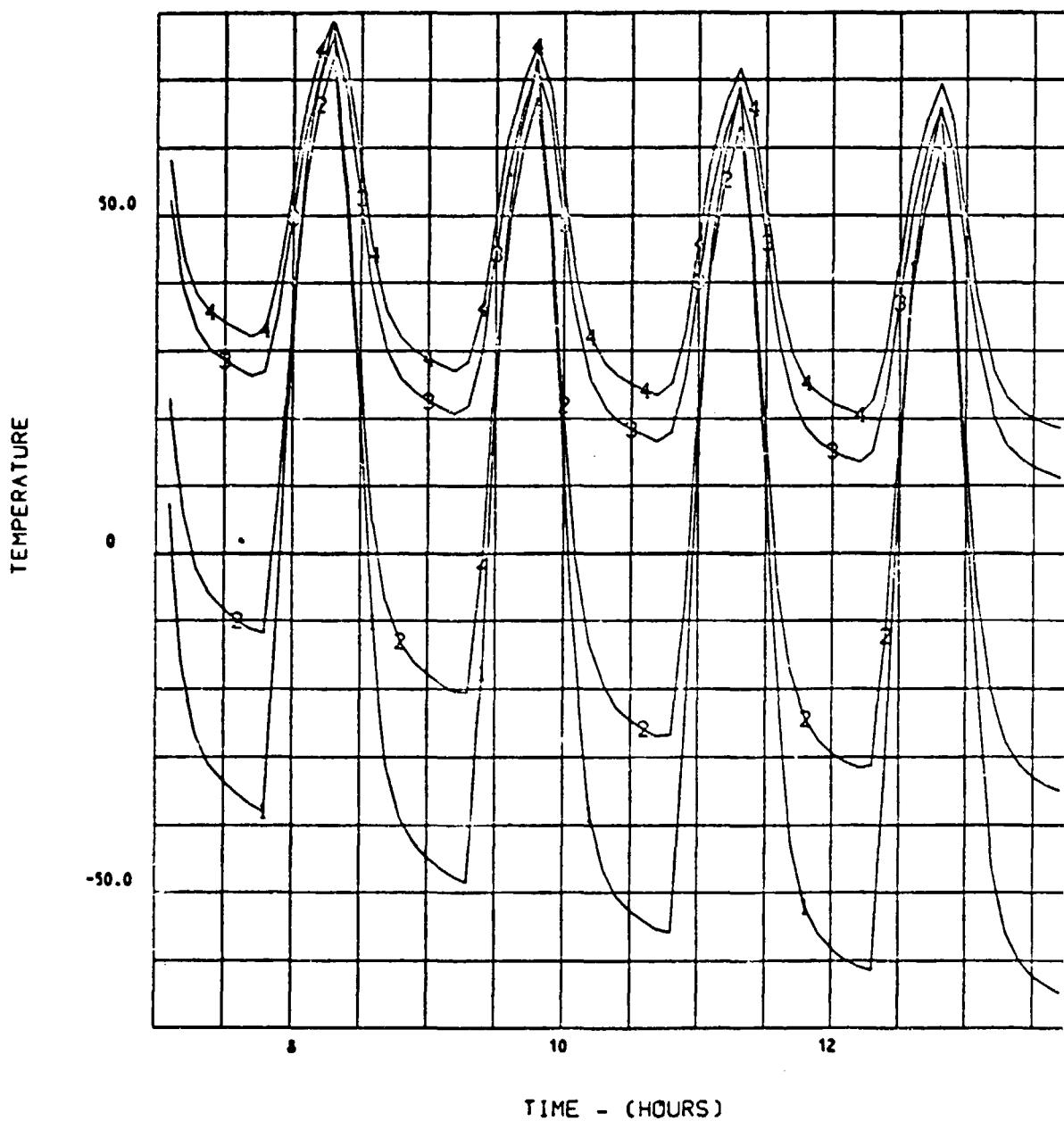


Figure 60

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD RADIATOR. PORT
[2] FWD RADIATOR. PORT
[3] AFT RADIATOR. PORT
[4] AFT RADIATOR. PORT

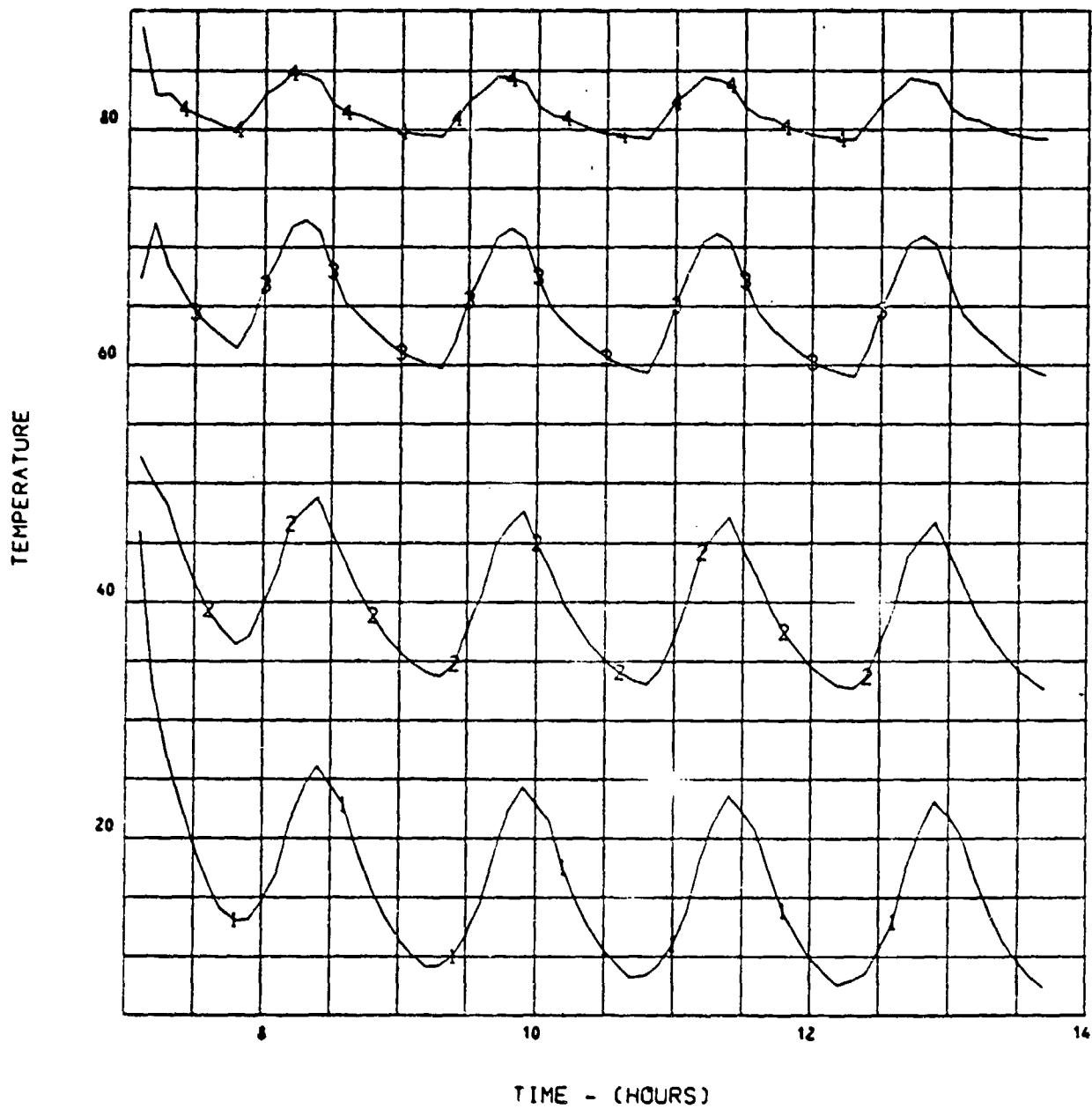


Figure 61

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD BULKHO BOTTOM
[2] FWD BULKHO TOP

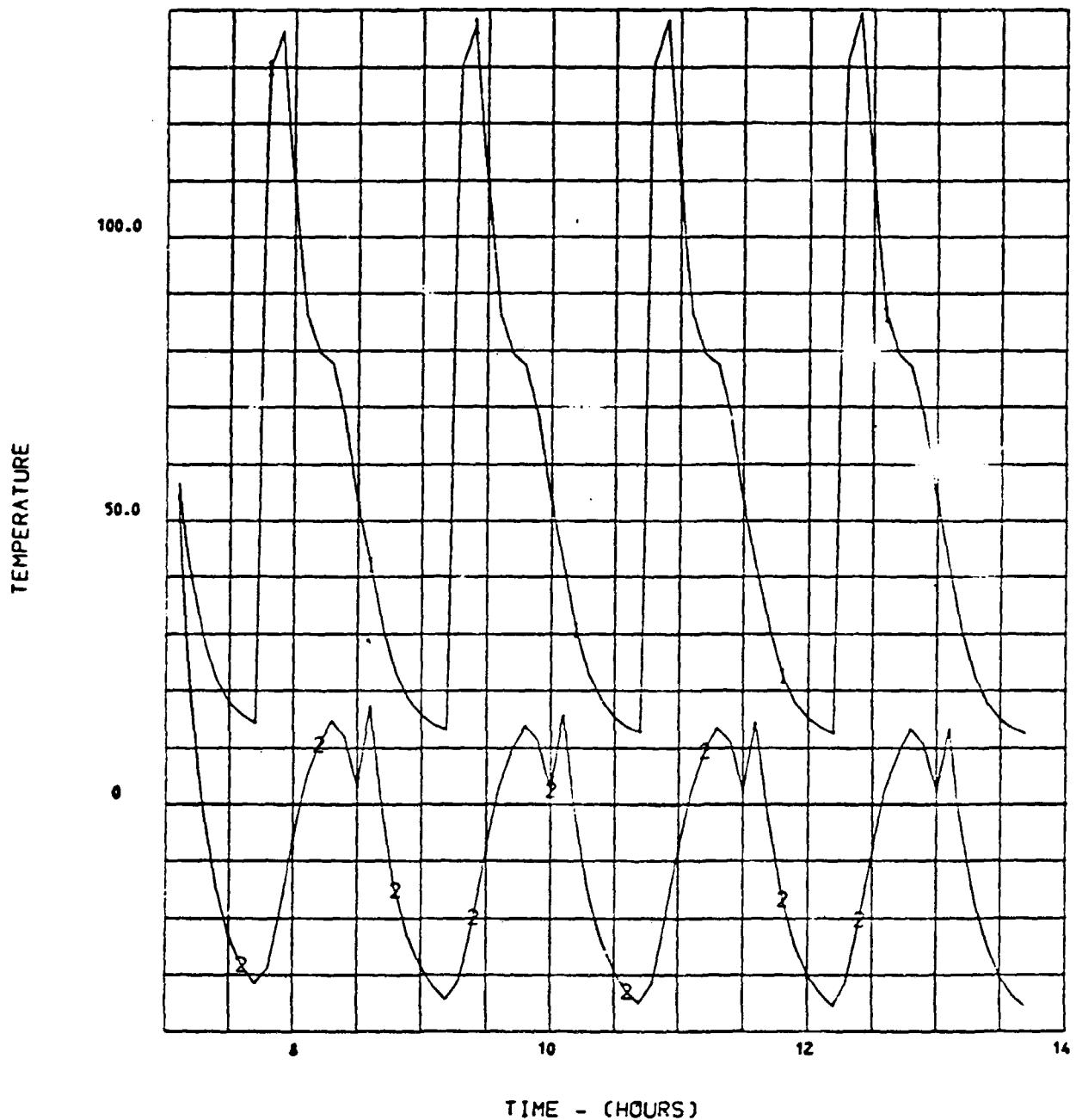


Figure 62

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] UPPER FWD PLB LINER. PORT
[2] UPPER FWD PLB LINER. PORT
[3] UPPER FWD PLB LINER. PORT
[4] UPPER FWD PLB LINER. PORT

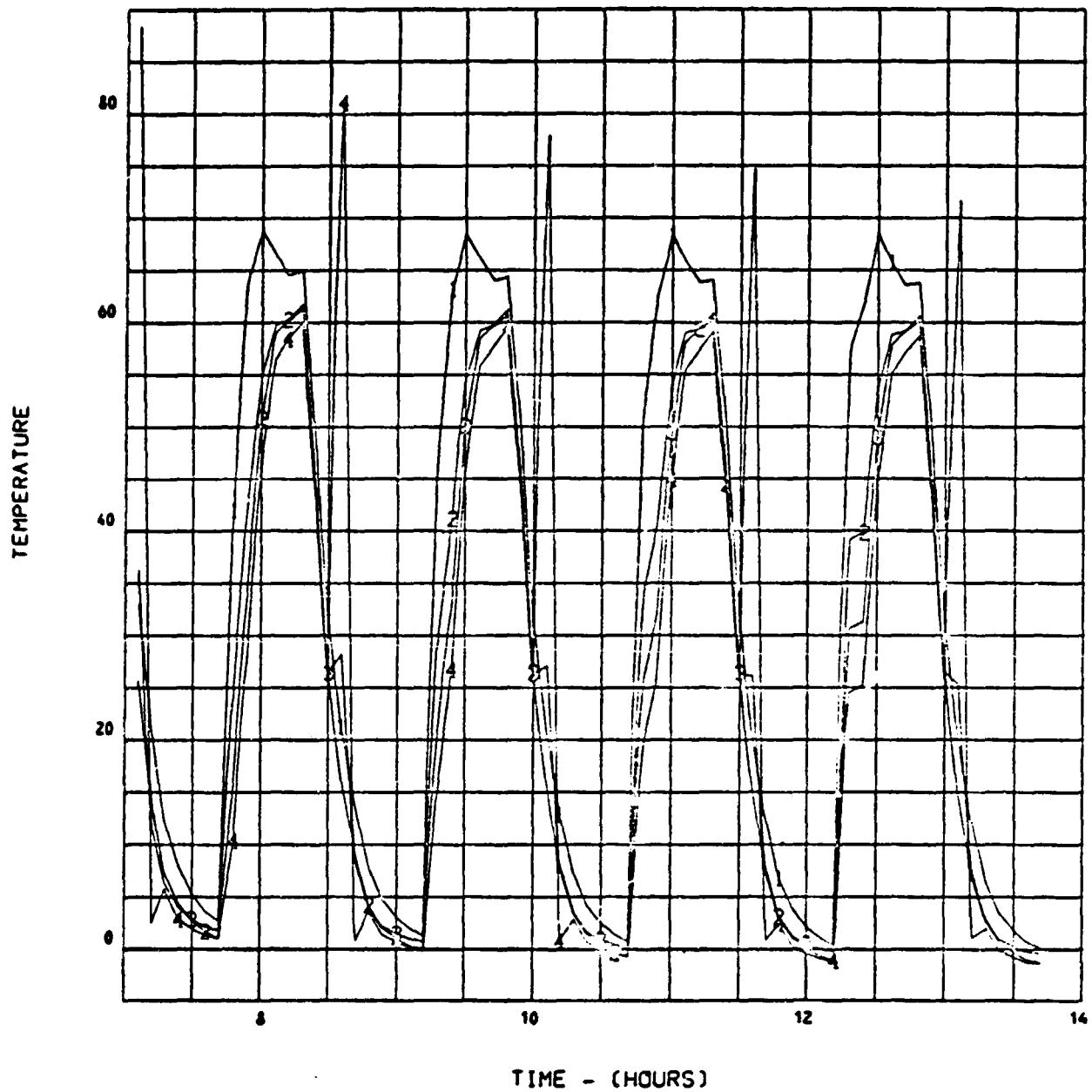
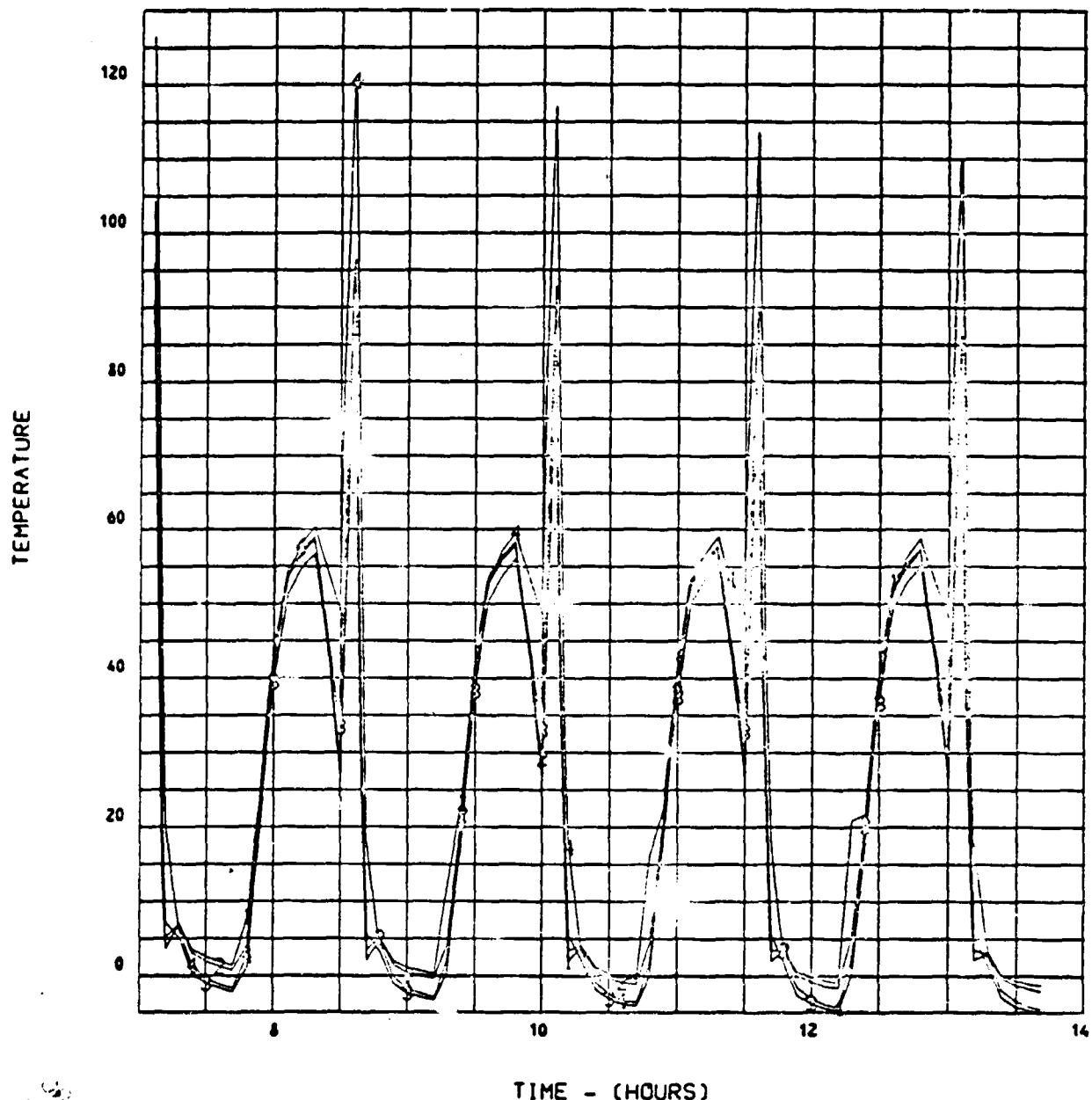


Figure 63

COMPARISON "136 NODE" PLB MODEL. OR3GEN NPT=6 VS TRJ.
[1] UPPER AFT PLB LINER. PORT
[2] UPPER AFT PLB LINER. PORT
[3] UPPER AFT PLB LINER. PORT
[4] UPPER AFT PLB LINER. PORT



TIME - (HOURS)

Figure 64

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.

- [1] LOWER FWD PLB LINER. PORT
- [2] LOWER FWD PLB LINER. PORT
- [3] LOWER FWD PLB LINER. PORT
- [4] LOWER FWD PLB LINER. PORT

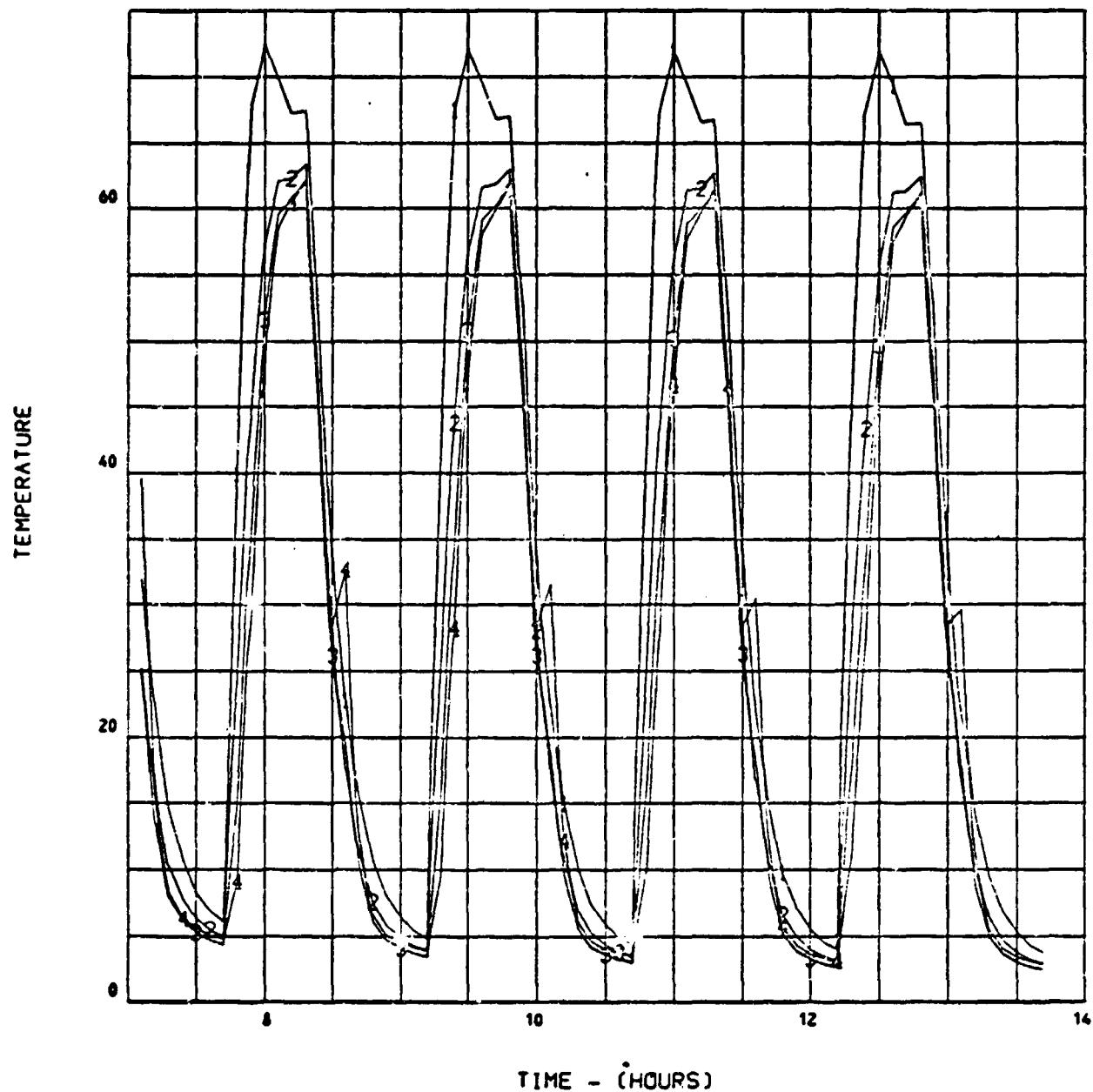


Figure 65

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.

- [1] LOWER AFT PLB LINER. PCRT
- [2] LOWER AFT PLB LINER. PCRT
- [3] LOWER AFT PLB LINER. PCRT
- [4] LOWER AFT PLB LINER. PCRT

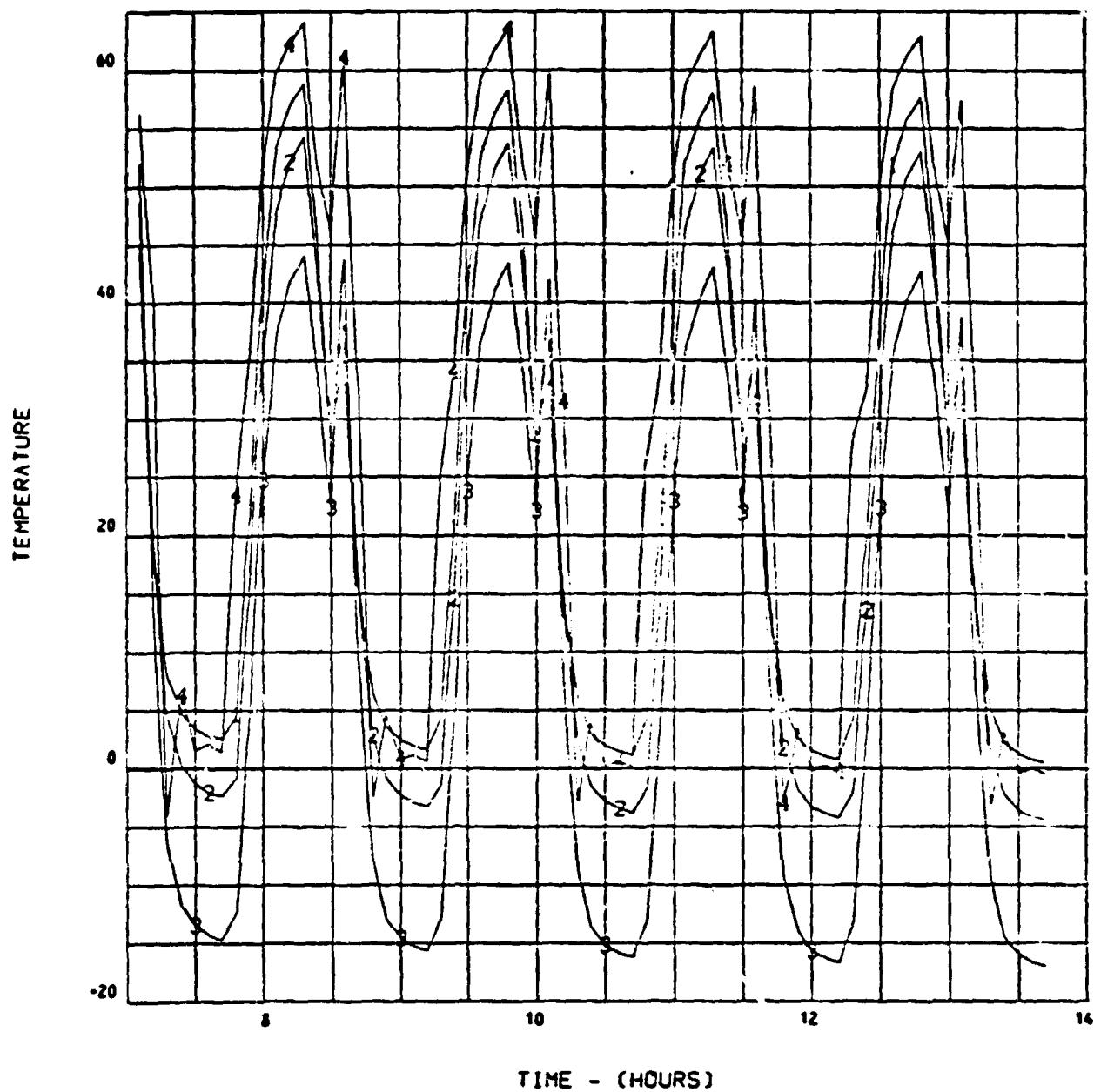


Figure 66

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD LONGERON. PORT
[2] AFT LONGERON. PORT

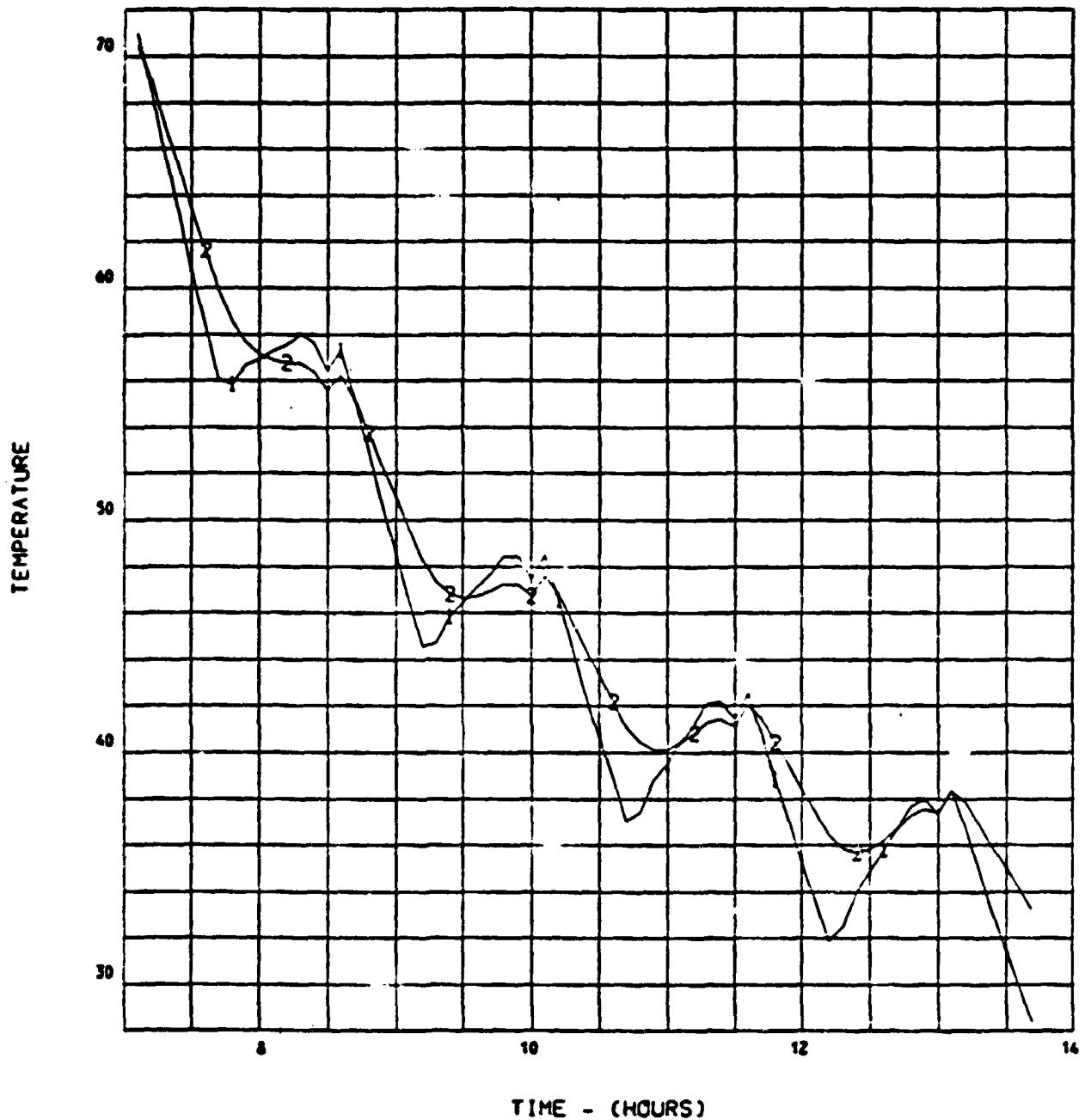


Figure 67

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] AFT BULKHD BOTTOM
[2] AFT BULKHD TOP

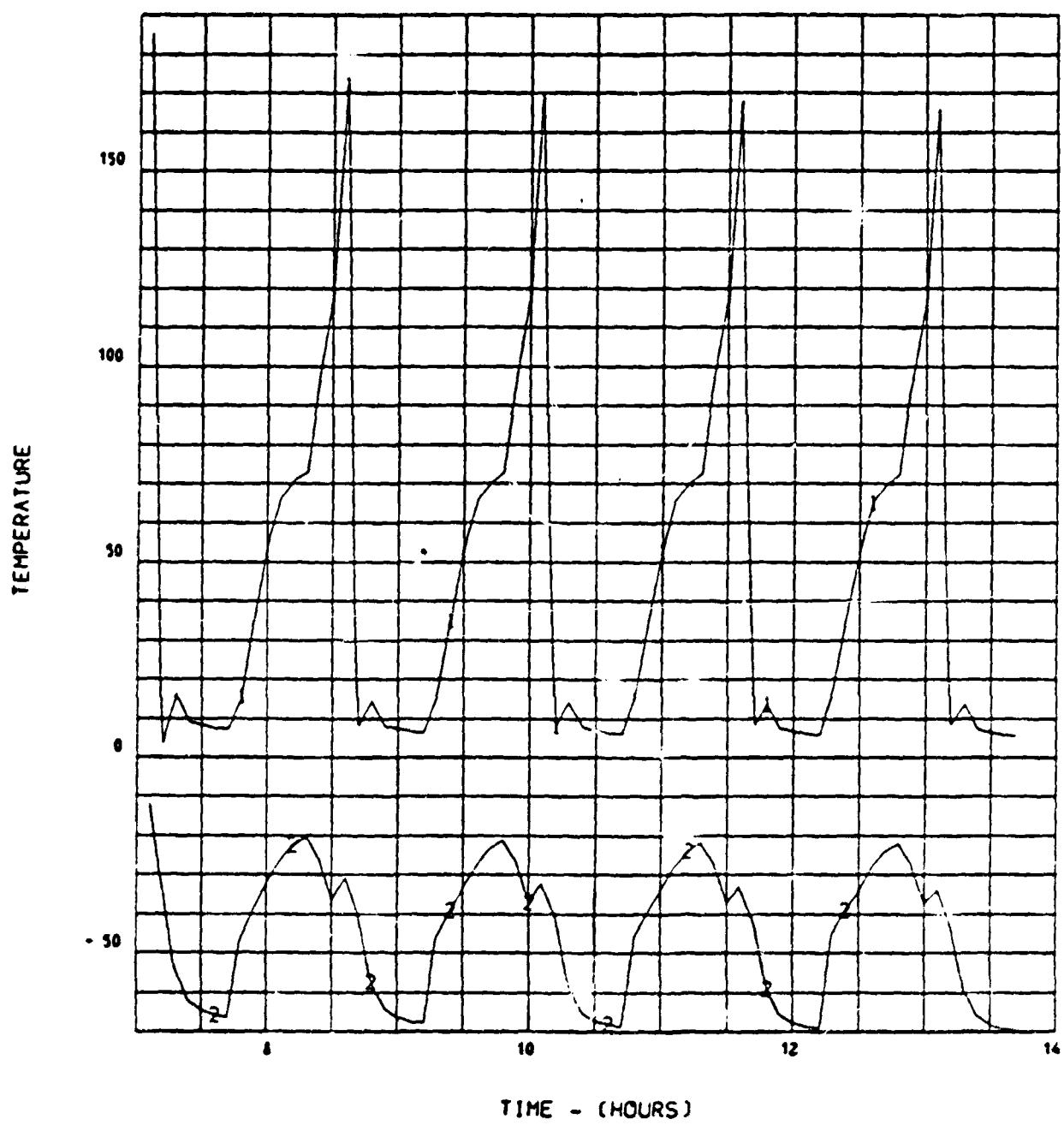


Figure 68

COMPARISON "136 NODE" PLB MODEL. OPBGEN NPT=6 VS TRJ.
(1) FWD BULKHD BOTTOM BELOW PLB LINER

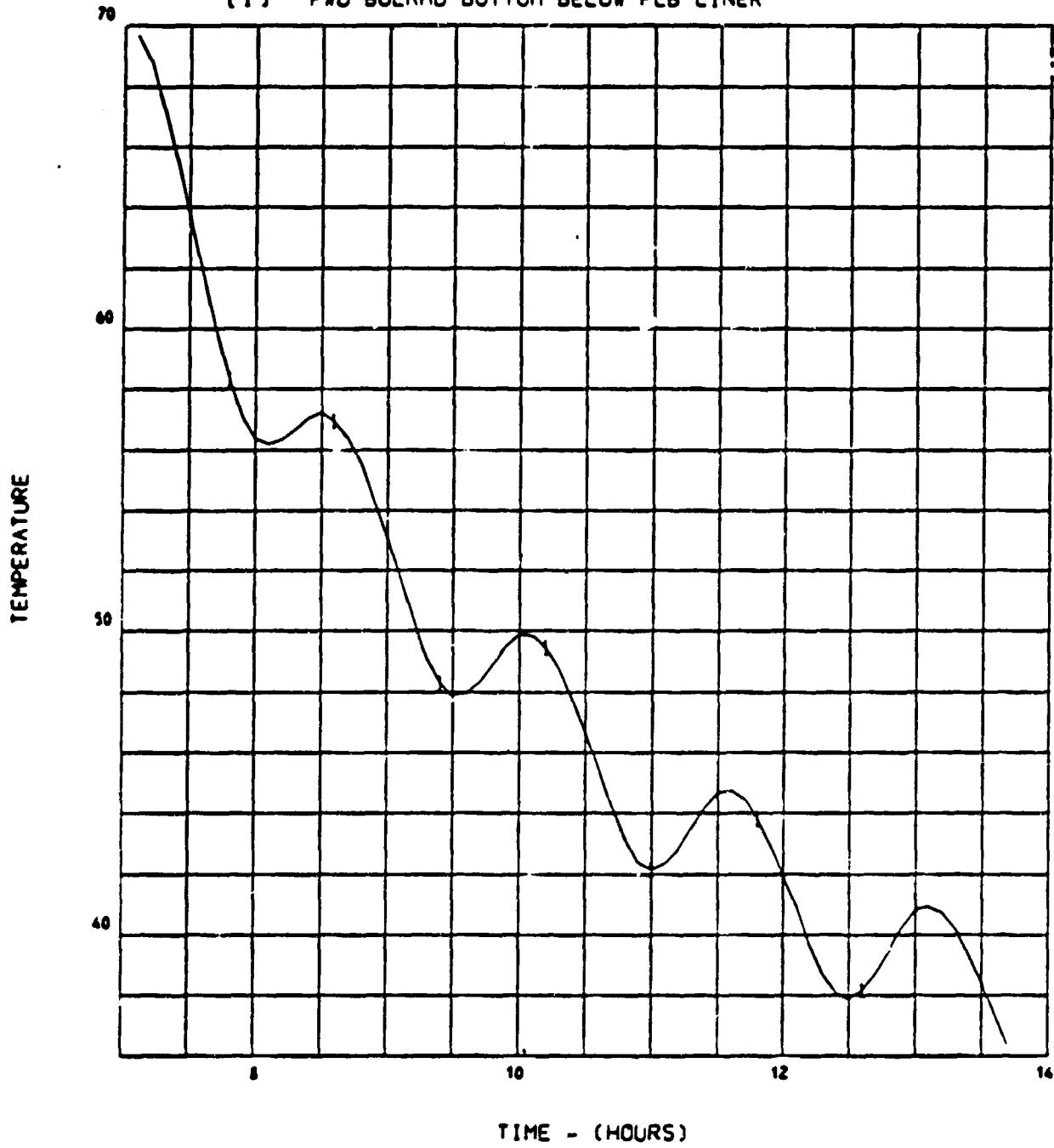
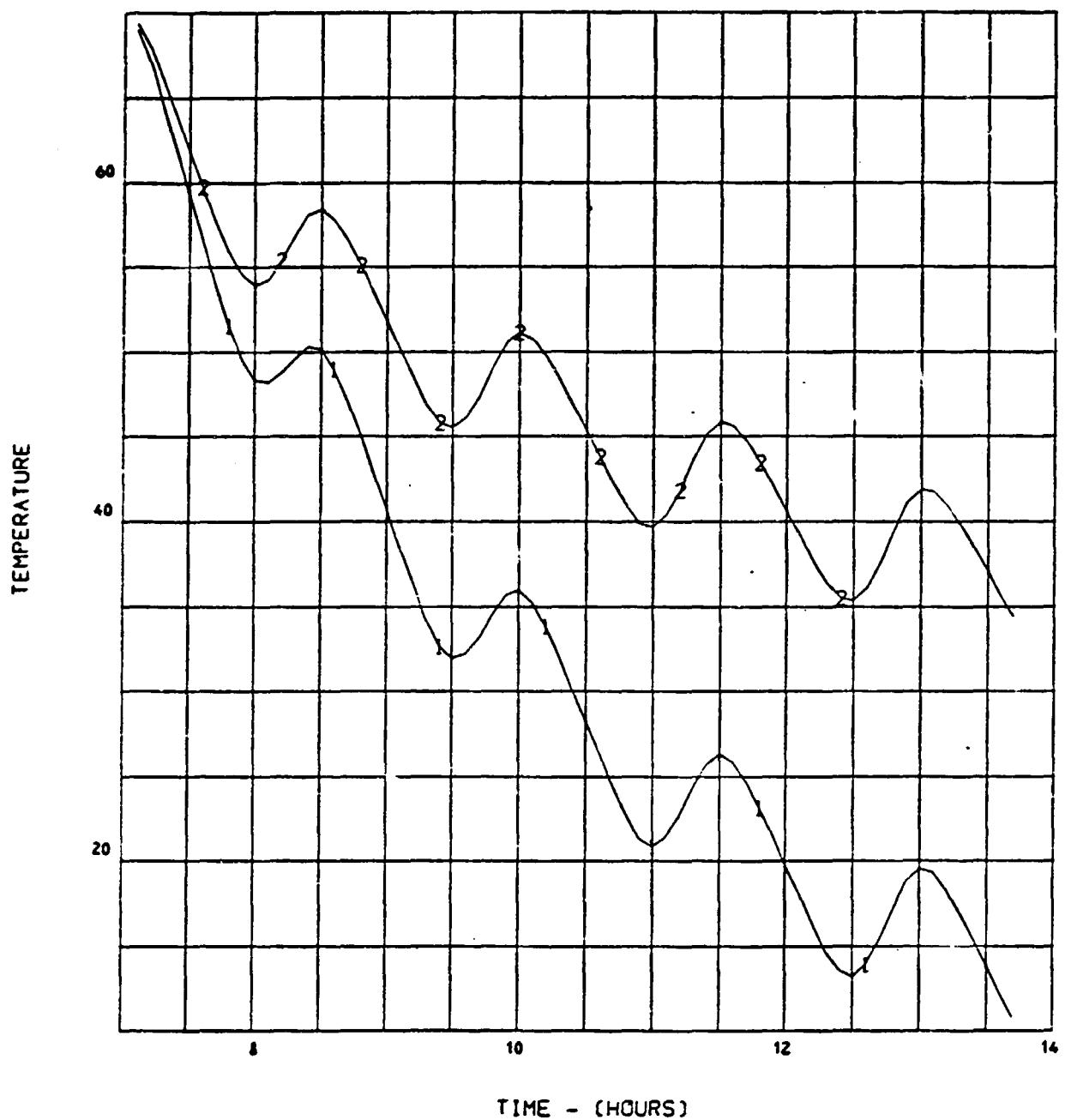


Figure 69

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD SIDE FUSELAGE STRUCTURE. PORT
[2] AFT SIDE FUSELAGE STRUCTURE. PORT



TIME - (HOURS)

Figure 70

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD BOTTOM FUSELAGE STRUCTURE. PORT
[2] AFT BOTTOM FUSELAGE STRUCTURE. PORT

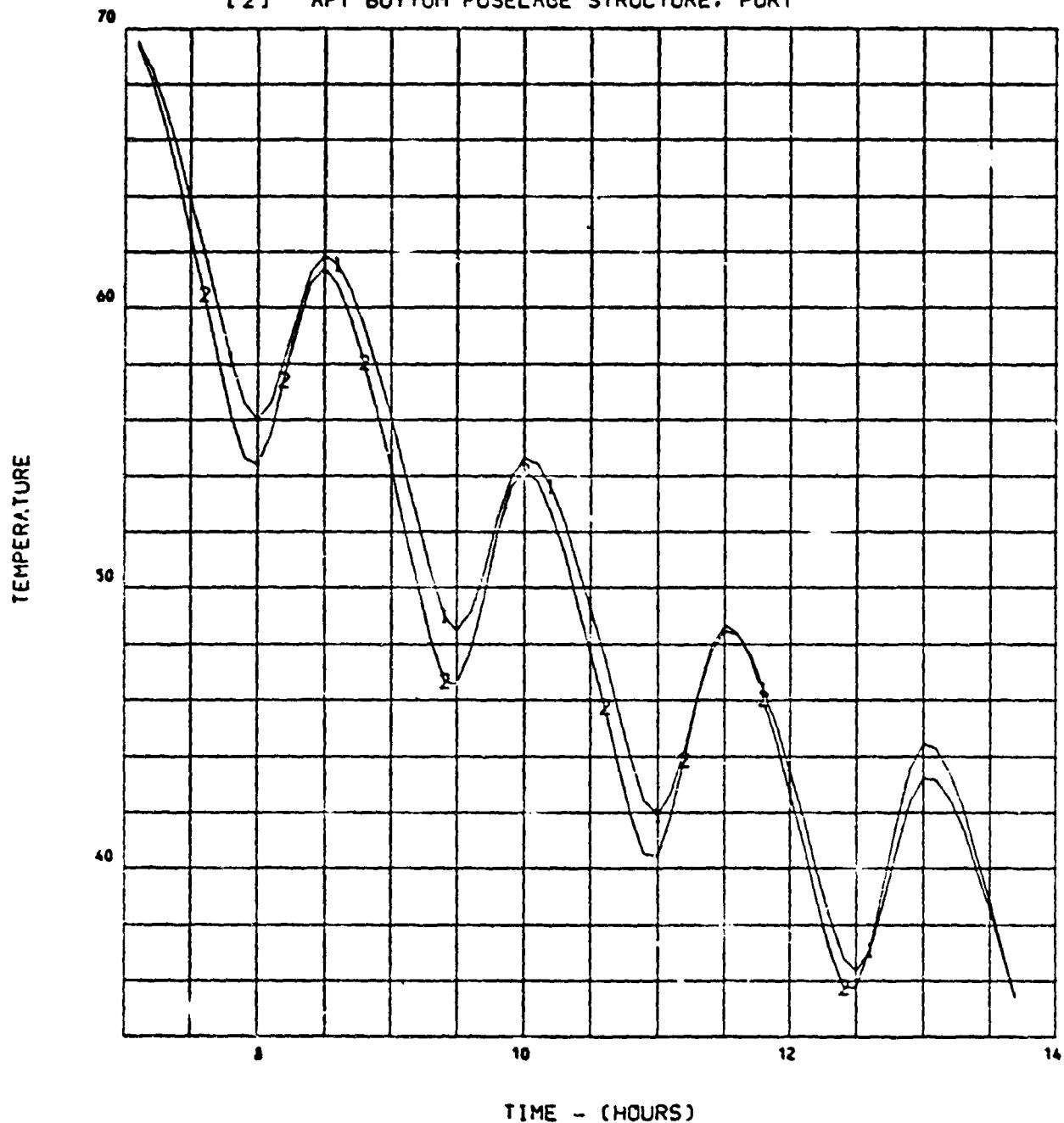


Figure 71

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD PLB DOORS STRUCTURE. PORT
[2] FWD PLB DOORS STRUCTURE. PORT
[3] AFT PLB DOORS STRUCTURE. PORT
[4] AFT PLB DOORS STRUCTURE. PORT

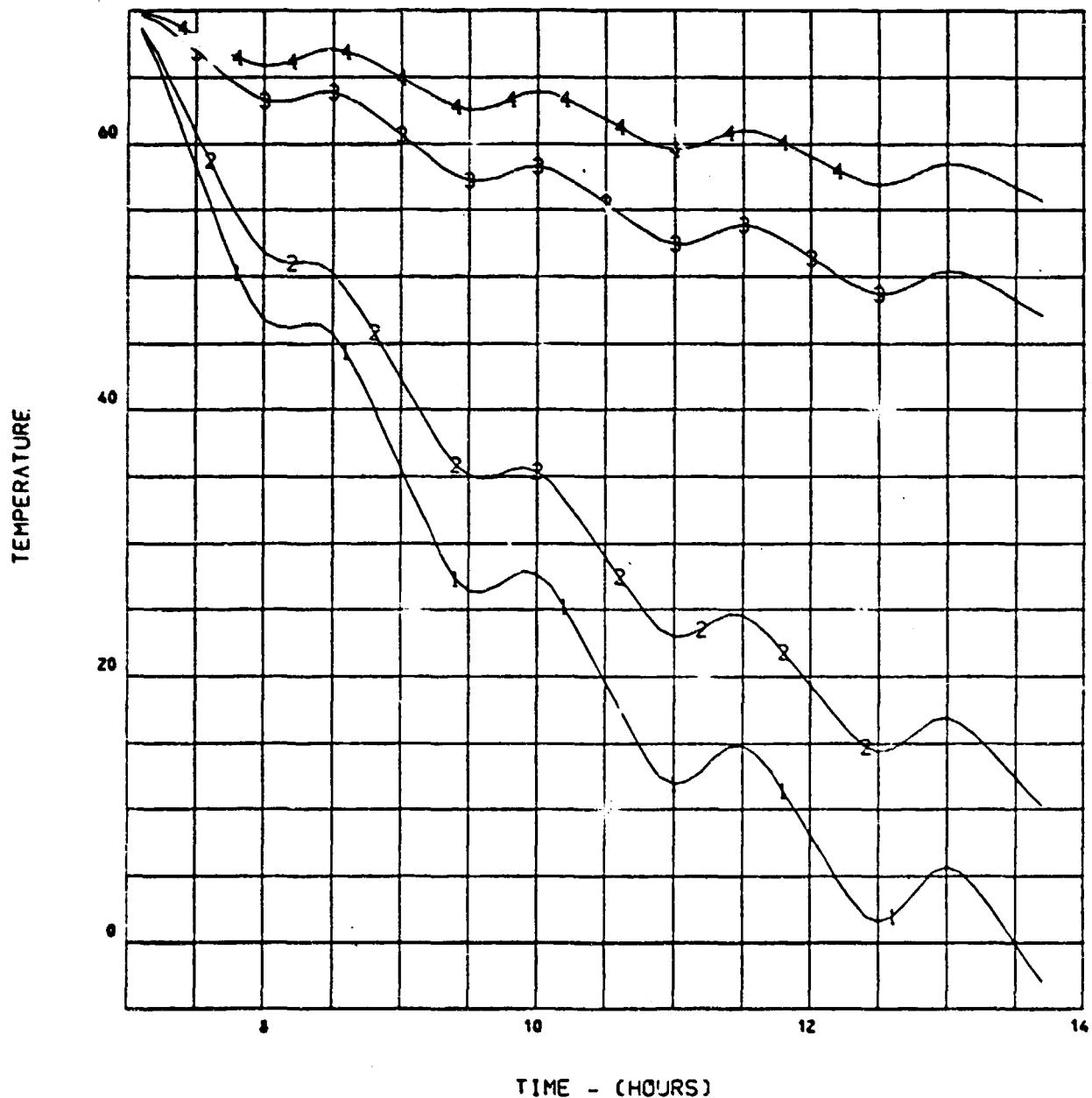


Figure 72

COMPARISON "136 NODE" PLS MODEL. ORBGEN NPT=6 VS TRJ.
[1] AFT BULKHD BOTTOM BELOW PLB LINER

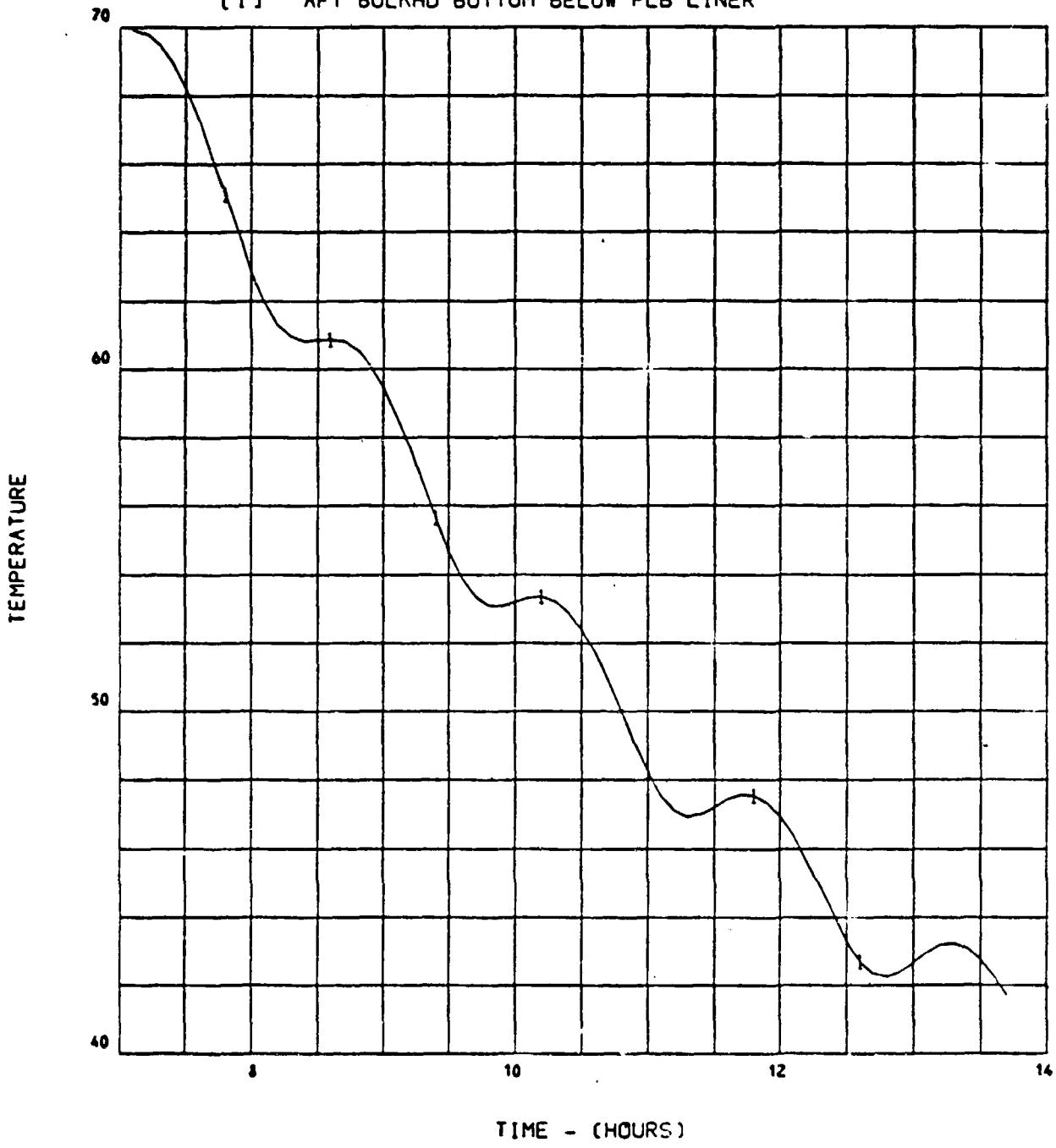


Figure 73

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT

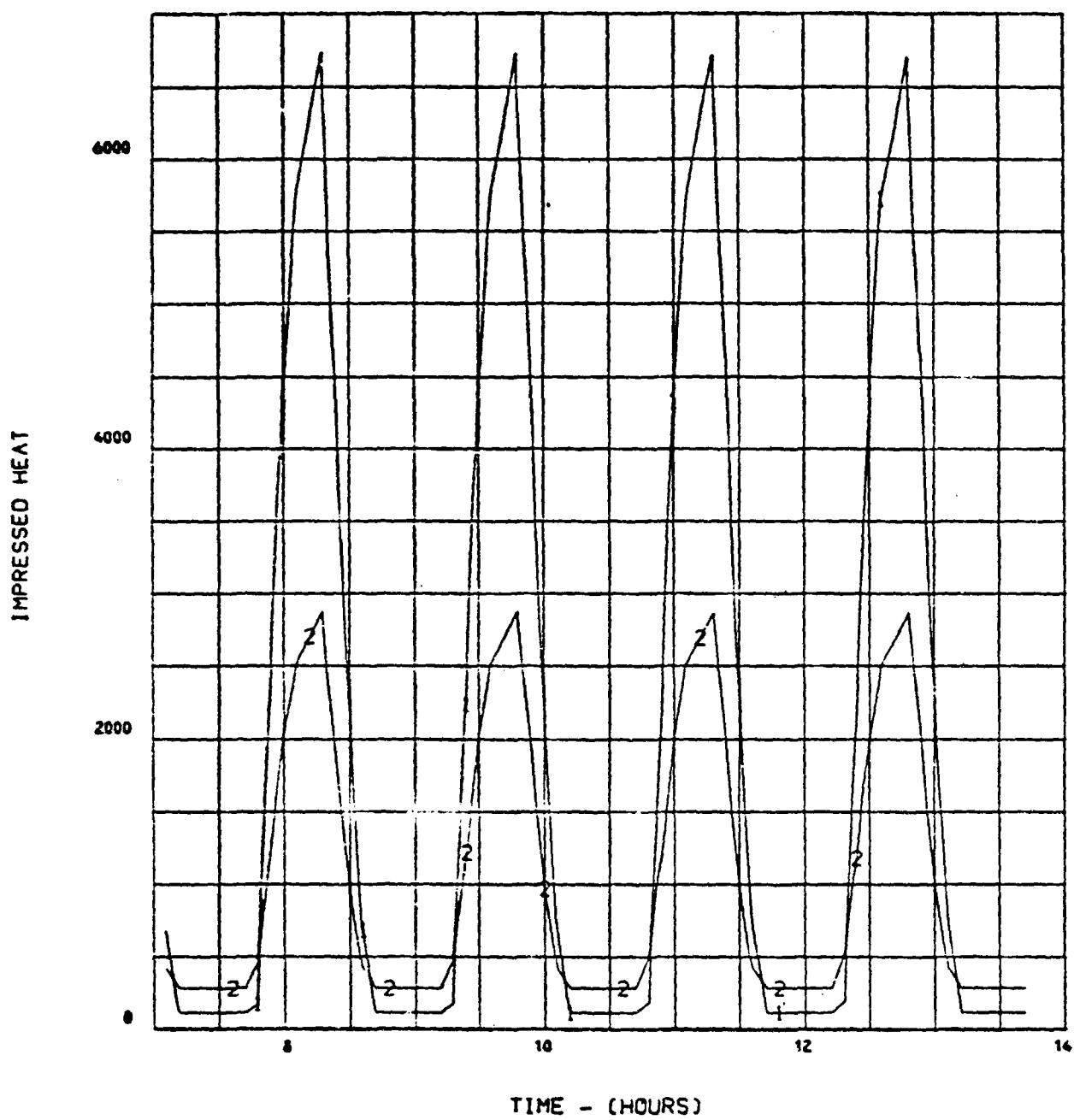


Figure 74

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

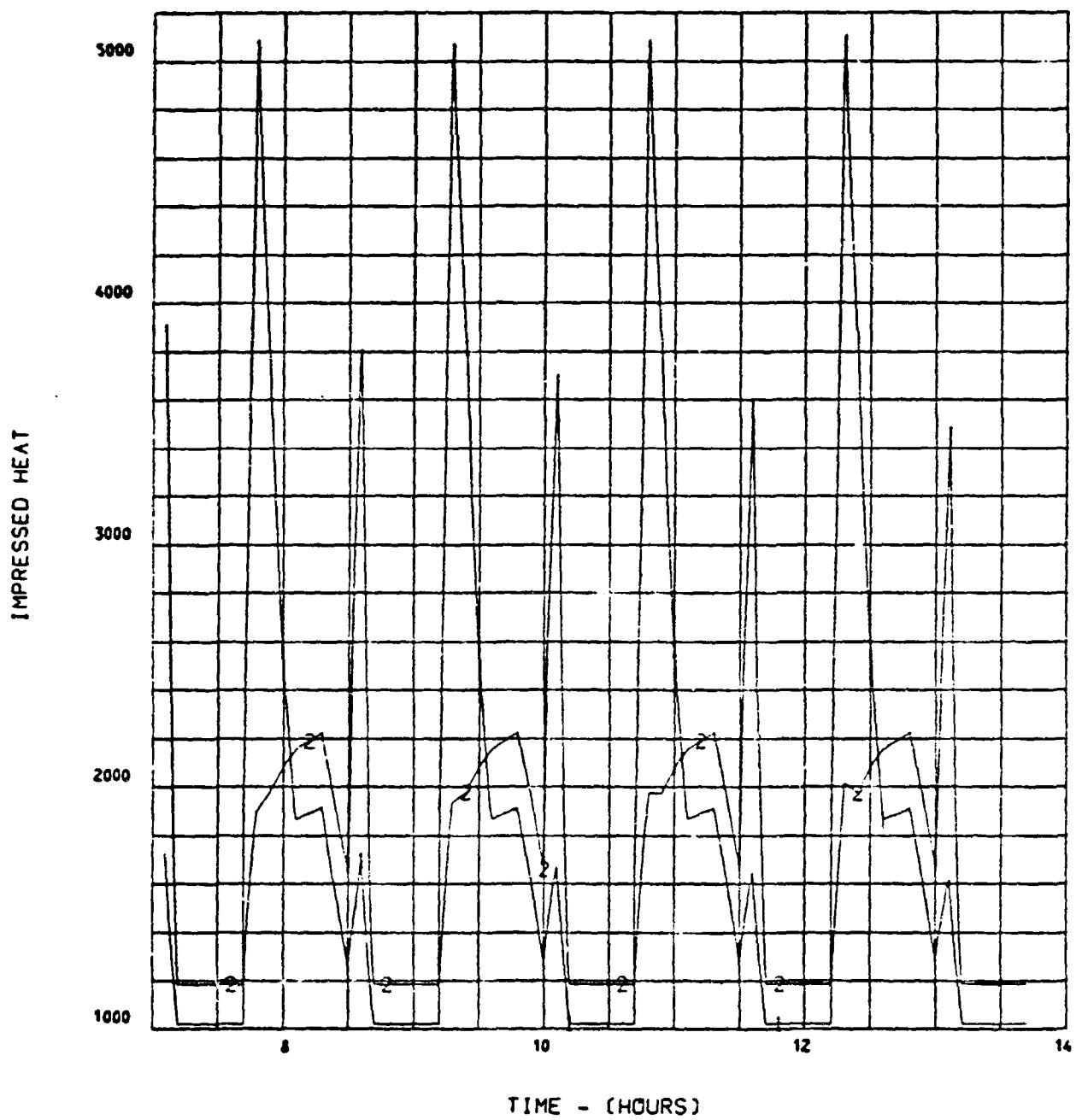


Figure 75

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD BOTTOM FUSELAGE. PORT
[2] AFT BOTTOM FUSELAGE. PORT

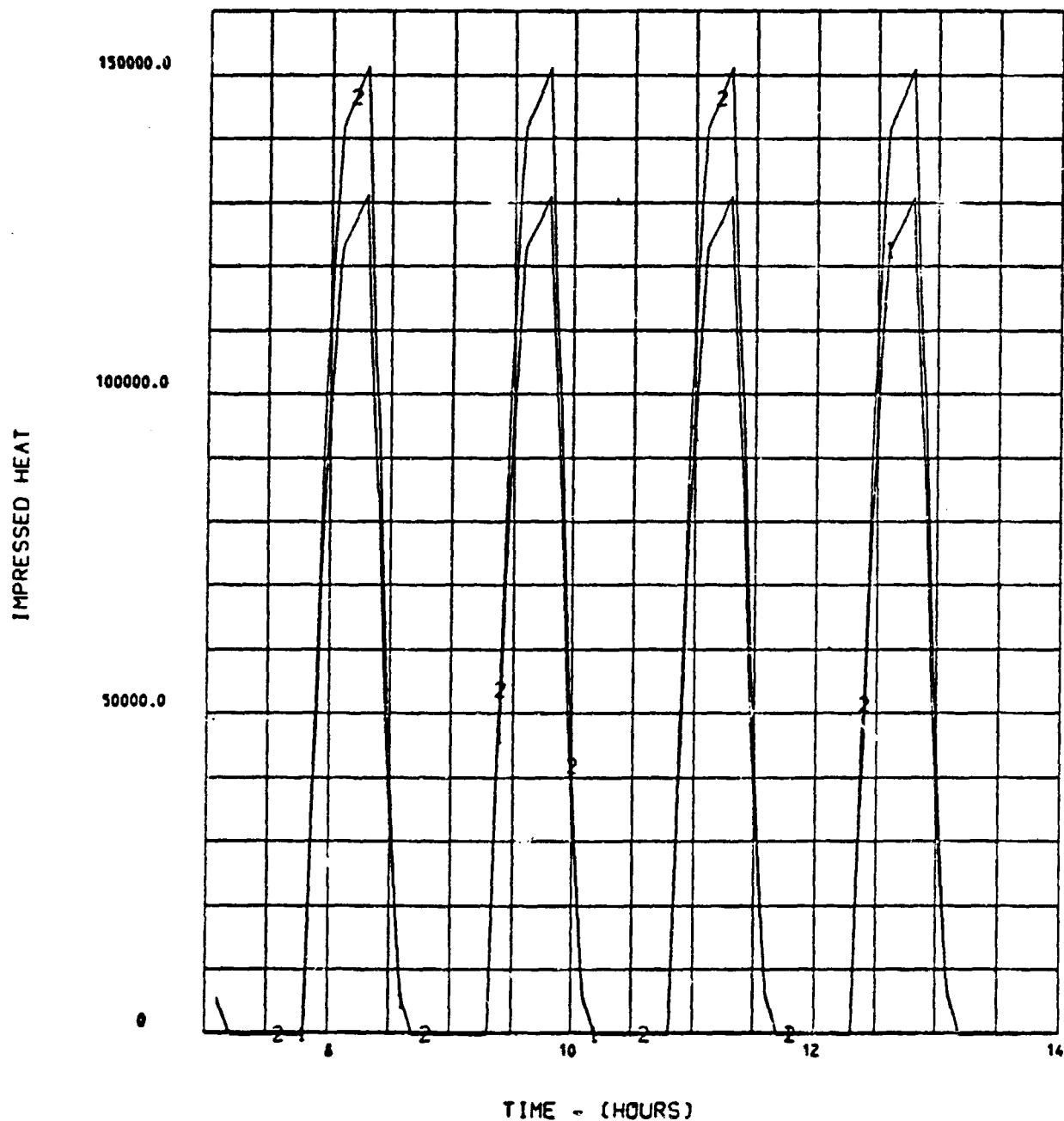


Figure 76

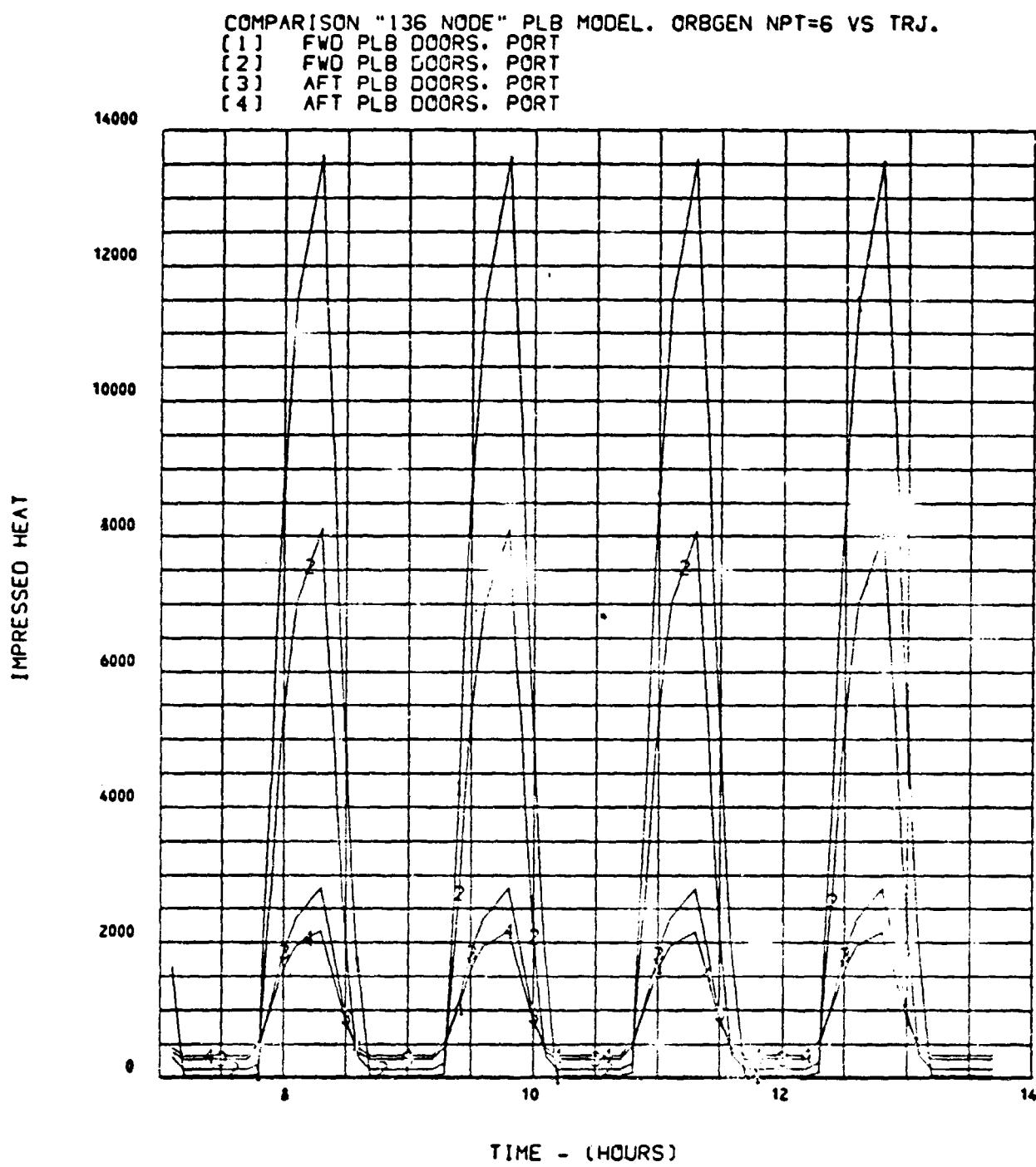


Figure 77

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD RADIATOR. PORT
[2] FWD RADIATOR. PORT
[3] AFT RADIATOR. PORT
[4] AFT RADIATOR. PORT

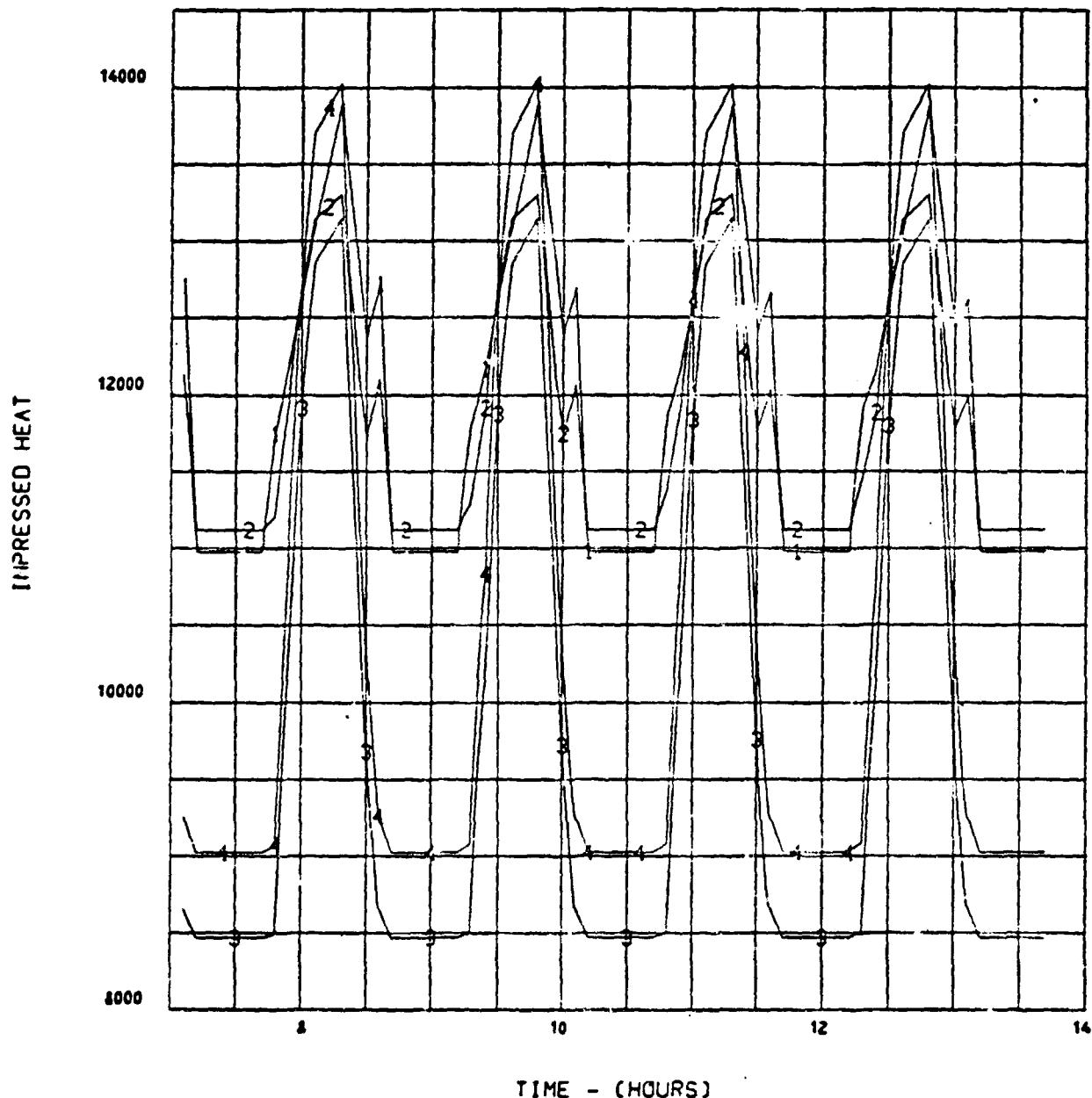


Figure 78

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD BULKHD TOP
[2] FWD BULKHD BOTTOM

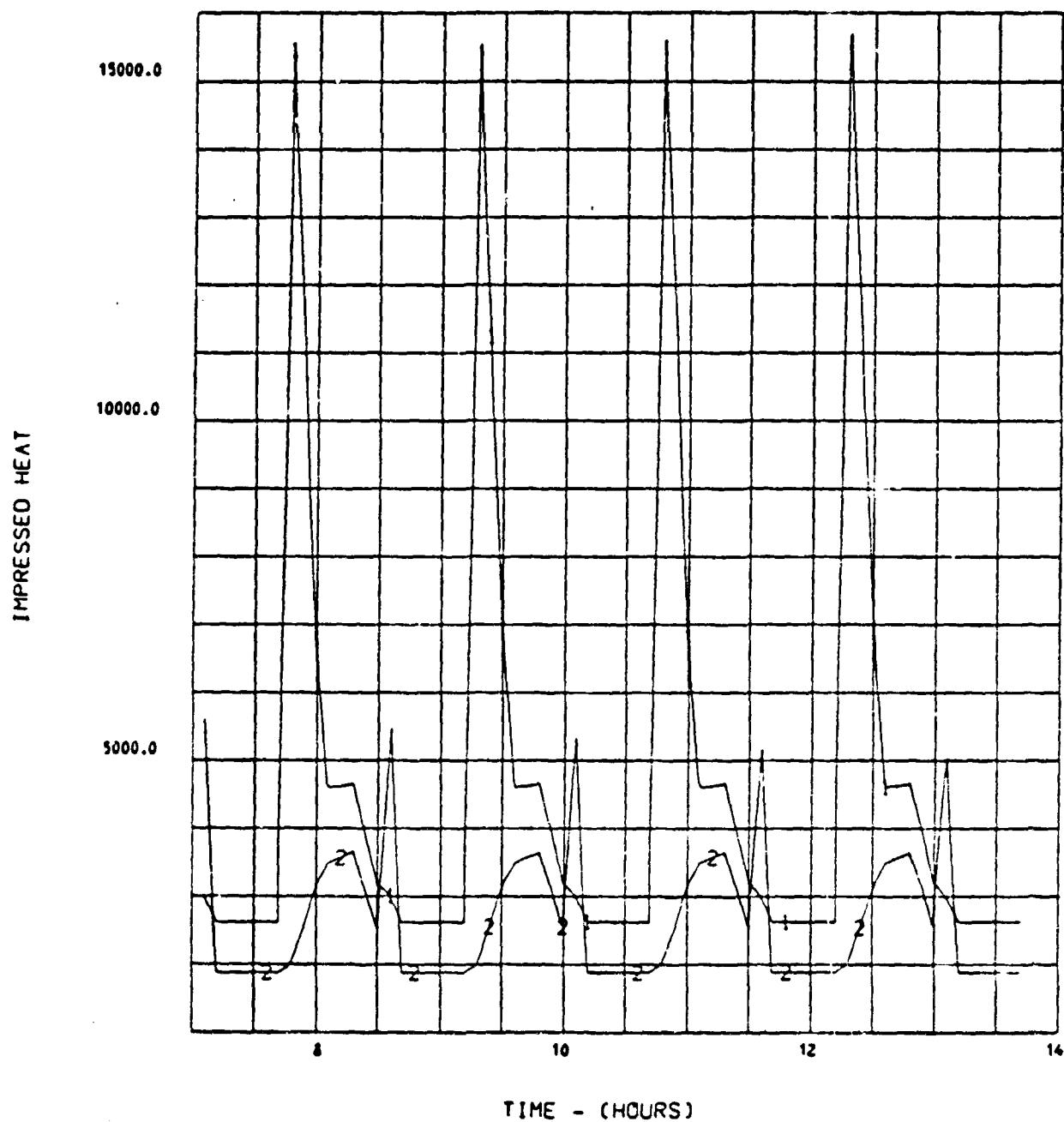


Figure 79

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS 1R.J.
[1] UPPER FWD PLB LINER. PORT
[2] UPPER FWD PLB LINER. PORT
[3] UPPER FWD PLB LINER. PORT
[4] UPPER FWD PLB LINER. PORT

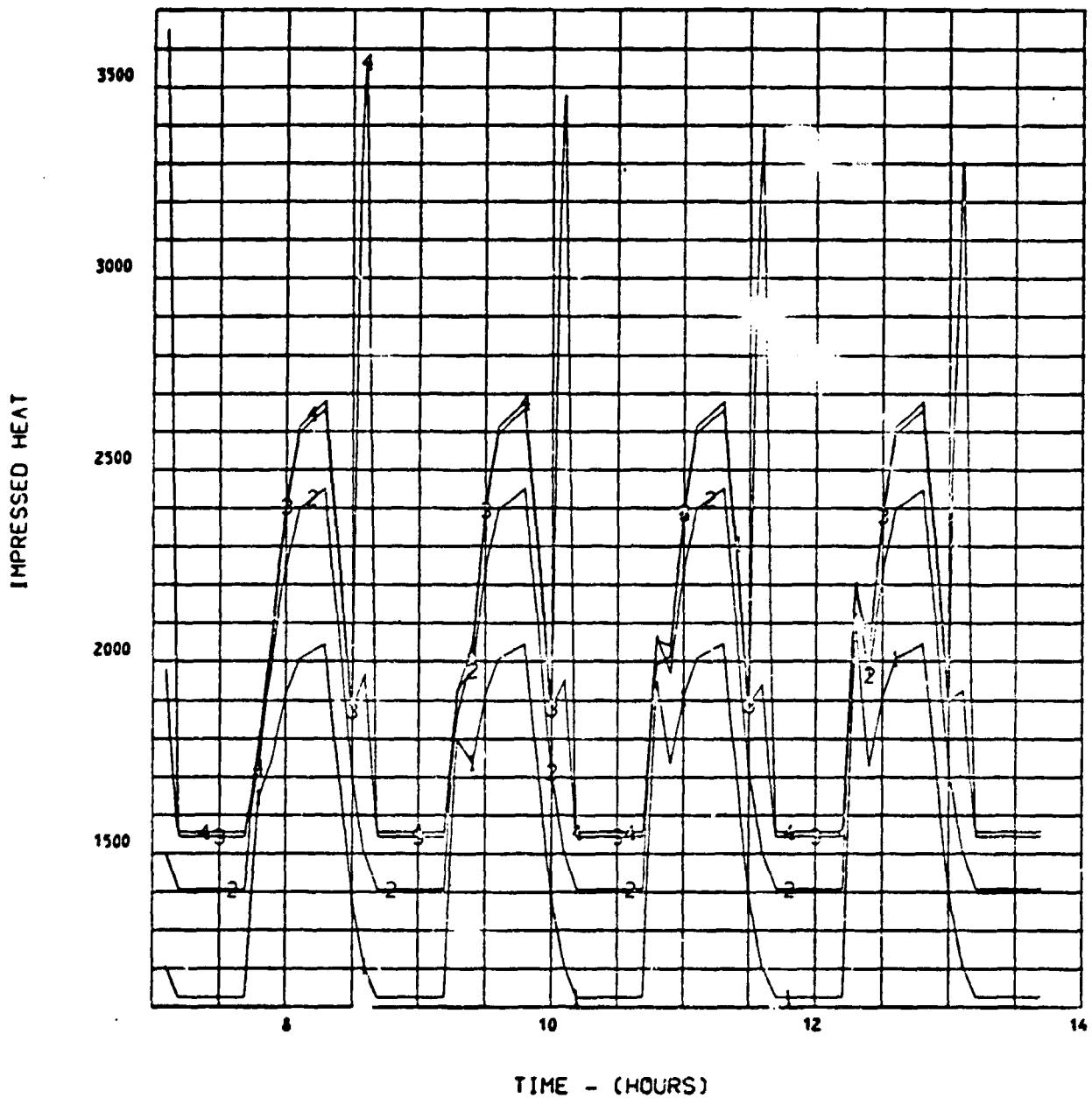


Figure 80

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.

- [1] UPPER AFT PLB LINER. PORT
- [2] UPPER AFT PLB LINER. PORT
- [3] UPPER AFT PLB LINER. PCRT
- [4] UPPER AFT PLB LINER. PORT

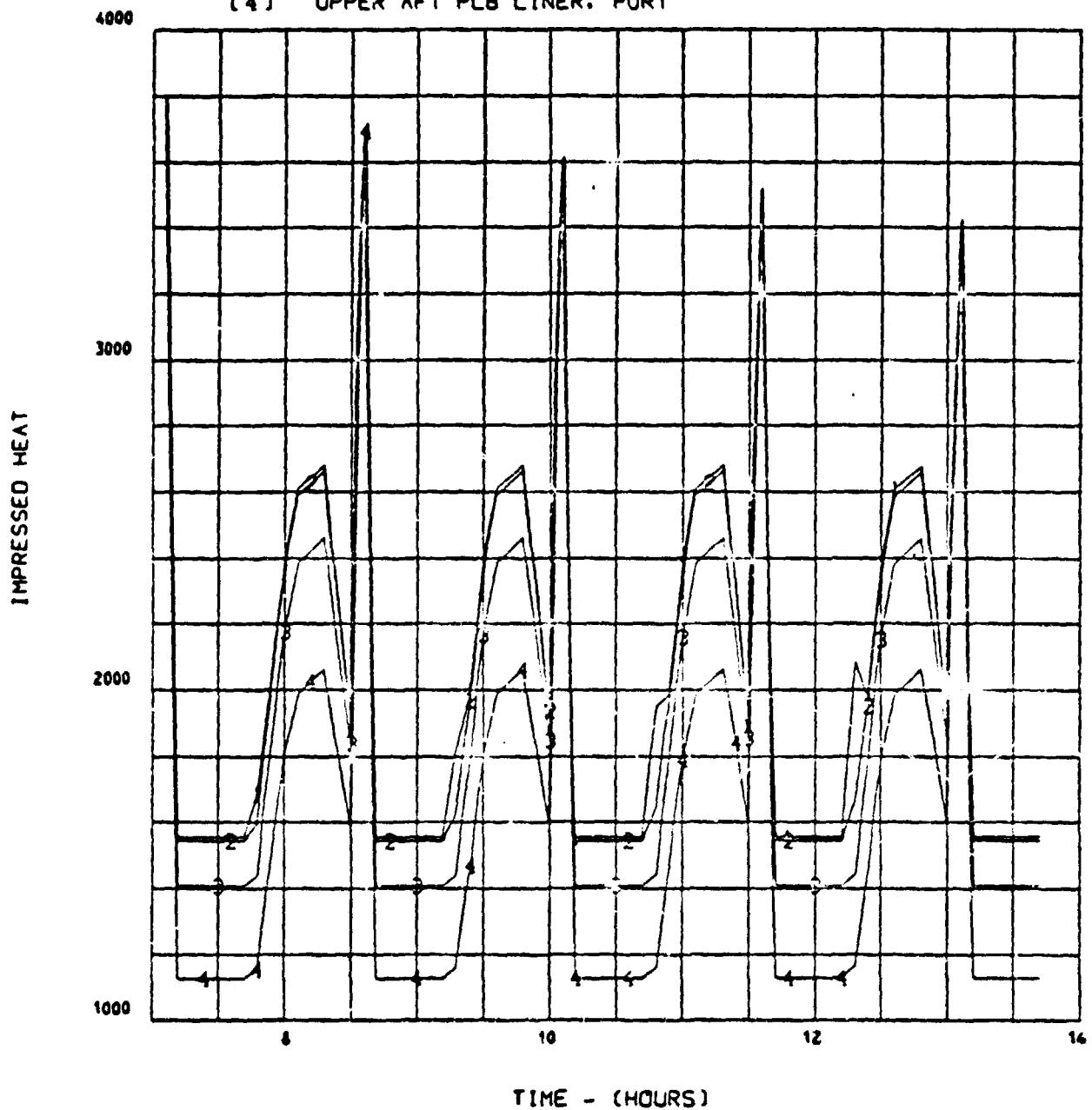


Figure 81

COMPARISON "136 NOD" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] LOWER FWD PLB LINER. PORT
[2] LOWER FWD PLB LINER. PORT
[3] LOWER FWD PLB LINER. PORT
[4] LOWER FWD PLB LINER. PORT

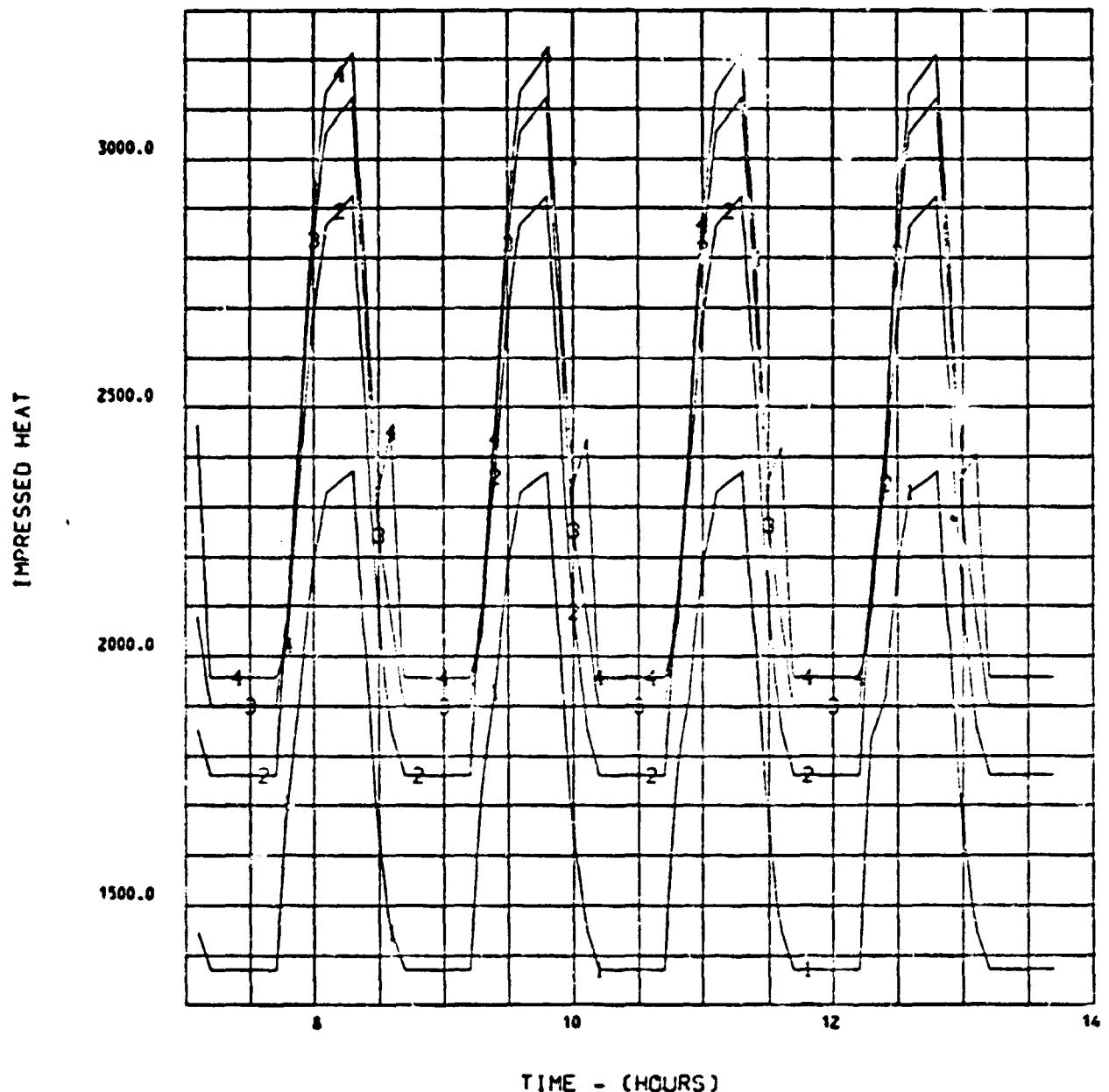


Figure 82

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] LOWER AFT PLB LINER. PC. 1
[2] LOWER AFT PLB LINER. PORT
[3] LOWER AFT PLB LINER. PORT
[4] LOWER AFT PLB LINER. PORT

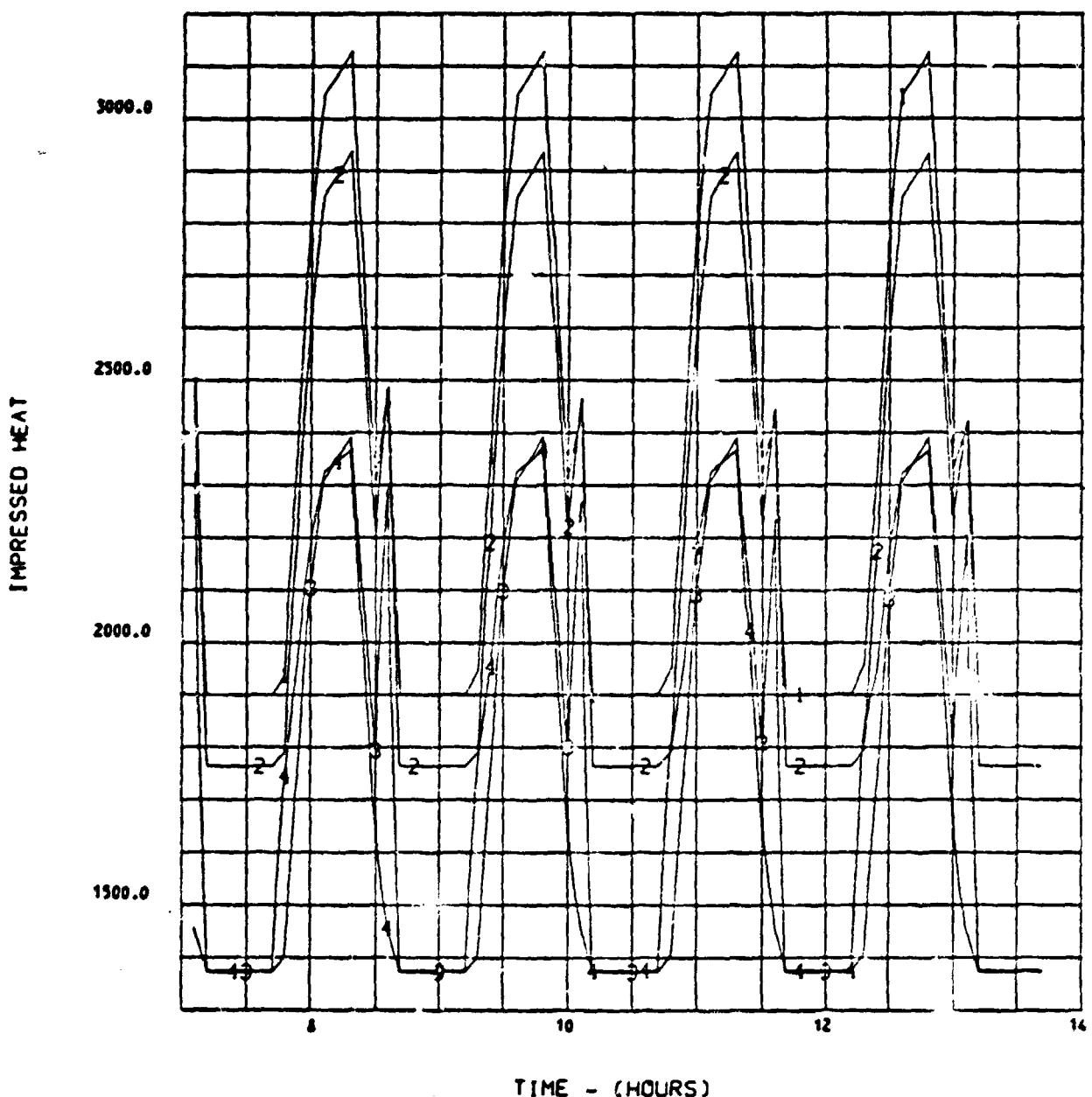


Figure 83

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=6 VS TRJ.
[1] FWD LONGERON, PORT
[2] AFT LONGERON, PORT

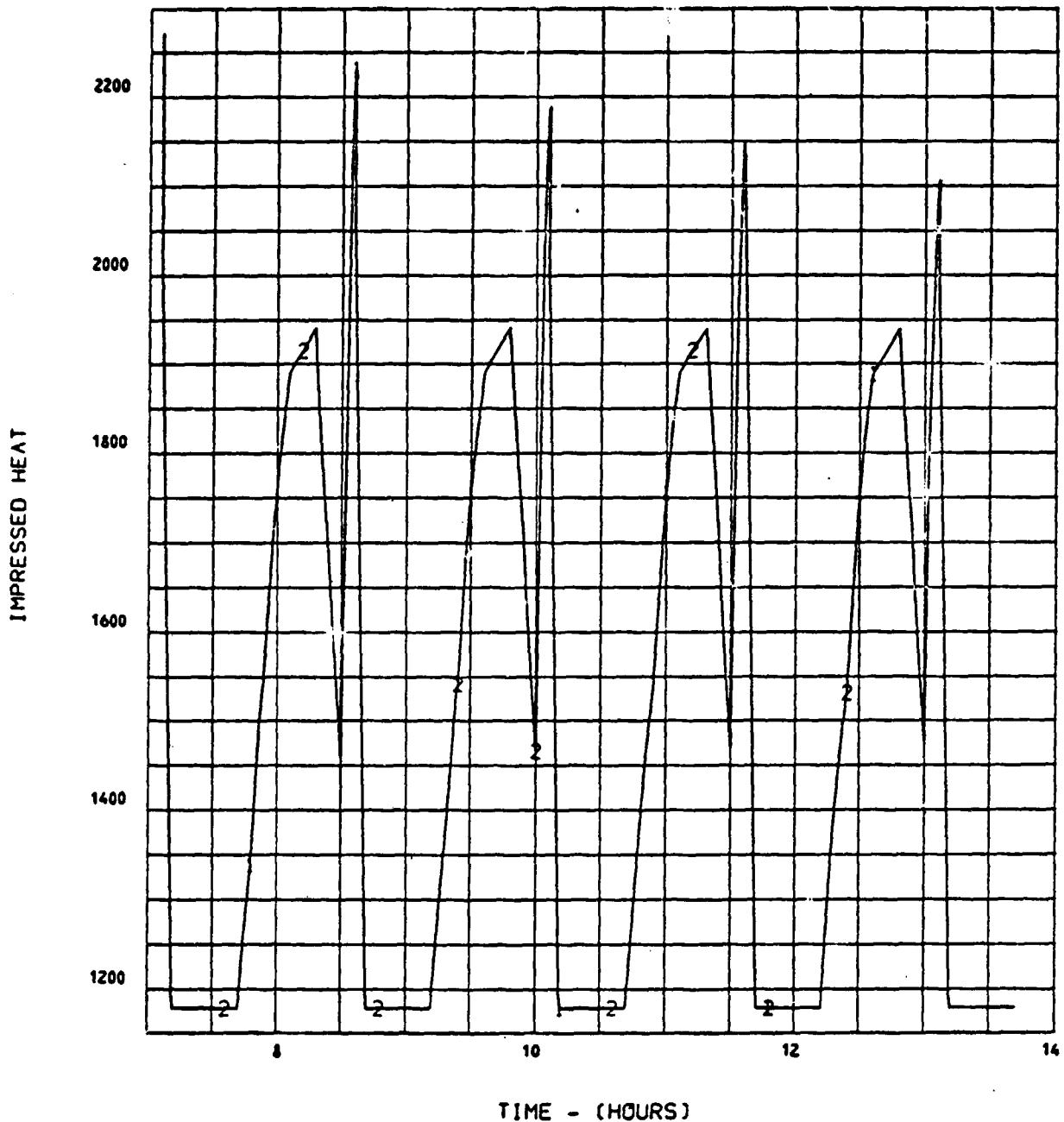


Figure 84

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT

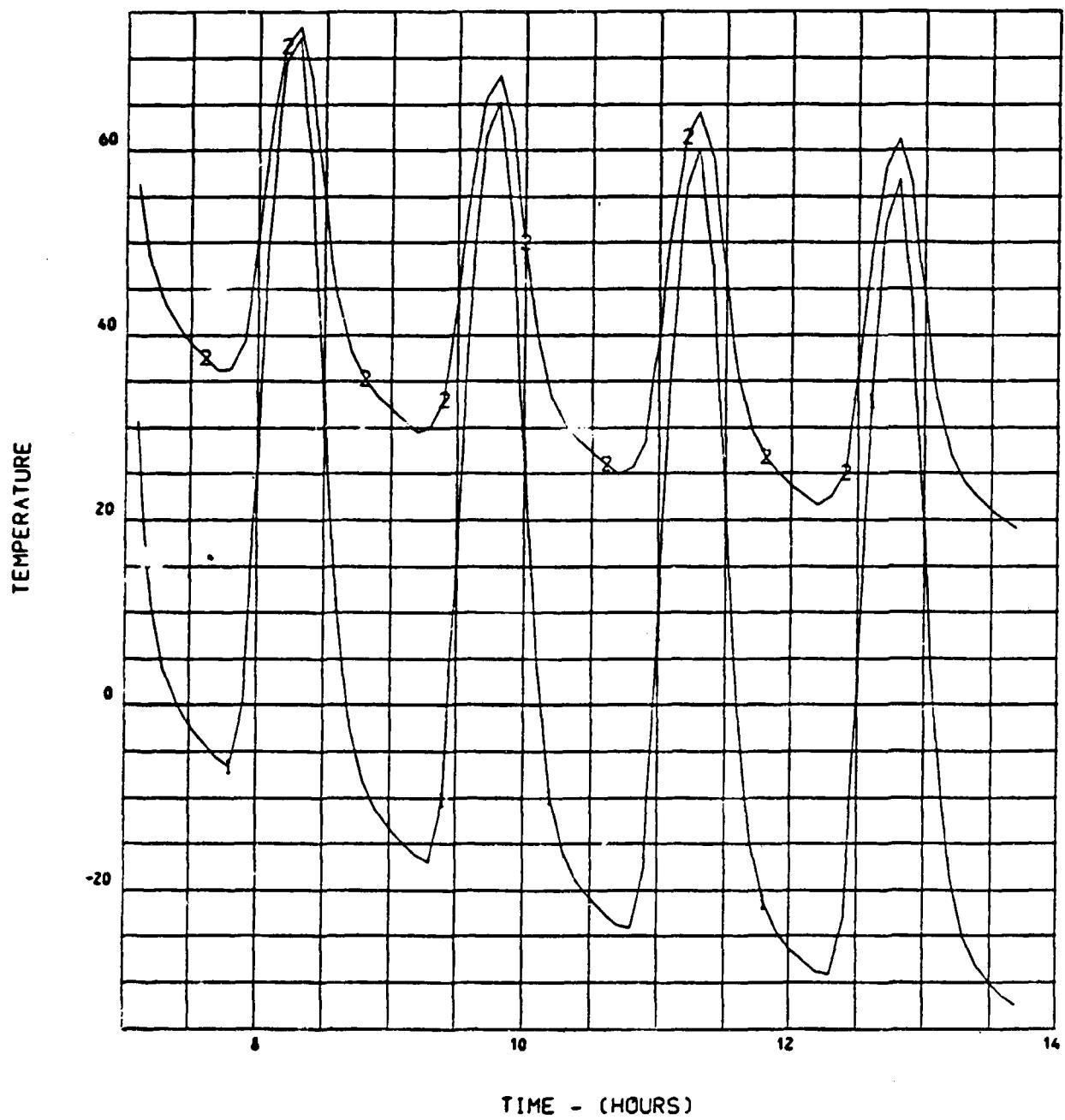


Figure 85

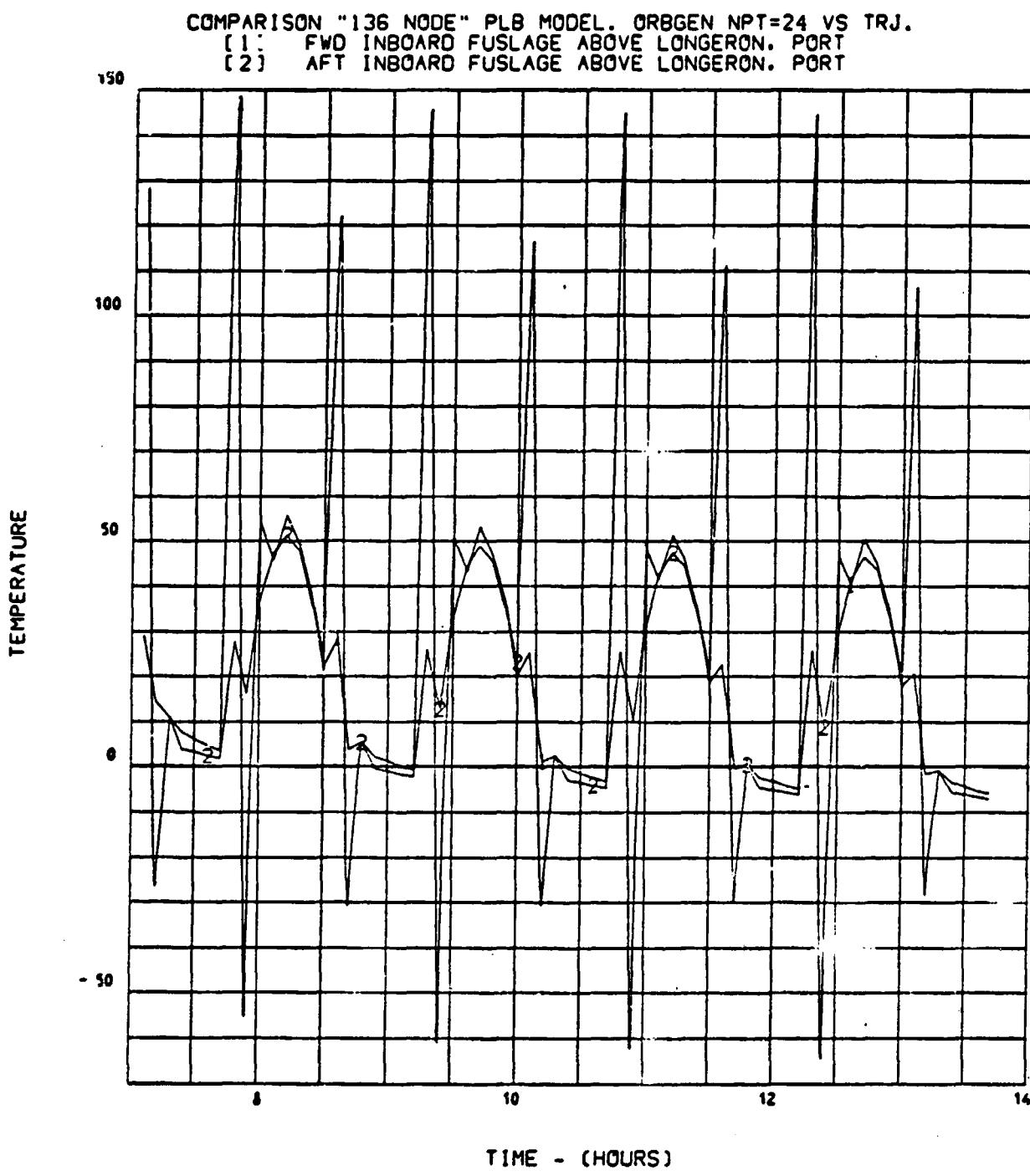


Figure 86

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
(1) FWD BOTTOM FUSELAGE. PORT
(2) AFT BOTTOM FUSELAGE. PORT

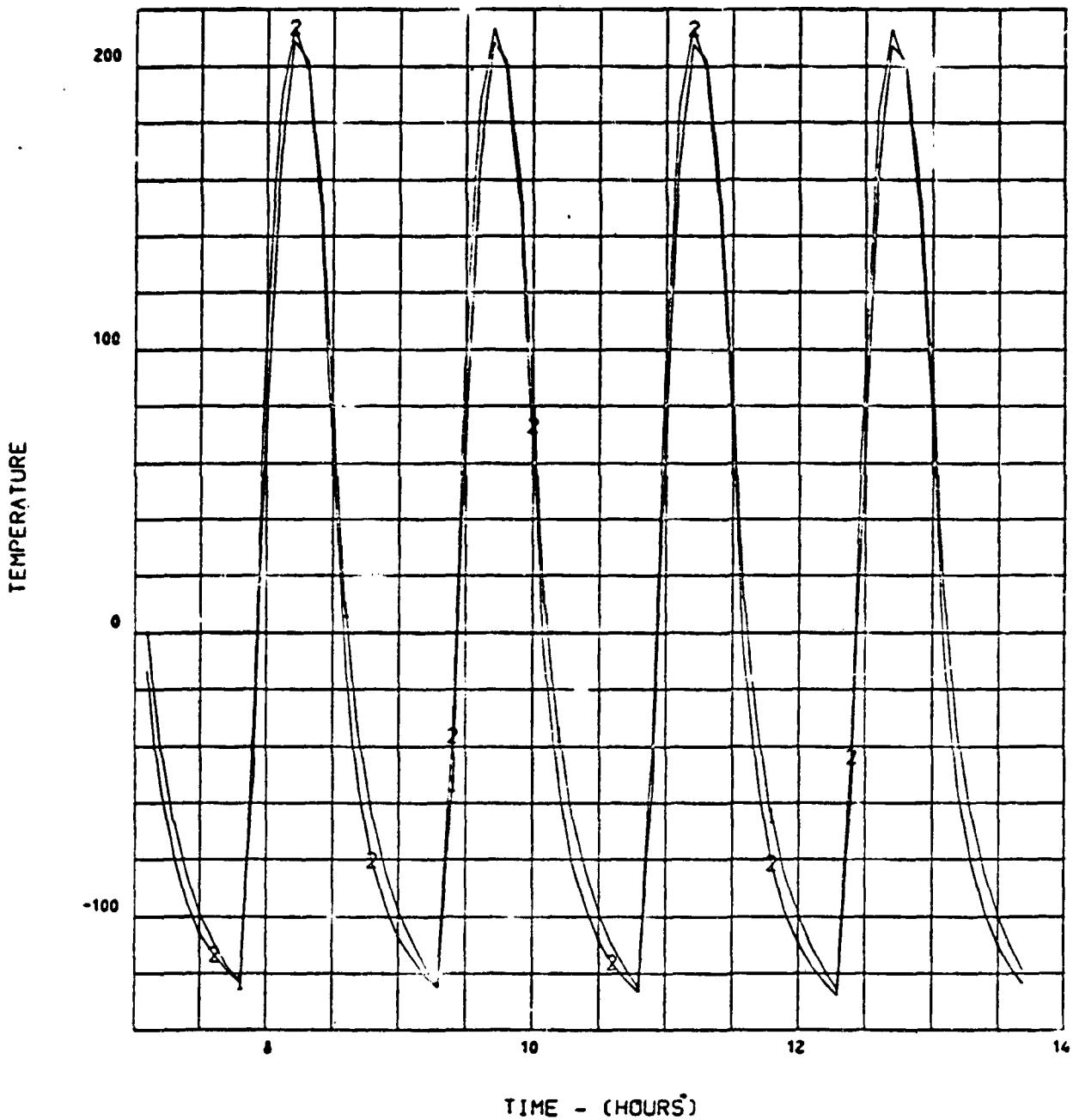


Figure 87

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] FWD PLB DOORS. PORT
- [2] FWD PLB DOORS. PORT
- [3] AFT PLB DOORS. PORT
- [4] AFT PLB DOORS. PORT

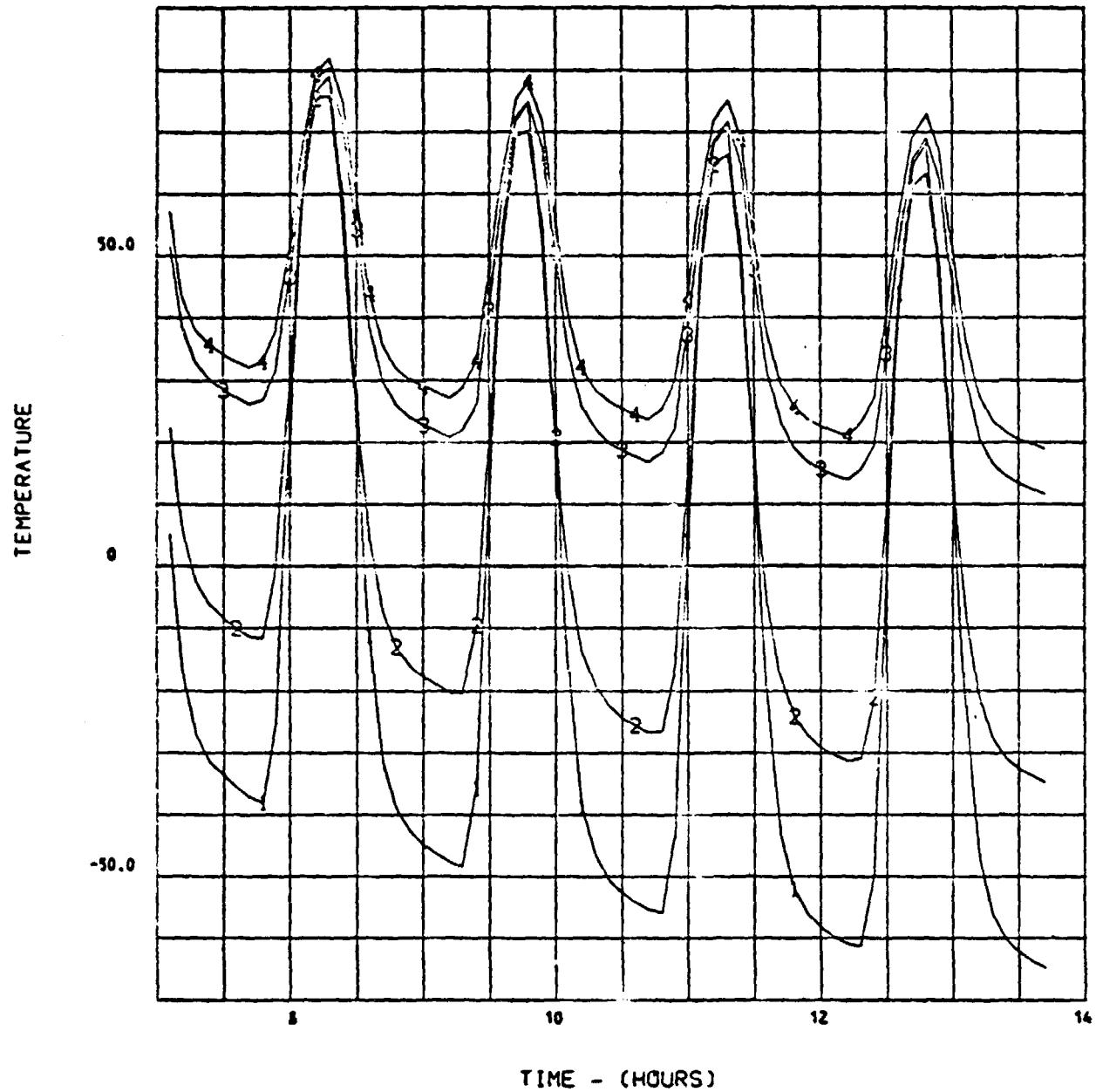


Figure 88

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] FWD RADIATOR. PORT
- [2] FWD RADIATOR. PORT
- [3] AFT RADIATOR. PORT
- [4] AFT RADIATOR. PORT

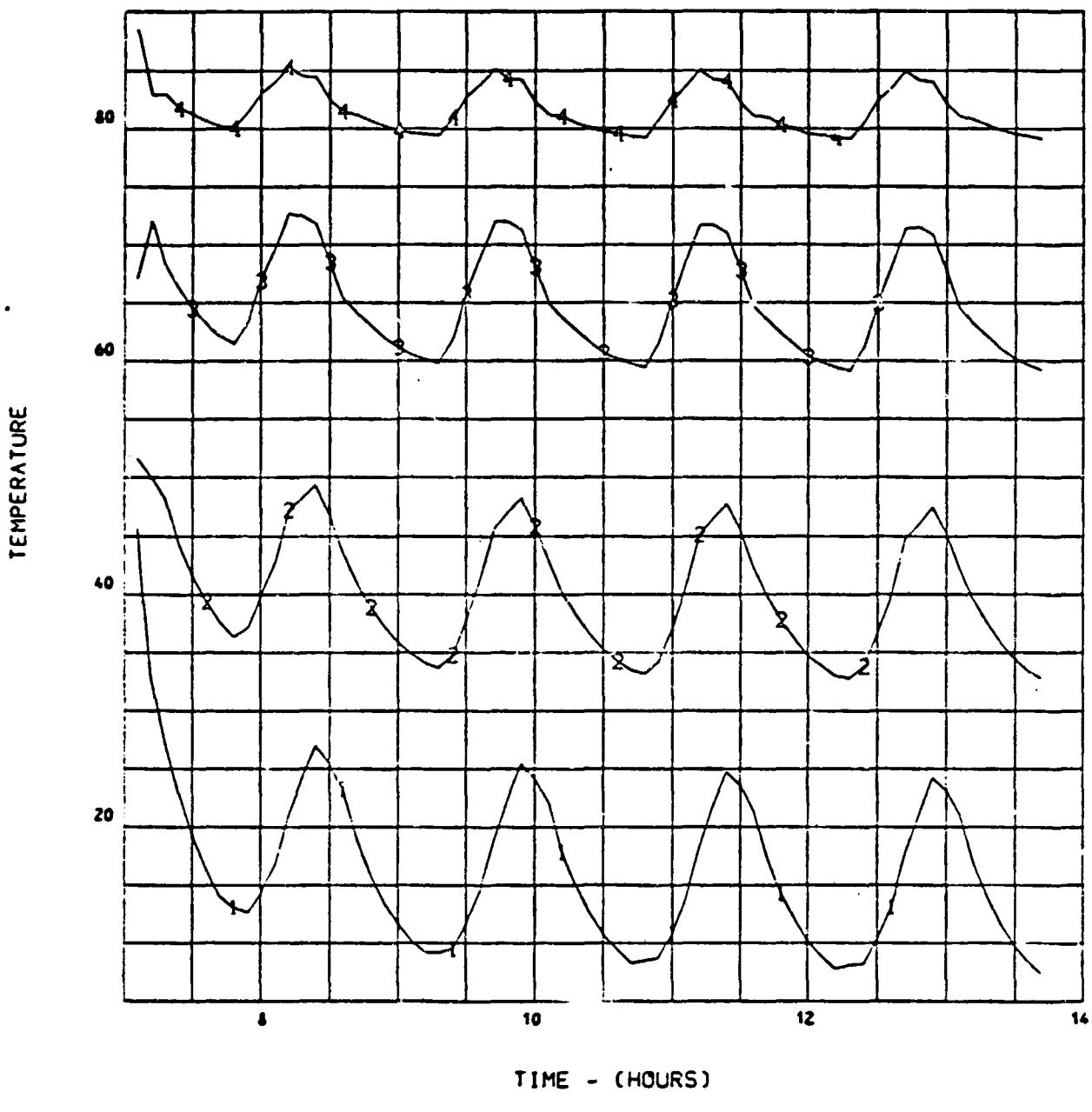


Figure 89

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD BULKHD BOTTOM
[2] FWD BULKHD TOP

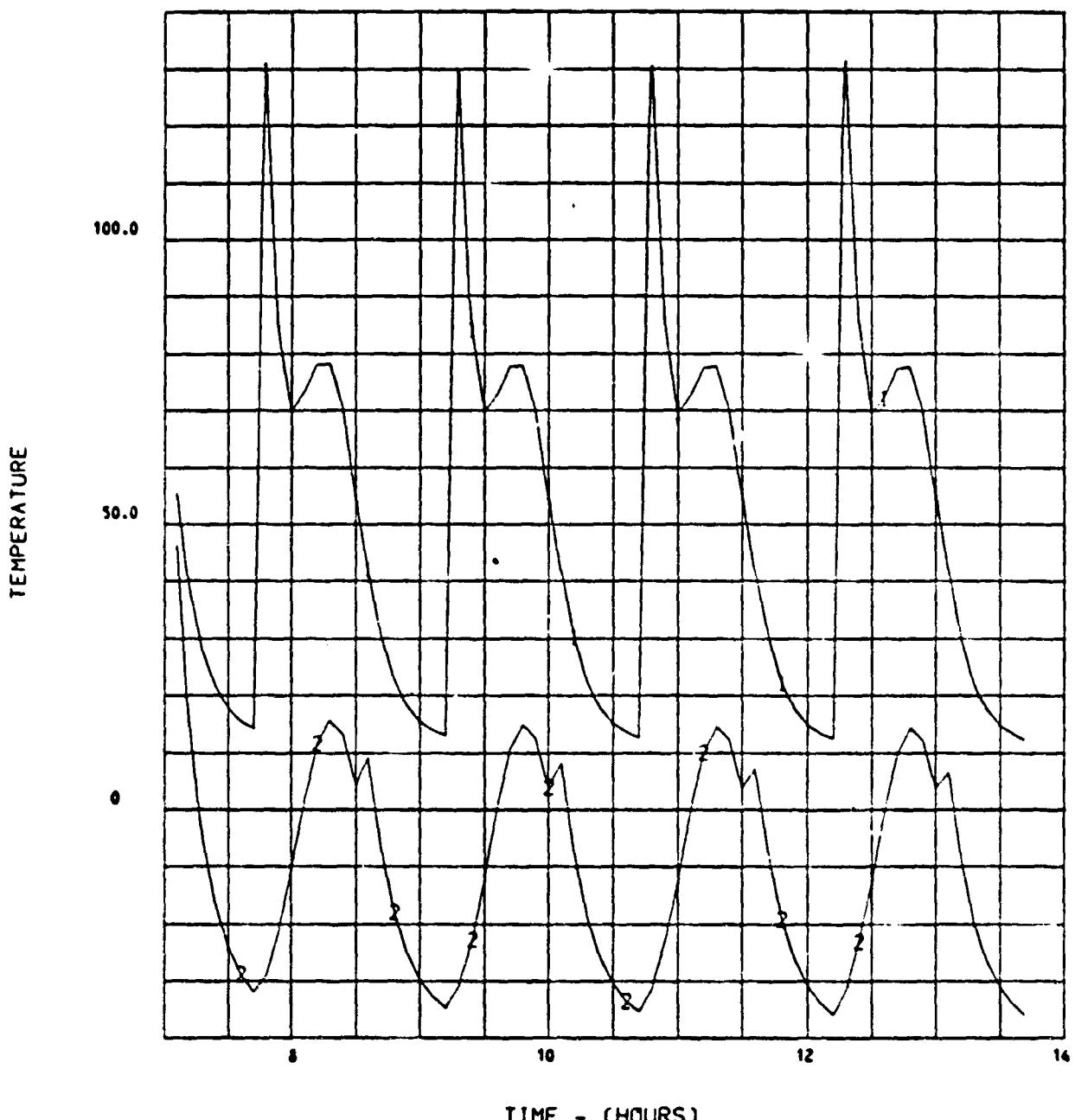


Figure 90

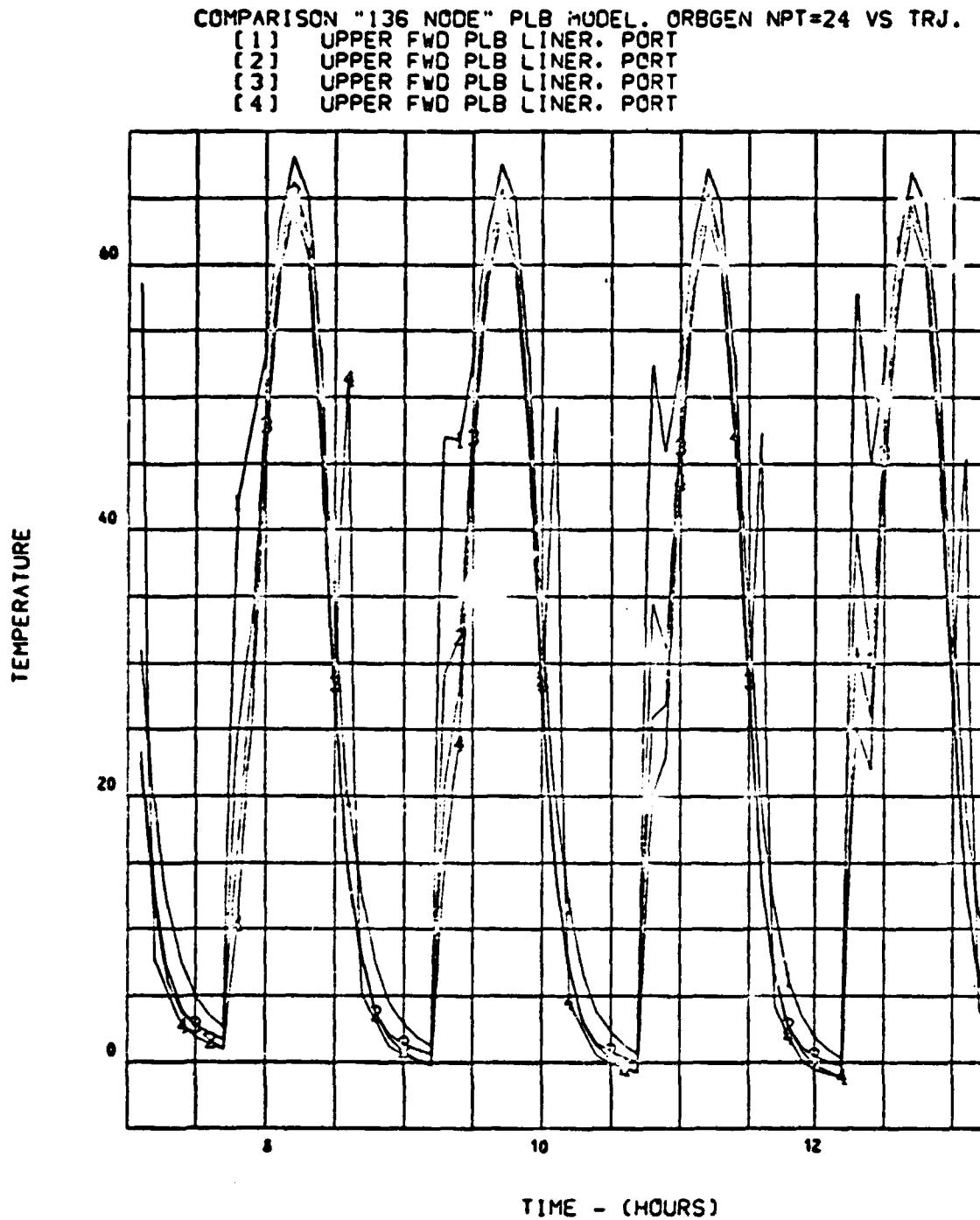
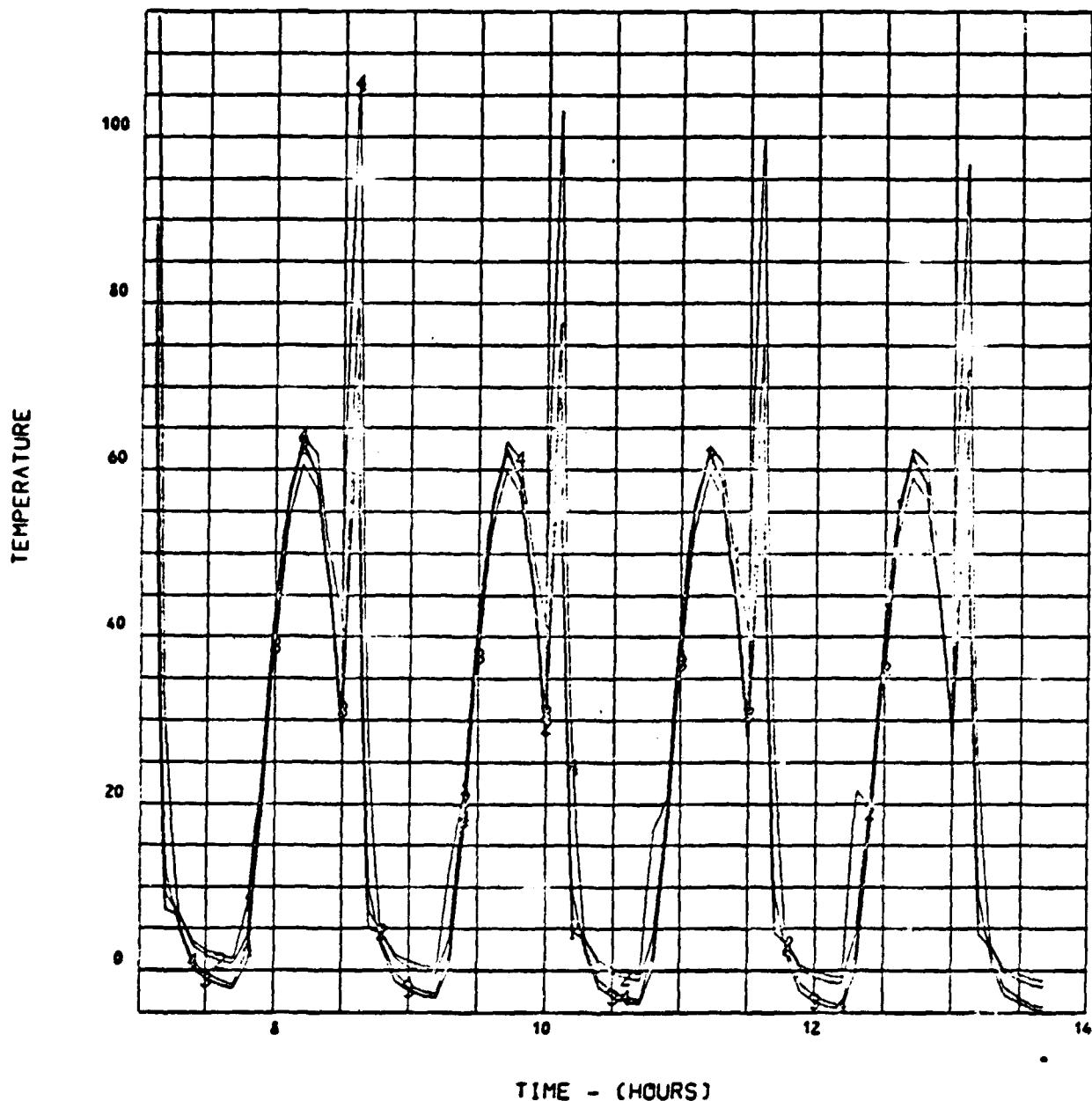


Figure 91

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] UPPER AFT PLB LINER. PORT
- [2] UPPER AFT PLB LINER. PORT
- [3] UPPER AFT PLB LINER. PORT
- [4] UPPER AFT PLB LINER. PORT



TIME - (HOURS)

Figure 92

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] LOWER FWD PLB LINER. PORT
- [2] LOWER FWD PLB LINER. PORT
- [3] LOWER FWD PLB LINER. PORT
- [4] LOWER FWD PLB LINER. PORT

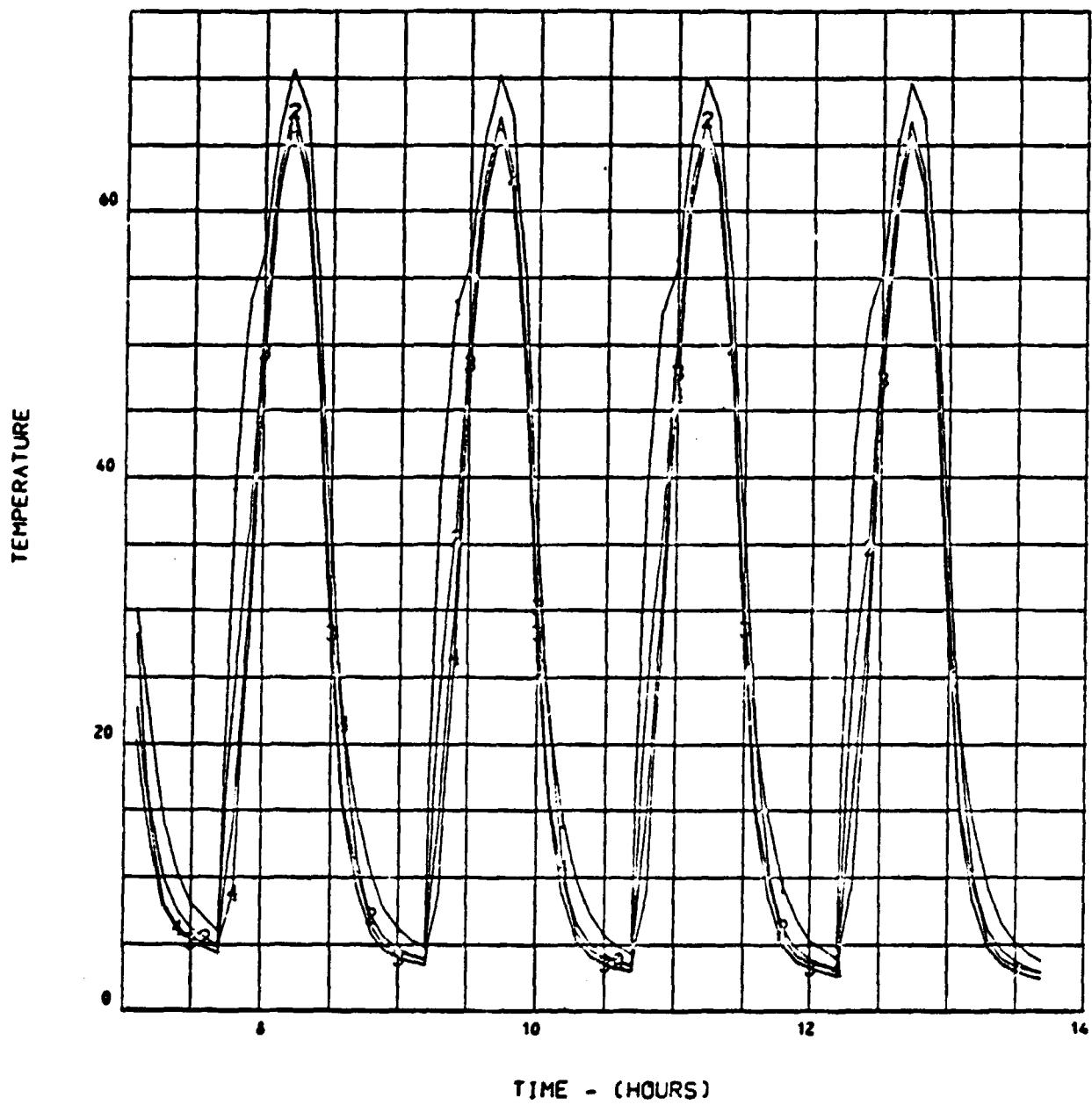


Figure 93

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] LOWER AFT PLB LINER. PORT
- [2] LOWER AFT PLB LINER. PORT
- [3] LOWER AFT PLB LINER. PORT
- [4] LOWER AFT PLB LINER. PORT

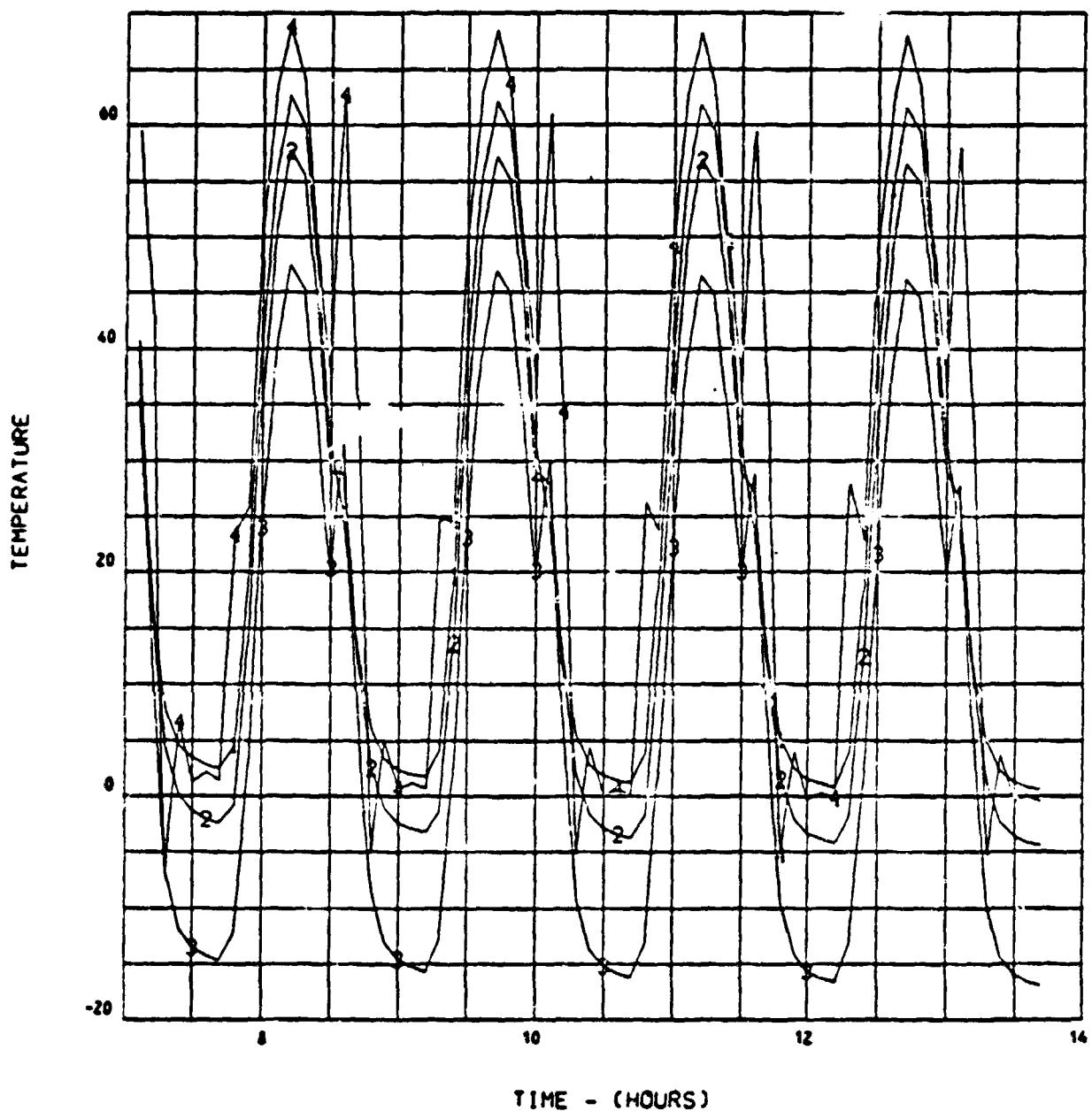
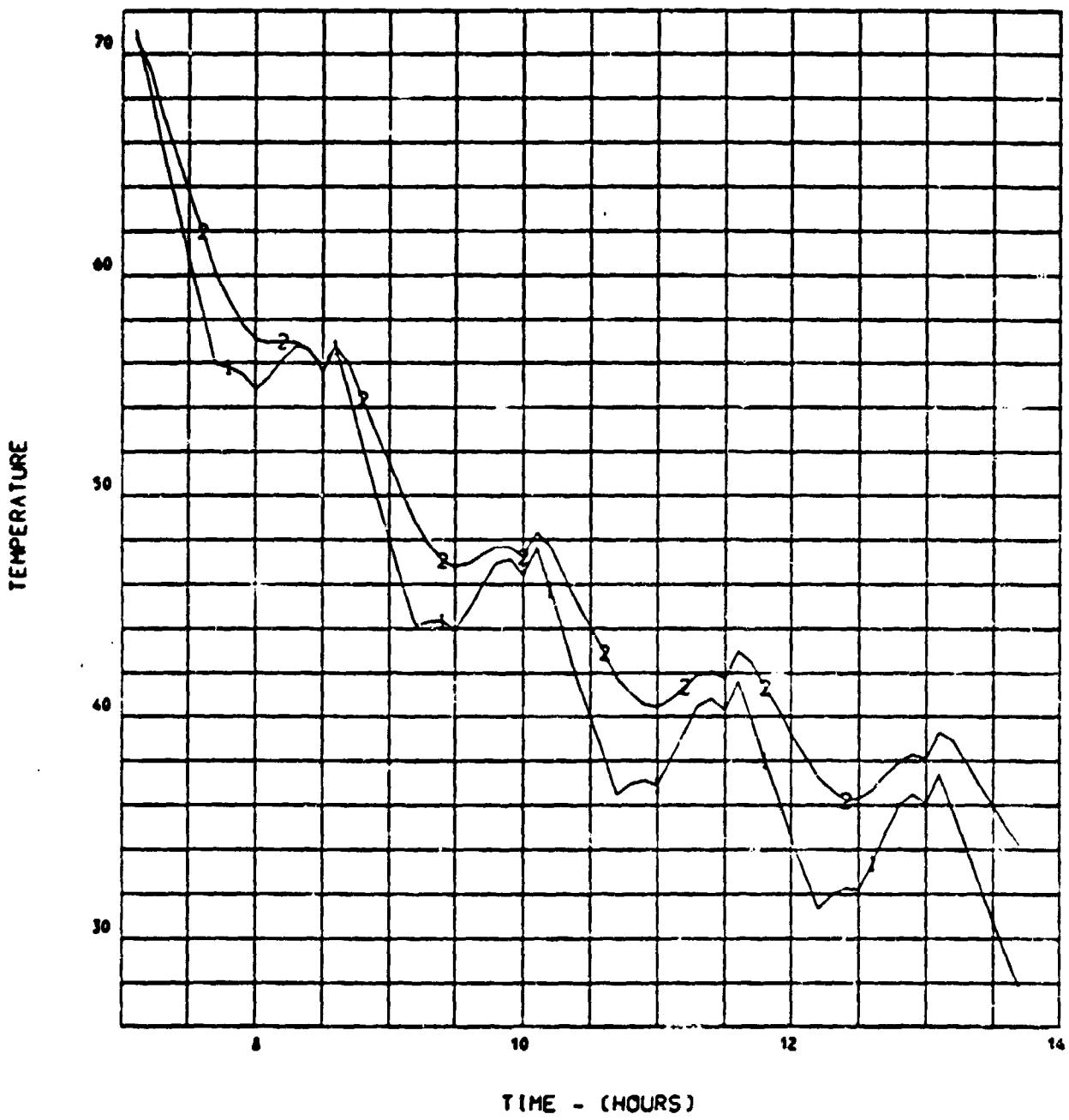


Figure 94

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD LONGERON. PORT
[2] AFT LONGERON. PORT



TIME - (HOURS)

Figure 95

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] AFT BULKHD BOTTOM
[2] AFT BULKHD TOP

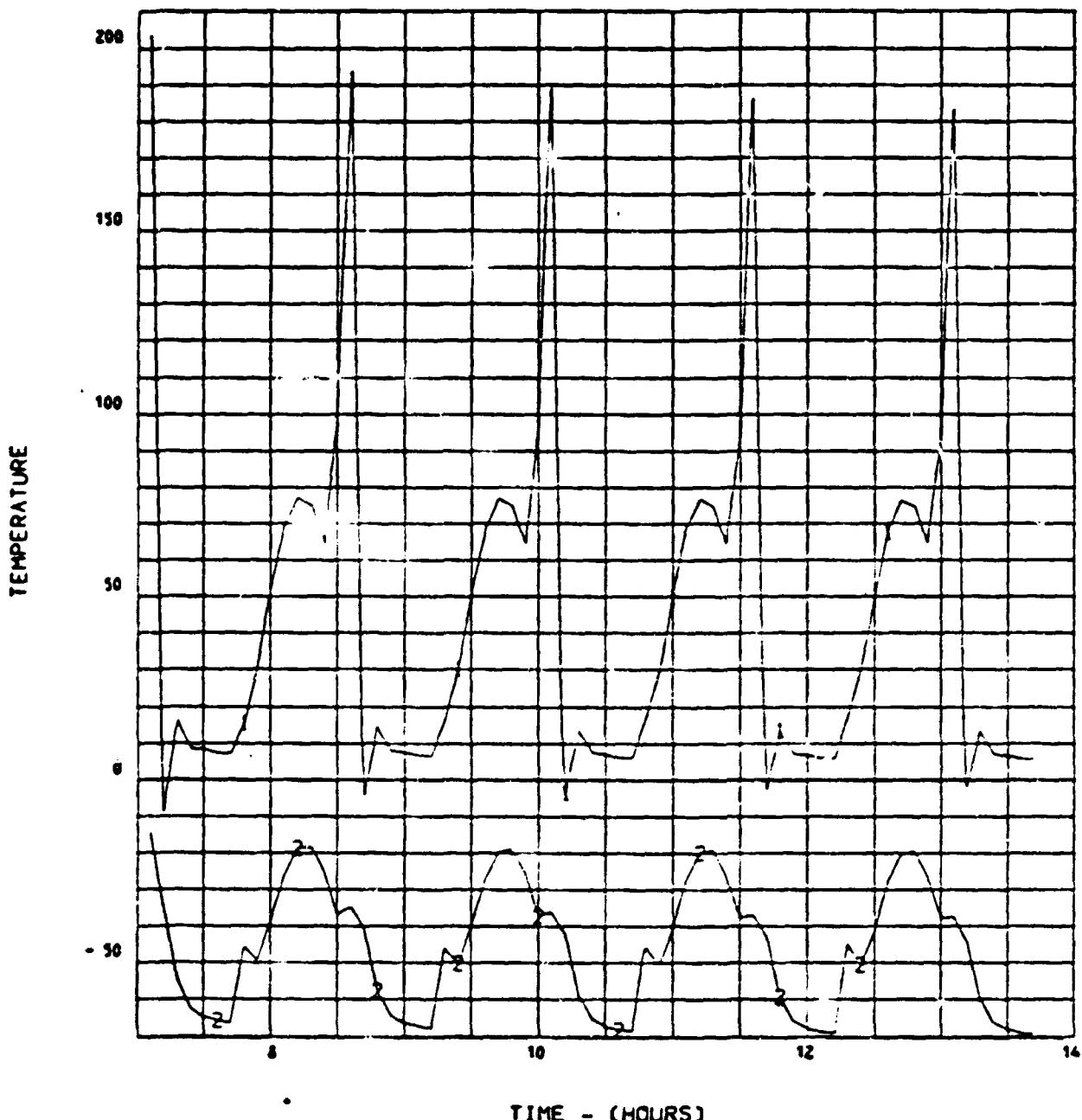


Figure 96

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD BULKHD BOTTOM BELOW PLB LINER

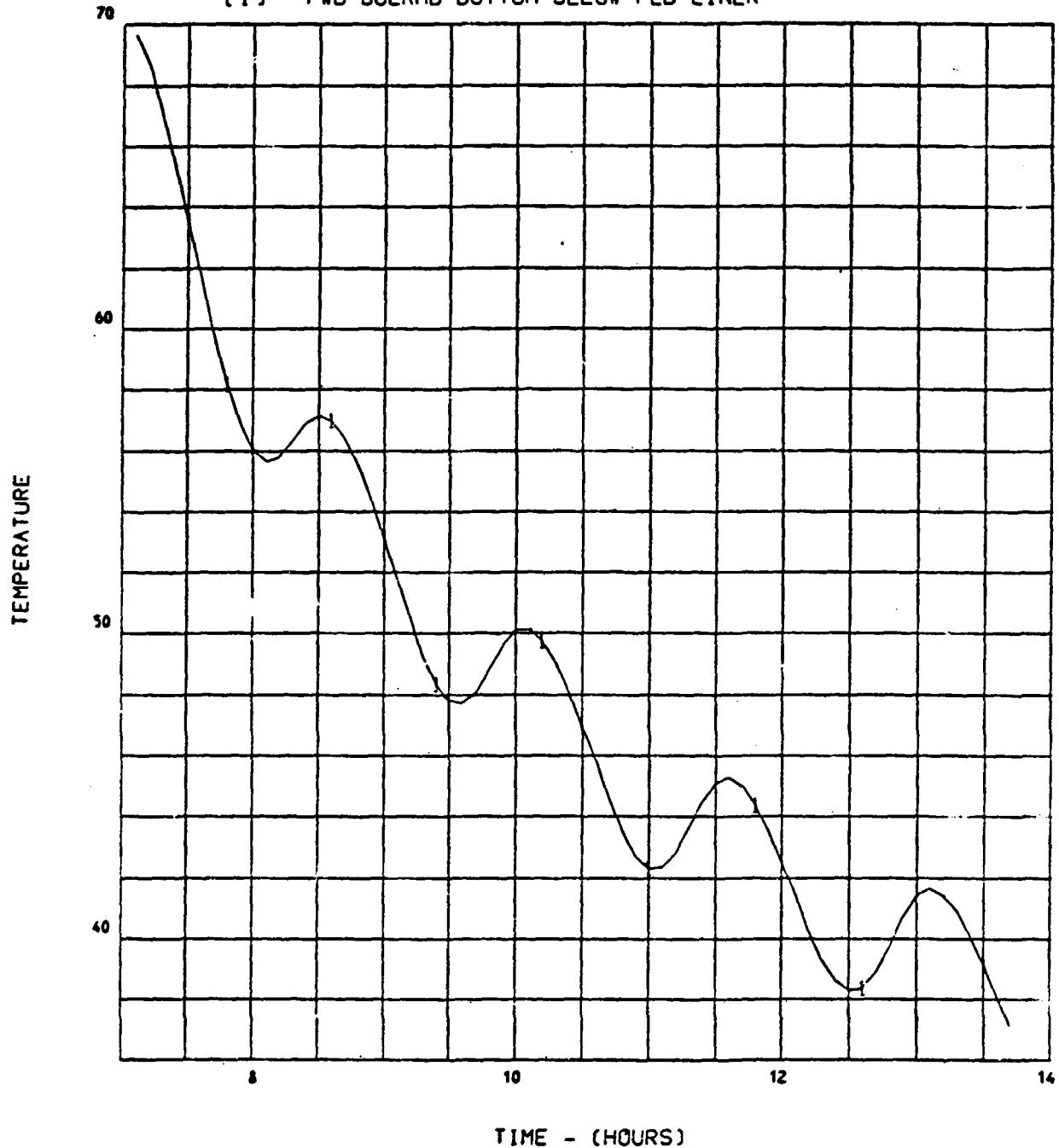


Figure 97

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD SIDE FUSELAGE STRUCTURE. PORT
[2] AFT SIDE FUSELAGE STRUCTURE. PORT

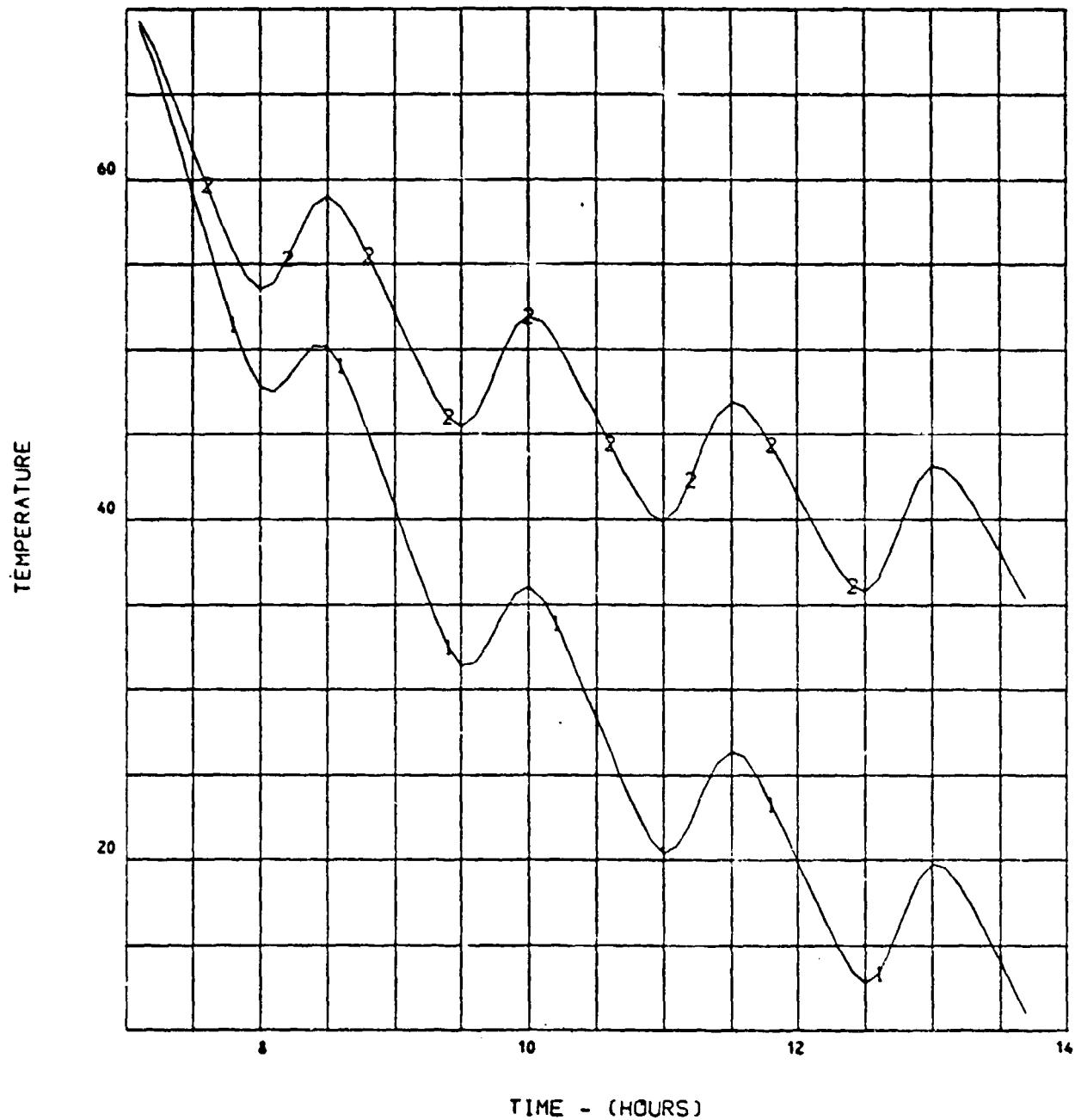


Figure 98

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD BOTTOM FUSELAGE STRUCTURE. PORT
[2] AFT BOTTOM FUSELAGE STRUCTURE. PORT

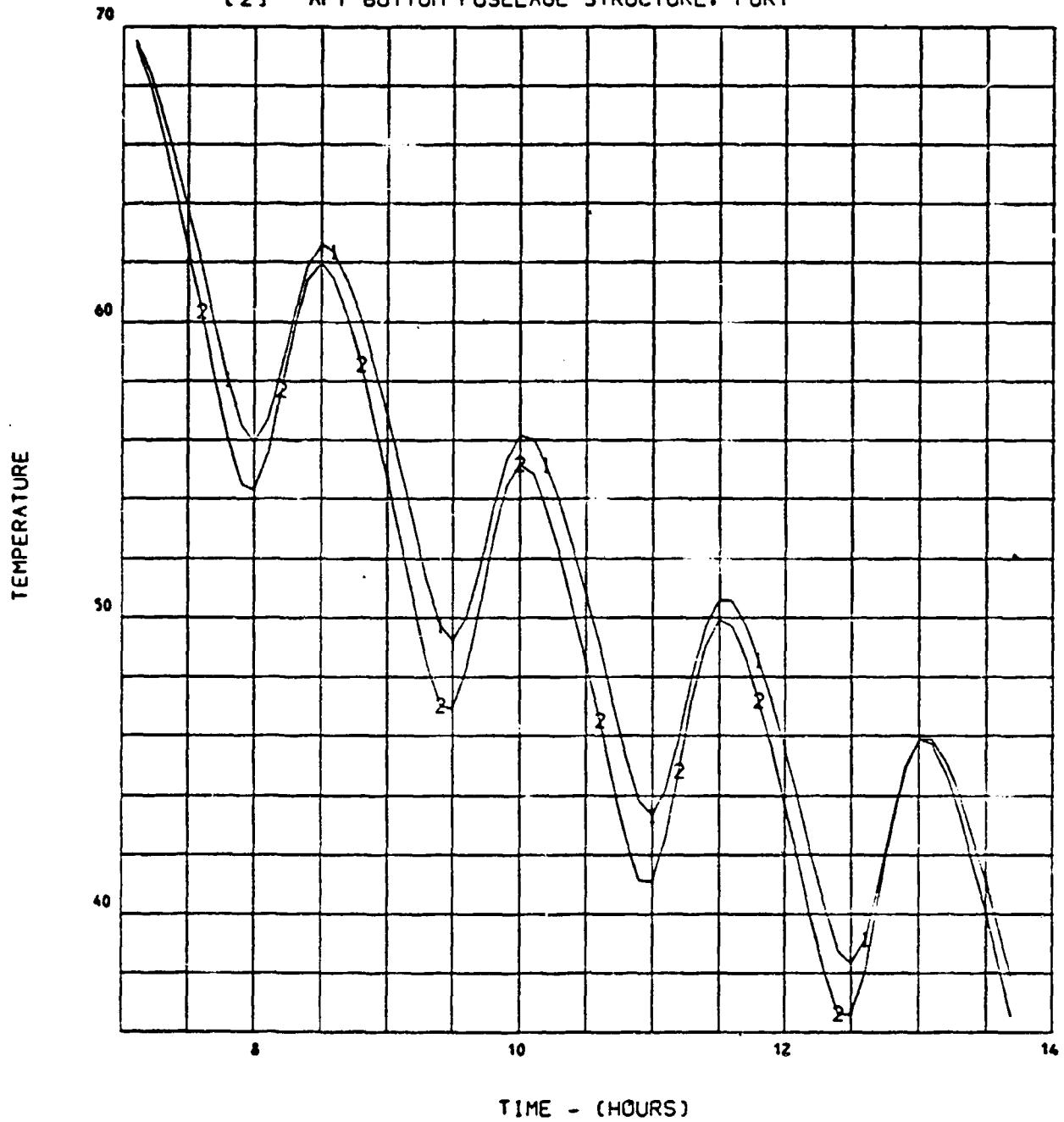


Figure 99

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] FWD PLB DOORS STRUCTURE. PORT
- [2] FWD PLB DOORS STRUCTURE. PORT
- [3] AFT PLB DOORS STRUCTURE. PORT
- [4] AFT PLB DOORS STRUCTURE. PORT

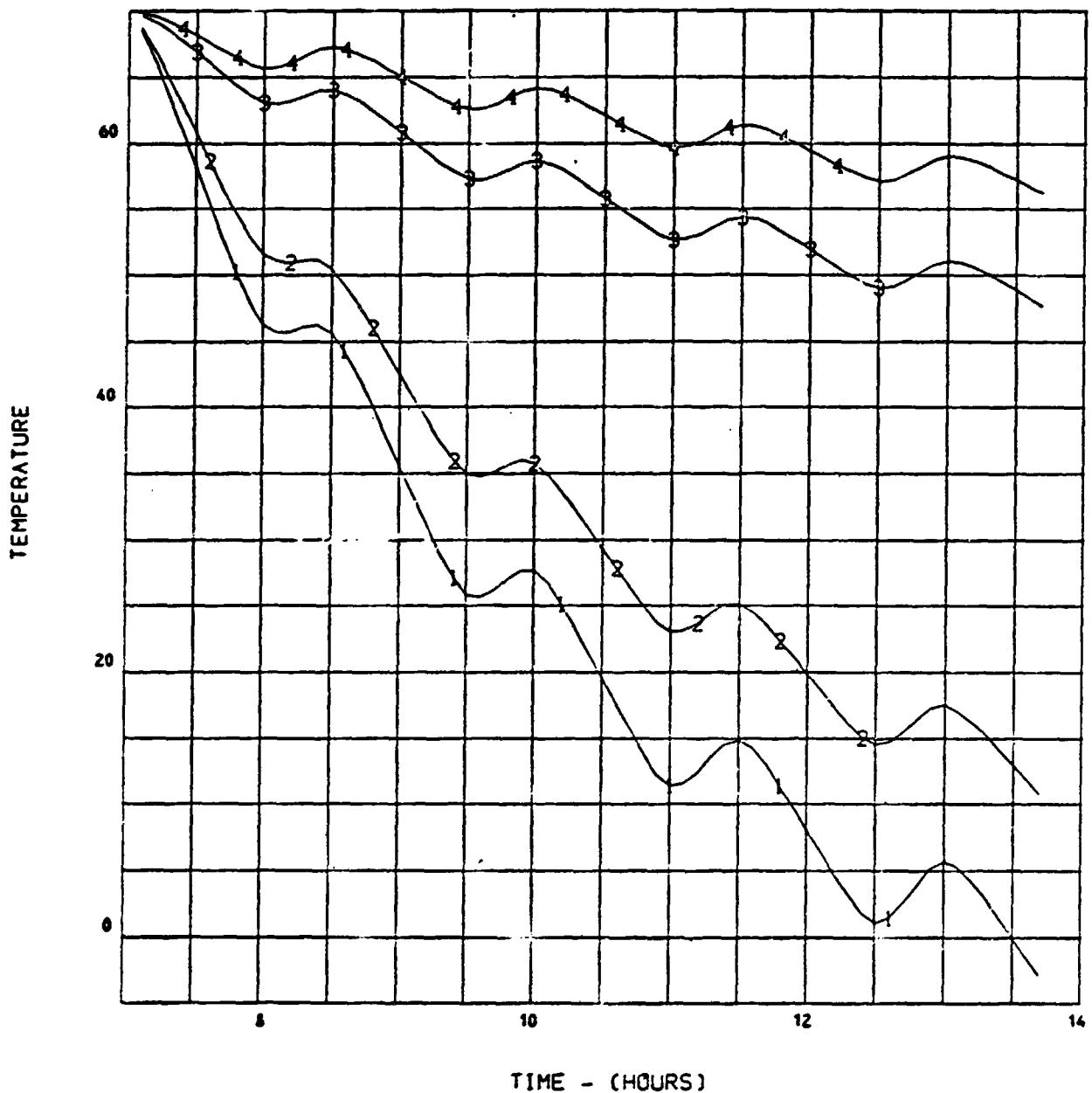


Figure 100

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] AFT BULKHD BOTTOM BELOW PLB LINER

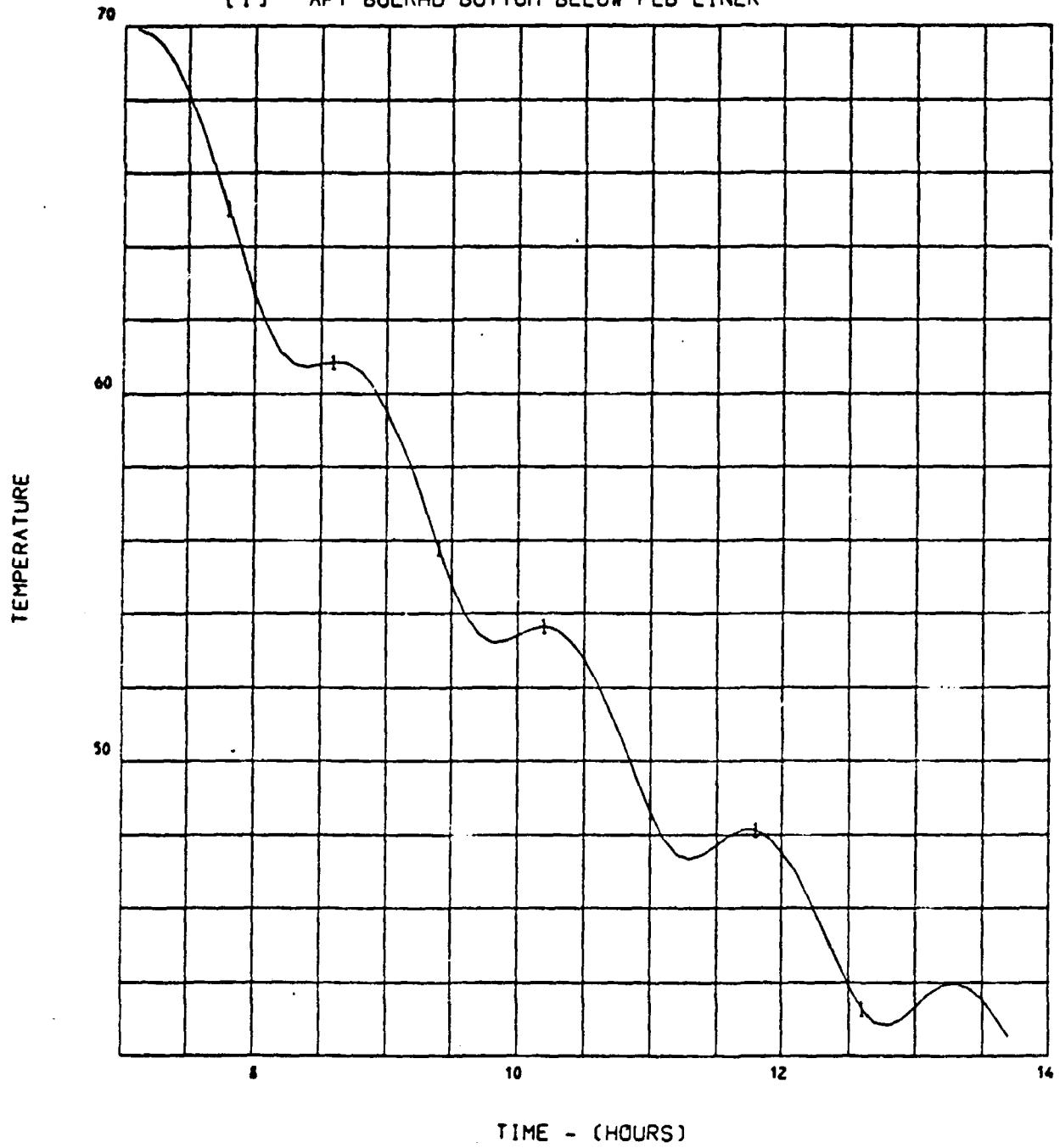


Figure 101

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD SIDE FUSELAGE. PORT
[2] AFT SIDE FUSELAGE. PORT

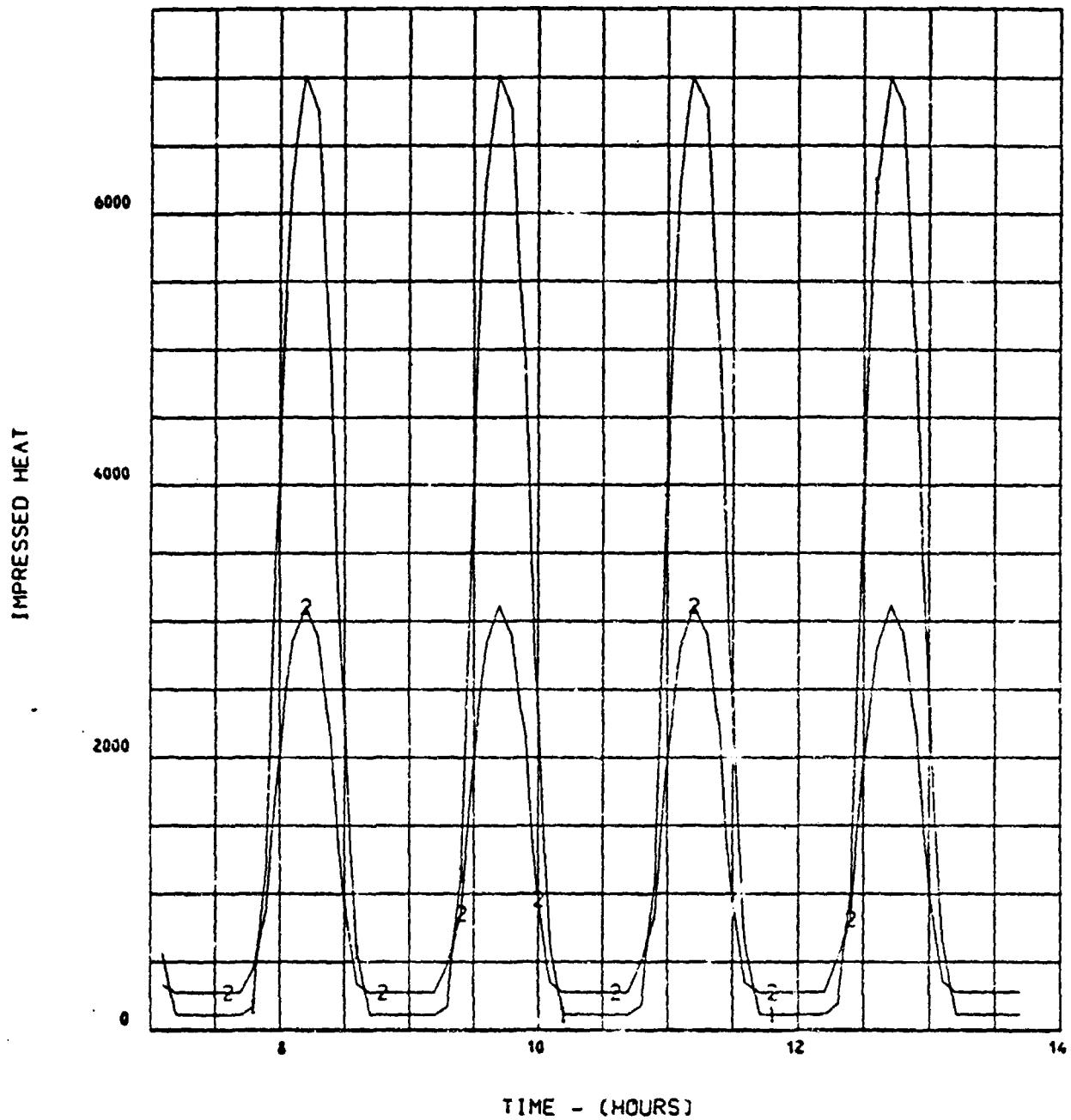
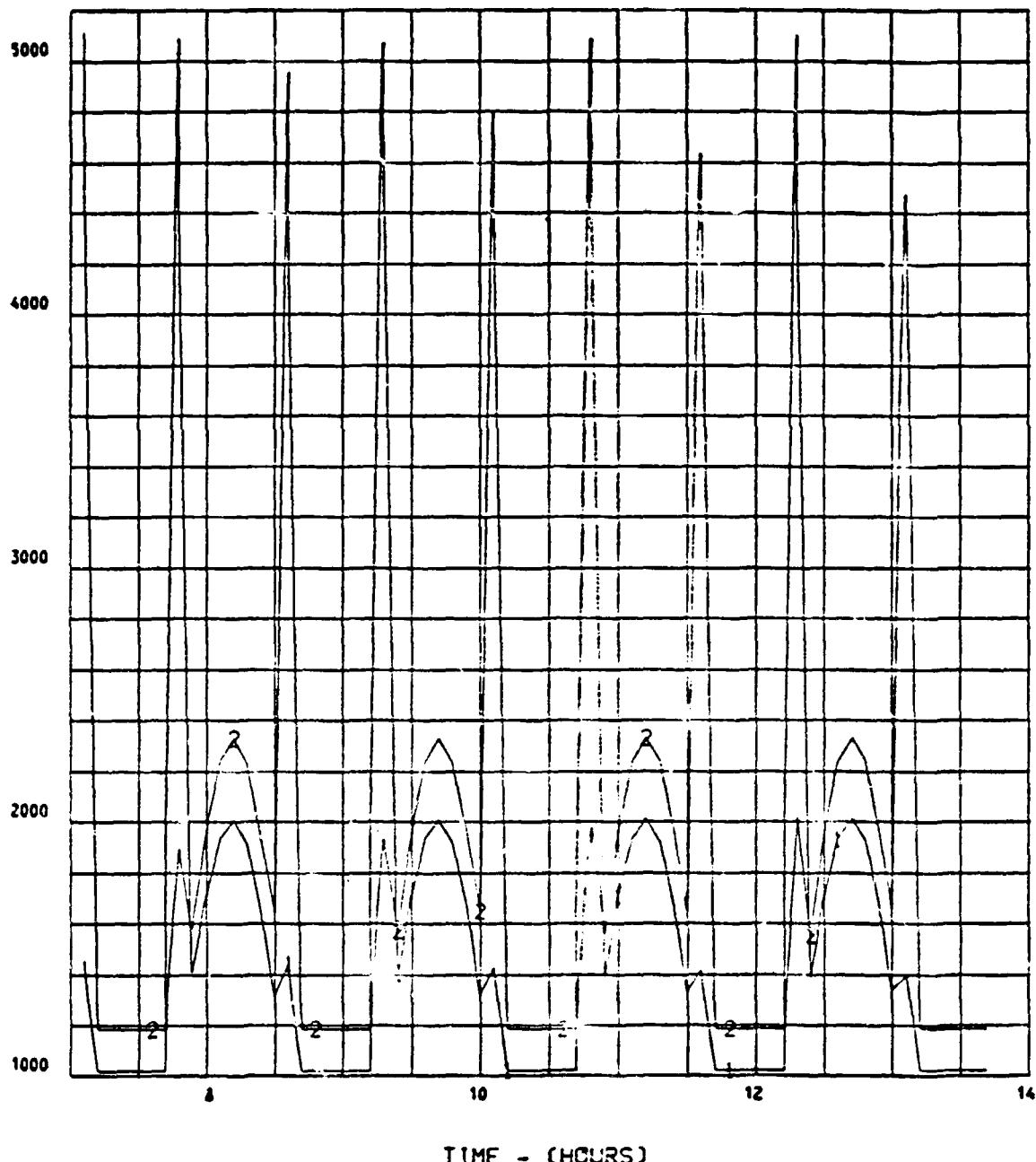


Figure 102

COMPARISON "136 NODE" PLB MODEL. CR8GEN NPT=24 VS TRJ.
[1] FWD INBOARD FUSLAGE ABOVE LONGERON. PORT
[2] AFT INBOARD FUSLAGE ABOVE LONGERON. PORT

IMPOSED HEAT



TIME - (HOURS)

Figure 103

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD BOTTOM FUSELAGE. PORT
[2] AFT BOTTOM FUSELAGE. PORT

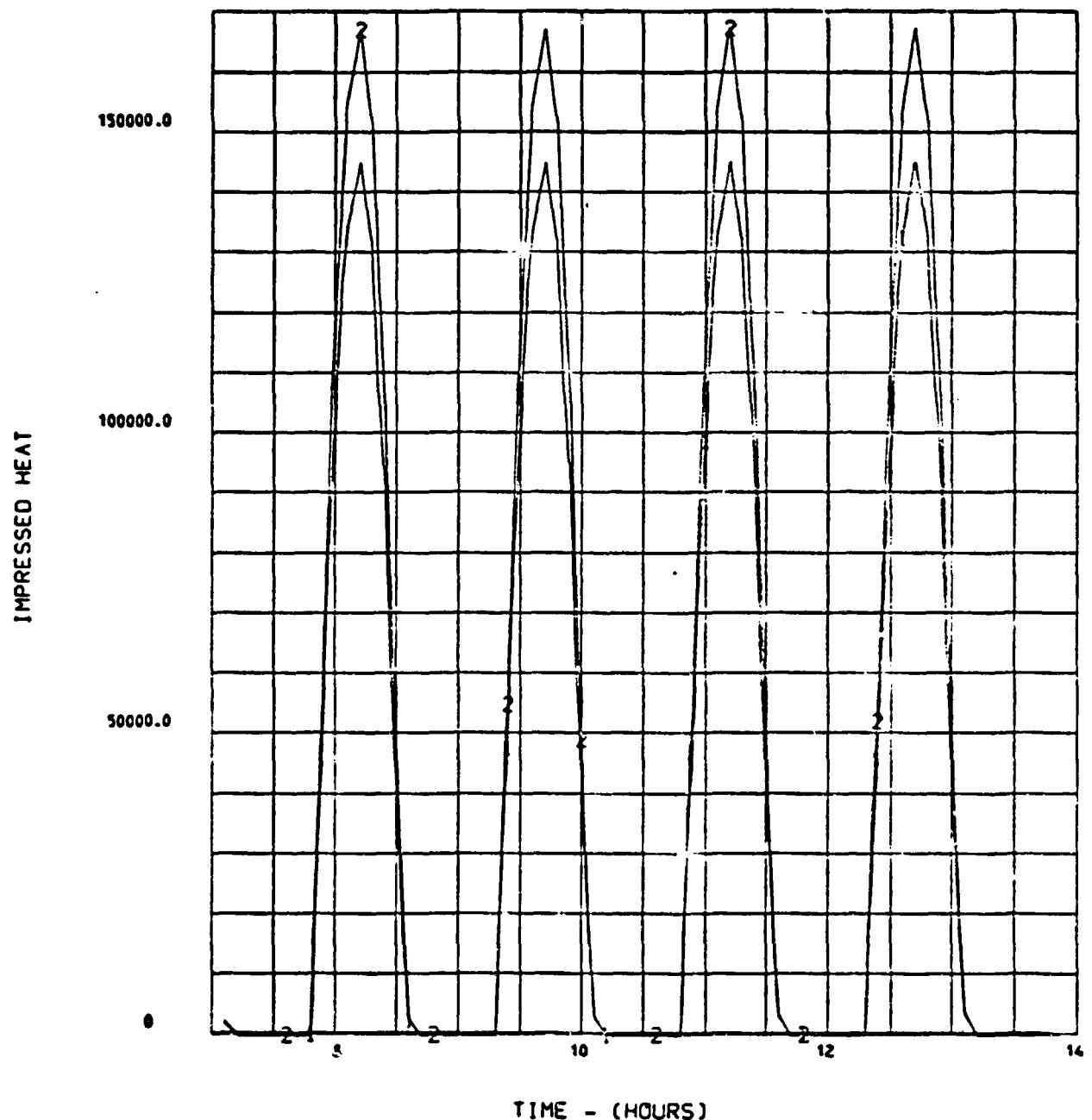


Figure 104

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] FWD PLB DOORS. PORT
- [2] FWD PLB DOORS. PORT
- [3] AFT PLB DOORS. PORT
- [4] AFT PLB DOORS. PORT

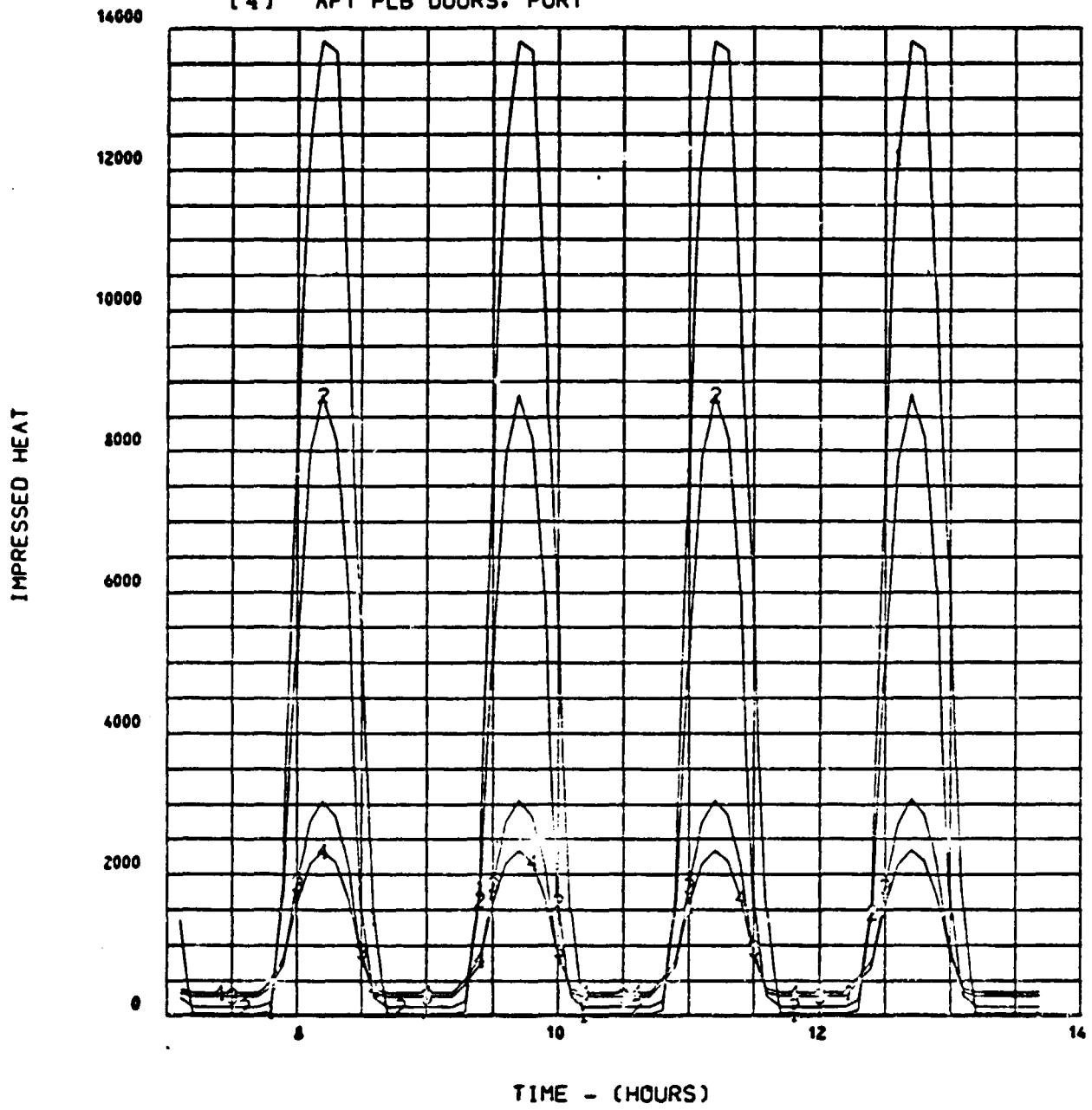


Figure 105

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] FWD RADIATOR. PORT
- [2] FWD RADIATOR. PORT
- [3] AFT RADIATOR. PORT
- [4] AFT RADIATOR. PORT

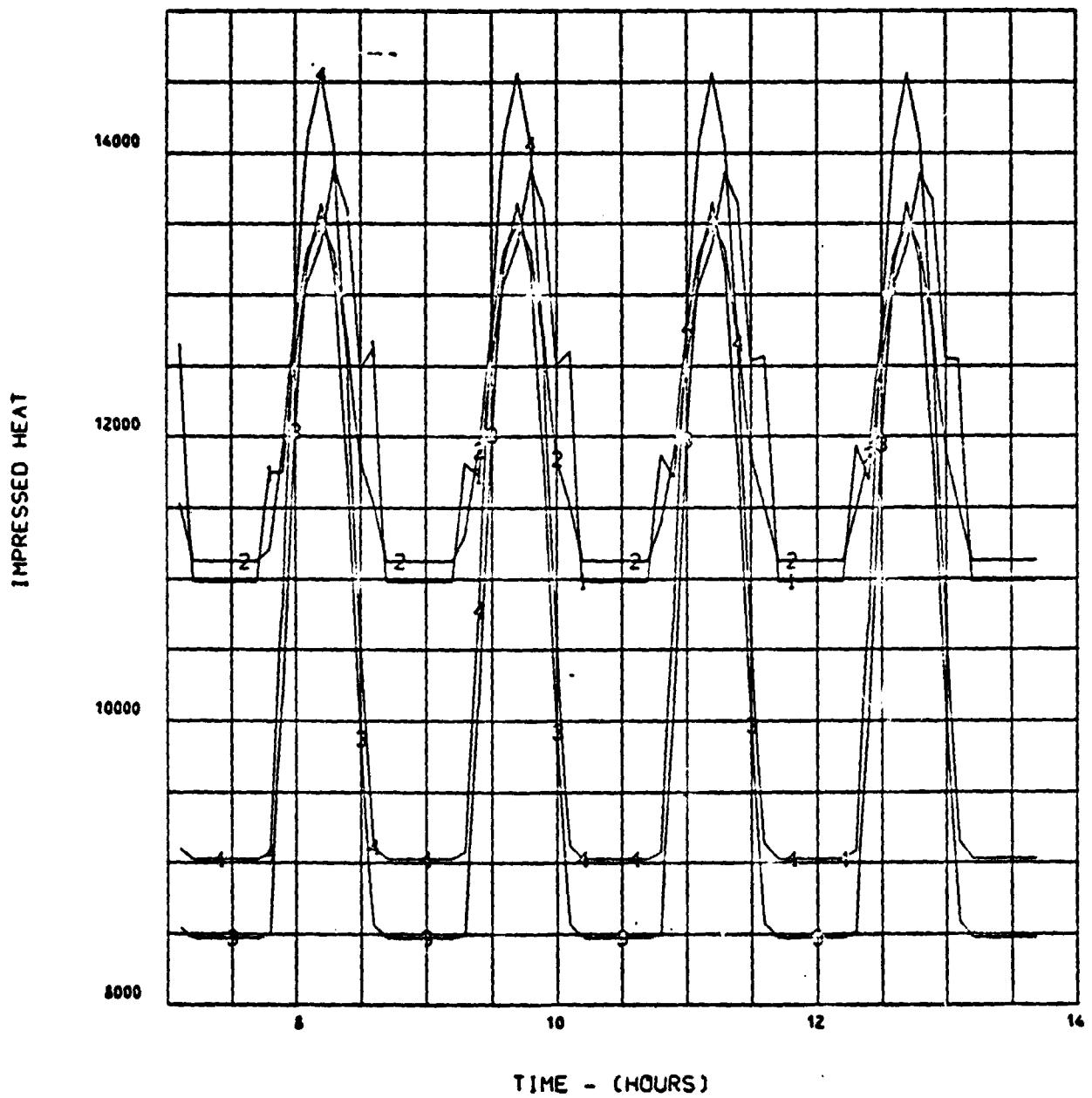


Figure 106

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD BULKHD TOP
[2] FWD BULKHD BOTTOM

IMPOSED HEAT

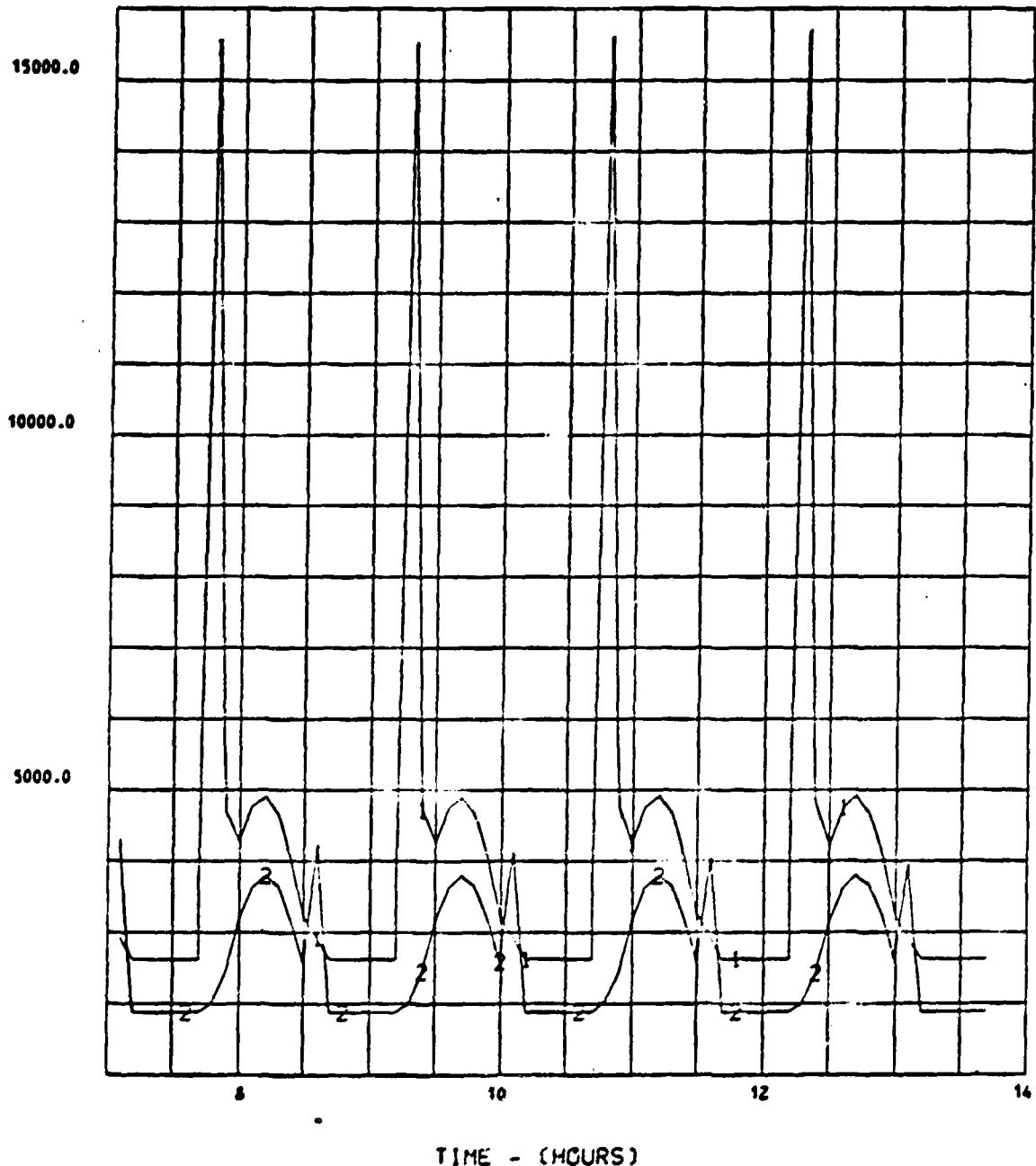


Figure 107

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] UPPER FWD PLB LINER. PORT
- [2] UPPER FWD PLB LINER. PORT
- [3] UPPER FWD PLB LINER. PORT
- [4] UPPER FWD PLB LINER. PORT

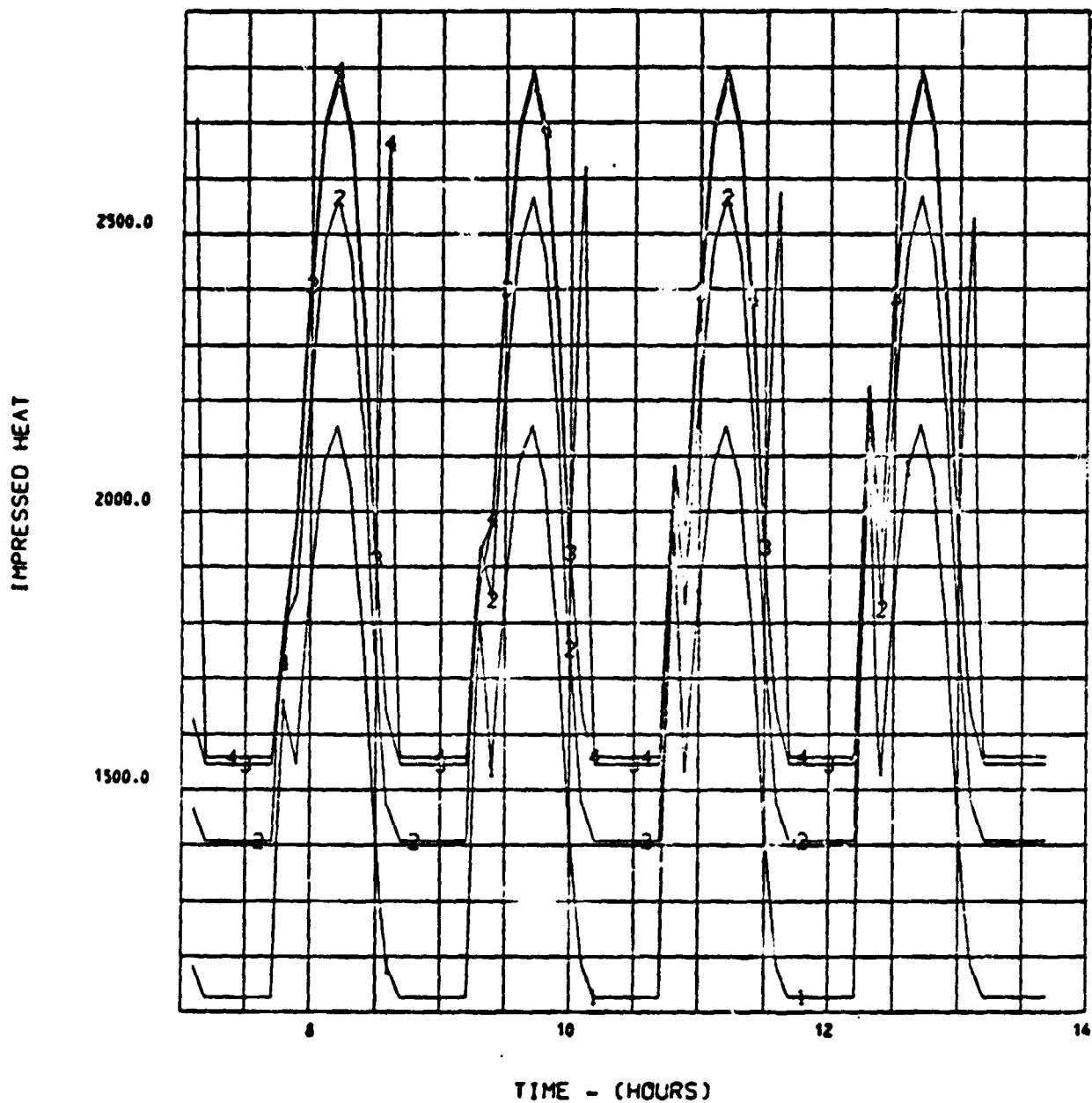


Figure 108

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] UPPER AFT PLB LINER. PORT
- [2] UPPER AFT PLB LINER. PCRT
- [3] UPPER AFT PLB LINER. PORT
- [4] UPPER AFT PLB LINER. PORT

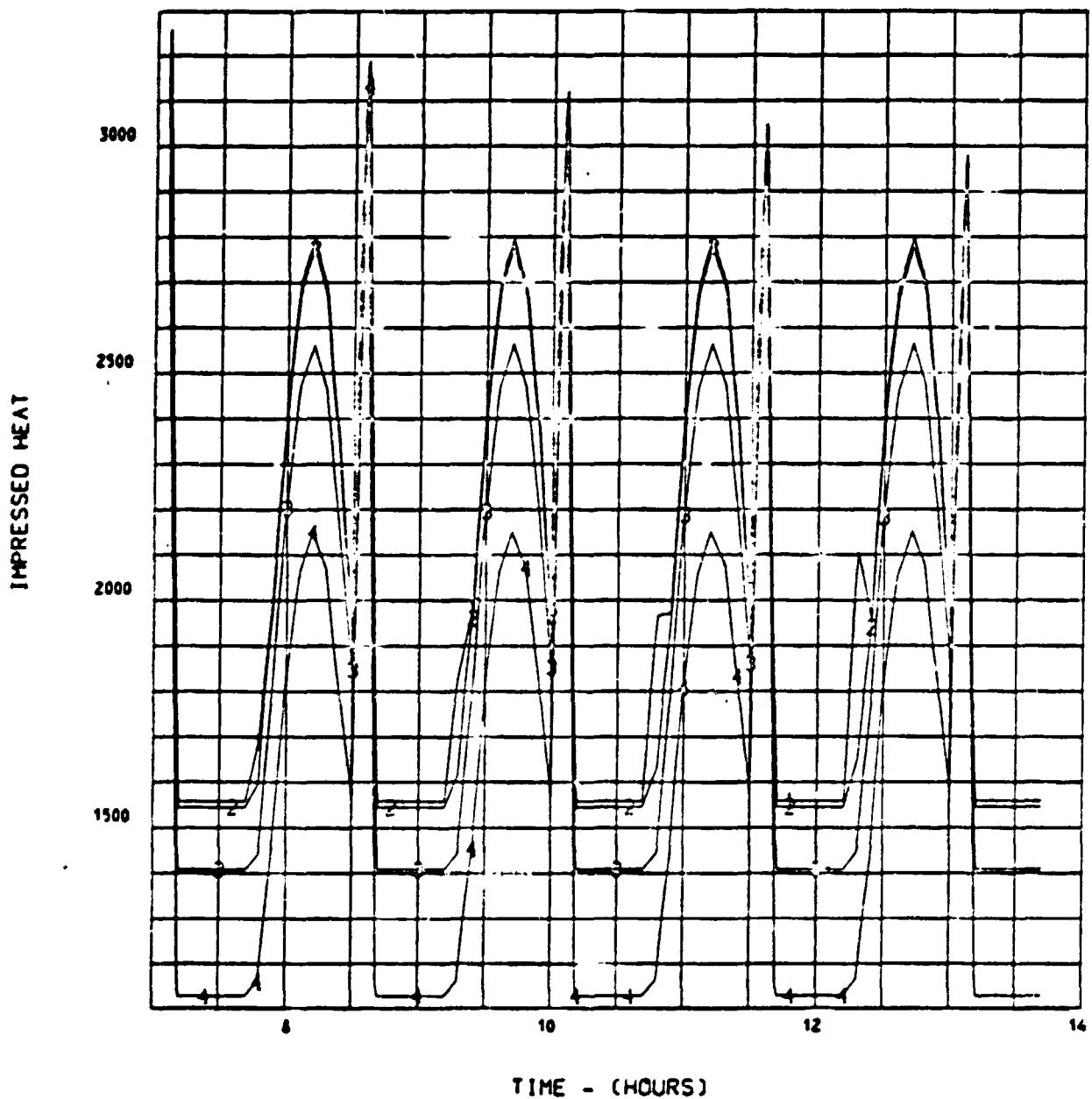


Figure 109

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] LOWER FWD PLB LINER. PORT
- [2] LOWER FWD PLB LINER. PORT
- [3] LOWER FWD PLB LINER. PORT
- [4] LOWER FWD PLB LINER. PORT

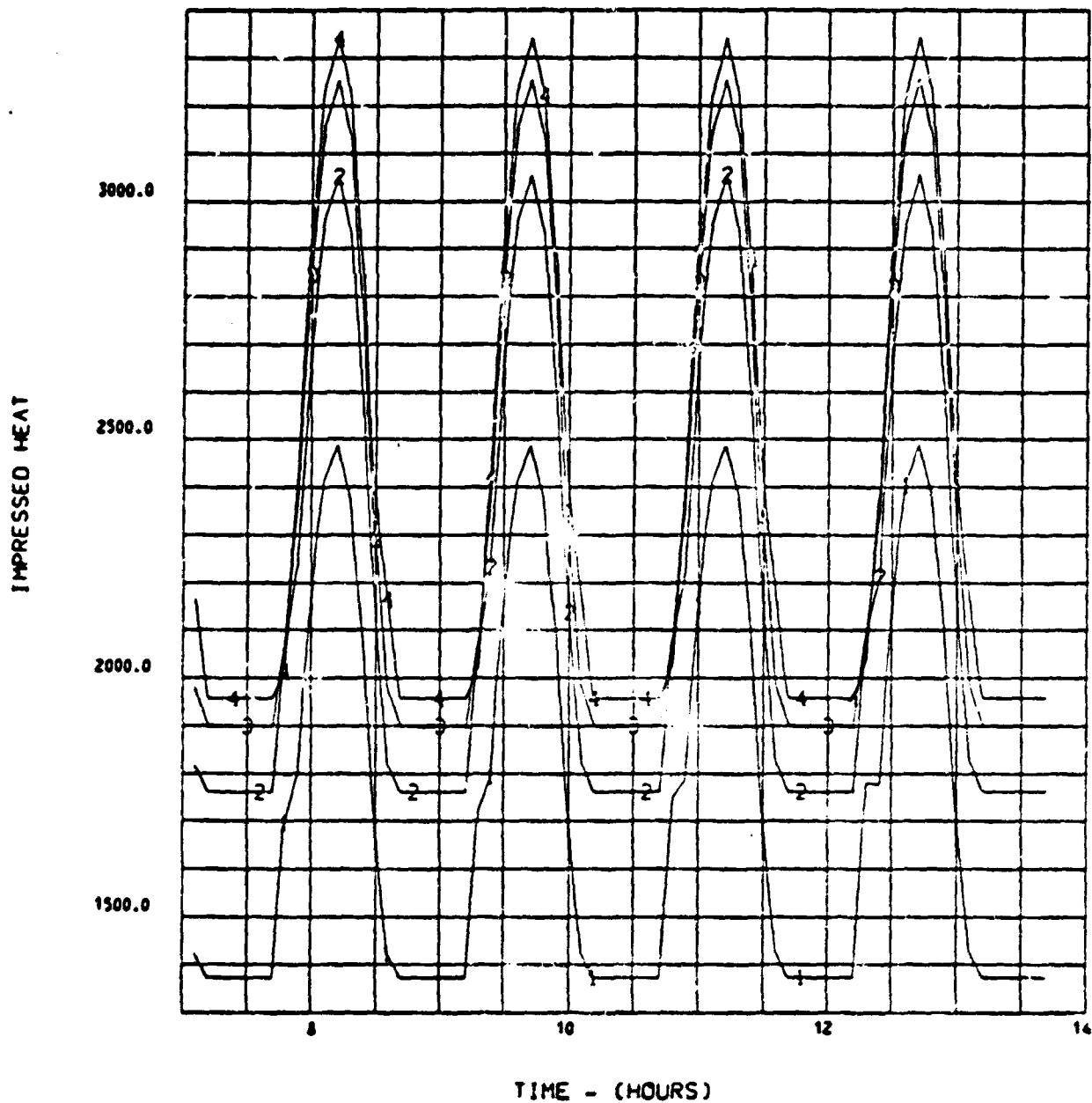


Figure 110

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.

- [1] LOWER AFT PLB LINER. PCRT
- [2] LOWER AFT PLB LINER. PORT
- [3] LOWER AFT PLB LINER. PORT
- [4] LOWER AFT PLB LINER. PCRT

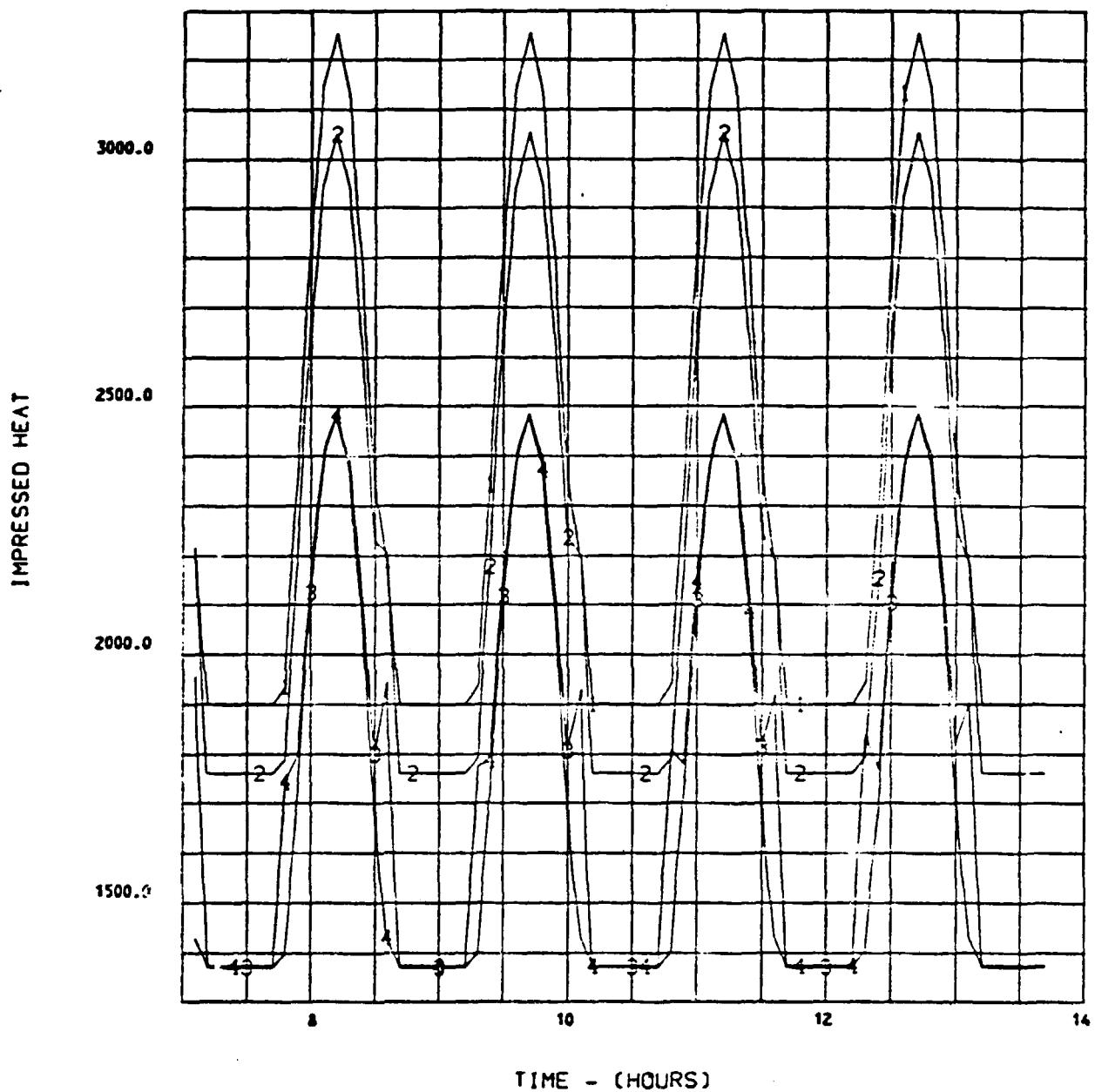


Figure 111

COMPARISON "136 NODE" PLB MODEL. ORBGEN NPT=24 VS TRJ.
[1] FWD LONGERON, PORT
[2] AFT LONGERON, PORT

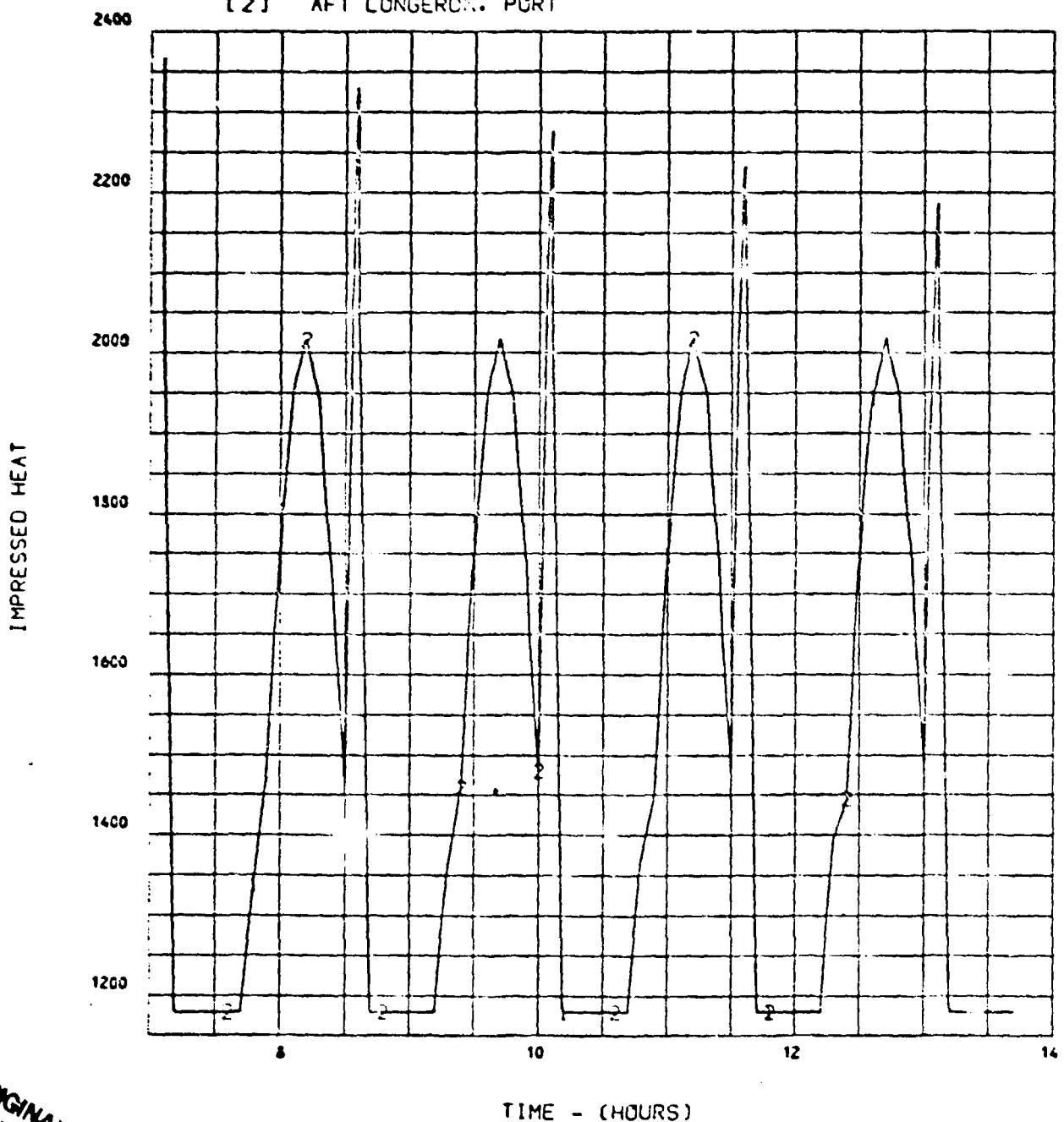


Figure 112

ORIGINAL PAGE IS
OF POOR QUALITY